

to discard certain parts, this AD requires removing those parts from service.

(5) While paragraph (2) of EASA AD 2020–0087 requires actions before next flight after each application of painting on the LH side stairway door or its external door handle, those actions are not required by this AD.

(6) Where paragraph (3) of EASA AD 2020–0087 requires reconditioning the locking safety mechanism, and the service information referenced in paragraph (3) of EASA AD 2020–0087 specifies contacting the Airbus Helicopters Support and Services Department if it is impossible to recondition the locking safety mechanism by moving the door handle, this AD requires moving the external door handle from the “Locked” to the “Unlocked” position to determine if the safety mechanism can lock automatically. If the safety mechanism does not lock automatically, this AD requires, before further flight accomplishing paragraph (5) of EASA AD 2020–0087 or accomplishing corrective action using a method approved by the Manager, International Validation Branch, FAA. The Manager’s approval letter must specifically refer to this AD.

(7) Where paragraph (5) of EASA AD 2020–0087 identifies the modification as required by paragraph (4) of EASA AD 2020–0087 as terminating action for the repetitive inspections as required by paragraph (2) of EASA AD 2020–0087 for that helicopter, the terminating action for the repetitive inspections as required by paragraph (2) of EASA AD 2020–0087 does not apply to this AD.

(8) This AD does not mandate compliance with the “Remarks” section of EASA AD 2020–0087.

(i) No Reporting Requirement

Although the service information referenced in EASA AD 2020–0087 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

(j) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(k) Related Information

For more information about this AD, contact Hal Jensen, Aerospace Engineer, Operational Safety Branch, FAA, 950 L’Enfant Plaza N SW, Washington, DC 20024; telephone (202) 267–9167; email hal.jensen@faa.gov.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following service information was approved for IBR on September 24, 2021 (86 FR 46771).

(i) European Union Aviation Safety Agency (EASA) AD 2020–0087, dated April 15, 2020.

(ii) [Reserved]

(4) For EASA AD 2020–0087, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>.

(5) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110. This material may be found in the AD docket at <https://www.regulations.gov> by searching for and locating FAA–2021–0785.

(6) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on September 8, 2021.

Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2021–20464 Filed 9–24–21; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2021–0723; Project Identifier MCAI–2020–00268–R; Amendment 39–21716; AD 2021–18–15]

RIN 2120–AA64

Airworthiness Directives; PZL Swidnik S.A. Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain PZL Swidnik S.A. Model PZL W–3A helicopters. This AD was prompted by a report that displaced teeth were detected on the moveable assemblies of a main rotor (MR) blade droop stop.

This AD requires removing from service the moveable assemblies from each affected MR blade droop stop and prohibits installation of an affected MR blade droop stop and moveable assemblies of affected MR blade droop stops. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD becomes effective October 12, 2021.

The Director of the Federal Register approved the incorporation by reference of a certain document listed in this AD as of October 12, 2021.

The FAA must receive comments on this AD by November 12, 2021.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- **Federal eRulemaking Portal:** Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** (202) 493–2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- **Hand Delivery:** Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this final rule, contact WSK “PZL-Swidnik” S.A., Al. Lotników Polskich 1, 21–045 Swidnik, Poland; telephone (+48) 81722 5716; fax (+48) 81722 5625; email: PL-CustomerSupport.AW@leonardocompany.com; or at <https://www.pzlswidnik.pl/en/home>. You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110. It is also available at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0723.

Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0723; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the European Union Aviation Safety Agency (EASA) AD, any comments received, and other information. The street address for Docket Operations is listed above.

FOR FURTHER INFORMATION CONTACT: Andrea Jimenez, Aerospace Engineer,

COS Program Management Section, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 1600 Stewart Ave., Suite 410, Westbury, NY 11590; telephone (516) 228-7330; email andrea.jimenez@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2019-0202, dated August 19, 2019 (EASA AD 2019-0202), to correct an unsafe condition for PZL Swidnik S.A. Model PZL W-3A and PZL W-3AS helicopters all manufacturer serial numbers. Model PZL W-3AS helicopters are not certificated by the FAA and are not included on the U.S. type certificate data sheet; this AD therefore does not include those helicopters in the applicability.

