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MANDATORY MANDATORY MANDATORY

ATA SYSTEM: 25

**TITLE: EQUIPMENT/FURNISHINGS - FORWARD FUSELAGE EQUIPMENT -
SPECIAL DETAILED INSPECTION OF THE RACK 80VU**

MODIFICATION No.:

None

ATA SYSTEM: 25

TITLE: EQUIPMENT/FURNISHINGS - FORWARD FUSELAGE EQUIPMENT - SPECIAL DETAILED INSPECTION OF THE RACK 80VU

REVISION TRANSMITTAL SHEET

This page transmits Revision No. 02 of Service Bulletin No. A320-25-1BKJ.

A. ADDITIONAL WORK

No additional work is required by this revision.

B. REASON

Revision No. 02 issued to remove the reference to the Service Bulletin A320-25-1BQF and to add the Repair task solution.

C. CHANGES

CONF 001

SUMMARY	
MATERIAL PRICE INFORMATION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
APPLICABILITY	Updated
NATURE OF THE WORK	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
MANPOWER	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
PLANNING INFORMATION	
APPLICABILITY	Applicability by MSN updated
DESCRIPTION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
COMPLIANCE	
Accomplishment Timescale	Table 01 through 03 reworked
MANPOWER	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
REFERENCES	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended References updated
INDUSTRY SUPPORT INFORMATION	Sentence added
MATERIAL INFORMATION	

MATERIAL APPLICABILITY	
Material for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
ACCOMPLISHMENT INSTRUCTIONS	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	
TASK SET ASSOCIATED INFORMATION	Man-hours updated
PREPARATION	
A320-A-25-XX-1BKJ-01001-540A-A Job Set-up	Working area added Man-hours updated References and worksteps added
PROCEDURE	
A320-A-25-XX-1BKJ-01001-346A-A Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FAAA	Flowchart modified
A320-A-25-XX-1BKJ-02001-346A-A Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FBAA	Flowchart modified
A320-A-25-XX-1BKJ-04001-346A-A Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FCAA	Flowchart modified
A320-A-25-XX-1BKJ-03001-90AA-A Complete the Inspection Report sheet	SUBTASK added
fig Inspection Report sheet	Illustration added, moved from APPENDIX
CLOSE-UP	
A320-A-25-XX-1BKJ-01001-740A-A Close-up	Working area added Man-hours updated References and worksteps added

CONF 002

SUMMARY	
MATERIAL PRICE INFORMATION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended

Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
APPLICABILITY	Updated
NATURE OF THE WORK	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
MANPOWER	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended Man-hours updated
PLANNING INFORMATION	
APPLICABILITY	Applicability by MSN updated
DESCRIPTION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
COMPLIANCE	
Accomplishment Timescale	Table 01 through 03 reworked
MANPOWER	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended Man-hours updated
REFERENCES	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended References updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended References updated
INDUSTRY SUPPORT INFORMATION	Sentence added
MATERIAL INFORMATION	
MATERIAL APPLICABILITY	
Material for task set A320-A-25-XX-1BKJ-01000-93BA-A	Title amended

Inspection of the Electronics Rack FIN 80VU	
Material for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
CONSUMABLES	
Consumables for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Repair Task added
LIST OF NEW PARTS	
Part kit 251BKJD01R00	Kit for Repair Task added
Part kit 251BKJD02R00	Kit for Repair Task added
Part kit 251BKJD05R00	Kit for Repair Task added
ACCOMPLISHMENT INSTRUCTIONS	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	
TASK SET ASSOCIATED INFORMATION	Man-hours updated
PREPARATION	
A320-A-25-XX-1BKJ-01001-540A-A Job Set-up	Working area added Man-hours updated References and worksteps modified and transfered to the Repair TASK
PROCEDURE	
A320-A-25-XX-1BKJ-01002-346A-A Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FAAB	Flowchart modified
A320-A-25-XX-1BKJ-02002-346A-A Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FBAB	Flowchart modified
A320-A-25-XX-1BKJ-04001-346A-A Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FCAA	Flowchart modified
A320-A-25-XX-1BKJ-03001-90AA-A Complete the Inspection Report sheet	SUBTASK added
fig Inspection Report sheet	Illustration added, moved from APPENDIX

TEST	SUBTASK removed
CLOSE-UP	
A320-A-25-XX-1BKJ-01002-740A-A Close-up	Working area added Man-hours updated References and worksteps modified and transfered to the Repair TASK
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Repair TASK added
PREPARATION	
Task A320-A-25-XX-1BKJ-02002-540A-A Job Set-up	
fig Principle jacking of the 80VU	Illustration added
Task A320-A-25-XX-1BKJ-03002-540A-A Remove the ducts	
fig Removal of the RH ducts	Illustration added
fig Removal of the LH duct	Illustration added
fig Removal of the upper LH duct	Illustration added
Task A320-A-25-XX-1BKJ-04002-540A-A Remove structure and struts	
fig Removal of the structure at FR24A	Illustration added
fig Removal of the LH strut at FR24A	Illustration added
fig Removal of the RH strut at FR24A	Illustration added
Task A320-A-25-XX-1BKJ-06002-540A-A Remove the support	
fig Removal of the support located under the 80VU	Illustration added
fig Removal of the support located under the 80VU	Illustration added
PROCEDURE	
Task A320-A-25-XX-1BKJ-01002-600A-A Accomplish the Repair Instruction	
fig Installation principle of the lower lateral fitting	Illustration added
CLOSE-UP	
Task A320-A-25-XX-1BKJ-02002-740A-A Install the support	
fig Installation of the support located under the 80VU	Illustration added
fig Installation of the support located under the 80VU	Illustration added
Task A320-A-25-XX-1BKJ-04002-740A-A Install structure and struts	
fig Installation of the structure at FR24A	Illustration added

fig Installation of the LH strut at FR24A	Illustration added
fig Installation of the RH strut at FR24A	Illustration added
Task A320-A-25-XX-1BKJ-05002-740A-A Install the ducts	
fig Installation of the upper LH duct	Illustration added
fig Installation of the RH ducts	Illustration added
fig Installation of the LH duct	Illustration added

CONF 003

SUMMARY	
MATERIAL PRICE INFORMATION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
APPLICABILITY	Updated
NATURE OF THE WORK	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
MANPOWER	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended Man-hours updated
PLANNING INFORMATION	
APPLICABILITY	Applicability by MSN updated
DESCRIPTION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
COMPLIANCE	
Accomplishment Timescale	Table 01 through 03 reworked
MANPOWER	

Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended Man-hours updated
REFERENCES	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended References updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended References updated
INDUSTRY SUPPORT INFORMATION	Sentence added
MATERIAL INFORMATION	
MATERIAL APPLICABILITY	
Material for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Material for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
CONSUMABLES	
Consumables for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Repair Task added
LIST OF NEW PARTS	
Part kit 251BKJD01R00	Kit for Repair Task added
Part kit 251BKJD02R00	Kit for Repair Task added
Part kit 251BKJD05R00	Kit for Repair Task added
ACCOMPLISHMENT INSTRUCTIONS	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	
TASK SET ASSOCIATED INFORMATION	Man-hours updated
PREPARATION	
A320-A-25-XX-1BKJ-01001-540A-A Job Set-up	Working area added Man-hours updated References and worksteps modified and transfered to the Repair TASK
PROCEDURE	
A320-A-25-XX-1BKJ-01003-346A-A Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	Working area modified Procedure modified

fig Flowchart FAAC	Flowchart modified
A320-A-25-XX-1BKJ-02003-346A-A Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FBAC	Flowchart modified
A320-A-25-XX-1BKJ-04001-346A-A Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FCAA	Flowchart modified
A320-A-25-XX-1BKJ-03001-90AA-A Complete the Inspection Report sheet	SUBTASK added
fig Inspection Report sheet	Illustration added, moved from APPENDIX
TEST	SUBTASK removed
CLOSE-UP	
A320-A-25-XX-1BKJ-01002-740A-A Close-up	Working area added Man-hours updated References and worksteps modified and transfered to the Repair TASK
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Repair TASK added
PREPARATION	
Task A320-A-25-XX-1BKJ-02002-540A-A Job Set-up	
fig Principle jacking of the 80VU	Illustration added
Task A320-A-25-XX-1BKJ-03002-540A-A Remove the ducts	
fig Removal of the RH ducts	Illustration added
fig Removal of the LH duct	Illustration added
fig Removal of the upper LH duct	Illustration added
Task A320-A-25-XX-1BKJ-04002-540A-A Remove structure and struts	
fig Removal of the structure at FR24A	Illustration added
fig Removal of the LH strut at FR24A	Illustration added
fig Removal of the RH strut at FR24A	Illustration added
Task A320-A-25-XX-1BKJ-06002-540A-A Remove the support	
fig Removal of the support located under the 80VU	Illustration added

fig Removal of the support located under the 80VU	Illustration added
PROCEDURE	
Task A320-A-25-XX-1BKJ-01002-600A-A Accomplish the Repair Instruction	
fig Installation principle of the lower lateral fitting	Illustration added
CLOSE-UP	
Task A320-A-25-XX-1BKJ-02002-740A-A Install the support	
fig Installation of the support located under the 80VU	Illustration added
fig Installation of the support located under the 80VU	Illustration added
Task A320-A-25-XX-1BKJ-04002-740A-A Install structure and struts	
fig Installation of the structure at FR24A	Illustration added
fig Installation of the LH strut at FR24A	Illustration added
fig Installation of the RH strut at FR24A	Illustration added
Task A320-A-25-XX-1BKJ-05002-740A-A Install the ducts	
fig Installation of the upper LH duct	Illustration added
fig Installation of the RH ducts	Illustration added
fig Installation of the LH duct	Illustration added

CONF 004

SUMMARY	
MATERIAL PRICE INFORMATION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
APPLICABILITY	Updated
NATURE OF THE WORK	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
MANPOWER	

Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended Man-hours updated
PLANNING INFORMATION	
APPLICABILITY	Applicability by MSN updated
DESCRIPTION	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
COMPLIANCE	
Accomplishment Timescale	Table 01 through 03 reworked
MANPOWER	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended Man-hours updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended Man-hours updated
REFERENCES	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended References updated
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended References updated
INDUSTRY SUPPORT INFORMATION	Sentence added
MATERIAL INFORMATION	
MATERIAL APPLICABILITY	
Material for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	Title amended
Material for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Title amended
CONSUMABLES	
Consumables for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Repair Task added
LIST OF NEW PARTS	
Part kit 251BKJD03R00	Kit for Repair Task added

Part kit 251BKJD04R00	Kit for Repair Task added
Part kit 251BKJD05R00	Kit for Repair Task added
ACCOMPLISHMENT INSTRUCTIONS	
Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU	
TASK SET ASSOCIATED INFORMATION	Man-hours updated
PREPARATION	
A320-A-25-XX-1BKJ-01001-540A-A Job Set-up	Working area added Man-hours updated References and worksteps modified and transfered to the Repair TASK
PROCEDURE	
A320-A-25-XX-1BKJ-01004-346A-A Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	Working area modified Man-hours updated Procedure modified
fig Flowchart FAAD	Flowchart modified
A320-A-25-XX-1BKJ-03002-346A-A Do an SDI of the Rods and Fittings of the Lower Central Support of the Electronics Rack FIN 80VU	Working area modified Man-hours updated Procedure modified
fig Flowchart FBAD	Flowchart modified
A320-A-25-XX-1BKJ-04001-346A-A Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	Working area modified Procedure modified
fig Flowchart FCAA	Flowchart modified
A320-A-25-XX-1BKJ-03001-90AA-A Complete the Inspection Report sheet	SUBTASK added
fig Inspection Report sheet	Illustration added, moved from APPENDIX
TEST	SUBTASK removed
CLOSE-UP	
A320-A-25-XX-1BKJ-01002-740A-A Close-up	Working area added Man-hours updated References and worksteps modified and transfered to the Repair TASK
Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU	Repair TASK added
PREPARATION	
Task A320-A-25-XX-1BKJ-02002-540A-A Job Set-up	

fig Principle jacking of the 80VU	Illustration added
Task A320-A-25-XX-1BKJ-03002-540A-A Remove the ducts	
fig Removal of the RH ducts	Illustration added
fig Removal of the LH duct	Illustration added
fig Removal of the upper LH duct	Illustration added
Task A320-A-25-XX-1BKJ-04002-540A-A Remove structure and struts	
fig Removal of the structure at FR24A	Illustration added
fig Removal of the LH strut at FR24A	Illustration added
fig Removal of the RH strut at FR24A	Illustration added
Task A320-A-25-XX-1BKJ-06002-540A-A Remove the support	
fig Removal of the support located under the 80VU	Illustration added
fig Removal of the support located under the 80VU	Illustration added
PROCEDURE	
Task A320-A-25-XX-1BKJ-01003-600A-A Accomplish the Repair Instruction	
fig Installation principle of the lower lateral fitting	Illustration added
CLOSE-UP	
Task A320-A-25-XX-1BKJ-02002-740A-A Install the support	
fig Installation of the support located under the 80VU	Illustration added
fig Installation of the support located under the 80VU	Illustration added
Task A320-A-25-XX-1BKJ-04002-740A-A Install structure and struts	
fig Installation of the structure at FR24A	Illustration added
fig Installation of the LH strut at FR24A	Illustration added
fig Installation of the RH strut at FR24A	Illustration added
Task A320-A-25-XX-1BKJ-05002-740A-A Install the ducts	
fig Installation of the upper LH duct	Illustration added
fig Installation of the RH ducts	Illustration added
fig Installation of the LH duct	Illustration added

CONF ALL**D. FILING INSTRUCTIONS**

This Service Bulletin has been generated electronically and is reissued as a complete document. Replace the complete document.

Put this Revision Transmittal Sheet in front of the Service Bulletin.

E. HISTORY OF PREVIOUS REVISIONS

Revision No. 01 is issued to transform Service Bulletin data from ATA iSpec 2200 norm to ASD S1000D norm.

F. REVISION SEQUENCE

ORIGINAL: Aug 23/18

REVISION No. : 01 - Jun 03/19

REVISION No. : 02 - Apr 09/20

ATA SYSTEM: 25

**TITLE: EQUIPMENT/FURNISHINGS - FORWARD FUSELAGE EQUIPMENT -
SPECIAL DETAILED INSPECTION OF THE RACK 80VU****SUMMARY**

This summary is for information only and is not approved for modification of the aircraft.

A. MODIFICATIONS**CONF 001**

MODIFICATION CLASSIFICATION	
MAJOR	None
MINOR	None

NOTE: As per EASA IR 21, a minor change is one that has no appreciable effect on the mass, balance, structural strength, reliability, operational characteristics affecting the airworthiness of the product. All other changes are major changes.

CONF 002

MODIFICATION CLASSIFICATION	
MAJOR	None
MINOR	None

NOTE: As per EASA IR 21, a minor change is one that has no appreciable effect on the mass, balance, structural strength, reliability, operational characteristics affecting the airworthiness of the product. All other changes are major changes.

CONF 003

MODIFICATION CLASSIFICATION	
MAJOR	None
MINOR	None

NOTE: As per EASA IR 21, a minor change is one that has no appreciable effect on the mass, balance, structural strength, reliability, operational characteristics affecting the airworthiness of the product. All other changes are major changes.

CONF 004

MODIFICATION CLASSIFICATION	
MAJOR	None
MINOR	None

NOTE: As per EASA IR 21, a minor change is one that has no appreciable effect on the mass, balance, structural strength, reliability, operational characteristics affecting the airworthiness of the product. All other changes are major changes.

CONF ALL**B. REASON**

(1) History

Further to in-service reports of worn Electronics Rack FIN 80VU lower lateral fittings, an inspection of the Electronics Rack FIN 80VU lower lateral fittings and the Electronics Rack FIN 80VU lower central support was required.

Service Bulletin No. Ref. SB A320-25-1555 was issued to inspect the Electronics Rack FIN 80VU and provide recommendations in case of findings. After Service Bulletin No. Ref. SB A320-25-1555 publication some findings have been reported on aircraft not listed in its applicability because they were post production MOD No. 34804P8794.

There have been reports as well of findings in aircraft having accomplished Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.

These Service Bulletins were a terminating action of Service Bulletin No. Ref. SB A320-25-1555.

NOTE: Service Bulletin No. Ref. SB A320-25-1557 is the corrective action for the lower lateral fittings.

NOTE: Service Bulletin No. Ref. SB A320-53-1215 is the corrective action for the central fitting.

Based on the above information, the following corrective actions cannot be considered as terminative actions anymore:

- Production MOD No. 34804P8794
- Service Bulletin No. Ref. SB A320-25-1557
- Service Bulletin No. Ref. SB A320-53-1215.

(2) Objective/Action

The objective of this Service Bulletin is to:

- Do a Special Detailed Inspection (SDI) of the complete Electronics Rack FIN 80VU (lower lateral fittings, the central support, the upper fittings, the central post, the shelves attachments, the lateral pyramid structure) for potential damage and provide an appropriate instruction in case of findings
- Extend the applicability in order to cover all aircraft defined in the applicability of Service Bulletin No. Ref. SB A320-25-1555 as well as post production MOD No. 34804P8794, post Service Bulletin No. Ref. SB A320-25-1557 and post Service Bulletin No. Ref. SB A320-53-1215
- Reflect that the production MOD No. 34804P8794 and the repair provided in Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215 cannot be considered as a terminative action
- Call out a new repair procedure on lower lateral fittings for aircraft post MOD No. 34804P8794, post Service Bulletin No. Ref. SB A320-25-1557 and post Service Bulletin No. Ref. SB A320-53-1215 via Repair Instruction (RI) R531-13174.

(3) Advantages

The accomplishment of this Service Bulletin will contribute to detect potential damage on the Electronics Rack FIN 80VU and its attachments to the fuselage.

NOTE: This Service Bulletin supersedes Ref. SB A320-25-1555.

(4) Operational/Maintenance Consequences

None

C. GENERAL EVALUATION

EVALUATION TABLE			
COMPLIANCE	MANDATORY	COST SAVING	NO
A/C OPERATION AFFECTED	NO	VENDOR SB	NO
CANCELS INSPECTION SB	NO	ETOPS AFFECTED	NO
PAX COMFORT AFFECTED	NO	POTENTIAL AD	YES
RELIABILITY AFFECTED	NO	STRUCTURAL LIFE EXTENSION	NO
LIFE LIMIT	NO	LINE MAINTENANCE AFFECTED	NO

NOTE: Service Bulletin must be accomplished.

D. MAINTENANCE CONFIGURATION
CONF 001

No specific aircraft maintenance configuration required.

CONF 002

No specific aircraft maintenance configuration required.

CONF 003

No specific aircraft maintenance configuration required.

CONF 004

No specific aircraft maintenance configuration required.

CONF ALL
E. MATERIAL PRICE INFORMATION
CONF 001
(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

No material

CONF 002
(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

No material

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MATERIAL PRICE INFORMATION TABLE			
MATERIAL SET	QTY PER A/C	PRICE PER A/C	MAIN PARTS
Part kit 251BKJD01R00	1	1400.00 USD	Forkend, Fitting, Attachment Parts
Part kit 251BKJD02R00	1	1350.00 USD	Forkend, Fitting, Attachment Parts

MATERIAL PRICE INFORMATION TABLE			
MATERIAL SET	QTY PER A/C	PRICE PER A/C	MAIN PARTS
Part kit 251BKJD05R00	1	50.00 USD	Attachment parts
TOTAL		2800.00 USD	

CONF 003
(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

No material

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MATERIAL PRICE INFORMATION TABLE			
MATERIAL SET	QTY PER A/C	PRICE PER A/C	MAIN PARTS
Part kit 251BKJD01R00	1	1400.00 USD	Forkend, Fitting, Attachment Parts
Part kit 251BKJD02R00	1	1350.00 USD	Forkend, Fitting, Attachment Parts
Part kit 251BKJD05R00	1	50.00 USD	Attachment parts
TOTAL		2800.00 USD	

CONF 004
(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

No material

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MATERIAL PRICE INFORMATION TABLE			
MATERIAL SET	QTY PER A/C	PRICE PER A/C	MAIN PARTS
Part kit 251BKJD03R00	1	2050.00 USD	Clevis, Fitting, Attachment Parts
Part kit 251BKJD04R00	1	2050.00 USD	Clevis, Fitting, Attachment Parts
Part kit 251BKJD05R00	1	50.00 USD	Attachment parts
TOTAL		4150.00 USD	

CONF ALL
F. APPLICABILITY

This Service Bulletin is applicable to this (these) operator(s):

01N 02K 03Q 06R 10N 11K 11S 11X 13Y 14S 15S 15T 16T 17M 17R 19R 20M 20Q 23R 25P
26L 27V 28M 29S 31V 31X 32Q 34K 34L 36R 39N 40N 40S 41N 41S 43S 46L 46O 46U 47P
48O 48P 49N 49O 50T 52L 53Z 56O 59Z 60U 60X 61W 66X 68M 69X 72T 73W 76W 78X 80X
82M 82U 82W 83O 83Q 83V 85T 86K 86P 86Z 87Q 87Z 89Q 91X 95V 95W 96O 96Q 96U
98W 98X 99W A1L AAF AAL AAR AAV AAW AAY ABL ABN ABQ ABY ACA ACI AEE AEG AFL AFR
AHY AIC AIJ ALK AMC AMU AMV ANA ANR ANZ APJ ART ARU ASA ASL ASV AUA AUH AVA AVJ
AWC AWW AXE AXM AZA B2V BAV BAW BDJ BEC BEL BGF BGH BJN BKP BNL BRS BSX BTK
BTN BUC BWJ C2C CBG CBJ CCA CCM CDC CEB CEF CES CFG CGH CHB CLH CQH CQN CRK
CSA CSC CSN CSZ CTB CTN CTV CWS CYP DAL DCS DKH DLH DNU DQA DRK EAF EDW EIN
ELB EOK ERN ETD ETJ EVA EVE EWE EWG EXS EZY F2I FAD FBS FFT FHM FHY FIA FIN FLI FOO
FRN G2X G4I G8E GAF GAP GBB GCR GFA GJT GLG GMI GOW GSH GSW GTR GWI GWR GZQ
HDA HES HFM HFY HIM HKE HKJ HVN HXA IAC IAW IBE IBS IGO IND IRA IRB IRC IRM ISR IYE
IZG JAD JAT JAV JBU JBW JCC JES JJP JOC JSA JST JZR KAC KAY KHV KIS KKK KME KNE KZR LAA
LAN LAO LBT LDM LFO LHT LKE LLM LMJ LMU LUR LVT LWA LYX LZB MAC MAU MEA MJF MJN
MKR MLD MLM MMA MMM MMO MRJ MSC MSR MXA MYX NIA NKS NMA NMB NUM NVD NWK
NWS O8K OHY OLY OMS ONE ORF OTC OTF OZW PAL PGT PIA PIC PKZ PLM QAF QDA QQE
QSM QTR R1T RBA RBG RIH RJA RJD RNA ROT RSY RWZ RYL RZO SAA SAS SAW SBI SDM SDR
SEK SFJ SFW SGQ SHJ SHU SIF SKU SLK SMC SND SNG SOL SSG SSV SUD SVA SVR SVW SWM
SWR SYR SZN T2R TAE TAI TAM TAP TAR TBA TBJ TBZ TCI TCN TDR TGW THD THY TNA TSC
TTW U6A UAE UAL UEA UKN UNA UZB VAW VIP VIV VJC VKG VLG VNL VOE VOI VRE VTI VVC
WHT WRC WZZ

G. CONCURRENT REQUIREMENTS

None

H. NATURE OF THE WORK

CONF 001

(1) **Inspection of the Electronics Rack FIN 80VU**

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT	YES
EQUIPMENT	NO

CONF 002

(1) **Inspection of the Electronics Rack FIN 80VU**

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT	YES
EQUIPMENT	NO

(2) **Repair of the Electronics Rack FIN 80VU**

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

AIRCRAFT	YES
EQUIPMENT	NO

CONF 003

(1) **Inspection of the Electronics Rack FIN 80VU**

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT	YES
EQUIPMENT	NO

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

AIRCRAFT	YES
EQUIPMENT	NO

CONF 004

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT	YES
EQUIPMENT	NO

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

AIRCRAFT	YES
EQUIPMENT	NO

CONF ALL

I. MANPOWER

CONF 001

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MANPOWER	
TOTAL MAN-HOURS	6.50 h
ELAPSED TIME (HOURS)	3.25 h

CONF 002

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MANPOWER	
TOTAL MAN-HOURS	6.50 h
ELAPSED TIME (HOURS)	3.25 h

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MANPOWER	
TOTAL MAN-HOURS	122.00 h
ELAPSED TIME (HOURS)	73.50 h

CONF 003

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MANPOWER	
TOTAL MAN-HOURS	6.50 h
ELAPSED TIME (HOURS)	3.25 h

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MANPOWER	
TOTAL MAN-HOURS	122.00 h
ELAPSED TIME (HOURS)	73.50 h

CONF 004

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MANPOWER	
TOTAL MAN-HOURS	7.50 h
ELAPSED TIME (HOURS)	3.75 h