EASA advises that an occurrence was reported where displaced teeth were detected on the moveable assemblies of an MR blade droop stop part number (P/N) 37.21.800.00.00). This condition, if not addressed, could result in erroneous operation of MR blade droop stop teeth during engine start-up or shut-down, or dynamic drop-down of an MR blade, resulting in contact of the affected MR blade with the tail boom, and possibly resulting in injury of occupants or persons on the ground.

Accordingly, EASA AD 2019-0202 requires removal of each affected part from any helicopter on which it is installed and prohibits installation of any affected MR blade droop stop or any moveable assembly of an affected MR blade droop stop on any helicopter.

FAA's Determination

These helicopters have been approved by EASA and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with the European Union, EASA has notified the FAA about the unsafe condition described in its AD. The FAA is issuing this AD after evaluating all known relevant information and determining that the unsafe condition described previously is likely to exist or develop on other helicopters of the same type design.

Related Service Information Under 1 CFR Part 51

The FAA reviewed WYTWÓRNIA SPRZĘTU KOMUNIKACYJNEGO "PZL-Swidnik" Spółka Akcyjna Mandatory Bulletin No. BO-37-18-302, Revision 1, dated July 11, 2019. This service information specifies procedures for removing the moveable assemblies of the MR blade droop stops. The service

information specifies that the MR blade droop stop consists of four retaining washers that are installed on the MR hub flapping hinges and four moveable MR blade droop stop assemblies installed on the retaining washers.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

Other Related Service Information

The FAA also reviewed WYTWÓRNIA SPRZĘTU KOMUNIKACYJNEGO "PZL-Swidnik" Spółka Akcyjna Mandatory Bulletin No. BO-37-18-302, dated June 19, 2019. This service information also specifies procedures for removing the moveable assemblies of the MR blade droop stops but does not include the detailed effectivity and scope of compliance that is included in WYTWÓRNIA SPRZĘTU KOMUNIKACYJNEGO "PZL-Swidnik" Spółka Akcyjna Mandatory Bulletin No. BO-37-18-302, Revision 1, dated July 11, 2019.

AD Requirements

This AD requires accomplishing the actions specified in the service information already described.

Justification for Immediate Adoption and Determination of the Effective Date

Section 553(b)(3)(B) of the Administrative Procedure Act (APA) (5 U.S.C. 551 *et seq.*) authorizes agencies to dispense with notice and comment procedures for rules when the agency, for "good cause," finds that those procedures are "impracticable, unnecessary, or contrary to the public interest." Under this section, an agency, upon finding good cause, may issue a final rule without providing notice and seeking comment prior to issuance. Further, section 553(d) of the APA authorizes agencies to make rules effective in less than thirty days, upon a finding of good cause.

There are no helicopters with this type certificate on the U.S. Registry. Accordingly, notice and opportunity for prior public comment are unnecessary, pursuant to 5 U.S.C. 553(b)(3)(B). In addition, for the foregoing reason, the FAA finds that good cause exists pursuant to 5 U.S.C. 553(d) for making this amendment effective in less than 30 days.

Comments Invited

The FAA invites you to send any written data, views, or arguments about this final rule. Send your comments to an address listed under ADDRESSES. Include "Docket No. FAA-2021-0723;

Project Identifier MCAI-2020-00268-R" at the beginning of your comments. The most helpful comments reference a specific portion of the final rule, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this final rule because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to <https://www.regulations.gov>, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this final rule.

Confidential Business Information

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this AD contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this AD, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as "PROPIN." The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this AD. Submissions containing CBI should be sent to Andrea Jimenez, Aerospace Engineer, COS Program Management Section, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 1600 Stewart Ave., Suite 410, Westbury, NY 11590; telephone (516) 228-7330; email andrea.jimenez@faa.gov. Any commentary that the FAA receives which is not specifically designated as CBI will be placed in the public docket for this rulemaking.

Regulatory Flexibility Act

The requirements of the Regulatory Flexibility Act (RFA) do not apply when an agency finds good cause pursuant to 5 U.S.C. 553 to adopt a rule without prior notice and comment. Because the FAA has determined that it has good cause to adopt this rule without prior notice and comment, RFA analysis is not required.

Costs of Compliance

There are no costs of compliance with this AD because there are no helicopters

with this type certificate on the U.S. Registry.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866, and
- (2) Will not affect intrastate aviation in Alaska.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2021–18–15 PZL Swidnik S.A.:
Amendment 39–21716; Docket No. FAA–2021–0723; Project Identifier MCAI–2020–00268–R.

(a) Effective Date

This airworthiness directive (AD) is effective October 12, 2021.

(b) Affected ADs

None.

(c) Applicability

This AD applies to PZL Swidnik S.A. Model PZL W–3A helicopters, certificated in any category, with movable assemblies of main rotor (MR) blade droop stop, part number (P/N) 37.21.800.00.00, installed.

(d) Subject

Joint Aircraft Service Component (JASC) Code: 6200, Main Rotor System.

(e) Unsafe Condition

This AD was prompted by a report that displaced teeth were detected on the moveable assemblies of an MR blade droop stop. The FAA is issuing this AD to address displaced teeth on the moveable assemblies of the MR blade droop stop. The unsafe condition, if not addressed, could result in erroneous operation of MR blade droop stop teeth during engine start-up or shut-down, or dynamic drop-down of an MR blade, resulting in contact of the affected MR blade with the tail boom, and possibly resulting in injury of occupants or persons on the ground.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

Within 30 days after the effective date of this AD, remove from service each moveable assembly of MR blade droop stop P/N 37.21.800.00.00 from all MR hub arms, in accordance with Chapter II of WYTWORNIA SPRZĘTU KOMUNIKACYJNEGO "PZL–Świdnik" Spółka Akcyjna Mandatory Bulletin No. BO–37–18–302, Revision 1, dated July 11, 2019.

(h) Part Installation Prohibition

As of the effective date of this AD, no person may install on any helicopter an MR blade droop stop, P/N 37.21.800.00.00, and do not install on any helicopter any movable assembly of an MR blade droop stop, P/N 37.21.800.00.00.

(i) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (g) of this AD, if those actions were performed before the effective date of this AD using WYTWORNIA SPRZĘTU KOMUNIKACYJNEGO "PZL–Świdnik" Spółka Akcyjna Mandatory Bulletin No. BO–37–18–302, dated June 19, 2019.

(j) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the

procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k)(1) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(k) Related Information

(1) For more information about this AD, contact Andrea Jimenez, Aerospace Engineer, COS Program Management Section, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 1600 Stewart Ave., Suite 410, Westbury, NY 11590; telephone (516) 228–7330; email andrea.jimenez@faa.gov.

(2) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (l)(3) and (4) of this AD.

(3) The subject of this AD is addressed in European Union Aviation Safety Agency (EASA) AD 2019–0202, dated August 19, 2019. You may view the EASA AD at <https://www.regulations.gov> in Docket No. FAA–2021–0723.

(l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) WYTWORNIA SPRZĘTU KOMUNIKACYJNEGO "PZL–Świdnik" Spółka Akcyjna Mandatory Bulletin No. BO–37–18–302, Revision 1, dated July 11, 2019.

(ii) [Reserved]

(3) For service information identified in this AD, contact WSK "PZL–Świdnik" S.A., Al. Lotników Polskich 1, 21–045 Świdnik, Poland; telephone (+48) 81722 5716; fax (+48) 81722 5625; email: PL-CustomerSupport.AW@leonardocompany.com; or at <https://www.pzlswidnik.pl/en/home>.

(4) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email: fr.inspection@nara.gov, or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on August 26, 2021.