(2) Repair of the Electronics Rack FIN 80VU

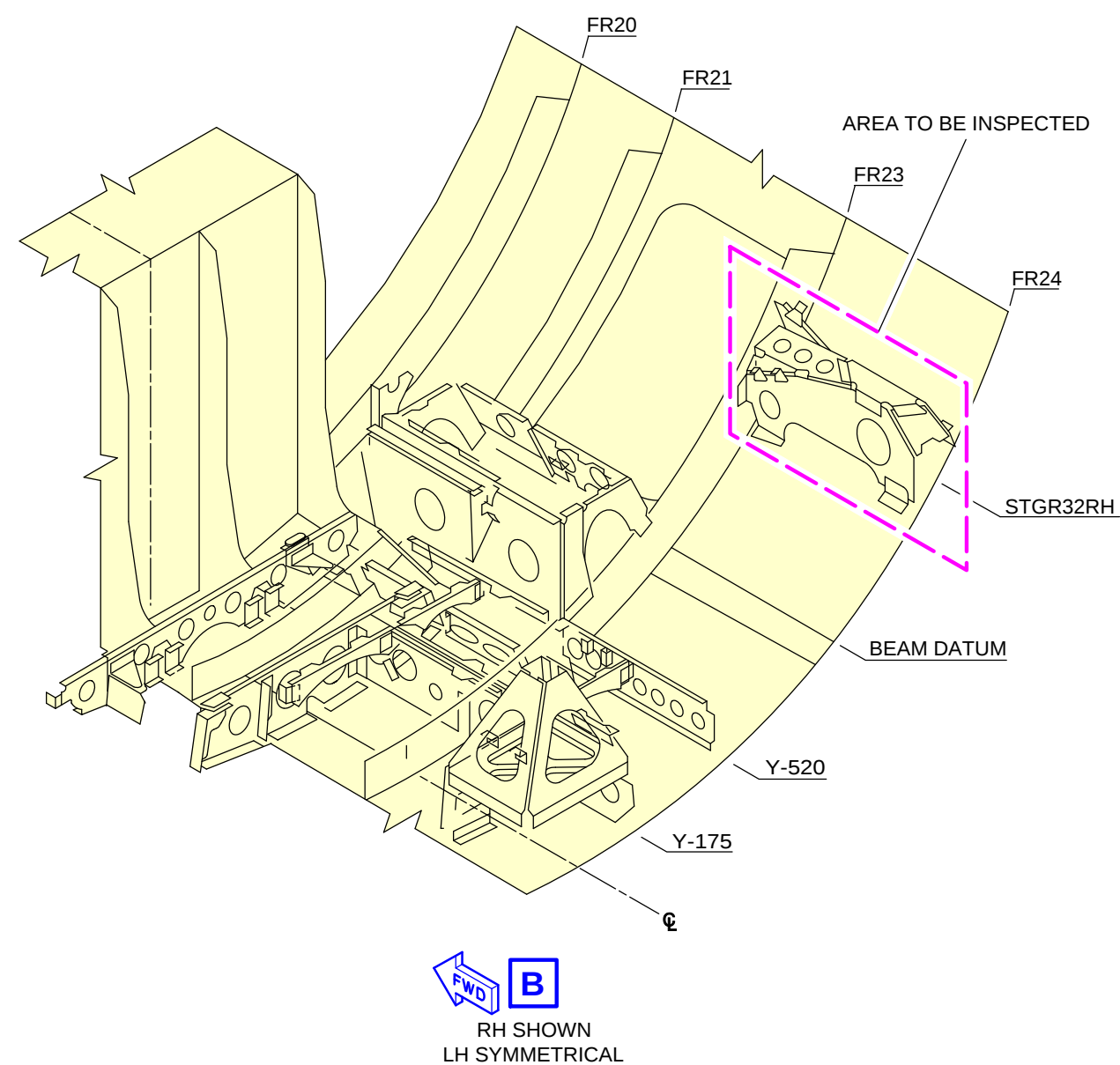
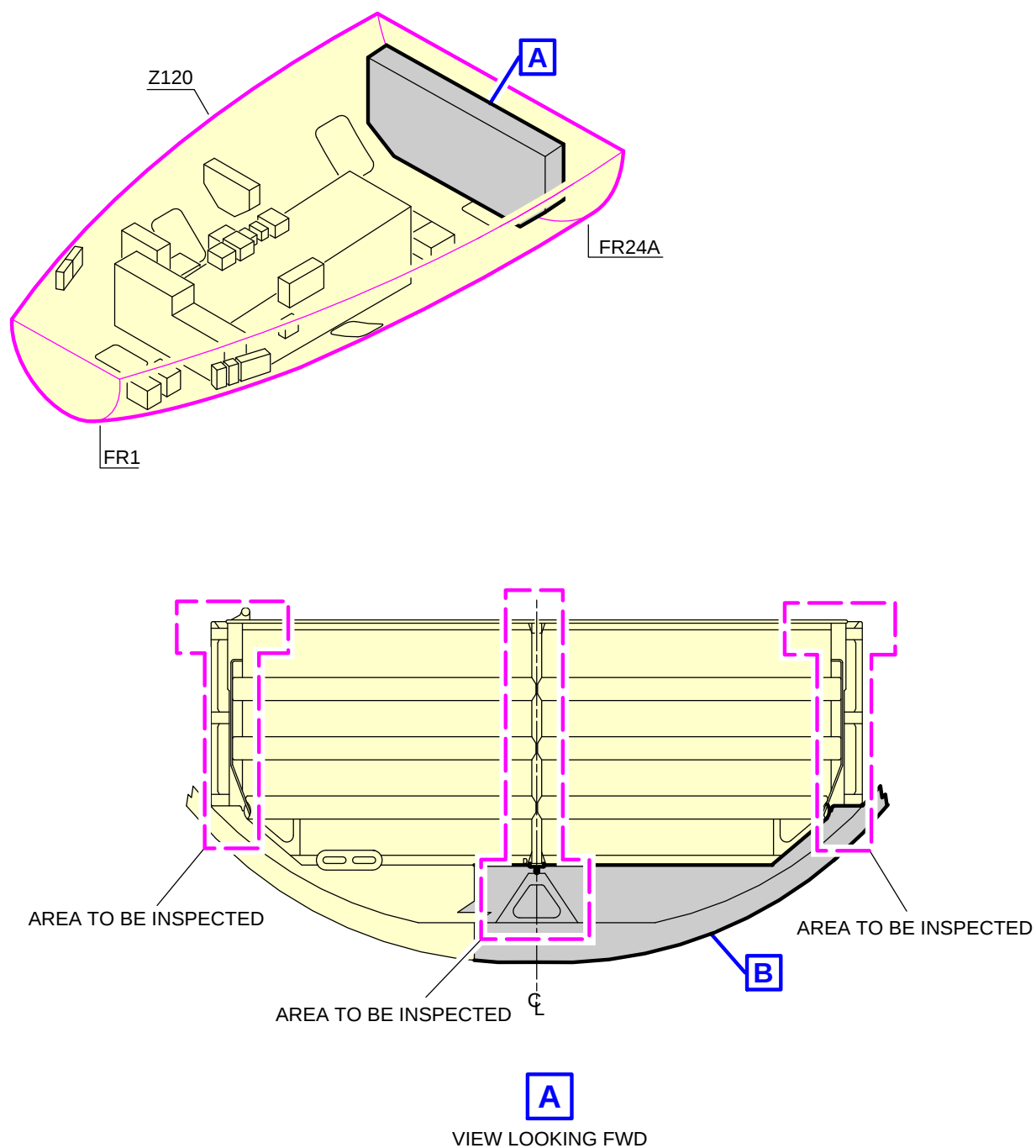
- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MANPOWER	
TOTAL MAN-HOURS	122.00 h
ELAPSED TIME (HOURS)	73.50 h

CONF ALL

J. GENERAL OVERVIEW

CONF ALL



ICN-A320-A-25XX1BKJ-A-FAPE3-00EP9-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP9-A-001-01 SHEET 1/1

General Overview

ATA SYSTEM: 25

**TITLE: EQUIPMENT/FURNISHINGS - FORWARD FUSELAGE EQUIPMENT -
SPECIAL DETAILED INSPECTION OF THE RACK 80VU****1. PLANNING INFORMATION****A. APPLICABILITY****(1) Models**

ON A/C 02035, 02059, 02071, 02081, 02100, 02109, 02317, 02328, 02333, 02344, 02350, 02358, 02367, 02377, 02582, 02601, 02686, 02750, 02918, 02931, 02951, 02955, 02967, 02972, 03009, 03220, 03225

318-111

ON A/C 01599, 02394, 02523, 02544, 02552, 02575, 02910, 03100, 03238, 03333, 03363, 03530, 03617, 03751, 03886, 03932, 03985, 04007, 04039, 04169, 04211, 04503, 04650, 04732, 04878, 05478, 05545

318-112

ON A/C 00750, 00755, 00763, 00790, 00821, 00837, 00906, 00917, 00933, 00938, 00979, 00985, 00998, 01000, 01020, 01025, 01034, 01036, 01100, 01106, 01120, 01127, 01151, 01165, 01169, 01176, 01180, 01190, 01201, 01216, 01247, 01267, 01271, 01344, 01362, 01404, 01415, 01444, 01449, 01471, 01488, 01505, 01524, 01560, 01583, 01616, 01622, 01640, 01645, 01654, 01677, 01679, 01684, 01699, 01733, 01743, 01753, 01759, 01761, 01803, 01841, 01851, 01863, 01876, 01890, 01938, 02010, 02030, 02037, 02043, 02050, 02052, 02053, 02062, 02069, 02072, 02091, 02093, 02103, 02119, 02129, 02170, 02176, 02179, 02181, 02184, 02186, 02196, 02213, 02214, 02222, 02228, 02230, 02236, 02240, 02241, 02243, 02245, 02249, 02251, 02253, 02258, 02260, 02264, 02279, 02283, 02287, 02289, 02300, 02311, 02318, 02332, 02353, 02360, 02365, 02370, 02380, 02382, 02385, 02392, 02400, 02402, 02412, 02436, 02456, 02460, 02463, 02465, 02471, 02477, 02481, 02492, 02495, 02497, 02503, 02512, 02514, 02516, 02527, 02528, 02538, 02548, 02556, 02565, 02578, 02586, 02588, 02605, 02625, 02636, 02638, 02646, 02677, 02681, 02691, 02700, 02702, 02715, 02716, 02727, 02735, 02742, 02744, 02754, 02763, 02765, 02769, 02777, 02779, 02782, 02786, 02788, 02790, 02795, 02803, 02806, 02812, 02818, 02821, 02827, 02829, 02831, 02843, 02852, 02854, 02860, 02866, 02870, 02873, 02876, 02884, 02889, 02897, 02923, 02938, 02946, 02959, 02961, 03003, 03036, 03041, 03053, 03059, 03061, 03078, 03082, 03084, 03088, 03090, 03118, 03122, 03134, 03137, 03176, 03179, 03184, 03209, 03255, 03281, 03298, 03320, 03336, 03380, 03411, 03413, 03426, 03432, 03442, 03443, 03466, 03537, 03544, 03555, 03569, 03571, 03608, 03615, 03634, 03651, 03657, 03675, 03683, 03702, 03720, 03735, 03742, 03744, 03746, 03754, 03763, 03774, 03788, 03794, 03799, 03808, 03810, 03824, 03838, 03844, 03845, 03854, 03888, 03942, 03998, 04004, 04040, 04048, 04056, 04069, 04076, 04080, 04087, 04125, 04129, 04132, 04313, 04327, 04380, 04385, 04418, 04425, 04427, 04437, 04444, 04451, 04624, 04635, 04640, 04667, 04693, 04705, 04717, 04744, 04759, 04764, 04778, 04781, 04785, 04787, 04837, 04859, 04875, 04910, 05018, 05130, 05294, 05383, 05424

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ON A/C 00578, 00612, 00621, 00629, 00646, 00654, 00713, 00727, 00734, 00767, 00794, 00833, 00885, 00890, 00896, 00904, 00922, 00929, 00946, 00949, 00972, 00997, 01016, 01018, 01019, 01029, 01033, 01038, 01040, 01046, 01048, 01051, 01055, 01069, 01073, 01077, 01084, 01086, 01089, 01095, 01097, 01102, 01107, 01109, 01122, 01124, 01135, 01136, 01147, 01155, 01160, 01170, 01172, 01182, 01184, 01203, 01205, 01209, 01245, 01252, 01254, 01263, 01265, 01269, 01275, 01277, 01283, 01285, 01287, 01289, 01297, 01301, 01303, 01305, 01309, 01311, 01313, 01315, 01317, 01319, 01326, 01328, 01331, 01336, 01340, 01342, 01348, 01352, 01354, 01358, 01360, 01364, 01369, 01371, 01377, 01378, 01382, 01388, 01389, 01391, 01393, 01429, 01494, 01515, 01562, 01577, 01579, 01592, 01598, 01601, 01603, 01612, 01618, 01625, 01630, 01634, 01668, 01673, 01706, 01718, 01722, 01740, 01742, 01745, 01750, 01756, 01765, 01770, 01778, 01779, 01781,

01786, 01791, 01805, 01808, 01846, 01853, 01866, 01872, 01882, 01884, 01886, 01901, 01916, 01925, 01936, 01963, 01980, 01997, 02012, 02019, 02033, 02057, 02066, 02074, 02078, 02083, 02086, 02101, 02113, 02122, 02124, 02126, 02127, 02131, 02174, 02198, 02209, 02224, 02262, 02265, 02266, 02271, 02319, 02378, 02387, 02398, 02406, 02416, 02420, 02427, 02442, 02446, 02448, 02450, 02483, 02494, 02518, 02547, 02593, 02607, 02611, 02617, 02624, 02628, 02629, 02632, 02644, 02652, 02662, 02709, 02729, 02733, 02746, 02751, 02762, 02773, 02774, 02811, 02836, 02849, 02857, 02878, 02879, 02905, 02907, 02913, 02925, 02935, 02954, 02985, 03019, 03024, 03028, 03043, 03054, 03094, 03096, 03102, 03108, 03139, 03169, 03171, 03181, 03188, 03202, 03204, 03245, 03271, 03285, 03288, 03300, 03305, 03309, 03324, 03331, 03344, 03346, 03347, 03364, 03377, 03385, 03388, 03403, 03406, 03407, 03415, 03417, 03436, 03445, 03447, 03452, 03533, 03551, 03560, 03564, 03586, 03589, 03604, 03620, 03660, 03661, 03684, 03686, 03687, 03689, 03700, 03704, 03727, 03728, 03733, 03790, 03818, 03822, 03834, 03853, 03865, 03872, 03874, 03892, 03895, 03905, 03918, 03950, 03970, 03983, 04020, 04029, 04038, 04052, 04071, 04089, 04121, 04127, 04141, 04164, 04182, 04258, 04268, 04275, 04332, 04455, 04511, 04663, 04691, 04713, 04734, 04756, 04796, 04841, 04871, 05005, 05085, 05097, 05239, 05284, 05293, 05366, 05400

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ON A/C 00598, 00600, 00618, 00625, 00637, 00644, 00647, 00660

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ON A/C 00546, 00572, 00609, 00610, 00616, 00623, 00627, 00634, 00636, 00639, 00641, 00649, 00651, 00656, 00670, 00672, 00679, 00682, 00688, 00689, 00691, 00693-00695, 00697, 00700, 00711, 00717, 00719, 00721, 00723, 00728, 00729, 00732, 00736, 00738, 00740, 00742, 00744, 00752, 00757, 00769, 00773, 00779, 00785, 00800, 00813, 00817, 00829, 00831, 00840, 00845, 00853, 00860, 00869, 00875, 00880, 01058, 01062, 01071, 01078, 01090, 01091, 01126, 01129, 01131, 01149, 01164, 01167, 01191, 01230, 01249, 01324, 01325, 01346, 01392, 01414, 01434, 01453, 01456, 01479, 01483, 01498, 01501, 01520, 01543, 01549, 01567, 01570, 01582, 01659, 01662, 01683, 01685, 01693, 01709, 01714, 01738, 01746, 01752, 01766, 01774, 01796, 01800, 01810, 01815, 01819, 01820, 01824, 01828, 01833, 01839, 01870, 01875, 01897, 01923, 01959, 01976, 01982, 01990, 02002, 02013, 02026, 02028, 02039, 02047, 02087, 02095, 02369, 02373, 02464, 02474, 02618, 02641

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ON A/C 01002, 01157, 01485, 01541, 01551, 01556, 01795, 01908, 01999, 02237, 02269, 02285, 02293, 02298, 02306, 02346, 02440, 02487, 02499, 02507, 02508, 02519, 02525, 02532, 02541, 02545, 02546, 02550, 02551, 02554, 02555, 02557, 02558, 02561, 02581, 02592, 02672, 02675, 02684, 02693, 02706, 02748, 02757, 02801, 02805, 02819, 02825, 02837, 02839, 02841, 02847, 02890, 02921, 02949, 03046, 03065, 03073, 03085, 03133, 03168, 03186, 03226, 03243, 03250, 03260, 03356, 03467, 03513, 03518, 03542, 03632, 03647, 03691, 03826, 03828, 03856, 03903, 03957, 03994, 04024, 04200, 04204, 04222, 04228, 04254, 04287, 04336, 04353, 04470, 04622, 04679, 04766, 04768, 04801, 04822, 04846, 04956, 05023, 05040, 05057, 05079, 05088, 05119, 05157, 05193, 05256, 05273, 05277, 05327, 05330, 05338, 05380, 05412, 05445, 05451, 05529, 05563, 05678, 05698, 05704, 05745, 05753, 05761, 05768, 05781, 05786, 05788, 05789, 05792, 05798, 05810, 05827, 05842, 05855, 05886, 05907, 05963, 05973, 06040, 06085, 06116, 06144, 06150, 06154, 06160, 06167, 06174, 06250, 06263, 06269, 06277, 06307, 06310, 06349, 06368, 06380, 06384, 06391, 06393, 06404, 06429, 06437, 06453, 06456, 06469, 06491, 06496, 06514, 06519, 06552, 06588, 06595, 06603, 06617, 06644, 06699, 06716, 06727, 06836, 06890, 07284, 07318, 07708, 07790, 07963, 07964, 07977, 08029, 08046, 08287, 08398, 08499, 08548, 08616, 08651, 08677, 08841, 09020, 09033, 09448, 10094, 10110, 10223, 10361

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ON A/C 00686, 00690, 00748, 00759, 00783, 00788, 00798, 00804, 00825, 00843, 00847, 00850, 00858, 00862, 00867, 00871, 00873, 00882, 00893, 00898, 00944, 00948, 00952, 00965, 00980, 00989, 01022, 01031, 01082, 01115, 01118, 01142, 01193, 01197, 01211,

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(3) MSN by configuration

CONF 001

MSN	CONFIGURATION
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MSN	CONFIGURATION
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00366 00367 00368 00370 00372 00376 00378 00380 00381	
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00406 00408 00410 00415 00417 00418 00420 00428 00429	
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MSN	CONFIGURATION
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CONF 002

MSN	CONFIGURATION
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MSN	CONFIGURATION
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01908 01909 01918 01921 01922 01923 01932 01934 01936	
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01968 01969 01972 01973 01975 01976 01982 01987 01988	
01990 01993 01994 02002 02005 02011 02013 02014 02016	
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02054 02057 02060 02069 02070 02072 02073 02074 02077	
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02094 02095 02098 02101 02112 02127 02130 02135 02137	
02141 02144 02148 02153 02157 02165 02173 02180 02190	
02192 02193 02194 02210 02211 02213 02216 02222 02225	
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CONF 003

MSN	CONFIGURATION
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00159 00160 00161 00162 00164 00165 00166 00168 00169	
00171 00172 00174 00175 00178 00181 00182 00183 00185	
00192 00194 00195 00197 00198 00201 00202 00203 00205	
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00242 00243 00248 00249 00251 00252 00253 00254 00255	
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00272 00273 00275 00276 00277 00279 00281 00282 00284	
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MSN	CONFIGURATION
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00381 00382 00385 00386 00387 00388 00390 00393 00395	
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00407 00408 00409 00410 00412 00414 00415 00416 00417	
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MSN	CONFIGURATION
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06822 06823 06824 06825 06826 06827 06828 06830 06831	
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MSN	CONFIGURATION
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MSN	CONFIGURATION
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08188 08189 08190 08191 08192 08200 08201 08205 08207	
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MSN	CONFIGURATION
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08479 08481 08485 08493 08499 08500 08502 08504 08509	
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08582 08585 08590 08592 08595 08597 08599 08600 08604	
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08734 08746 08748 08763 08765 08778 08784 08786 08791	
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08838 08841 08842 08845 08847 08853 08854 08856 08858	
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08916 08925 08931 08942 08943 08947 08950 08951 08960	
08964 08974 08975 08986 08994 09009 09010 09020 09025	
09027 09028 09033 09037 09043 09052 09057 09058 09066	
09067 09081 09104 09122 09125 09126 09128 09130 09176	
09185 09279 09292 09304 09313 09319 09329 09360 09409	
09448 09485 09538 09582 09588 10003 10009 10015 10018	
10030 10085 10088 10094 10100 10110 10121 10122 10132	
10133 10140 10182 10183 10185 10223 10233 10234 10236	
10251 10252 10297 10300 10315 10361	

CONF ALL
(4) Configuration definition
CONF 001
Configuration 001

CONF 001 is applicable for all the aircraft on which:

- MOD No. 34804P8794 is not embodied
- Repair Instruction No. R53113174000 LH and R53113174001 RH are not accomplished
- Service Bulletin No. A320-25-1557 is not accomplished
- Service Bulletin No. A320-53-1215 is not accomplished.

CONF 002
Configuration 002

CONF 002 is applicable for all the aircraft on which:

- MOD No. 34804P8794 is not embodied
- Repair Instruction No. R53113174000 LH and R53113174001 RH are not accomplished
- Service Bulletin No. A320-25-1557 rev00 or rev01 is accomplished
- Service Bulletin No. A320-53-1215 is not accomplished.

CONF 003
Configuration 003

CONF 003 is applicable for all the aircraft on which:

- MOD No. 34804P8794 is not embodied

- Service Bulletin No. A320-25-1557 is accomplished
 - Service Bulletin No. A320-53-1215 is accomplished.
- Or
- MOD No. 34804P8794 is not embodied
 - Repair Instruction No. R53113174000 LH and R53113174001 RH are accomplished
 - Service Bulletin No. A320-53-1215 is accomplished.

CONF 004**Configuration 004**

CONF 004 is applicable for all the aircraft on which:

- MOD No. 34804P8794 is embodied.

CONF ALL**B. CONCURRENT REQUIREMENTS**

None

C. REASON**(1) History**

Further to in-service reports of worn Electronics Rack FIN 80VU lower lateral fittings, an inspection of the Electronics Rack FIN 80VU lower lateral fittings and the Electronics Rack FIN 80VU lower central support was required.

Service Bulletin No. Ref. SB A320-25-1555 was issued to inspect the Electronics Rack FIN 80VU and provide recommendations in case of findings. After Service Bulletin No. Ref. SB A320-25-1555 publication some findings have been reported on aircraft not listed in its applicability because they were post production MOD No. 34804P8794.

There have been reports as well of findings in aircraft having accomplished Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.

These Service Bulletins were a terminating action of Service Bulletin No. Ref. SB A320-25-1555.

NOTE: Service Bulletin No. Ref. SB A320-25-1557 is the corrective action for the lower lateral fittings.

NOTE: Service Bulletin No. Ref. SB A320-53-1215 is the corrective action for the central fitting.

Based on the above information, the following corrective actions cannot be considered as terminative actions anymore:

- Production MOD No. 34804P8794
- Service Bulletin No. Ref. SB A320-25-1557
- Service Bulletin No. Ref. SB A320-53-1215.

(2) Objective/Action

The objective of this Service Bulletin is to:

- Do a Special Detailed Inspection (SDI) of the complete Electronics Rack FIN 80VU (lower lateral fittings, the central support, the upper fittings, the central post, the

shelves attachments, the lateral pyramid structure) for potential damage and provide an appropriate instruction in case of findings

- Extend the applicability in order to cover all aircraft defined in the applicability of Service Bulletin No. Ref. SB A320-25-1555 as well as post production MOD No. 34804P8794, post Service Bulletin No. Ref. SB A320-25-1557 and post Service Bulletin No. Ref. SB A320-53-1215
- Reflect that the production MOD No. 34804P8794 and the repair provided in Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215 cannot be considered as a terminative action
- Call out a new repair procedure on lower lateral fittings for aircraft post MOD No. 34804P8794, post Service Bulletin No. Ref. SB A320-25-1557 and post Service Bulletin No. Ref. SB A320-53-1215 via Repair Instruction (RI) R531-13174.

(3) Advantages

The accomplishment of this Service Bulletin will contribute to detect potential damage on the Electronics Rack FIN 80VU and its attachments to the fuselage.

NOTE: This Service Bulletin supersedes Ref. SB A320-25-1555.

(4) Operational/Maintenance Consequences

None

D. DESCRIPTION

To accomplish this Service Bulletin it is necessary to:

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

CONF 001

1 Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU

CONF 002

1 Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU

CONF 003

1 Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU

CONF 004

1 Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU

2 Do an SDI of the Rods and Fittings of the Lower Central Support of the Electronics Rack FIN 80VU

CONF 001

3 Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU

CONF 002

3 Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU

CONF 003

3 Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU

CONF ALL

4 Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU

5 Complete the Inspection Report sheet

CONF 002-004

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

CONF 002 003

- 1 Accomplish the Repair Instruction

CONF 004

- 1 Accomplish the Repair Instruction

CONF ALL

E. COMPLIANCE

CONF 001

(1) Classification

MANDATORY: Service Bulletin must be accomplished.

(2) Accomplishment Timescale

This Service Bulletin has been classified Mandatory. It is expected that an Airworthiness Directive (AD) published by the European Aviation Safety Agency (EASA) will be issued to confirm this classification. It is anticipated that the AD will confirm the 19500 FC given but reference should be made to the AD for confirmation of the mandated accomplishment timescale.

Table 01: CONF 001 - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Lower Lateral Fittings in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A	19500 Flight Cycle(s) from aircraft first flight	500 Flight Cycle(s)	None
If one or both fittings are broken or if one or both bolts are missing	Accomplish the Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Before next Flight Cycle(s)	None	None
Service Bulletins accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the fittings are not broken	Check for contact between the Electronics Rack	Before next Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
and the bolts are not missing	FIN 80VU and the structure			
If the Electronics Rack FIN 80VU is in contact with the structure	Accomplish the Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If the Electronics Rack FIN 80VU is in contact with the structure	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Service Bulletin No. A320-25-1557 and Service Bulletin No. A320-53-1215 are accomplished
Service Bulletins accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the Electronics Rack FIN 80VU is not in contact with the structure	Check for bush migration or burred material	Before next Flight Cycle(s)	None	None
If the bush has migrated or material is burred	Accomplish the Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If the bush has migrated or material is burred	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Service Bulletin No. A320-25-1557 and Service Bulletin No. A320-53-1215 are accomplished

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
Service Bulletins accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the bush has not migrated and the material is not burred	Accomplish Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Within 500 Flight Cycle(s)	None	None
	OR	None	None	None
If the bush has not migrated and the material is not burred	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) from last inspection
Service Bulletins accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

Table 02: CONF 001 - Do an SDI of the Lower Central Support of the Electronics Rack
 Electronics Rack FIN 80VU.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Lower Central Support in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A	19500 Flight Cycle(s) from aircraft first flight	500 Flight Cycle(s)	None
If any crack(s) is(are) found on the upper sheet	Accomplish the Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Before next Flight Cycle(s)	None	None
Service Bulletins accomplished	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If no crack(s) is(are) found on the upper sheet	Check the front/rear sheets for crack(s)	Before next Flight Cycle(s)	None	None
If crack(s) is(are) found on the front/rear sheets with more than 40 mm (1.575 in.)	Accomplish the Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Before next Flight Cycle(s)	None	None
Service Bulletins accomplished	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If crack(s) is(are) found on the front/rear sheets with less than or equal to 40 mm (1.575 in.)	Accomplish the Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
If crack(s) is(are) found on the front/rear sheets with less than or equal to 40 mm (1.575 in.)	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Service Bulletin No. A320-25-1557 and Service Bulletin No. A320-53-1215 are accomplished
Service Bulletins accomplished	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If no crack(s) is(are) found on the front/rear sheets	Accomplish the Service Bulletin No. Ref. SB A320-53-1215 and Service Bulletin No. Ref. SB A320-25-1557	Within 500 Flight Cycle(s)	None	None
	OR	None	None	None
If no crack(s) is(are) found on the front/rear sheets	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) from last inspection
Service Bulletins accomplished	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-	Within 90 day(s) after completion of this Service Bulletin	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	A-25-XX-1BKJ-03ZZZ-90AZ-A			

Table 03: CONF 001 - Do an SDI of the Central Post and Shelves attachment of the Electronics Rack Electronics Rack FIN 80VU

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Central Post and Shelves in accordance with Ref. Task A320-A-25-XX-1BKJ-04ZZZ-346Z-A	19500 Flight Cycle(s) from aircraft first flight	500 Flight Cycle(s)	None
If any crack(s) is(are) found on central post	Replace the damaged fittings with new ones which have the same PN	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on central post	Do an SDI of the upper fittings	Before next Flight Cycle(s)	None	None
If the upper fittings are found loose	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If the upper fittings are not found loose	Do an SDI of the lateral shelves	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral shelves	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the lateral shelves	Do an SDI of the lateral pyramid structure side	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral pyramid structure side	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the lateral pyramid structure side	Repeat the inspection of the Central Post and Shelves attachment in accordance with Ref. Task A320-	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	A-25-XX-1BKJ-04ZZZ-346Z-A			
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

CONF 002
(1) Classification

MANDATORY: Service Bulletin must be accomplished.

(2) Accomplishment Timescale

This Service Bulletin has been classified Mandatory. It is expected that an Airworthiness Directive (AD) published by the European Aviation Safety Agency (EASA) will be issued to confirm this classification. It is anticipated that the AD will confirm the 19500 FC given but reference should be made to the AD for confirmation of the mandated accomplishment timescale.

Table 01: CONF 002 - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Lower Lateral Fittings in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A	19500 Flight Cycle(s) from accomplishment of Service Bulletin No. A320-25-1557	500 Flight Cycle(s)	None
If one or both fittings are broken or if one or both bolts are missing	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Service Bulletin No. Ref. SB A320-53-1215	Before next Flight Cycle(s)	None	None
Repair Instruction and Service Bulletin accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-	After 19500 Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	25-XX-1BKJ-01ZZZ-93BZ-A			
If the fittings are not broken and the bolts are not missing	Check for contact between the Electronics Rack FIN 80VU and the structure	Before next Flight Cycle(s)	None	None
If the Electronics Rack FIN 80VU is in contact with the structure	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Service Bulletin No. Ref. SB A320-53-1215	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If the Electronics Rack FIN 80VU is in contact with the structure	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Repair Instruction R53113174000 and Repair Instruction R53113174001 and Service Bulletin No. A320-53-1215 are accomplished
Repair Instruction and Service Bulletin accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the Electronics Rack FIN 80VU is not in contact with the structure	Check for bush migration or burred material	Before next Flight Cycle(s)	None	None
If the bush has migrated or material is burred	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Service Bulletin No. Ref. SB A320-53-1215	Within 4500 Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	AND	None	None	None
If the bush has migrated or material is burred	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Repair Instruction R53113174000 and Repair Instruction R53113174001 and Service Bulletin No. A320-53-1215 are accomplished
Repair Instruction and Service Bulletin accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the bush has not migrated and the material is not burred	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Service Bulletin No. Ref. SB A320-53-1215	Within 500 Flight Cycle(s)	None	None
	OR	None	None	None
If the bush has not migrated and the material is not burred	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) from last inspection
Repair Instruction and Service Bulletin accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
All inspections completed	Complete and send the	Within 90 day(s) after completion	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	of this Service Bulletin		

Table 02: CONF 002 - Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Lower Central Support in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A	19500 Flight Cycle(s) from accomplishment of Service Bulletin No. A320-25-1557	500 Flight Cycle(s)	None
If any crack(s) is(are) found on the upper sheet	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Ref. SB A320-53-1215	Before next Flight Cycle(s)	None	None
Repair Instruction and Service Bulletin accomplished	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If no crack(s) is(are) found on the upper sheet	Check the front/rear sheets for crack(s)	Before next Flight Cycle(s)	None	None
If crack(s) is(are) found on the front/rear sheets with more than 40 mm (1.575 in.)	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Ref. SB A320-53-1215	Before next Flight Cycle(s)	None	None
Repair Instruction and Service	Repeat the inspection of the Lower	After 19500 Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
Bulletin accomplished	Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A			
If crack(s) is(are) found on the front/rear sheets with less than or equal to 40 mm (1.575 in.)	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Ref. SB A320-53-1215	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If crack(s) is(are) found on the front/rear sheets with less than or equal to 40 mm (1.575 in.)	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Repair Instruction R53113174000 and Repair Instruction R53113174001 and Service Bulletin No. A320-53-1215 are accomplished
Repair Instruction and Service Bulletin accomplished	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If no crack(s) is(are) found on the front/rear sheets	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001 and Ref. SB A320-53-1215	Within 500 Flight Cycle(s)	None	None
	OR	None	None	None
If no crack(s) is(are) found on the front/rear sheets	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) from last inspection

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	25-XX-1BKJ-01ZZZ-93BZ-A			
Repair Instruction and Service Bulletin accomplished	Repeat the inspection of the Lower Central Support in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

Table 03: CONF 002 - Do an SDI of the Central Post and Shelves attachment of the Electronics Rack Electronics Rack FIN 80VU

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Central Post and Shelves in accordance with Ref. Task A320-A-25-XX-1BKJ-04ZZZ-346Z-A	19500 Flight Cycle(s) from accomplishment of Service Bulletin No. A320-25-1557	500 Flight Cycle(s)	None
If any crack(s) is(are) found on central post	Replace the damaged fittings with new ones which have the same PN	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on central post	Do an SDI of the upper fittings	Before next Flight Cycle(s)	None	None
If the upper fittings are found loose	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If the upper fittings are not found loose	Do an SDI of the lateral shelves	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral shelves	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
If no crack(s) is(are) found on the lateral shelves	Do an SDI of the lateral pyramid structure side	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral pyramid structure side	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the lateral pyramid structure side	Repeat the inspection of the Central Post and Shelves attachment in accordance with Ref. Task A320-A-25-XX-1BKJ-04ZZZ-346Z-A	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

CONF 003
(1) Classification

MANDATORY: Service Bulletin must be accomplished.

(2) Accomplishment Timescale

This Service Bulletin has been classified Mandatory. It is expected that an Airworthiness Directive (AD) published by the European Aviation Safety Agency (EASA) will be issued to confirm this classification. It is anticipated that the AD will confirm the 19500 FC given but reference should be made to the AD for confirmation of the mandated accomplishment timescale.

Table 01: CONF 003 - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Lower Lateral Fittings in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A	19500 Flight Cycle(s) from accomplishment of Service Bulletin No. A320-25-1557 and Service Bulletin No.	500 Flight Cycle(s)	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
		A320-53-1215 or 500 Flight Cycle(s) from the first issue of this SB		
If one or both fittings are broken or if one or both bolts are missing	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001	Before next Flight Cycle(s)	None	None
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the fittings are not broken and the bolts are not missing	Check for contact between the Electronics Rack FIN 80VU and the structure	Before next Flight Cycle(s)	None	None
If the Electronics Rack FIN 80VU is in contact with the structure	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If the Electronics Rack FIN 80VU is in contact with the structure	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Repair Instruction R53113174000 and Repair Instruction R53113174001 are accomplished
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
If the Electronics Rack FIN 80VU is not in contact with the structure	Check for bush migration or burred material	Before next Flight Cycle(s)	None	None
If the bush has migrated or material is burred	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If the bush has migrated or material is burred	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Repair Instruction R53113174000 and Repair Instruction R53113174001 are accomplished
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the bush has not migrated and the material is not burred	Accomplish the Repair Instruction R53113174000 and Repair Instruction R53113174001	Within 1000 Flight Cycle(s)	None	None
	OR	None	None	None
If the bush has not migrated and the material is not burred	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance	After 19500 Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A			
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

Table 02: CONF 003 - Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU for cracks using borescope.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Lower Central Support in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A	19500 Flight Cycle(s) from accomplishment of Service Bulletin No. A320-25-1557 and Service Bulletin No. A320-53-1215 or 500 Flight Cycle(s) from the first issue of this SB	500 Flight Cycle(s)	None
If any crack(s) is(are) found on the upper sheet	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the upper sheet	Check the front/rear sheets for crack(s)	Before next Flight Cycle(s)	None	None
If crack(s) are found on the front/rear sheets	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) are found on the front/rear sheets	Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)
All inspections completed	Complete and send the Inspection Report sheet in accordance with	Within 90 day(s) after completion of this Service Bulletin	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A			

Table 03: CONF 003 - Do an SDI of the Central Post and Shelves attachment of the Electronics Rack Electronics Rack FIN 80VU

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Central Post and Shelves in accordance with Ref. Task A320-A-25-XX-1BKJ-04ZZZ-346Z-A	19500 Flight Cycle(s) from accomplishment of Service Bulletin No. A320-25-1557 and Service Bulletin No. A320-53-1215 or 500 Flight Cycle(s) from the first issue of this SB	500 Flight Cycle(s)	None
If any crack(s) is(are) found on central post	Replace the damaged fittings with new ones which have the same PN	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on central post	Do an SDI of the upper fittings	Before next Flight Cycle(s)	None	None
If the upper fittings are found loose	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If the upper fittings are not found loose	Do an SDI of the lateral shelves	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral shelves	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the lateral shelves	Do an SDI of the lateral pyramid structure side	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral pyramid structure side	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
If no crack(s) is(are) found on the lateral pyramid structure side	Repeat the inspection of the Central Post and Shelves attachment in accordance with Ref. Task A320-A-25-XX-1BKJ-04ZZZ-346Z-A	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

CONF 004
(1) Classification

MANDATORY: Service Bulletin must be accomplished.

(2) Accomplishment Timescale

This Service Bulletin has been classified Mandatory. It is expected that an Airworthiness Directive (AD) published by the European Aviation Safety Agency (EASA) will be issued to confirm this classification. It is anticipated that the AD will confirm the 19500 FC given but reference should be made to the AD for confirmation of the mandated accomplishment timescale.

Table 01: CONF 004 - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Lower Lateral Fittings in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A	19500 Flight Cycle(s) from aircraft first flight or 500 Flight Cycle(s) from the first issue of this SB	500 Flight Cycle(s)	None
If one or both fittings are broken or if one or both bolts are missing	Accomplish the Repair Instruction R53113174002 and Repair Instruction R53113174003	Before next Flight Cycle(s)	None	None
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings	After 19500 Flight Cycle(s)	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A			
If the fittings are not broken and the bolts are not missing	Check for contact between the Electronics Rack FIN 80VU and the structure	Before next Flight Cycle(s)	None	None
If the Electronics Rack FIN 80VU is in contact with the structure	Accomplish the Repair Instruction R53113174002 and Repair Instruction R53113174003	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If the Electronics Rack FIN 80VU is in contact with the structure	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Repair Instruction R53113174002 and Repair Instruction R53113174003 are accomplished
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the Electronics Rack FIN 80VU is not in contact with the structure	Check for bush migration or burred material	Before next Flight Cycle(s)	None	None
If the bush has migrated or material is burred	Accomplish the Repair Instruction R53113174002 and Repair Instruction R53113174003	Within 4500 Flight Cycle(s)	None	None
	AND	None	None	None
If the bush has migrated	Repeat the inspection of the Lower	Within 500 Flight Cycle(s)	None	500 Flight Cycle(s) until Repair Instruction

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
or material is burred	Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A			R53113174002 and Repair Instruction R53113174003 are accomplished
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
If the bush has not migrated and the material is not burred	Accomplish the Repair Instruction R53113174002 and Repair Instruction R53113174003	Within 1000 Flight Cycle(s)	None	None
	OR	None	None	None
If the bush has not migrated and the material is not burred	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)
Repair Instruction accomplished	Repeat the inspection of the Lower Lateral Fittings in accordance with Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93BZ-A	After 19500 Flight Cycle(s)	None	None
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

Table 02: CONF 004 - Do an SDI of the Rods and Fittings and the Lower Central Support of the Electronics Rack FIN 80VU.

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the 2 rods and their fittings located behind Lower Central Support in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A	19500 Flight Cycle(s) from aircraft first flight or 500 Flight Cycle(s) from the first issue of this SB	500 Flight Cycle(s)	None
If the Rods and Fittings are not found in good condition	Contact AIRBUS and follow their instructions AND Do the inspection of the Lower Central Support in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-346Z-A	Before next Flight Cycle(s)	None	None
If the Rods and Fittings are found in good condition	Do the inspection of the Lower Central Support in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-346Z-A	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the upper sheet	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the upper sheet	Check the front/rear sheets for crack(s)	Before next Flight Cycle(s)	None	None
If crack(s) are found on the front/rear sheets	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) are found on the front/rear sheets	Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-	Within 90 day(s) after completion of this Service Bulletin	None	None

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	A-25-XX-1BKJ-03ZZZ-90AZ-A			

Table 03: CONF 004 - Do an SDI of the Central Post and Shelves attachment of the Electronics Rack Electronics Rack FIN 80VU

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
All aircraft	Do the inspection of the Central Post and Shelves in accordance with Ref. Task A320-A-25-XX-1BKJ-04ZZZ-346Z-A	19500 Flight Cycle(s) from aircraft first flight or 500 Flight Cycle(s) from the first issue of this SB	500 Flight Cycle(s)	None
If any crack(s) is(are) found on central post	Replace the damaged fittings with new ones which have the same PN	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on central post	Do an SDI of the upper fittings	Before next Flight Cycle(s)	None	None
If the upper fittings are found loose	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If the upper fittings are not found loose	Do an SDI of the lateral shelves	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral shelves	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the lateral shelves	Do an SDI of the lateral pyramid structure side	Before next Flight Cycle(s)	None	None
If any crack(s) is(are) found on the lateral pyramid structure side	Contact AIRBUS and follow their instructions	Before next Flight Cycle(s)	None	None
If no crack(s) is(are) found on the lateral pyramid structure side	Repeat the inspection of the Central Post and Shelves attachment in accordance with Ref. Task A320-	Within 1000 Flight Cycle(s)	None	1000 Flight Cycle(s)

CONDITIONS	ACTION	COMPLIANCE TIME		REPETITIVE INTERVAL
		THRESHOLD	GRACE PERIOD	
	A-25-XX-1BKJ-04ZZZ-346Z-A			
All inspections completed	Complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A	Within 90 day(s) after completion of this Service Bulletin	None	None

CONF ALL
F. APPROVAL

The technical content of this document is approved under the authority of Design Organisation Approval No. EASA.21J.031.

If an aircraft listed in the applicability has a modification or repair embodied that is not of AIRBUS origin, and which affects the content of this Service Bulletin, the operator is responsible for obtaining approval by its airworthiness authority for any adaptation necessary before incorporation of the Service Bulletin.

G. MANPOWER
CONF 001

The manpower estimates given in this Service Bulletin are based on the direct labor cost to do the work. These estimates assume that the work will be done by experienced personnel, and may need to be revised upwards to suit operator's circumstances. The estimates do not include the time to prepare, plan or inspect the work. Manufacture and procurement of parts and tools, drying times for paints, sealants, etc., and general administration work are also not included.

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

PREPARATION		
Job Set-up	1.50 h	
PROCEDURE		
Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	0.75 h	
Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU	1.00 h	
Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	1.50 h	
Complete the Inspection Report sheet	0.25 h	
PREPARATION FOR TEST		
None		
TEST		
None		
CLOSE-UP		

Close-up	1.50 h	
TOTAL MAN-HOURS	6.50 h	
ELAPSED TIME (HOURS)	3.25 h	

NOTE: The minimum man-hours necessary to accomplish this Service Bulletin are those for the Inspection. Depending on the inspection result, additional man-hours can apply.

CONF 002

The manpower estimates given in this Service Bulletin are based on the direct labor cost to do the work. These estimates assume that the work will be done by experienced personnel, and may need to be revised upwards to suit operator's circumstances. The estimates do not include the time to prepare, plan or inspect the work. Manufacture and procurement of parts and tools, drying times for paints, sealants, etc., and general administration work are also not included.

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

PREPARATION		
Job Set-up	1.50 h	
PROCEDURE		
Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	0.75 h	
Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU	1.00 h	
Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	1.50 h	
Complete the Inspection Report sheet	0.25 h	
PREPARATION FOR TEST		
None		
TEST		
None		
CLOSE-UP		
Close-up	1.50 h	
TOTAL MAN-HOURS	6.50 h	
ELAPSED TIME (HOURS)	3.25 h	

NOTE: The minimum man-hours necessary to accomplish this Service Bulletin are those for the Inspection. Depending on the inspection result, additional man-hours can apply.

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

PREPARATION		
Job Set-up	8.25 h	
Remove the ducts	2.00 h	

Remove structure and struts	5.75 h	
Remove the support	1.00 h	
PROCEDURE		
Accomplish the Repair Instruction	67.00 h	
PREPARATION FOR TEST		
Preparation for Test	4.00 h	
TEST		
Test	20.00 h	
CLOSE-UP		
Install the support	1.00 h	
Install structure and struts	5.75 h	
Install the ducts	2.00 h	
Close-up	5.25 h	
TOTAL MAN-HOURS	122.00 h	
ELAPSED TIME (HOURS)	73.50 h	

CONF 003

The manpower estimates given in this Service Bulletin are based on the direct labor cost to do the work. These estimates assume that the work will be done by experienced personnel, and may need to be revised upwards to suit operator's circumstances. The estimates do not include the time to prepare, plan or inspect the work. Manufacture and procurement of parts and tools, drying times for paints, sealants, etc., and general administration work are also not included.

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

PREPARATION		
Job Set-up	1.50 h	
PROCEDURE		
Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	0.75 h	
Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU	1.00 h	
Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	1.50 h	
Complete the Inspection Report sheet	0.25 h	
PREPARATION FOR TEST		
None		
TEST		
None		
CLOSE-UP		
Close-up	1.50 h	
TOTAL MAN-HOURS	6.50 h	
ELAPSED TIME (HOURS)	3.25 h	

NOTE: The minimum man-hours necessary to accomplish this Service Bulletin are those for the Inspection. Depending on the inspection result, additional man-hours can apply.

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

PREPARATION		
Job Set-up	8.25 h	
Remove the ducts	2.00 h	
Remove structure and struts	5.75 h	
Remove the support	1.00 h	
PROCEDURE		
Accomplish the Repair Instruction	67.00 h	
PREPARATION FOR TEST		
Preparation for Test	4.00 h	
TEST		
Test	20.00 h	
CLOSE-UP		
Install the support	1.00 h	
Install structure and struts	5.75 h	
Install the ducts	2.00 h	
Close-up	5.25 h	
TOTAL MAN-HOURS	122.00 h	
ELAPSED TIME (HOURS)	73.50 h	

CONF 004

The manpower estimates given in this Service Bulletin are based on the direct labor cost to do the work. These estimates assume that the work will be done by experienced personnel, and may need to be revised upwards to suit operator's circumstances. The estimates do not include the time to prepare, plan or inspect the work. Manufacture and procurement of parts and tools, drying times for paints, sealants, etc., and general administration work are also not included.

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

PREPARATION		
Job Set-up	1.50 h	
PROCEDURE		
Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU	0.75 h	
Do an SDI of the Rods and Fittings of the Lower Central Support of the Electronics Rack FIN 80VU	2.00 h	
Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU	1.50 h	
Complete the Inspection Report sheet	0.25 h	

PREPARATION FOR TEST		
None		
TEST		
None		
CLOSE-UP		
Close-up	1.50 h	
TOTAL MAN-HOURS	7.50 h	
ELAPSED TIME (HOURS)	3.75 h	

NOTE: The minimum man-hours necessary to accomplish this Service Bulletin are those for the Inspection. Depending on the inspection result, additional man-hours can apply.

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

PREPARATION		
Job Set-up	8.25 h	
Remove the ducts	2.00 h	
Remove structure and struts	5.75 h	
Remove the support	1.00 h	
PROCEDURE		
Accomplish the Repair Instruction	67.00 h	
PREPARATION FOR TEST		
Preparation for Test	4.00 h	
TEST		
Test	20.00 h	
CLOSE-UP		
Install the support	1.00 h	
Install structure and struts	5.75 h	
Install the ducts	2.00 h	
Close-up	5.25 h	
TOTAL MAN-HOURS	122.00 h	
ELAPSED TIME (HOURS)	73.50 h	

CONF ALL

H. WEIGHT AND BALANCE

Not changed

I. ELECTRICAL LOAD DATA

Not changed

J. REFERENCES

CONF 001

(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT MAINTENANCE MANUAL (AMM)	
06-20-00, Page Block 001	06-31-53, Page Block 001
Task 12-34-24-869-002	Task 25-54-11-000-001 Task 25-54-11-400-001
Task 52-30-00-860-001	Task 52-41-00-010-002
Task 52-30-00-860-002	Task 52-41-00-410-002
ELECTRICAL STANDARD PRACTICES MANUAL (ESPM)	
20-55-00	
IN SERVICE INFORMATION (ISI)	
00.00.00179	
NON DESTRUCTIVE TEST MANUAL (NTM)	
51-90-00	
SERVICE BULLETIN (SB)	
A320-25-1555	A320-25-1557
A320-53-1215	
TECHNICAL FOLLOW UP (TFU)	
25.70.00.007	

CONSUMABLE MATERIAL LIST (CML)

CONF 002
(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT MAINTENANCE MANUAL (AMM)	
06-20-00, Page Block 001	06-31-53, Page Block 001
Task 12-34-24-869-002	Task 25-54-11-000-001 Task 25-54-11-400-001
Task 52-30-00-860-001	Task 52-41-00-010-002
Task 52-30-00-860-002	Task 52-41-00-410-002
ELECTRICAL STANDARD PRACTICES MANUAL (ESPM)	
20-55-00	
IN SERVICE INFORMATION (ISI)	
00.00.00179	
NON DESTRUCTIVE TEST MANUAL (NTM)	
51-90-00	

SERVICE BULLETIN (SB)	
A320-25-1555	A320-25-1557
A320-53-1215	
TECHNICAL FOLLOW UP (TFU)	
25.70.00.007	
REPAIR DRAWINGS	SHEET(S)
R53113174000	all sheets
R53113174001	all sheets

STANDARDS MANUAL (SM)

CONSUMABLE MATERIAL LIST (CML)

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

AIRCRAFT MAINTENANCE MANUAL (AMM)	
06-20-00, Page Block 000	06-31-53, Page Block 000
Task 12-34-24-869-002	Task 20-21-11-911-001
20-28-00, Page Block 200	Task 21-20-00-010-001
	Task 21-20-00-410-001
21-26-00, Page Block 500	21-26-34, Page Block 400
	21-26-34, Page Block 401
22-66-34, Page Block 400	22-83-34, Page Block 400
22-66-34, Page Block 401	22-83-34, Page Block 401
22-96-00, Page Block 500	22-97-00, Page Block 500
23-11-33, Page Block 400	23-12-33, Page Block 400
23-11-33, Page Block 401	23-12-33, Page Block 401
23-24-34, Page Block 400	23-28-34, Page Block 400
23-24-34, Page Block 401	23-28-34, Page Block 401
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23-34-34, Page Block 401	23-51-34, Page Block 401
23-72-31, Page Block 400	23-73-34, Page Block 400
23-72-31, Page Block 401	23-73-34, Page Block 401
Task 24-38-51-000-001	Task 25-54-11-400-001
Task 25-54-12-000-003	26-12-34, Page Block 400

AIRCRAFT MAINTENANCE MANUAL (AMM)	
Task 25-54-12-400-003	26-12-34, Page Block 401
26-13-34, Page Block 400	27-51-34, Page Block 400
26-13-34, Page Block 401	27-51-34, Page Block 401
27-93-34, Page Block 400	27-94-34, Page Block 400
27-93-34, Page Block 401	27-94-34, Page Block 401
27-95-34, Page Block 400	31-32-34, Page Block 400
27-95-34, Page Block 401	31-32-34, Page Block 401
31-33-34, Page Block 400	31-33-52, Page Block 400
31-33-34, Page Block 401	31-33-52, Page Block 401
31-36-34, Page Block 400	31-36-52, Page Block 400
31-36-34, Page Block 401	31-36-52, Page Block 401
31-38-34, Page Block 400	31-50-00, Page Block 500
31-38-34, Page Block 401	
31-53-34, Page Block 400	31-55-34, Page Block 400
31-53-34, Page Block 401	31-55-34, Page Block 401
31-60-00, Page Block 500	31-63-34, Page Block 400
	31-63-34, Page Block 401
33-13-00, Page Block 500	33-14-00, Page Block 500
34-35-34, Page Block 400	34-43-34, Page Block 400
34-35-34, Page Block 401	34-43-34, Page Block 401
34-48-34, Page Block 400	34-51-33, Page Block 400
34-48-34, Page Block 401	34-51-33, Page Block 401
34-52-33, Page Block 400	34-52-35, Page Block 400
34-52-33, Page Block 401	34-52-35, Page Block 401
34-53-31, Page Block 400	34-55-31, Page Block 400
34-53-31, Page Block 401	34-55-31, Page Block 401
46-21-34, Page Block 400	Task 52-30-00-860-002
46-21-34, Page Block 401	
Task 52-41-00-410-002	Task 53-12-13-000-001
	Task 53-12-13-400-001
Task 53-22-14-000-001	73-25-34, Page Block 400
Task 53-22-14-400-001	73-25-34, Page Block 401
77-32-34, Page Block 400	
77-32-34, Page Block 401	
ELECTRICAL STANDARD PRACTICES MANUAL (ESPM)	
20-51-20	20-52-10
20-54-50	20-55-00

SERVICE BULLETIN (SB)	
A320-25-1555	A320-25-1557
A320-53-1215	
STRUCTURAL REPAIR MANUAL (SRM)	
51-24-00	51-42-00
51-43-00	51-44-00
51-46-00	51-47-00
51-75-12	
TECHNICAL FOLLOW UP (TFU)	
25.70.00.007	
REPAIR DRAWINGS	SHEET(S)
R53113174000	all sheets
R53113174001	all sheets

STANDARDS MANUAL (SM)

CONSUMABLE MATERIAL LIST (CML)

CONF 003
(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT MAINTENANCE MANUAL (AMM)	
06-20-00, Page Block 001	06-31-53, Page Block 001
Task 12-34-24-869-002	Task 25-54-11-000-001
	Task 25-54-11-400-001
Task 52-30-00-860-001	Task 52-41-00-010-002
Task 52-30-00-860-002	Task 52-41-00-410-002
ELECTRICAL STANDARD PRACTICES MANUAL (ESPM)	
20-55-00	
IN SERVICE INFORMATION (ISI)	
00.00.00179	
NON DESTRUCTIVE TEST MANUAL (NTM)	
51-90-00	
SERVICE BULLETIN (SB)	
A320-25-1555	A320-25-1557
A320-53-1215	
TECHNICAL FOLLOW UP (TFU)	
25.70.00.007	

REPAIR DRAWINGS	SHEET(S)
R53113174000	all sheets
R53113174001	all sheets

STANDARDS MANUAL (SM)

CONSUMABLE MATERIAL LIST (CML)

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

AIRCRAFT MAINTENANCE MANUAL (AMM)	
06-20-00, Page Block 000	06-31-53, Page Block 000
Task 12-34-24-869-002	Task 20-21-11-911-001
20-28-00, Page Block 200	Task 21-20-00-010-001
	Task 21-20-00-410-001
21-26-00, Page Block 500	21-26-34, Page Block 400
	21-26-34, Page Block 401
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22-66-34, Page Block 401	22-83-34, Page Block 401
22-96-00, Page Block 500	22-97-00, Page Block 500
23-11-33, Page Block 400	23-12-33, Page Block 400
23-11-33, Page Block 401	23-12-33, Page Block 401
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23-24-34, Page Block 401	23-28-34, Page Block 401
23-32-34, Page Block 400	23-32-35, Page Block 400
23-32-34, Page Block 401	23-32-35, Page Block 401
23-33-31, Page Block 400	23-33-37, Page Block 400
23-33-31, Page Block 401	23-33-37, Page Block 401
23-33-39, Page Block 400	23-33-52, Page Block 400
23-33-39, Page Block 401	23-33-52, Page Block 401
23-34-34, Page Block 400	23-51-34, Page Block 400
23-34-34, Page Block 401	23-51-34, Page Block 401
23-72-31, Page Block 400	23-73-34, Page Block 400
23-72-31, Page Block 401	23-73-34, Page Block 401
Task 24-38-51-000-001	Task 25-54-11-400-001
Task 25-54-12-000-003	26-12-34, Page Block 400
Task 25-54-12-400-003	26-12-34, Page Block 401
26-13-34, Page Block 400	27-51-34, Page Block 400
26-13-34, Page Block 401	27-51-34, Page Block 401
27-93-34, Page Block 400	27-94-34, Page Block 400
27-93-34, Page Block 401	27-94-34, Page Block 401
27-95-34, Page Block 400	31-32-34, Page Block 400

AIRCRAFT MAINTENANCE MANUAL (AMM)	
27-95-34, Page Block 401	31-32-34, Page Block 401
31-33-34, Page Block 400	31-33-52, Page Block 400
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31-36-34, Page Block 400	31-36-52, Page Block 400
31-36-34, Page Block 401	31-36-52, Page Block 401
31-38-34, Page Block 400	31-50-00, Page Block 500
31-38-34, Page Block 401	
31-53-34, Page Block 400	31-55-34, Page Block 400
31-53-34, Page Block 401	31-55-34, Page Block 401
31-60-00, Page Block 500	31-63-34, Page Block 400
	31-63-34, Page Block 401
33-13-00, Page Block 500	33-14-00, Page Block 500
34-35-34, Page Block 400	34-43-34, Page Block 400
34-35-34, Page Block 401	34-43-34, Page Block 401
34-48-34, Page Block 400	34-51-33, Page Block 400
34-48-34, Page Block 401	34-51-33, Page Block 401
34-52-33, Page Block 400	34-52-35, Page Block 400
34-52-33, Page Block 401	34-52-35, Page Block 401
34-53-31, Page Block 400	34-55-31, Page Block 400
34-53-31, Page Block 401	34-55-31, Page Block 401
46-21-34, Page Block 400	Task 52-30-00-860-002
46-21-34, Page Block 401	
Task 52-41-00-410-002	Task 53-12-13-000-001
	Task 53-12-13-400-001
Task 53-22-14-000-001	73-25-34, Page Block 400
Task 53-22-14-400-001	73-25-34, Page Block 401
77-32-34, Page Block 400	
77-32-34, Page Block 401	
ELECTRICAL STANDARD PRACTICES MANUAL (ESPM)	
20-51-20	20-52-10
20-54-50	20-55-00
SERVICE BULLETIN (SB)	
A320-25-1555	A320-25-1557
A320-53-1215	
STRUCTURAL REPAIR MANUAL (SRM)	
51-24-00	51-42-00
51-43-00	51-44-00
51-46-00	51-47-00

STRUCTURAL REPAIR MANUAL (SRM)	
51-75-12	
TECHNICAL FOLLOW UP (TFU)	
25.70.00.007	
REPAIR DRAWINGS	SHEET(S)
R53113174000	all sheets
R53113174001	all sheets

STANDARDS MANUAL (SM)

CONSUMABLE MATERIAL LIST (CML)

CONF 004
(1) Inspection of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

AIRCRAFT MAINTENANCE MANUAL (AMM)	
06-20-00, Page Block 001	06-31-53, Page Block 001
Task 12-34-24-869-002	Task 25-54-11-000-001
	Task 25-54-11-400-001
Task 52-30-00-860-001	Task 52-41-00-010-002
Task 52-30-00-860-002	Task 52-41-00-410-002

ELECTRICAL STANDARD PRACTICES MANUAL (ESPM)	
20-55-00	

IN SERVICE INFORMATION (ISI)	
00.00.00179	

NON DESTRUCTIVE TEST MANUAL (NTM)	
51-90-00	

SERVICE BULLETIN (SB)	
A320-25-1555	A320-25-1557
A320-53-1215	

TECHNICAL FOLLOW UP (TFU)	
25.70.00.007	

REPAIR DRAWINGS	SHEET(S)
R53113174002	all sheets
R53113174003	all sheets

STANDARDS MANUAL (SM)

CONSUMABLE MATERIAL LIST (CML)

(2) Repair of the Electronics Rack FIN 80VU

- (a) Task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

AIRCRAFT MAINTENANCE MANUAL (AMM)	
06-20-00, Page Block 000	06-31-53, Page Block 000
Task 12-34-24-869-002	Task 20-21-11-911-001
20-28-00, Page Block 200	Task 21-20-00-010-001
	Task 21-20-00-410-001
21-26-00, Page Block 500	21-26-34, Page Block 400
	21-26-34, Page Block 401
22-66-34, Page Block 400	22-83-34, Page Block 400
22-66-34, Page Block 401	22-83-34, Page Block 401
22-96-00, Page Block 500	22-97-00, Page Block 500
23-11-33, Page Block 400	23-12-33, Page Block 400
23-11-33, Page Block 401	23-12-33, Page Block 401
23-24-34, Page Block 400	23-28-34, Page Block 400
23-24-34, Page Block 401	23-28-34, Page Block 401
23-32-34, Page Block 400	23-32-35, Page Block 400
23-32-34, Page Block 401	23-32-35, Page Block 401
23-33-31, Page Block 400	23-33-37, Page Block 400
23-33-31, Page Block 401	23-33-37, Page Block 401
23-33-39, Page Block 400	23-33-52, Page Block 400
23-33-39, Page Block 401	23-33-52, Page Block 401
23-34-34, Page Block 400	23-51-34, Page Block 400
23-34-34, Page Block 401	23-51-34, Page Block 401
23-72-31, Page Block 400	23-73-34, Page Block 400
23-72-31, Page Block 401	23-73-34, Page Block 401
Task 24-38-51-000-001	Task 25-54-11-400-001
Task 25-54-12-000-003	26-12-34, Page Block 400
Task 25-54-12-400-003	26-12-34, Page Block 401
26-13-34, Page Block 400	27-51-34, Page Block 400
26-13-34, Page Block 401	27-51-34, Page Block 401
27-93-34, Page Block 400	27-94-34, Page Block 400
27-93-34, Page Block 401	27-94-34, Page Block 401
27-95-34, Page Block 400	31-32-34, Page Block 400
27-95-34, Page Block 401	31-32-34, Page Block 401
31-33-34, Page Block 400	31-33-52, Page Block 400
31-33-34, Page Block 401	31-33-52, Page Block 401
31-36-34, Page Block 400	31-36-52, Page Block 400
31-36-34, Page Block 401	31-36-52, Page Block 401
31-38-34, Page Block 400	31-50-00, Page Block 500

AIRCRAFT MAINTENANCE MANUAL (AMM)	
31-38-34, Page Block 401	
31-53-34, Page Block 400	31-55-34, Page Block 400
31-53-34, Page Block 401	31-55-34, Page Block 401
31-60-00, Page Block 500	31-63-34, Page Block 400
	31-63-34, Page Block 401
33-13-00, Page Block 500	33-14-00, Page Block 500
34-35-34, Page Block 400	34-43-34, Page Block 400
34-35-34, Page Block 401	34-43-34, Page Block 401
34-48-34, Page Block 400	34-51-33, Page Block 400
34-48-34, Page Block 401	34-51-33, Page Block 401
34-52-33, Page Block 400	34-52-35, Page Block 400
34-52-33, Page Block 401	34-52-35, Page Block 401
34-53-31, Page Block 400	34-55-31, Page Block 400
34-53-31, Page Block 401	34-55-31, Page Block 401
46-21-34, Page Block 400	Task 52-30-00-860-002
46-21-34, Page Block 401	
Task 52-41-00-410-002	Task 53-12-13-000-001
	Task 53-12-13-400-001
Task 53-22-14-000-001	73-25-34, Page Block 400
Task 53-22-14-400-001	73-25-34, Page Block 401
77-32-34, Page Block 400	
77-32-34, Page Block 401	
ELECTRICAL STANDARD PRACTICES MANUAL (ESPM)	
20-51-20	20-52-10
20-54-50	20-55-00
SERVICE BULLETIN (SB)	
A320-25-1555	A320-25-1557
A320-53-1215	
STRUCTURAL REPAIR MANUAL (SRM)	
51-24-00	51-42-00
51-43-00	51-44-00
51-46-00	51-47-00
51-75-12	
TECHNICAL FOLLOW UP (TFU)	
25.70.00.007	
REPAIR DRAWINGS	SHEET(S)
R53113174002	all sheets

REPAIR DRAWINGS	SHEET(S)
R53113174003	all sheets

STANDARDS MANUAL (SM)

CONSUMABLE MATERIAL LIST (CML)

CONF ALL

K. PUBLICATION AFFECTED

None

L. INDUSTRY SUPPORT INFORMATION

This Service Bulletin is the subject of an AIRBUS monitored retrofit campaign.

For the organization of this retrofit, you are kindly requested to contact AIRBUS retrofit management department: monitored.retrofit@airbus.com.

2. MATERIAL INFORMATION

A. MATERIAL APPLICABILITY

CONF 001

Material for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MATERIAL	QTY	SEE NOTES
Referenced tools	See SB	

CONF 002

Material for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MATERIAL	QTY	SEE NOTES
Referenced tools	See SB	

Material for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MATERIAL	QTY	SEE NOTES
Consumables	See SB	
Part kit 251BKJD01R00	1	
Part kit 251BKJD02R00	1	
Part kit 251BKJD05R00	1	

CONF 003

Material for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MATERIAL	QTY	SEE NOTES
Referenced tools	See SB	

Material for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MATERIAL	QTY	SEE NOTES
Consumables	See SB	
Part kit 251BKJD01R00	1	
Part kit 251BKJD02R00	1	
Part kit 251BKJD05R00	1	

CONF 004

Material for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

MATERIAL	QTY	SEE NOTES
Referenced tools	See SB	

Material for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

MATERIAL	QTY	SEE NOTES
Consumables	See SB	
Part kit 251BKJD03R00	1	

MATERIAL	QTY	SEE NOTES
Part kit 251BKJD04R00	1	
Part kit 251BKJD05R00	1	

CONF ALL

B. TOOLING

CONF 001

Referenced tools for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

To accomplish this Service Bulletin it is necessary to use the tools listed in the subsequent referenced tasks:

Ref. NTM 51-90-00

CONF 002

Referenced tools for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

To accomplish this Service Bulletin it is necessary to use the tools listed in the subsequent referenced tasks:

Ref. NTM 51-90-00

CONF 003

Referenced tools for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

To accomplish this Service Bulletin it is necessary to use the tools listed in the subsequent referenced tasks:

Ref. NTM 51-90-00

CONF 004

Referenced tools for task set A320-A-25-XX-1BKJ-01000-93BA-A Inspection of the Electronics Rack FIN 80VU

To accomplish this Service Bulletin it is necessary to use the tools listed in the subsequent referenced tasks:

Ref. NTM 51-90-00

CONF ALL

C. CONSUMABLES

CONF 002

Consumables for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	06AAA1	Polysulfide Sealant - General Purpose Brushable		
	06AAC1	Polysulfide Sealant - General Purpose Interfay		
	06AEB2	Polysulfide Sealant - Low Adhesion Fillet Structure		
	06BHG1	Silicone Sealant - Firewall High Temp.		

ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	08BAA9	Non Aqueous Cleaner - General		
	14SBA1	Textile - Lint free Cotton		

CONF 003
Consumables for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	06AAA1	Polysulfide Sealant - General Purpose Brushable		
	06AAC1	Polysulfide Sealant - General Purpose Interfay		
	06AEB2	Polysulfide Sealant - Low Adhesion Fillet Structure		
	06BHG1	Silicone Sealant - Firewall High Temp.		
	08BAA9	Non Aqueous Cleaner - General		
	14SBA1	Textile - Lint free Cotton		

CONF 004
Consumables for task set A320-A-25-XX-1BKJ-01000-93CA-A Repair of the Electronics Rack FIN 80VU

ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	06AAA1	Polysulfide Sealant - General Purpose Brushable		
	06AAC1	Polysulfide Sealant - General Purpose Interfay		
	06AEB2	Polysulfide Sealant - Low Adhesion Fillet Structure		
	06BHG1	Silicone Sealant - Firewall High Temp.		
	08BAA9	Non Aqueous Cleaner - General		
	14SBA1	Textile - Lint free Cotton		

CONF 001

No consumables

CONF ALL
D. LIST OF NEW PARTS
CONF 002 003
Part kit 251BKJD01R00

Cost	1400.00 USD
Availability	240 d (calendar days from receipt of order)
NOTE(S)	

The sales terms (costs and availability) are estimated in relation to economic conditions. The latest valid price and lead time for ordering purposes shall be obtained through the AIRBUS Spares portal or through a request for quotation to airbus.kit@airbus.com.

The availability given above is the standard lead time from the date of your purchase order. If items are required before this time, please include a retrofit planning schedule with your order so that we can try to comply with your requirements.

Procurement address	D4296	AIRBUS Material, Logistics and Suppliers ATP II B, Hein-Sass-Weg 24 21129 HAMBURG GERMANY Phone: +49 40 743 51XXX E-mail: airbus.kit@airbus.com Internet: http://spares.airbus.com
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NOTE(S)

Customers with aircraft shown in the applicability of this Service Bulletin should send a purchase order to AIRBUS at the address given above and quote the number of this Service Bulletin.

ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
0	D5391065720000	1	FORKEND		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
634	NAS6603-4	4	BOLT		
644	NAS6604-4	4	BOLT		
813	NAS1726-3E	12	NUT		
814	NAS1726-4E	25	NUT		

NOTE(S)

Refer to the AIRBUS Ref. SM if you find part numbers of hardware components which you cannot identify. The SM will give you the correct part number to part number relationship.

Part kit 251BKJD02R00

Cost	1350.00 USD
Availability	240 d (calendar days from receipt of order)

NOTE(S)

The sales terms (costs and availability) are estimated in relation to economic conditions. The latest valid price and lead time for ordering purposes shall be obtained through the AIRBUS Spares portal or through a request for quotation to airbus.kit@airbus.com.

The availability given above is the standard lead time from the date of your purchase order. If items are required before this time, please include a retrofit planning schedule with your order so that we can try to comply with your requirements.

Procurement address	D4296	AIRBUS
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		Material, Logistics and Suppliers ATP II B, Hein-Sass-Weg 24 21129 HAMBURG GERMANY Phone: +49 40 743 51XXX E-mail: airbus.kit@airbus.com Internet: http://spares.airbus.com
NOTE(S)		
Customers with aircraft shown in the applicability of this Service Bulletin should send a purchase order to AIRBUS at the address given above and quote the number of this Service Bulletin.		

ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
1	D5391065720100	1	FORKEND		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
634	NAS6603-4	4	BOLT		
644	NAS6604-4	4	BOLT		
813	NAS1726-3E	12	NUT		
814	NAS1726-4E	25	NUT		

NOTE(S)					
Refer to the AIRBUS Ref. SM if you find part numbers of hardware components which you cannot identify. The SM will give you the correct part number to part number relationship.					

CONF 004
Part kit 251BKJD03R00

Cost	2050.00 USD
Availability	240 d (calendar days from receipt of order)

NOTE(S)	
The sales terms (costs and availability) are estimated in relation to economic conditions. The latest valid price and lead time for ordering purposes shall be obtained through the AIRBUS Spares portal or through a request for quotation to airbus.kit@airbus.com.	
The availability given above is the standard lead time from the date of your purchase order. If items are required before this time, please include a retrofit planning schedule with your order so that we can try to comply with your requirements.	

Procurement address	D4296	AIRBUS Material, Logistics and Suppliers ATP II B, Hein-Sass-Weg 24 21129 HAMBURG GERMANY Phone: +49 40 743 51XXX E-mail: airbus.kit@airbus.com Internet: http://spares.airbus.com
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NOTE(S)		
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Customers with aircraft shown in the applicability of this Service Bulletin should send a purchase order to AIRBUS at the address given above and quote the number of this Service Bulletin.

ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
0	D5392408320090	1	CLEVIS		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
644	NAS6604-4	4	BOLT		
657	ABS0114T4-7	4	BOLT		
813	NAS1726-3E	8	NUT		
814	NAS1726-4E	25	NUT		

NOTE(S)

Refer to the AIRBUS Ref. SM if you find part numbers of hardware components which you cannot identify. The SM will give you the correct part number to part number relationship.

Part kit 251BKJD04R00

Cost	2050.00 USD
Availability	240 d (calendar days from receipt of order)

NOTE(S)

The sales terms (costs and availability) are estimated in relation to economic conditions. The latest valid price and lead time for ordering purposes shall be obtained through the AIRBUS Spares portal or through a request for quotation to airbus.kit@airbus.com.

The availability given above is the standard lead time from the date of your purchase order. If items are required before this time, please include a retrofit planning schedule with your order so that we can try to comply with your requirements.

Procurement address	D4296	AIRBUS Material, Logistics and Suppliers ATP II B, Hein-Sass-Weg 24 21129 HAMBURG GERMANY Phone: +49 40 743 51XXX E-mail: airbus.kit@airbus.com Internet: http://spares.airbus.com
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NOTE(S)

Customers with aircraft shown in the applicability of this Service Bulletin should send a purchase order to AIRBUS at the address given above and quote the number of this Service Bulletin.

ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
1	D5392408320190	1	CLEVIS		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		

ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
644	NAS6604-4	4	BOLT		
657	ABS0114T4-7	4	BOLT		
813	NAS1726-3E	8	NUT		
814	NAS1726-4E	25	NUT		
NOTE(S)					
Refer to the AIRBUS Ref. SM if you find part numbers of hardware components which you cannot identify. The SM will give you the correct part number to part number relationship.					

CONF 002-004
Part kit 251BKJD05R00

Cost	50.00 USD	
Availability	210 d (calendar days from receipt of order)	
NOTE(S)		
The sales terms (costs and availability) are estimated in relation to economic conditions. The latest valid price and lead time for ordering purposes shall be obtained through the AIRBUS Spares portal or through a request for quotation to airbus.kit@airbus.com. The availability given above is the standard lead time from the date of your purchase order. If items are required before this time, please include a retrofit planning schedule with your order so that we can try to comply with your requirements.		
Procurement address	D4296	AIRBUS Material, Logistics and Suppliers ATP II B, Hein-Sass-Weg 24 21129 HAMBURG GERMANY Phone: +49 40 743 51XXX E-mail: airbus.kit@airbus.com Internet: http://spares.airbus.com
NOTE(S)		
Customers with aircraft shown in the applicability of this Service Bulletin should send a purchase order to AIRBUS at the address given above and quote the number of this Service Bulletin.		

ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
4	NAS1149F0332P	53	WASHER		
87	NSA8203-258	1	ORING		
93	ASNA2528-3	34	NUT		
101	NAS1169DD10	11	WASHER		
102	ASNA2531-3	11	NUT		

ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
111	ASNA2529-3	2	NUT		
NOTE(S)					
Refer to the AIRBUS Ref. SM if you find part numbers of hardware components which you cannot identify. The SM will give you the correct part number to part number relationship.					

CONF 001

No parts

CONF ALL

E. LIST OF EXISTING PARTS

No parts

3. ACCOMPLISHMENT INSTRUCTIONS

Task set **A320-A-25-XX-1BKJ-01000-93BA-A**

Inspection of the Electronics Rack FIN 80VU

WARNING: MAKE SURE THAT YOU OBEY ALL THE WARNINGS AND ALL THE CAUTIONS INCLUDED IN THE REFERENCED PROCEDURES.

CAUTION: ALWAYS OBEY THE PRECAUTIONS THAT FOLLOW TO KEEP ELECTRICAL WIRING IN A SATISFACTORY CONDITION (ELECTRICALLY AND MECHANICALLY SERVICEABLE). WHEN YOU DO MAINTENANCE WORK, REPAIRS MODIFICATIONS, ALWAYS KEEP ELECTRICAL WIRING, COMPONENTS AND THE WORK AREA AS CLEAN AS POSSIBLE. TO DO THIS:

- **PUT PROTECTION, SUCH AS PLASTIC SHEETING, CLOTHS, ETC., AS NECESSARY ON WIRING AND COMPONENTS.**
- **REGULARLY REMOVE ALL SHAVINGS, UNWANTED MATERIAL AND OTHER CONTAMINATION.**

**THESE PRECAUTIONS WILL DECREASE THE RISK OF CONTAMINATION AND DAMAGE TO THE ELECTRICAL WIRING INSTALLATION.
IF THERE IS CONTAMINATION, REFER TO Ref. ESPM 20-55-00.**

NOTE: The accomplishment instructions of this Service Bulletin include procedures given in other documents or in other sections of the Service Bulletin. When the words "refer to" are used and the operator has a procedure accepted by the local authority he belongs to, the accepted alternative procedure can be used. When the words "in accordance with" are used then the given procedure must be followed.

NOTE: The access and close-up instructions, not comprising return to service tests, in this Service Bulletin do not constitute or affect the technical intent of the Service Bulletin. Operators can therefore, as deemed necessary, omit or add access and/or close-up steps to add flexibility to their maintenance operations as long as the technical intent of the Service Bulletin is met within the set parameters.

NOTE: Manual titles given in the accomplishment instructions are referred to by acronyms. Refer to paragraph 1.J., References, for the definition of acronyms.

NOTE: This Service Bulletin contains the instructions for the on-aircraft maintenance necessary to ensure the continued airworthiness of the aircraft.
For any deviations to the instructions, including RC paragraph contained herein, contact AIRBUS for further instructions and approval.

NOTE: This Service Bulletin is classified mandatory or expected to be classified mandatory by an Airworthiness Directive (AD). The paragraphs 3.C. and 3.E. in these accomplishment instructions are Required for Compliance (RC) and must be done to comply with the AD. To allow more flexibility, the rule "refer to" and "in accordance with" will apply to these paragraphs 3.C. and 3.E. Other paragraphs are recommended and may be deviated from, done as part of other actions or done with accepted methods different from those given in the Service Bulletin, as long as the RC paragraphs can be done and the aircraft can be put back into a serviceable condition.

NOTE: The purpose of flowcharts is to supplement the information given in the Procedure and Compliance paragraphs and not to serve as the primary source for tasks or compliance times given in this Service Bulletin.

A. TASK SET ASSOCIATED INFORMATION

Task Manpower

CONF 001

TOTAL MAN-HOURS

6.50 h

CONF 002

TOTAL MAN-HOURS 6.50 h

CONF 003

TOTAL MAN-HOURS 6.50 h

CONF 004

TOTAL MAN-HOURS 7.50 h

CONF 001

ELAPSED TIME (HOURS) 3.25 h

CONF 002

ELAPSED TIME (HOURS) 3.25 h

CONF 003

ELAPSED TIME (HOURS) 3.25 h

CONF 004

ELAPSED TIME (HOURS) 3.75 h

CONF ALL

B. PREPARATION

(1) Task A320-A-25-XX-1BKJ-01001-540A-A - Job Set-up

Job Set-Up Information

Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.50 h
Minimum Number of person	2
Elapsed Time	1.00 h

Referenced Information

Manuals/Drawings	References/Sheets
AMM	Task 12-34-24-869-002 Task 25-54-11-000-001 Task 52-30-00-860-001 Task 52-41-00-010-002

Procedure

NOTE: For the basic Aircraft configuration, refer to AMM Chapter "Introduction - Aircraft status for maintenance" in Manual Front Matter section.

- Make sure that the aircraft is electrically grounded, refer to Ref. AMM Task 12-34-24-869-002.
- Open the avionics compartment door, Door 824, refer to Ref. AMM Task 52-41-00-010-002.
- Open the FWD cargo compartment door, Door 825, refer to Ref. AMM Task 52-30-00-860-001.

- (d) Remove the FWD cargo compartment panels, Panel 131DW and/or Panel 131CW and/or Panel 132EW, refer to Ref. AMM Task 25-54-11-000-001.

C. PROCEDURE
CONF 001
(1) Task A320-A-25-XX-1BKJ-01001-346A-A - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	0.75 h
Minimum Number of person	1
Elapsed Time	0.75 h
Skills	Non-Destructive Testing

Referenced Information

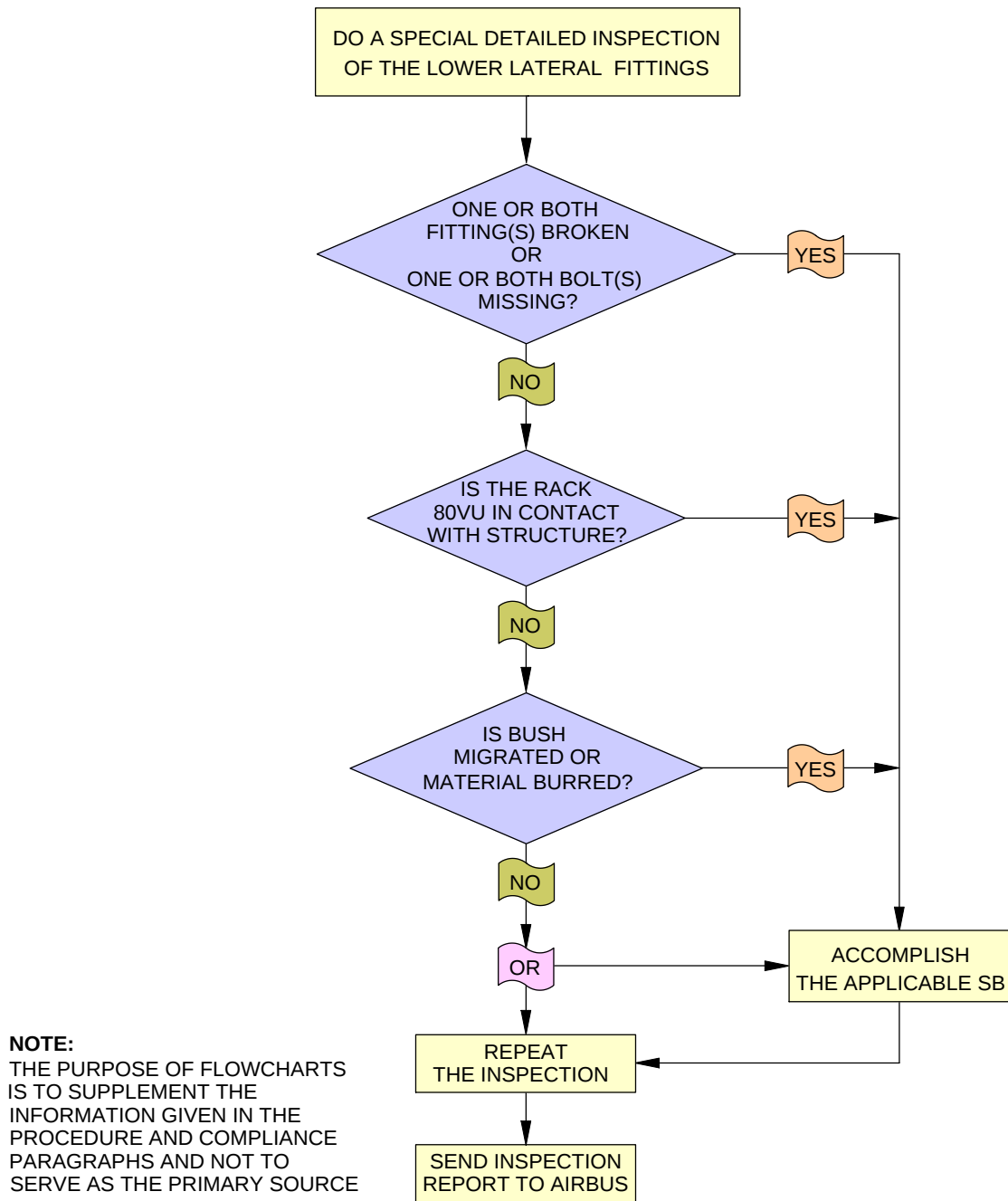
Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00
SB	A320-25-1557 A320-53-1215

Procedure

- (a) Standard Practices:
- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
 - 2 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 001.
 - 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.
- (b) Do an SDI of the lower lateral fittings of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).
Refer to Ref. NTM 51-90-00
Refer to Ref. Fig. Flowchart
Refer to Ref. Fig. SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU
Refer to Ref. Appendix Inspection Condition of the Lateral Support
- 1 If one or both fittings are broken or if one or both bolts are missing:
 - a Accomplish the Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.

- b Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- 2 If the fittings are not broken and the bolts are not missing:
 - a Check for contact between the Electronics Rack FIN 80VU and the structure:
 - <1> If the Electronics Rack FIN 80VU is in contact with the structure:
 - <a> Accomplish the Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.
 - Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
 - <2> If the Electronics Rack FIN 80VU is not in contact with the structure:
 - <a> Check for bush migration or burred material:
 - [1] If the bush has migrated or material is burred:
 - [a] Accomplish the Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.
 - [b] Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
 - [2] If the bush has not migrated and material is not burred:
 - [a] Accomplish the Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.
 - [b] Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF 001



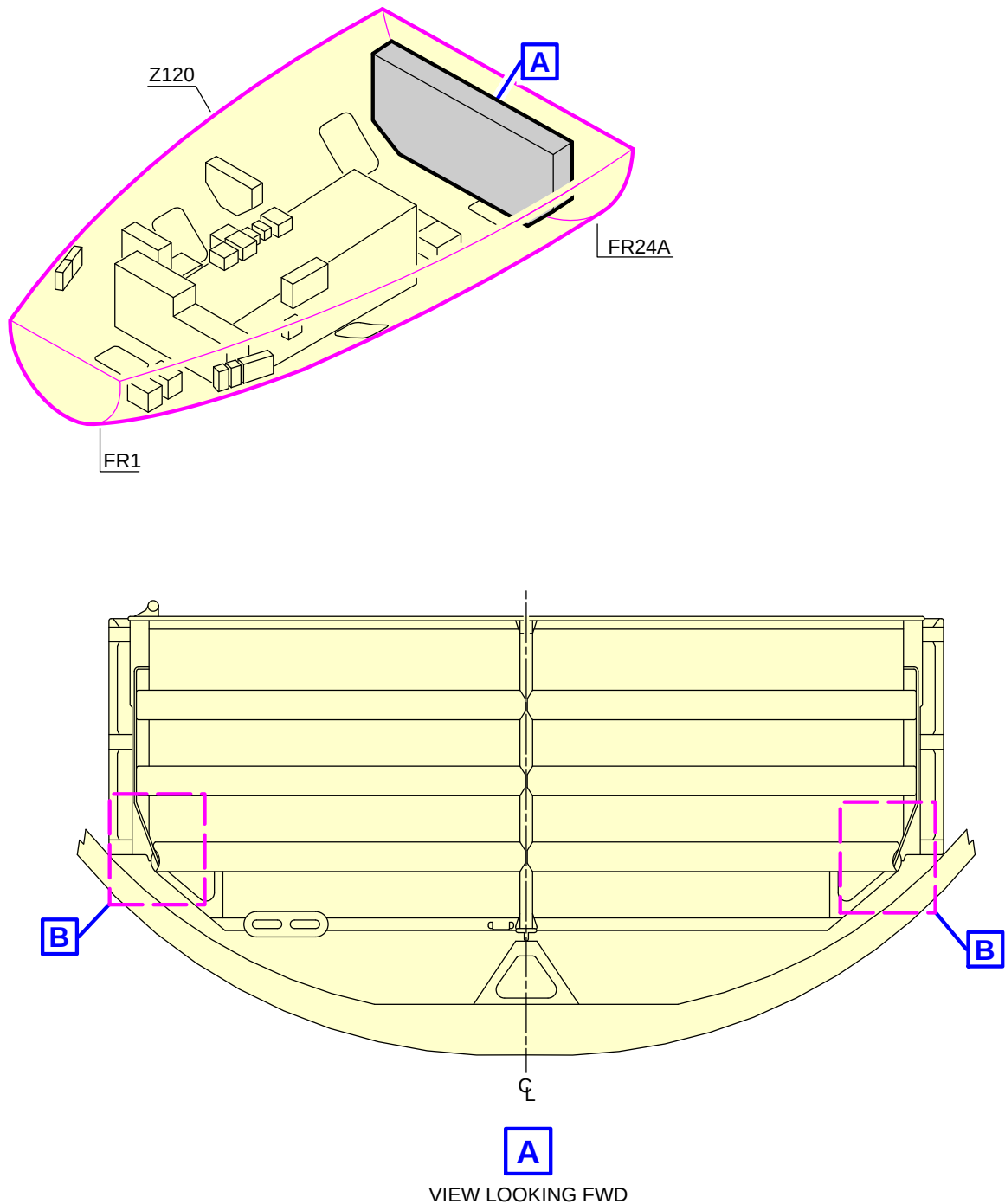
NOTE:
THE PURPOSE OF FLOWCHARTS IS TO SUPPLEMENT THE INFORMATION GIVEN IN THE PROCEDURE AND COMPLIANCE PARAGRAPHS AND NOT TO SERVE AS THE PRIMARY SOURCE FOR TASKS OR COMPLIANCE TIMES GIVEN IN THIS SERVICE BULLETIN.

FOR THE COMPLIANCE TIMES, REFER TO THE ACCOMPLISHMENT TIMESCALE PARA 1.E.(2).

ICN-A320-A-25XX1BKJ-A-FAPE3-00EP0-A-005-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP0-A-005-01 SHEET 1/1

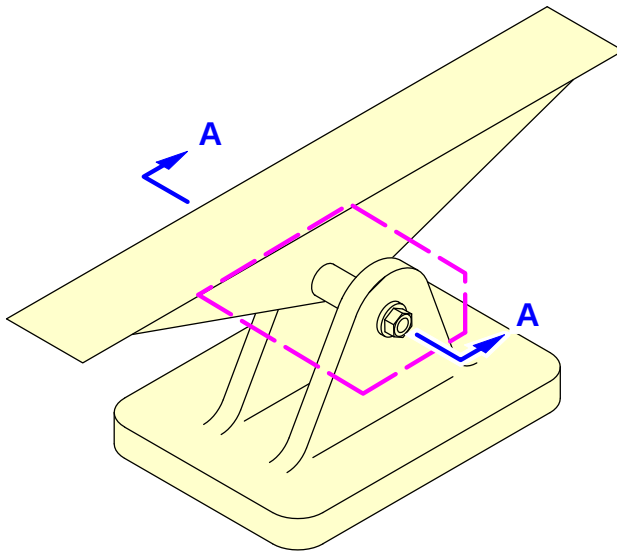
CONF 001



ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01

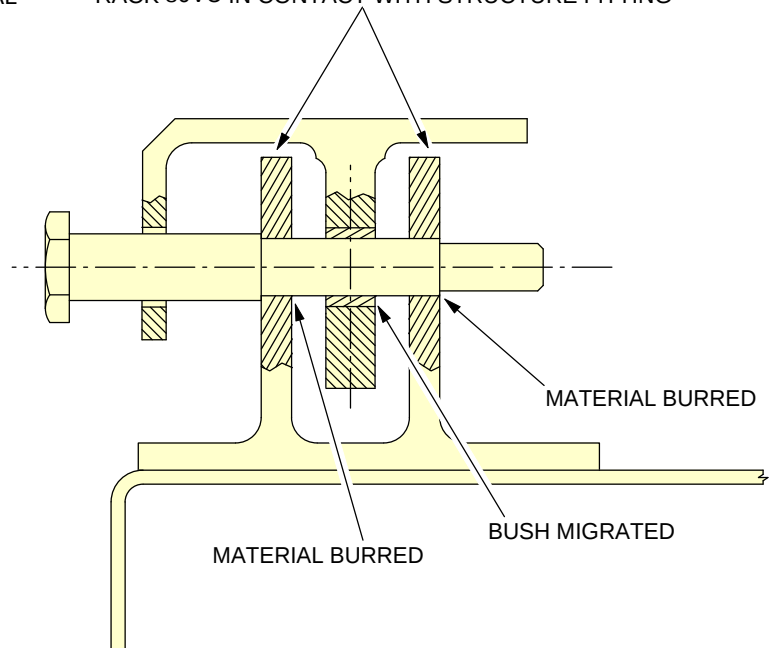
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01 SHEET 1/2

CONF 001



LH SHOWN
RH SYMMETRICAL

RACK 80VU IN CONTACT WITH STRUCTURE FITTING



A-A

NOTE:



INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01 SHEET 2/2

CONF 002
(1) Task A320-A-25-XX-1BKJ-01002-346A-A - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	0.75 h
Minimum Number of person	1
Elapsed Time	0.75 h
Skills	Non-Destructive Testing

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00
SB	A320-53-1215
R53113174000	all sheets
R53113174001	all sheets

Procedure
(a) Standard Practices:

- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
- 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.

(b) Do an SDI of the lower lateral fittings of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).

Refer to Ref. NTM 51-90-00

Refer to Ref. Fig. Flowchart

Refer to Ref. Fig. SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU

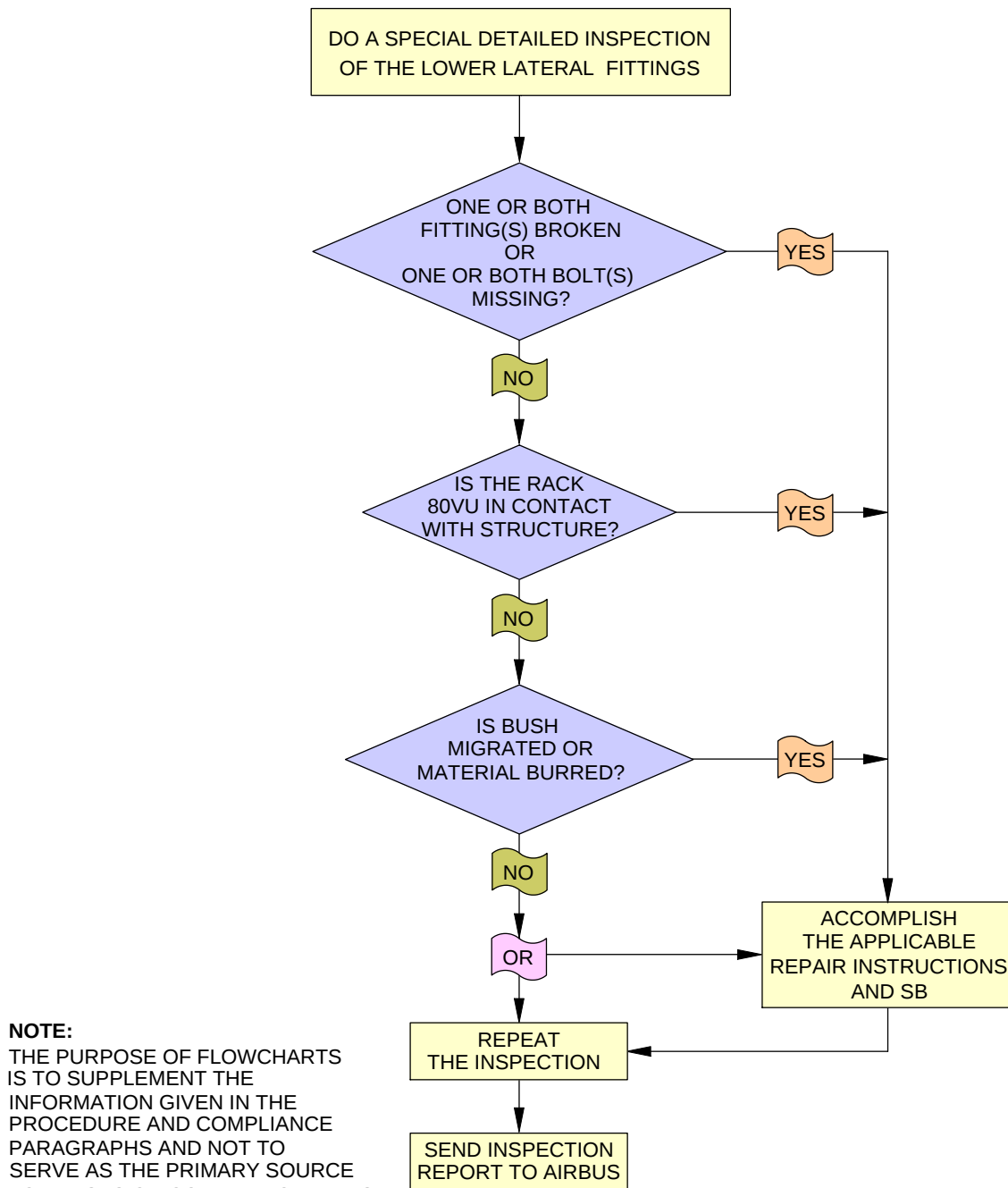
Refer to Ref. Appendix Inspection Condition of the Lateral Support

1 If one or both fittings are broken or if one or both bolts are missing:

- a Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A and accomplish Service Bulletin No. Ref. SB A320-53-1215.

- b Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- 2 If the fittings are not broken and the bolts are not missing:
- a Check for contact between the Electronics Rack FIN 80VU and the structure:
- <1> If the Electronics Rack FIN 80VU is in contact with the structure:
- <a> Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A and accomplish Service Bulletin No. Ref. SB A320-53-1215.
- Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- <2> If the Electronics Rack FIN 80VU is not in contact with the structure:
- <a> Check for bush migration or burred material:
- [1] If the bush has migrated or material is burred:
- [a] Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A and accomplish Service Bulletin No. Ref. SB A320-53-1215.
- [b] Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- [2] If the bush has not migrated and material is not burred:
- [a] Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A and accomplish Service Bulletin No. Ref. SB A320-53-1215.
- [b] Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF 002



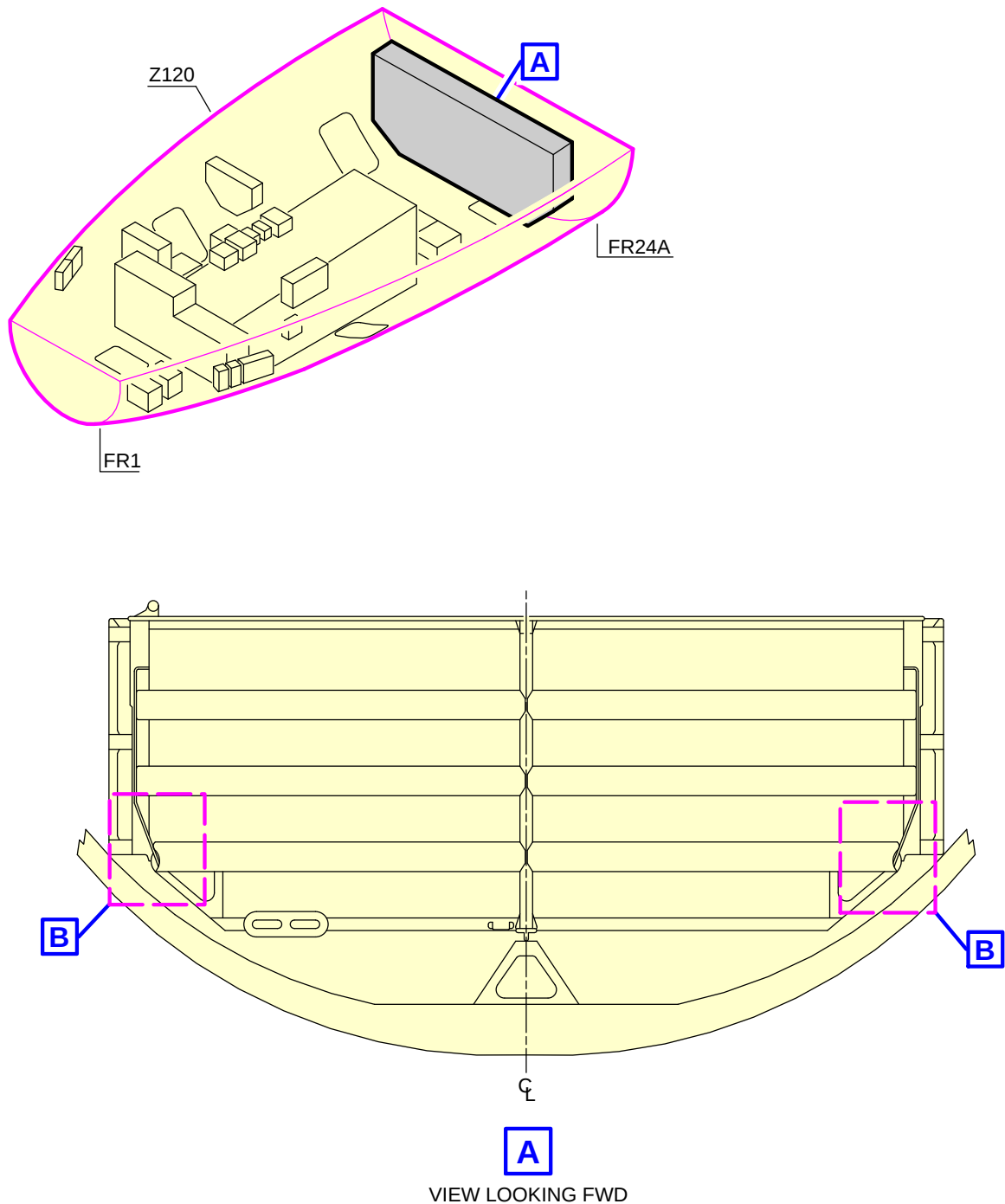
NOTE:
THE PURPOSE OF FLOWCHARTS IS TO SUPPLEMENT THE INFORMATION GIVEN IN THE PROCEDURE AND COMPLIANCE PARAGRAPHS AND NOT TO SERVE AS THE PRIMARY SOURCE FOR TASKS OR COMPLIANCE TIMES GIVEN IN THIS SERVICE BULLETIN.

FOR THE COMPLIANCE TIMES, REFER TO THE ACCOMPLISHMENT TIMESCALE PARA 1.E.(2).

ICN-A320-A-25XX1BKJ-A-FAPE3-00EP1-A-008-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP1-A-008-01 SHEET 1/1

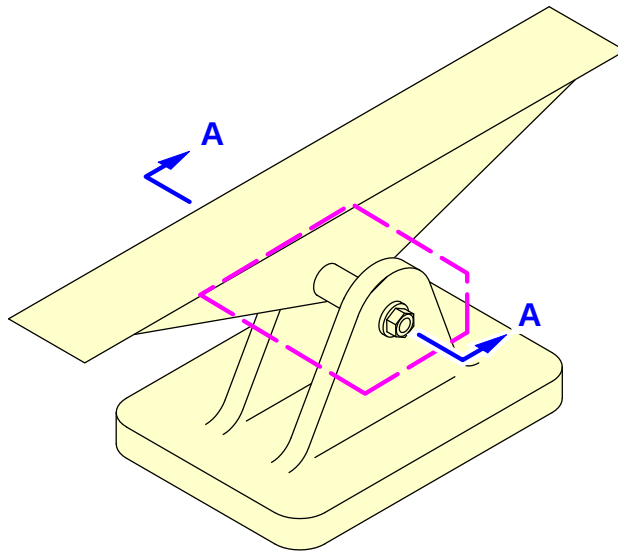
CONF 002



ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01

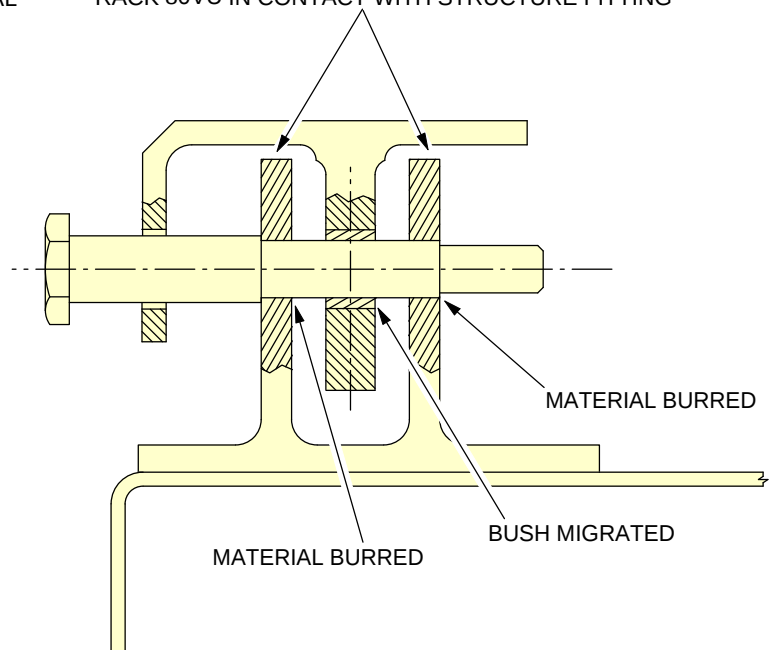
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01 SHEET 1/2

CONF 002



LH SHOWN
RH SYMMETRICAL

RACK 80VU IN CONTACT WITH STRUCTURE FITTING



A-A

NOTE:



INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01 SHEET 2/2

CONF 003
(1) Task A320-A-25-XX-1BKJ-01003-346A-A - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	0.75 h
Minimum Number of person	1
Elapsed Time	0.75 h
Skills	Non-Destructive Testing

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00
R53113174000	all sheets
R53113174001	all sheets

Procedure
(a) Standard Practices:

- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
- 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.

(b) Do an SDI of the lower lateral fittings of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).

Refer to Ref. NTM 51-90-00

Refer to Ref. Fig. Flowchart

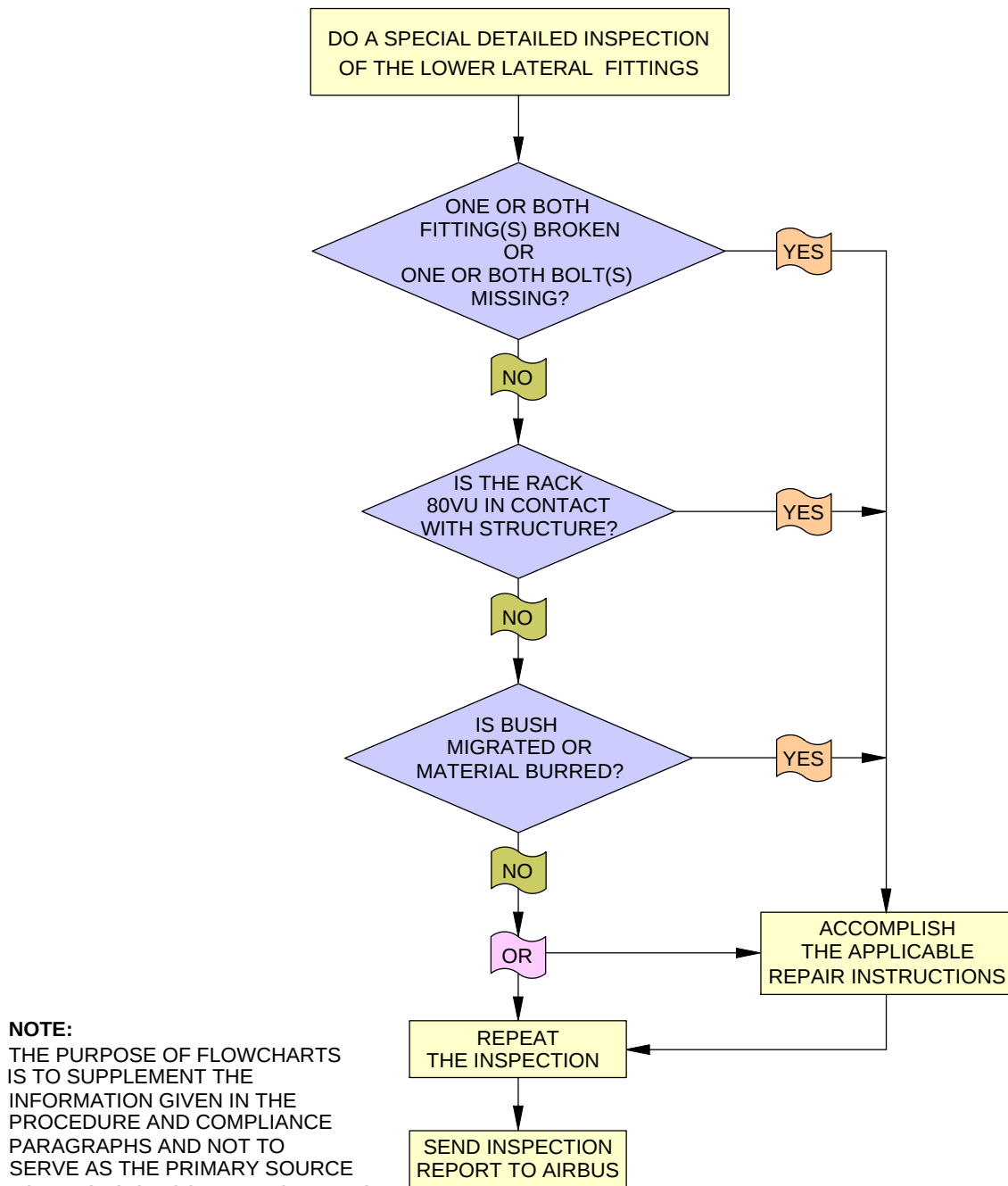
Refer to Ref. Fig. SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU

Refer to Ref. Appendix Inspection Condition of the Lateral Support

- 1 If one or both fittings are broken or if one or both bolts are missing:
 - a Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A.
 - b Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.

- 2 If the fittings are not broken and the bolts are not missing:
 - a Check for contact between the Electronics Rack FIN 80VU and the structure:
 - <1> If the Electronics Rack FIN 80VU is in contact with the structure:
 - <a> Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A).
 - Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
 - <2> If the Electronics Rack FIN 80VU is not in contact with the structure:
 - <a> Check for bush migration or burred material.
 - [1] If the bush has migrated or material is burred:
 - [a] Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A.
 - [b] Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
 - [2] If the bush has not migrated and material is not burred:
 - [a] Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A).
 - [b] Repeat the inspection within the interval given in PLANNING INFORMATION paragraph E.(2) in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
 - (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF 003



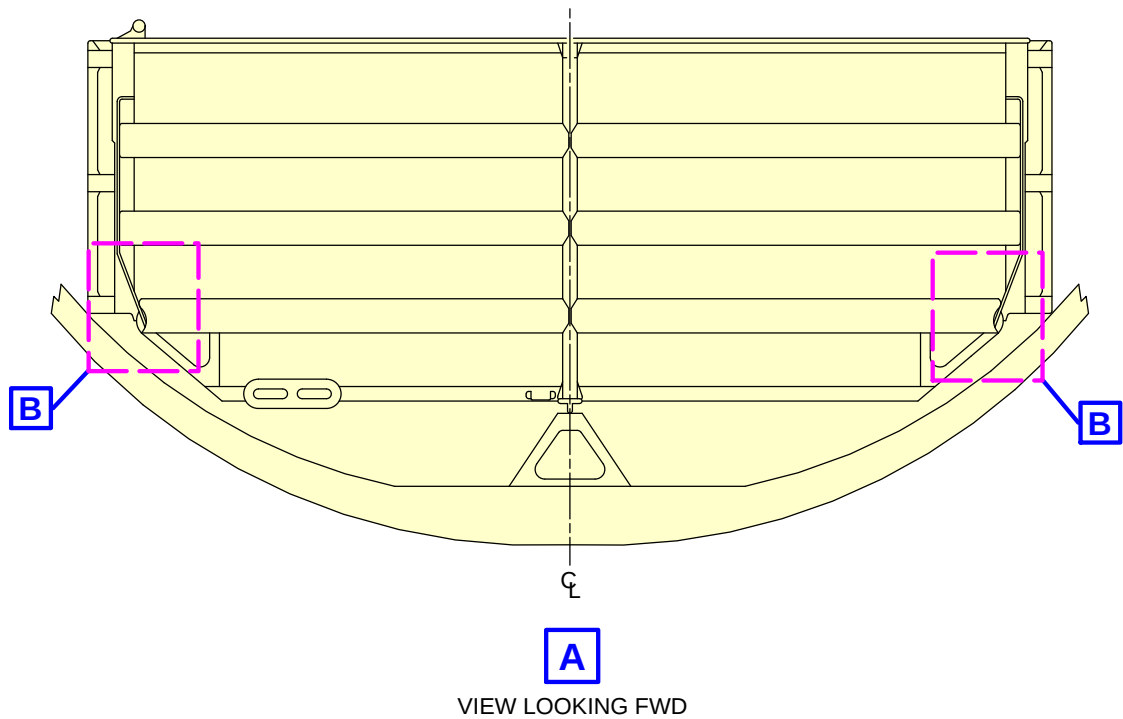
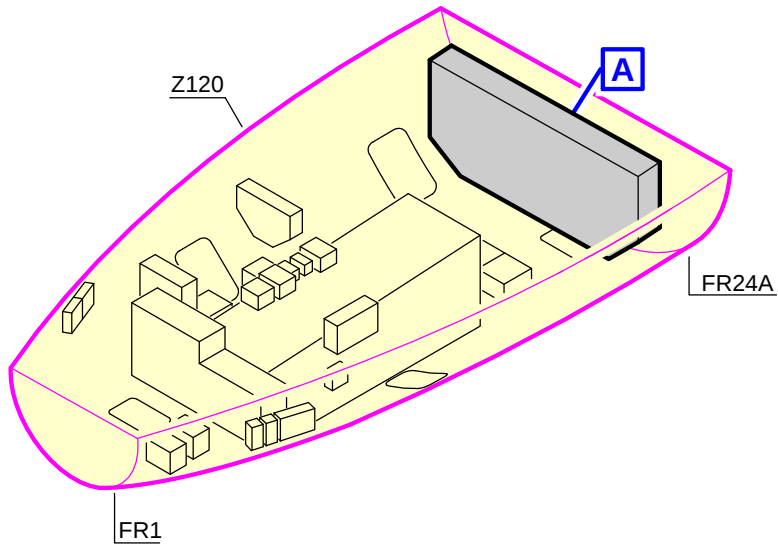
NOTE:
THE PURPOSE OF FLOWCHARTS
IS TO SUPPLEMENT THE
INFORMATION GIVEN IN THE
PROCEDURE AND COMPLIANCE
PARAGRAPHS AND NOT TO
SERVE AS THE PRIMARY SOURCE
FOR TASKS OR COMPLIANCE TIMES
GIVEN IN THIS SERVICE BULLETIN.

FOR THE COMPLIANCE TIMES,
REFER TO THE ACCOMPLISHMENT
TIMESCALE PARA 1.E.(2).

ICN-A320-A-25XX1BKJ-A-FAPE3-00EP2-A-006-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP2-A-006-01 SHEET 1/1

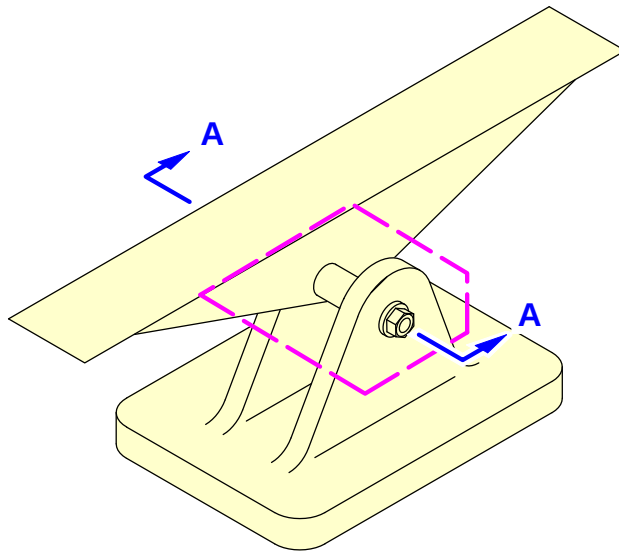
CONF 003



ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01

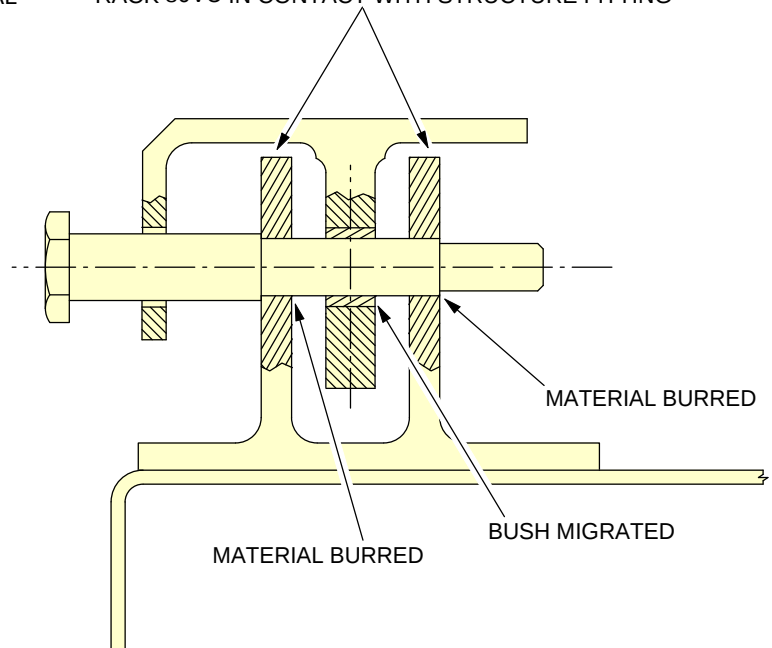
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01 SHEET 1/2

CONF 003



LH SHOWN
RH SYMMETRICAL

RACK 80VU IN CONTACT WITH STRUCTURE FITTING



A-A

NOTE:



INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01 SHEET 2/2

CONF 004
(1) Task A320-A-25-XX-1BKJ-01004-346A-A - Do an SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	0.75 h
Minimum Number of person	1
Elapsed Time	0.75 h
Skills	Airframe

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00
R53113174002	all sheets
R53113174003	all sheets

Procedure
(a) Standard Practices:

- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
- 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.

(b) Do an SDI of the lower lateral fittings of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).

Refer to Ref. NTM 51-90-00

Refer to Ref. Fig. Flowchart

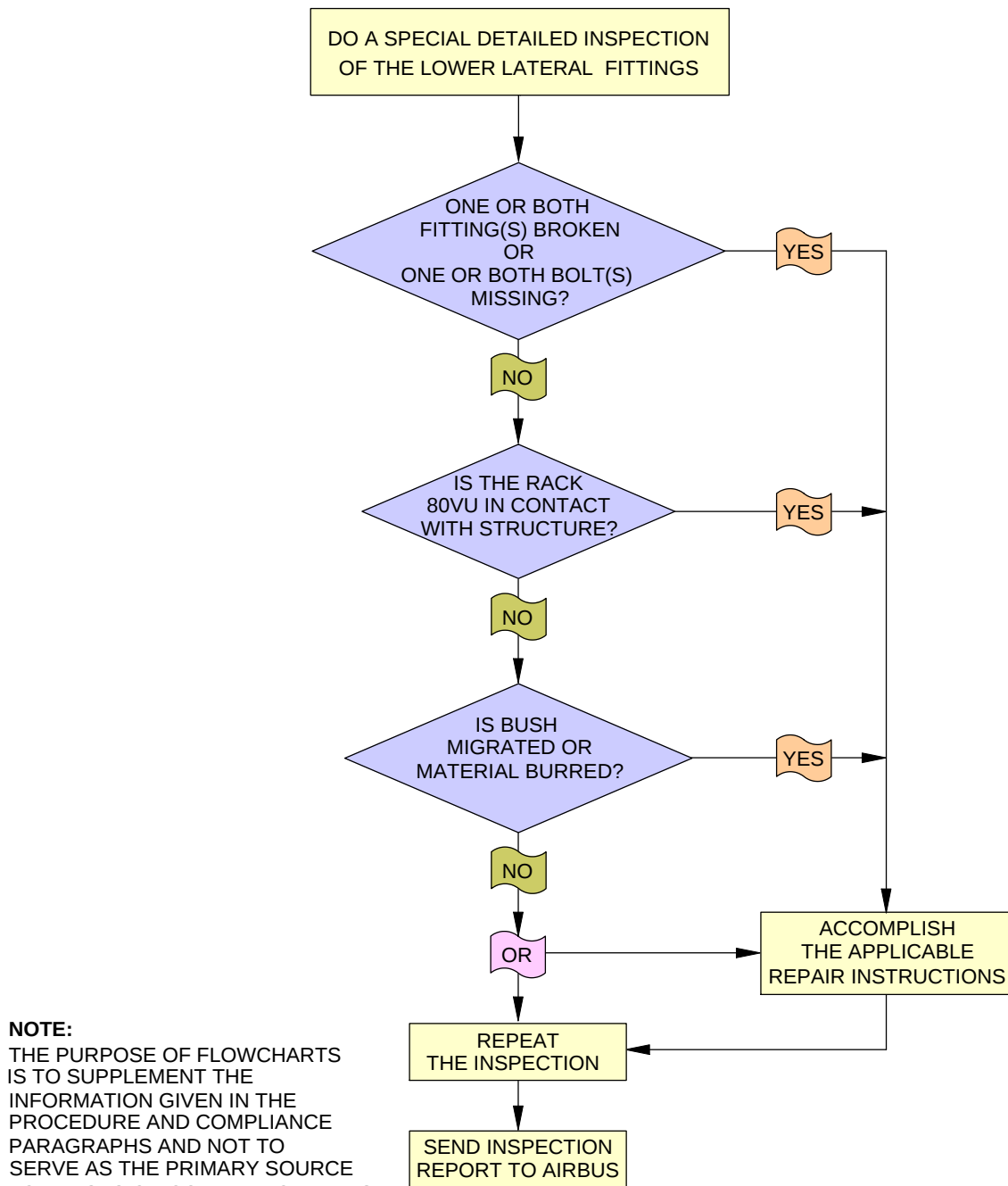
Refer to Ref. Fig. SDI of the Lower Lateral Fittings of the Electronics Rack FIN 80VU

Refer to Ref. Appendix Inspection Condition of the Lateral Support

- 1 If one or both fittings are broken or if one or both bolts are missing:
 - a Repair in accordance with Repair Instruction Ref. R53113174002 all sheets and Ref. R53113174003 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A.
 - b Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.

- 2 If the fittings are not broken and the bolts are not missing:
- a Check for contact between the Electronics Rack FIN 80VU and the structure:
- <1> If the Electronics Rack FIN 80VU is in contact with the structure:
- <a> Repair in accordance with Repair Instruction Ref. R53113174002 all sheets and Ref. R53113174003 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A).
- Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- <2> If the Electronics Rack FIN 80VU is not in contact with the structure:
- <a> Check for bush migration or burred material.
- [1] If the bush has migrated or material is burred:
- [a] Repair in accordance with Repair Instruction Ref. R53113174002 all sheets and Ref. R53113174003 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A.
- [b] Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- [2] If the bush has not migrated and material is not burred:
- [a] Repair in accordance with Repair Instruction Ref. R53113174002 all sheets and Ref. R53113174003 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A.
- [b] Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
- (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF 004



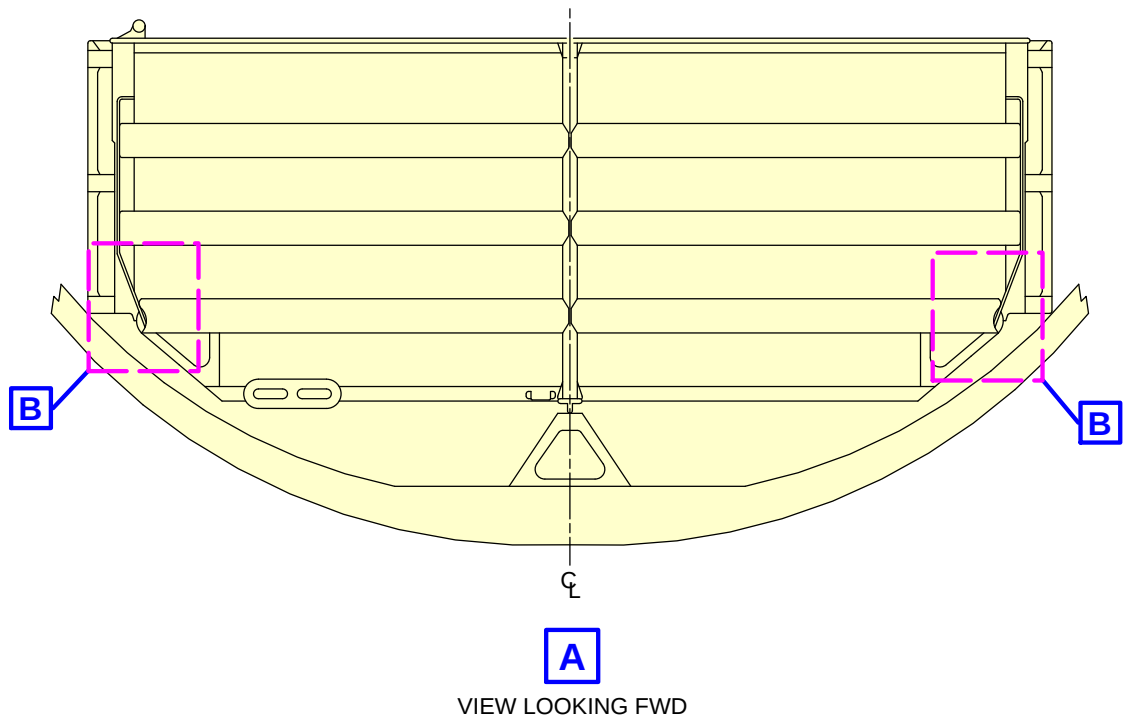
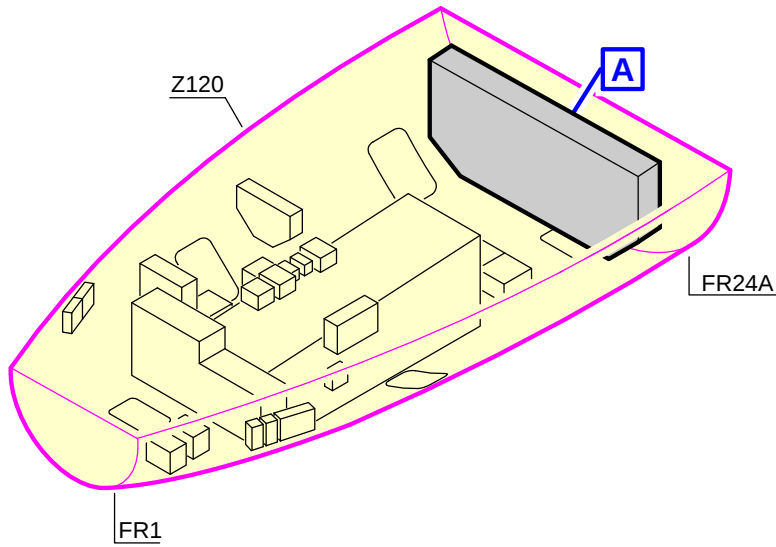
NOTE:
THE PURPOSE OF FLOWCHARTS IS TO SUPPLEMENT THE INFORMATION GIVEN IN THE PROCEDURE AND COMPLIANCE PARAGRAPHS AND NOT TO SERVE AS THE PRIMARY SOURCE FOR TASKS OR COMPLIANCE TIMES GIVEN IN THIS SERVICE BULLETIN.

FOR THE COMPLIANCE TIMES,
REFER TO THE ACCOMPLISHMENT
TIMESCALE PARA 1.E.(2).

ICN-A320-A-25XX1BKJ-A-FAPE3-00EP3-A-005-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP3-A-005-01 SHEET 1/1

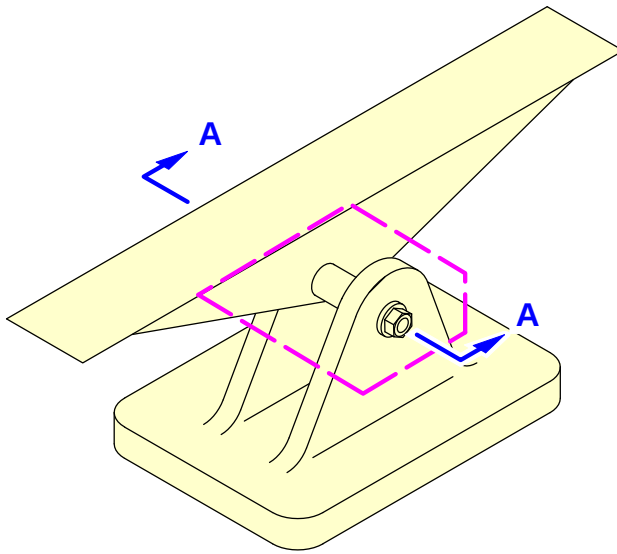
CONF 004



ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01

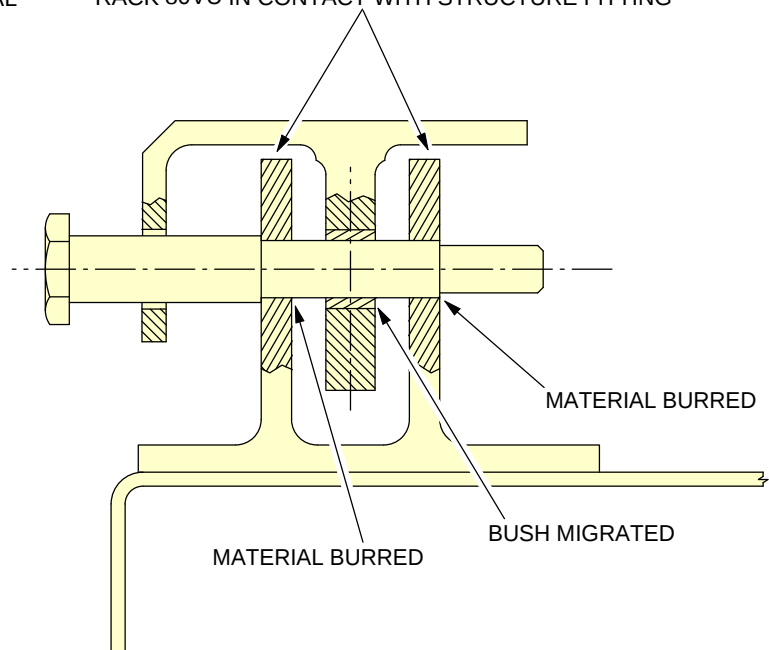
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOS-A-001-01 SHEET 1/2

CONF 004



LH SHOWN
RH SYMMETRICAL

RACK 80VU IN CONTACT WITH STRUCTURE FITTING



A-A

NOTE:



INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOT-A-001-01 SHEET 2/2

(2) Task A320-A-25-XX-1BKJ-03002-346A-A - Do an SDI of the Rods and Fittings of the Lower Central Support of the Electronics Rack FIN 80VU

Job Set-Up Information

Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	2.00 h
Minimum Number of person	1
Elapsed Time	2.00 h
Skills	Non-Destructive Testing

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00

Procedure

(a) Standard Practices:

- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
- 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.

(b) Do an SDI of the rods and fittings of the lower central support of the Electronics Rack FIN 80VU within the threshold given in the PLANNING INFORMATION paragraph E.(2).

Refer to Ref. Fig. Flowchart

Refer to Ref. Fig. SDI of the Rods and fittings of the Lower Central Support of the Electronics Rack FIN 80VU

- 1 If the rods and fittings are found in good condition:
 - a Do an SDI of the lower central support using a borescope refer to Para (c) below.
- 2 If the rods and fittings are not found in good condition:
 - a Contact AIRBUS before next flight and follow their instructions.
 - b Do an SDI of the lower central support using a borescope, refer to Para (c) below.

(c) Do an SDI of the lower central support of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).

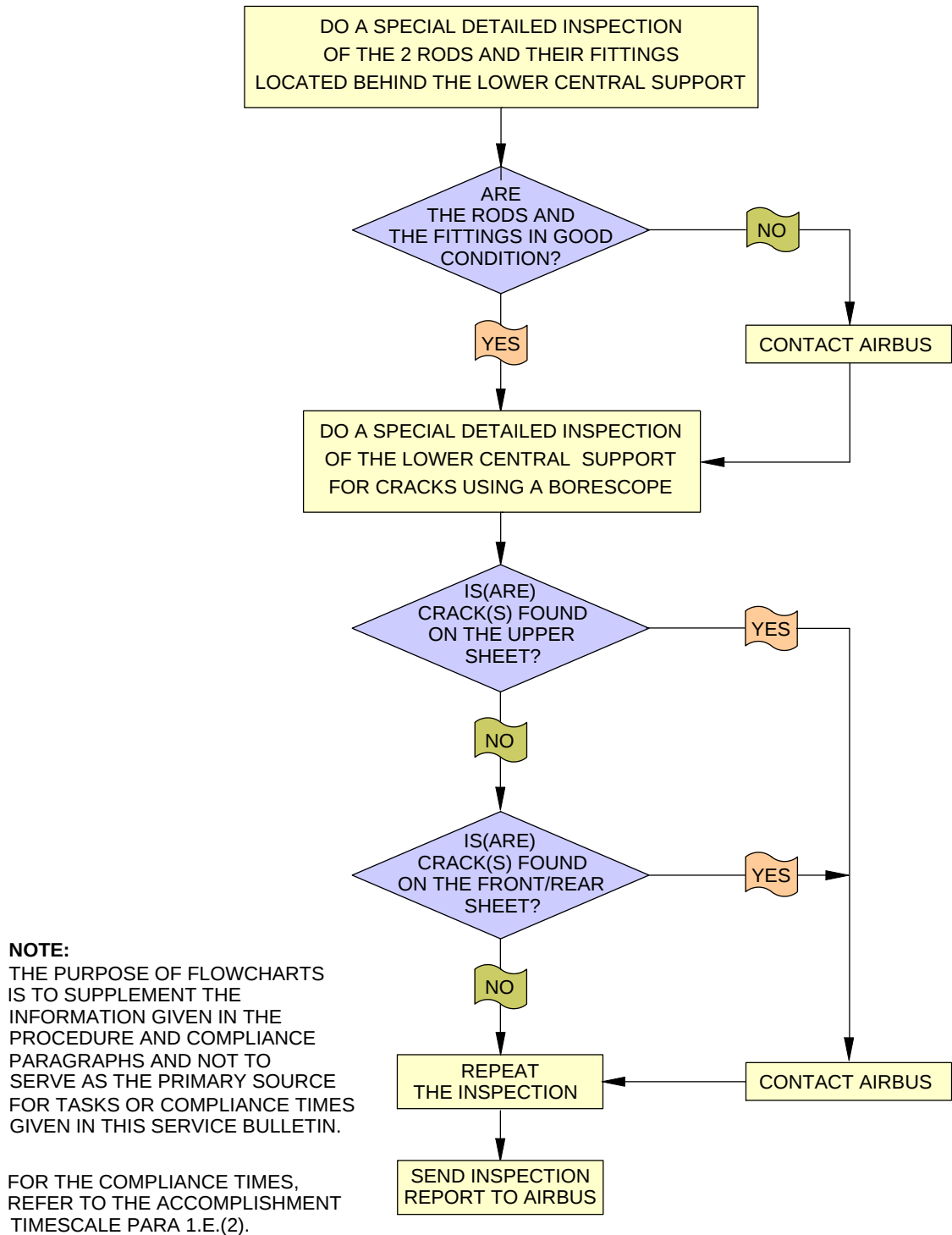
Refer to Ref. NTM 51-90-00

Refer to Ref. Fig. SDI of the Lower Central Support of the Electronics Rack FIN 80VU

Refer to Ref. Fig. Flowchart

- 1 If any crack(s) is(are) found on the upper sheet:
 - a Contact AIRBUS before next flight and follow their instructions.
 - 2 If no crack(s) is(are) found on the upper sheet:
 - a Check the front/rear sheets for crack(s):
 - <1> If any crack(s) is(are) found on the front/rear sheets:
 - <a> Contact AIRBUS before next flight and follow their instructions.
 - <2> If no crack(s) is(are) found on the front/rear sheets:
 - <a> Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-346Z-A.
- (d) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

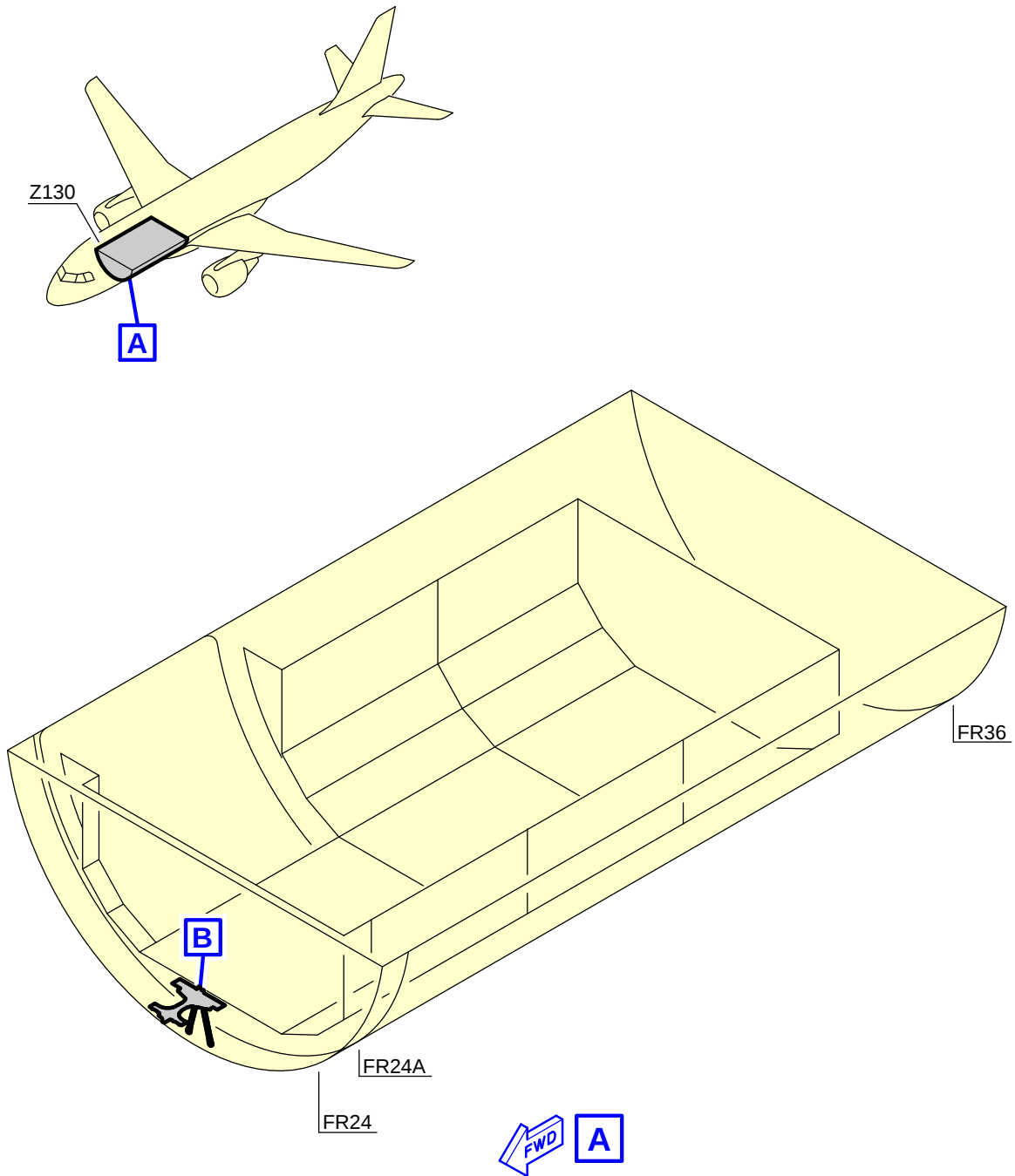
CONF 004



ICN-A320-A-25XX1BKJ-A-FAPE3-00EP7-A-007-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP7-A-007-01 SHEET 1/1

CONF 004



ICN-A320-A-25XX1BKJ-A-FAPE3-00EOY-A-001-01

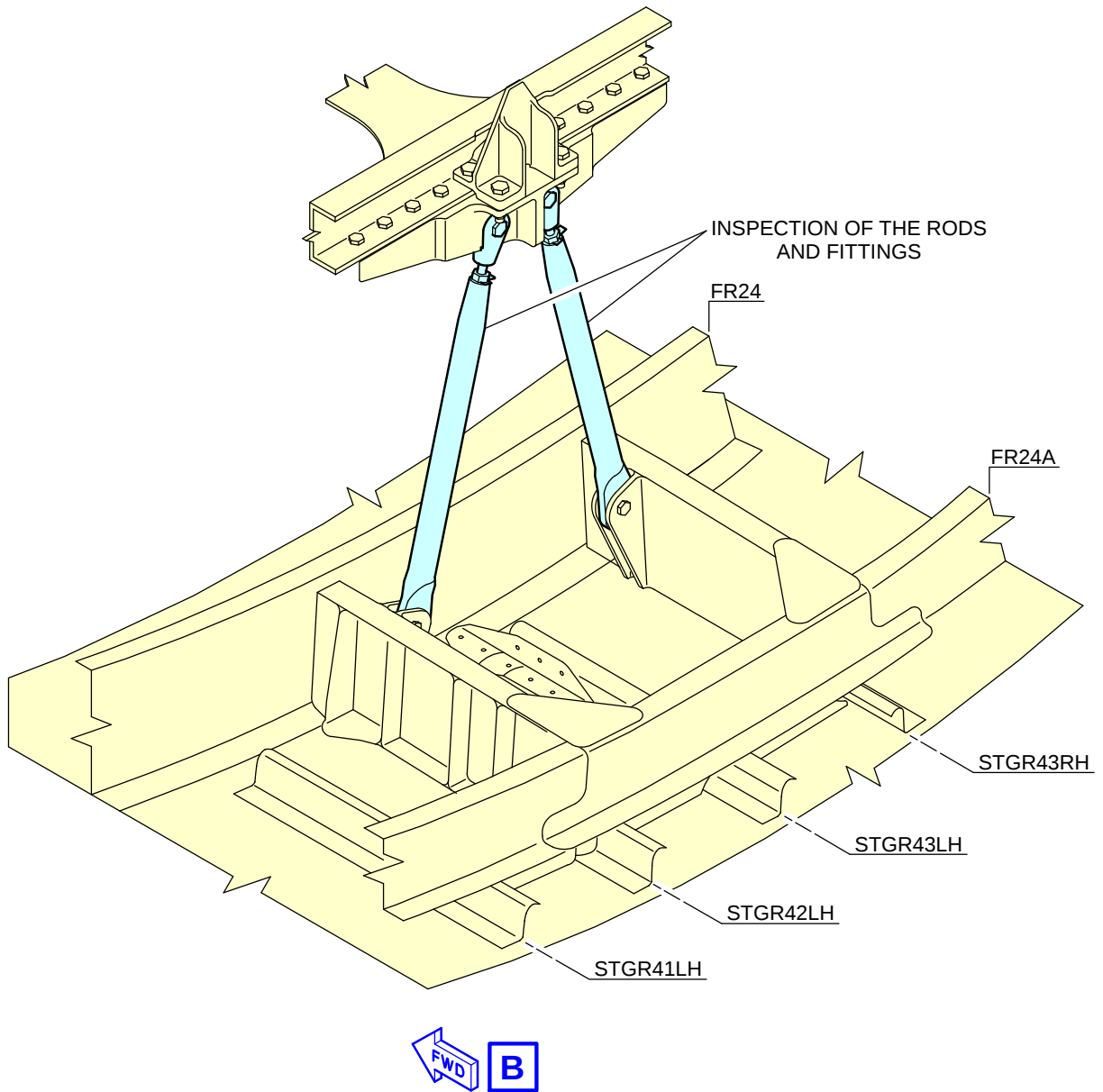
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOY-A-001-01 SHEET 1/2

REV 00 Aug 23/18
REV 02 Apr 09/20

SDI of the Rods and fittings of the Lower Central
Support of the Electronics Rack FIN 80VU

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ICN-A320-A-25XX1BKJ-A-FAPE3-00EOZ-A-001-01

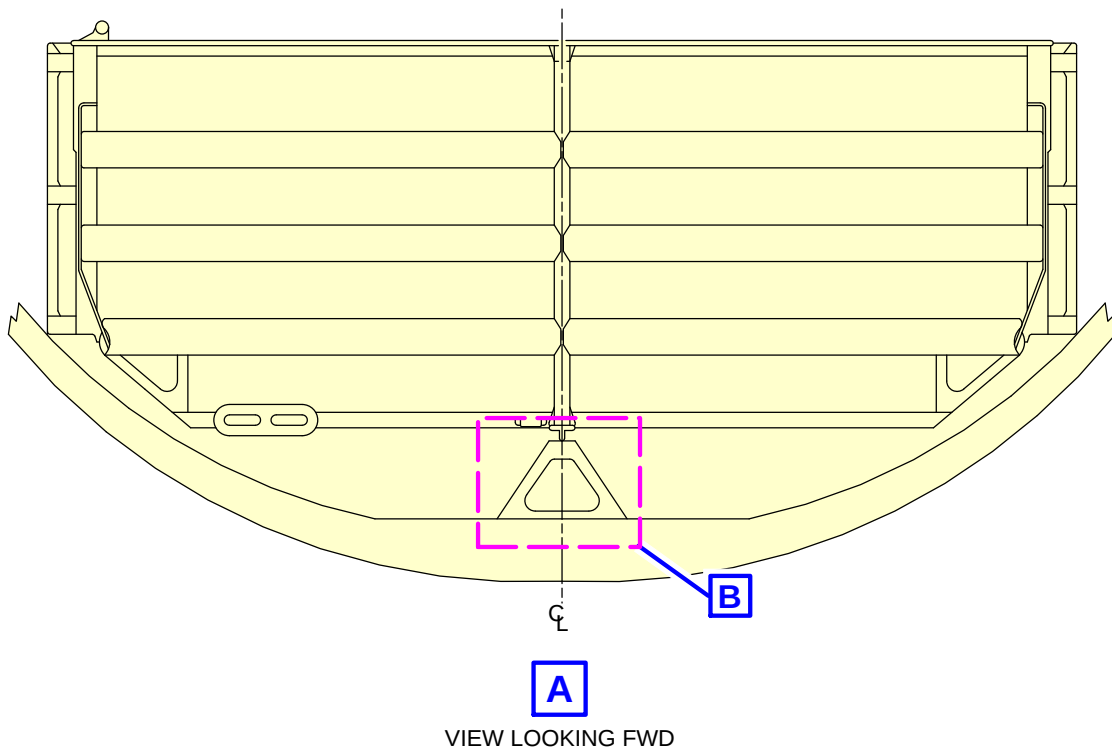
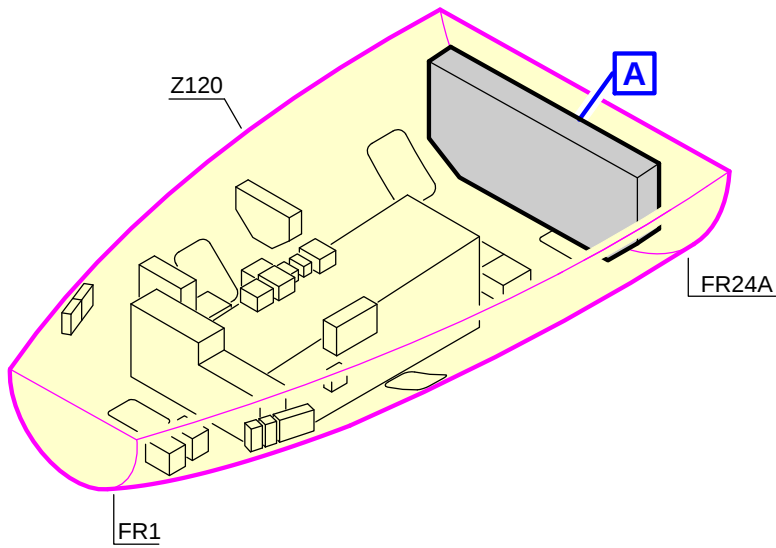
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOZ-A-001-01 SHEET 2/2

REV 00 Aug 23/18
REV 02 Apr 09/20

SDI of the Rods and fittings of the Lower Central
Support of the Electronics Rack FIN 80VU

A320-25-1BKJ
Page 129

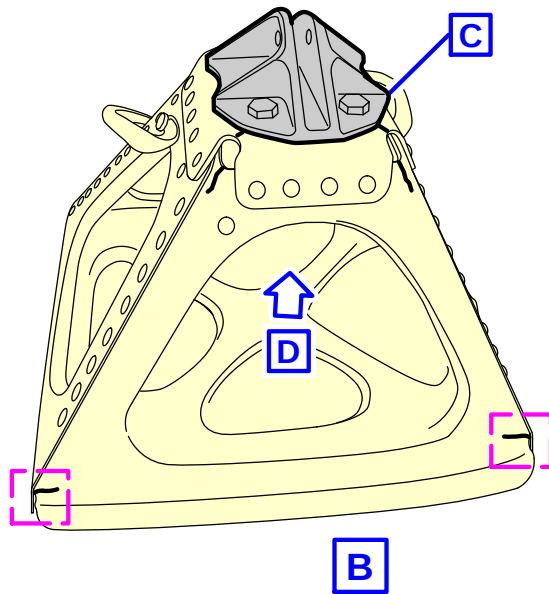
CONF 004



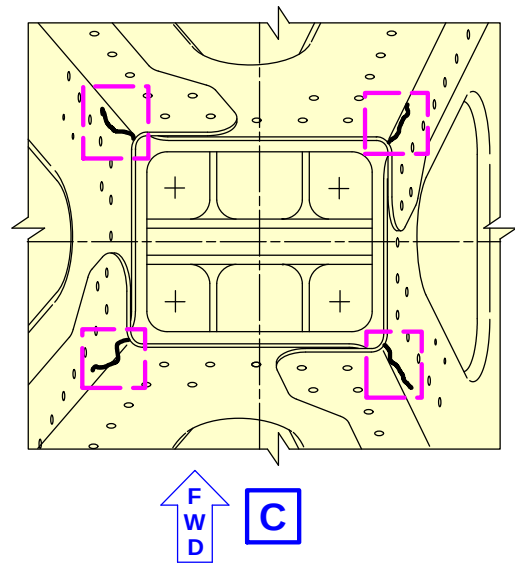
ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01 SHEET 1/2

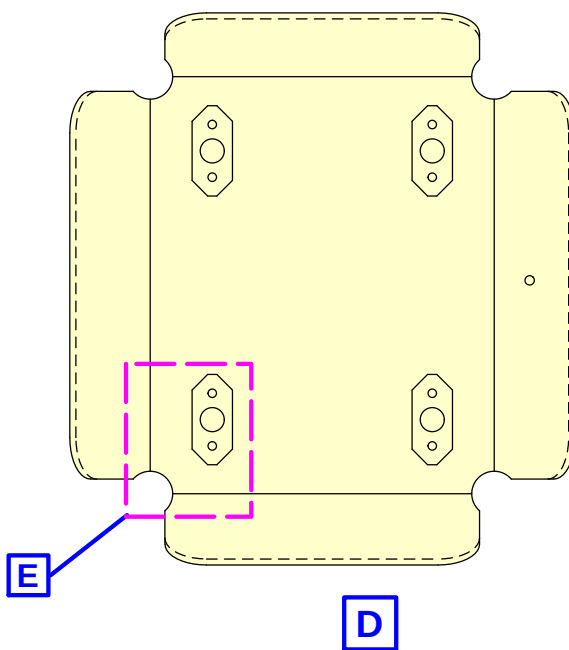
CONF 004



VIEW LOOKING FWD



VIEW LOOKING DOWN



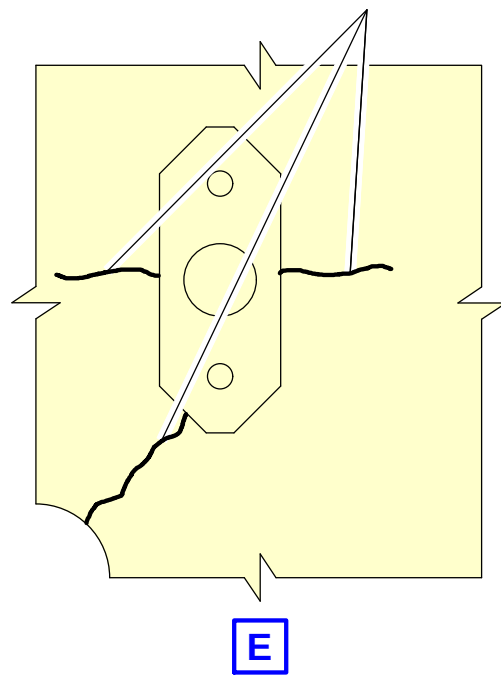
VIEW LOOKING UP

NOTE:



INSPECTION AREA

ANY CRACKS ARE NOT ACCEPTABLE



VIEW LOOKING UP

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01 SHEET 2/2

CONF 001
(3) Task A320-A-25-XX-1BKJ-02001-346A-A - Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.00 h
Minimum Number of person	1
Elapsed Time	1.00 h
Skills	Non-Destructive Testing

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00
SB	A320-25-1557 A320-53-1215

Procedure
(a) Standard Practices:

- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
- 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.

(b) Do an SDI of the lower central support of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).

Refer to Ref. NTM 51-90-00

Refer to Ref. Fig. Flowchart

Refer to Ref. Fig. SDI of the Lower Central Support of the Electronics Rack FIN 80VU

- 1 If any crack(s) is(are) found on the upper sheet:
 - a Accomplish Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.
 - b Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A.
- 2 If no crack(s) is(are) found on the upper sheet:
 - a Check the front/rear sheets for crack(s):

<1> If any crack(s) is(are) found on the front/rear sheets:

<a> Accomplish the Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.

 Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A

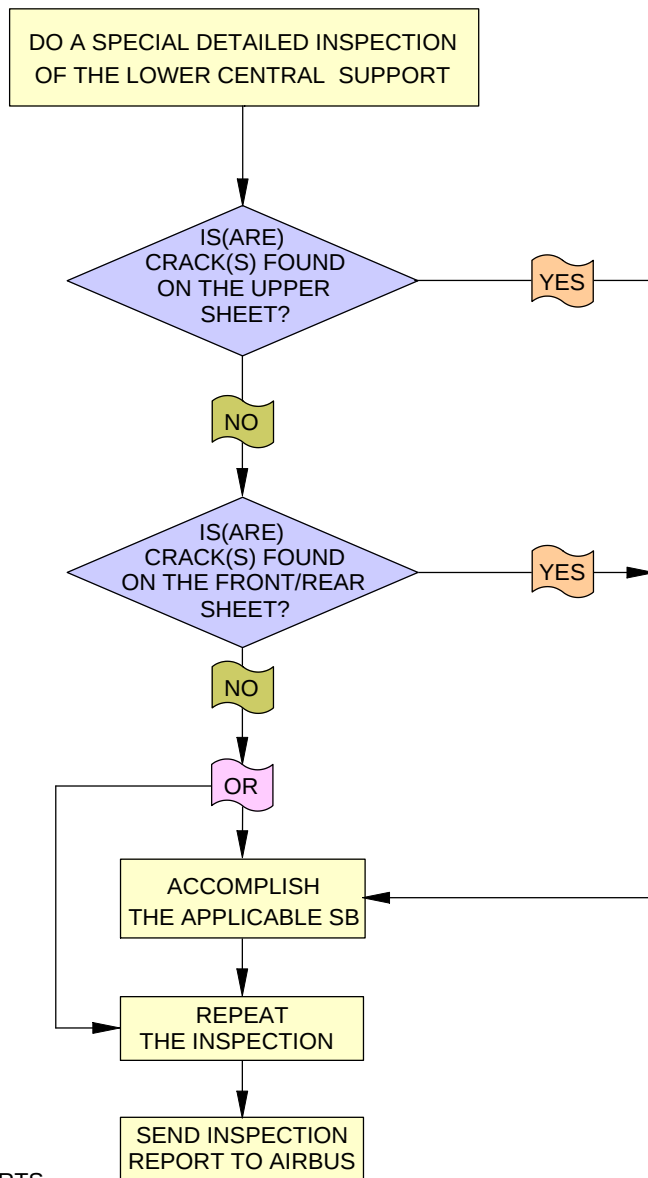
<2> If no crack(s) is(are) found on the front/rear sheets:

<a> Accomplish the Service Bulletin No. Ref. SB A320-25-1557 and Service Bulletin No. Ref. SB A320-53-1215.
OR

 Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A.

- (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF 001



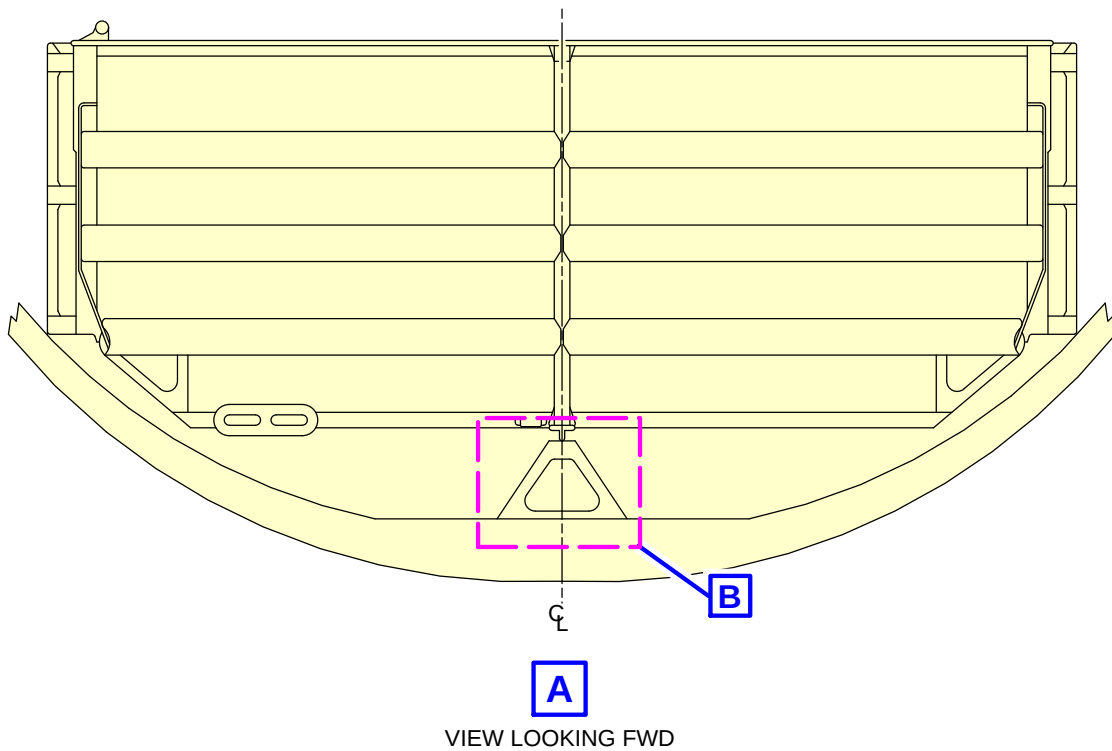
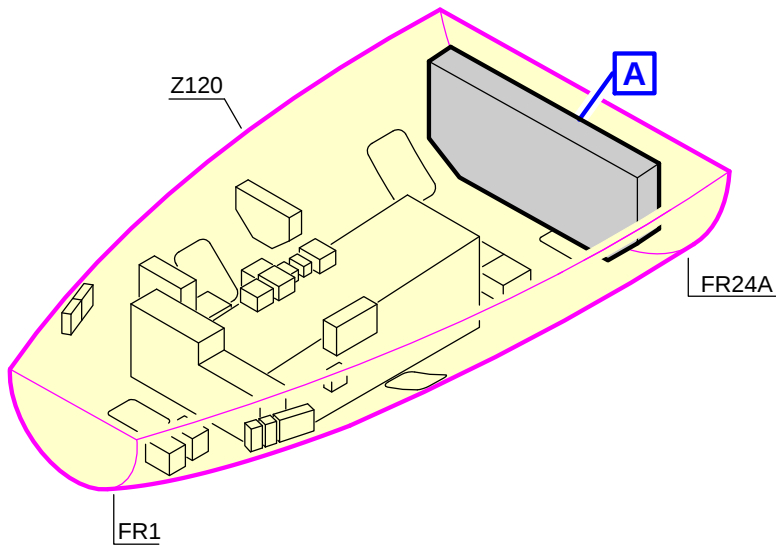
NOTE:
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INFORMATION GIVEN IN THE
PROCEDURE AND COMPLIANCE
PARAGRAPHS AND NOT TO
SERVE AS THE PRIMARY SOURCE
FOR TASKS OR COMPLIANCE TIMES
GIVEN IN THIS SERVICE BULLETIN.

FOR THE COMPLIANCE TIMES,
REFER TO THE ACCOMPLISHMENT
TIMESCALE PARA 1.E.(2).

ICN-A320-A-25XX1BKJ-A-FAPE3-00EP4-A-006-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP4-A-006-01 SHEET 1/1

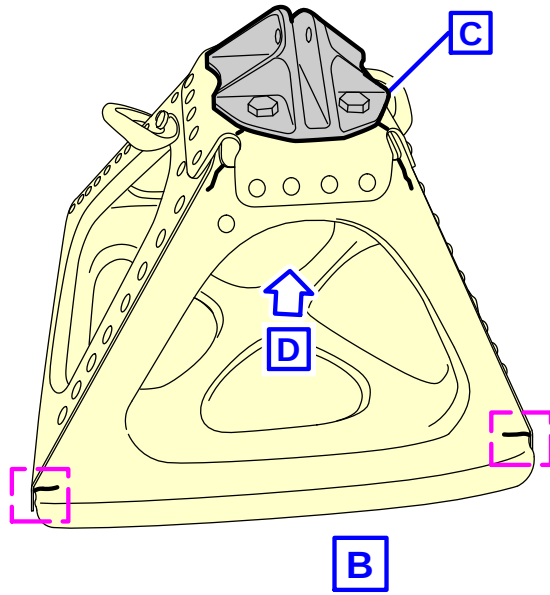
CONF 001



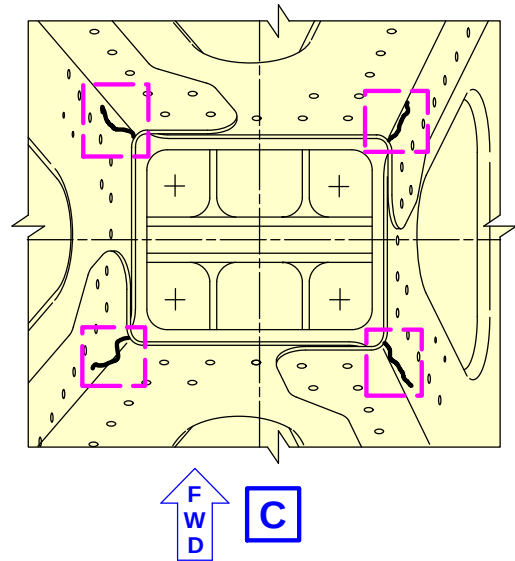
ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01 SHEET 1/2

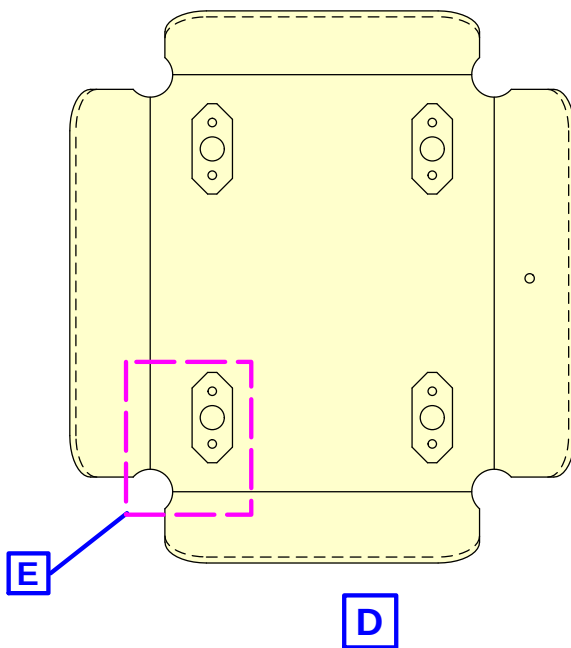
CONF 001



VIEW LOOKING FWD



VIEW LOOKING DOWN



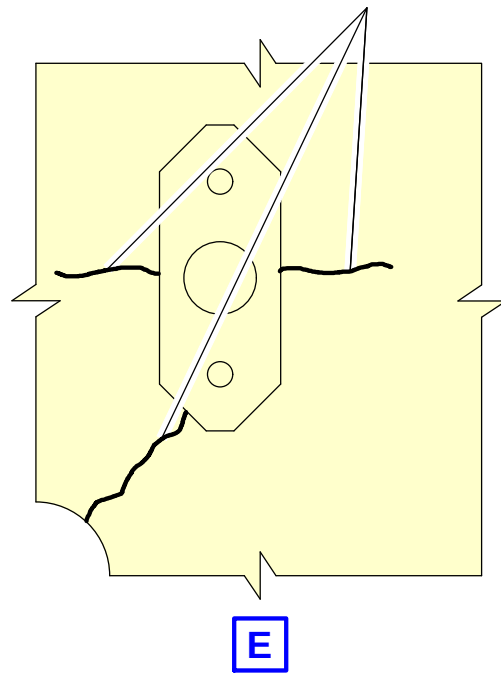
VIEW LOOKING UP

NOTE:



INSPECTION AREA

ANY CRACKS ARE NOT ACCEPTABLE



VIEW LOOKING UP

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01 SHEET 2/2

CONF 002
(3) Task A320-A-25-XX-1BKJ-02002-346A-A - Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.00 h
Minimum Number of person	1
Elapsed Time	1.00 h
Skills	Non-Destructive Testing

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00
SB	A320-53-1215
R53113174000	all sheets
R53113174001	all sheets

Procedure
(a) Standard Practices:

- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
- 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.

(b) Do an SDI of the lower central support of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).

Refer to Ref. NTM 51-90-00

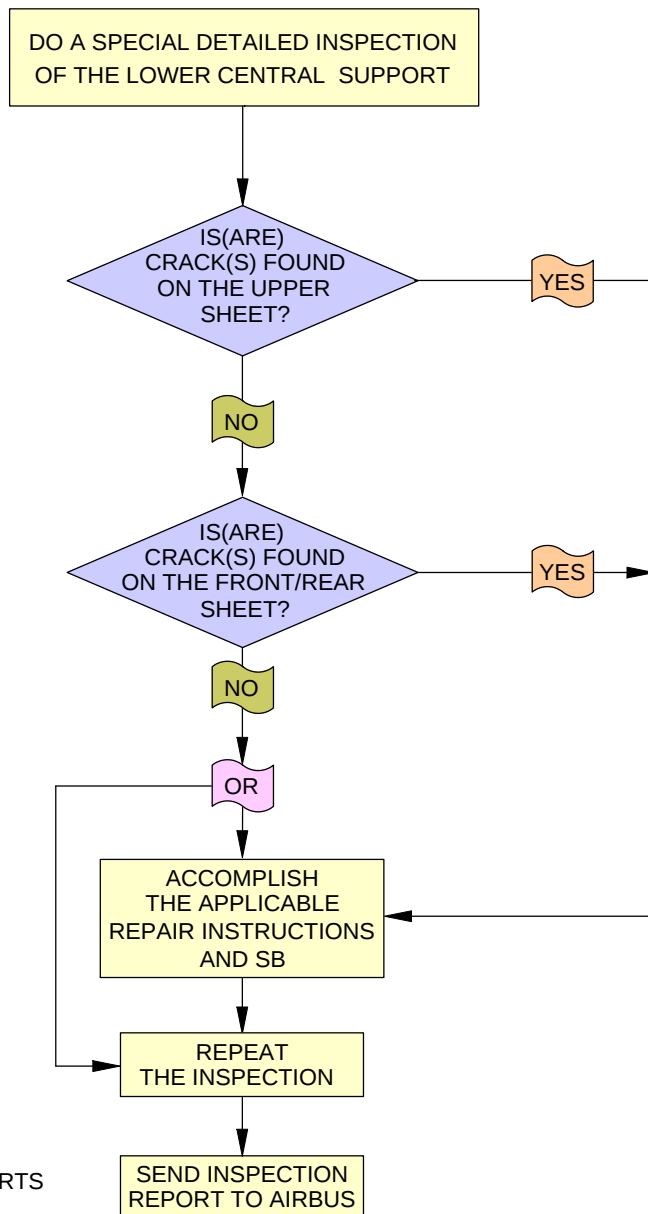
Refer to Ref. Fig. Flowchart

Refer to Ref. Fig. SDI of the Lower Central Support of the Electronics Rack FIN 80VU

- 1 If any crack(s) is(are) found on the upper sheet:
 - a Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A and accomplish Service Bulletin No. Ref. SB A320-53-1215.
 - b Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A.

- 2 If no crack(s) is(are) found on the upper sheet:
- a Check the front/rear sheets for crack(s):
- <1> If any crack(s) is(are) found on the front/rear sheets:
- <a> Repair in accordance with refer to Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A and accomplish Service Bulletin No. Ref. SB A320-53-1215.
- Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A
- <2> If no crack(s) is(are) found on the front/rear sheets:
- <a> Repair in accordance with Repair Instruction Ref. R53113174000 all sheets and Ref. R53113174001 all sheets, refer to Ref. Task set A320-A-25-XX-1BKJ-01ZZZ-93CZ-A and accomplish Service Bulletin No. Ref. SB A320-53-1215.
OR
- Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-02ZZZ-346Z-A.
- (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF 002



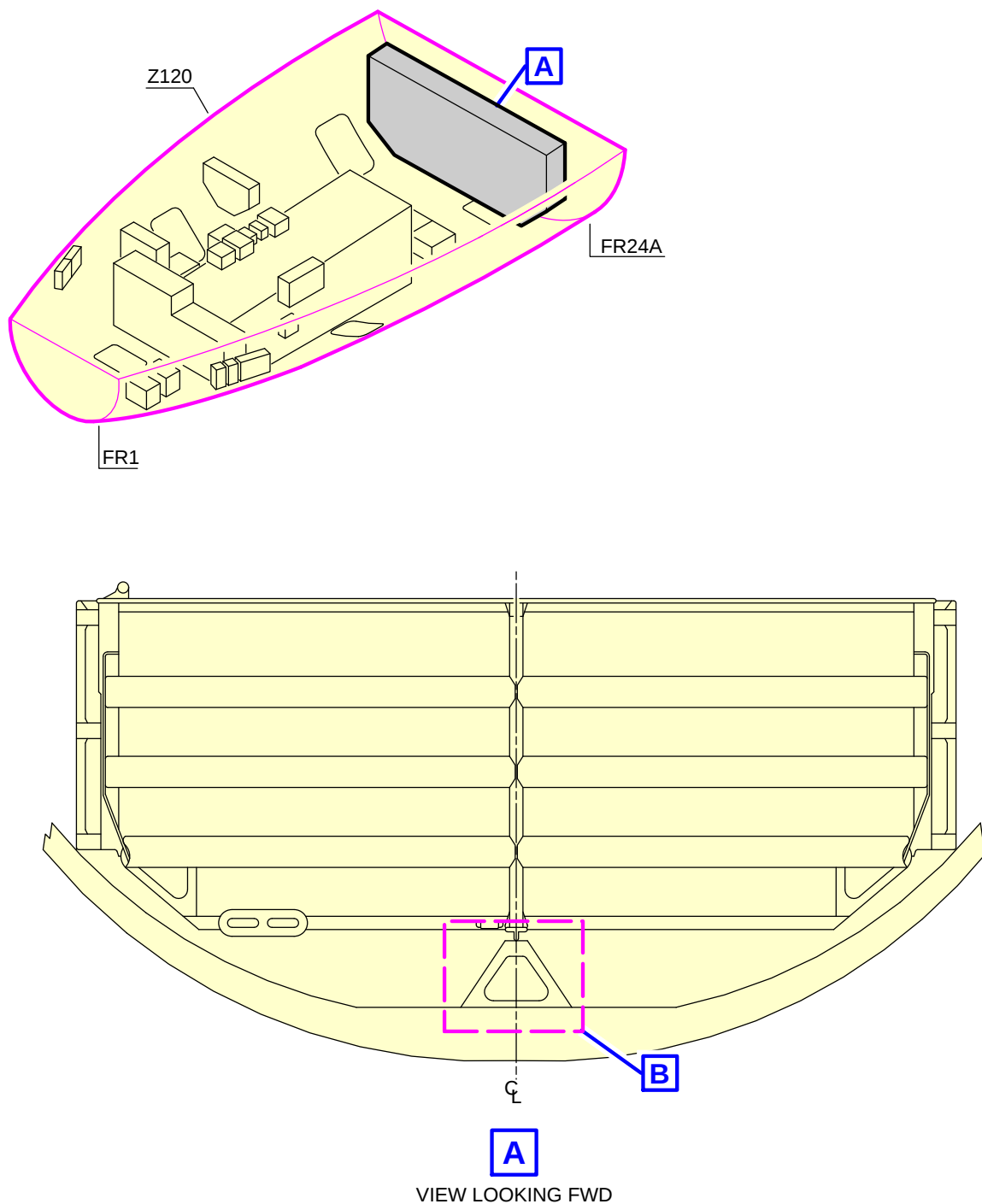
NOTE:
THE PURPOSE OF FLOWCHARTS
IS TO SUPPLEMENT THE
INFORMATION GIVEN IN THE
PROCEDURE AND COMPLIANCE
PARAGRAPHS AND NOT TO
SERVE AS THE PRIMARY SOURCE
FOR TASKS OR COMPLIANCE TIMES
GIVEN IN THIS SERVICE BULLETIN.

FOR THE COMPLIANCE TIMES,
REFER TO THE ACCOMPLISHMENT
TIMESCALE PARA 1.E.(2).

ICN-A320-A-25XX1BKJ-A-FAPE3-00EP5-A-008-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP5-A-008-01 SHEET 1/1

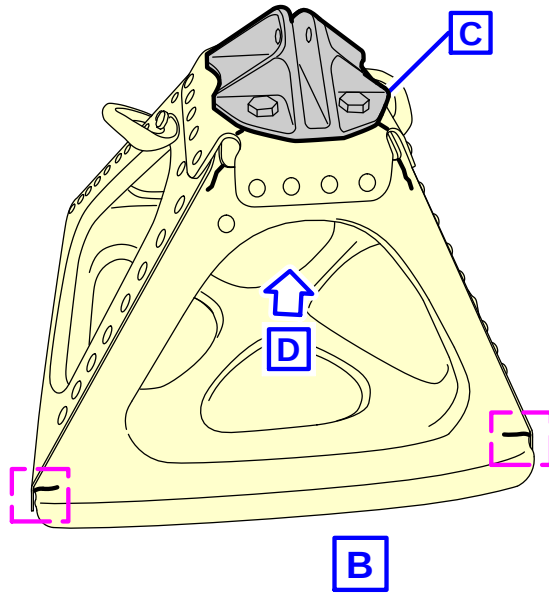
CONF 002



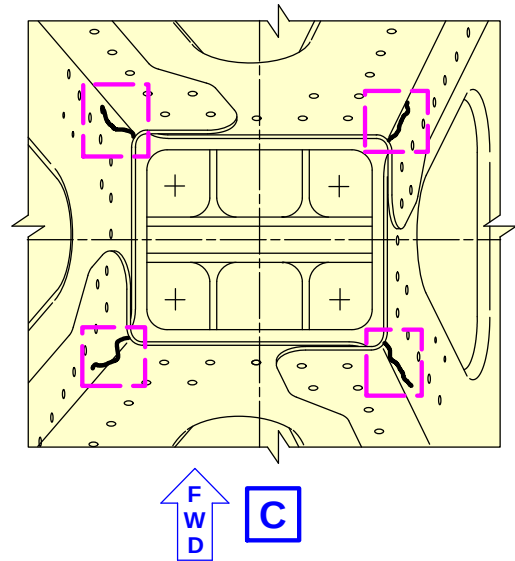
ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01 SHEET 1/2

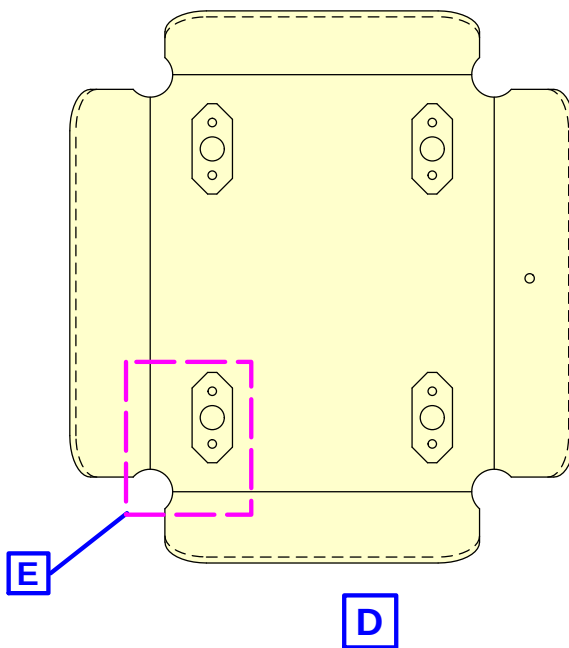
CONF 002



VIEW LOOKING FWD



VIEW LOOKING DOWN