Lance T. Gant,

*Director, Compliance & Airworthiness
Division, Aircraft Certification Service.*

[FR Doc. 2021-20830 Filed 9-24-21; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2021-0513; Project Identifier 2018-SW-116-AD; Amendment 39-21717; AD 2021-18-16]

RIN 2120-AA64

Airworthiness Directives; Bell Textron Canada Limited (Type Certificate Previously Held by Bell Helicopter Textron Canada Limited) Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Bell Textron Canada Limited (Bell) Model 429 helicopters. This AD was prompted by reports of tail rotor gearbox assemblies found loose on the gearbox support. This AD requires repetitive torque checks of the tail rotor gearbox attachment hardware, and corrective action if necessary. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective November 1, 2021.

The Director of the Federal Register approved the incorporation by reference of a certain document listed in this AD as of November 1, 2021.

ADDRESSES: For service information identified in this final rule, contact Bell Textron Canada Limited, 12,800 Rue de l'Avenir, Mirabel, Quebec J7J 1R4, Canada; telephone 1-450-437-2862 or 1-800-363-8023; fax 1-450-433-0272; email productsupport@bellflight.com; or at <https://www.bellflight.com/support/contact-support>. You may view the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817-222-5110. It is also available at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0513.

Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0513; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the Transport Canada AD, any comments received, and other information. The street address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

Darren Gassetto, Aerospace Engineer, COS Program Management Section, FAA, Operational Safety Branch, Compliance & Airworthiness Division, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone (516) 228-7323; email Darren.Gassetto@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain Bell Textron Canada Limited (Bell) Model 429 helicopters. The NPRM published in the **Federal Register** on June 28, 2021 (86 FR 33918). In the NPRM, the FAA proposed to require repetitive torque checks of the tail rotor gearbox attachment hardware, and corrective action if necessary. The NPRM was prompted by Canadian AD CF-2018-35, dated December 19, 2018 (Canadian AD CF-2018-35), issued by Transport Canada, which is the aviation authority for Canada, to correct an unsafe condition for Bell Textron Canada Limited Model 429 helicopters. Transport Canada advises that there have been reports of tail rotor gearbox assemblies found loose on the gearbox support. Transport Canada issued Emergency Canadian Airworthiness Directive CF-2018-18, dated July 11, 2018, which corresponds to FAA AD 2018-16-51, Amendment 39-19421 (83 FR 53171, October 22, 2018), to address the immediate safety concern. An ongoing investigation determined that this condition-loose tail rotor gearbox assemblies-could return even after the corrective actions by the previous AD have been completed. This condition, if not addressed, could result in structural

damage and possible loss of control of the helicopter.

Accordingly, Canadian AD CF-2018-35 requires repetitive torque checks of the tail rotor gearbox attachment hardware and corrective actions if necessary. The corrective action is doing additional repetitive torque checks at intervals of 10 to 25 hours air time until the torque stabilizes on all the nuts.

Discussion of Final Airworthiness Directive

Comments

The FAA received no comments on the NPRM or on the determination of the costs.

Conclusion

These helicopters have been approved by the aviation authority of Canada and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with Canada, Transport Canada, its technical representative, has notified the FAA of the unsafe condition described in its AD. The FAA reviewed the relevant data and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on these helicopters. Except for minor editorial changes, this AD is adopted as proposed in the NPRM.

Related Service Information Under 14 CFR Part 51

The FAA reviewed Bell Alert Service Bulletin 429-18-41, dated July 24, 2018. This service information specifies procedures for repetitive torque checks of the tail rotor gearbox attachment hardware.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Differences Between This AD and the Transport Canada AD

Where Canadian AD CF-2018-35 refers to "200-hour" inspections and "10 to 25 hours air time" for the torque checks, for this AD use "time-in-service" instead.

Costs of Compliance

The FAA estimates that this AD affects 98 helicopters of U.S. Registry. The FAA estimates the following costs to comply with this AD: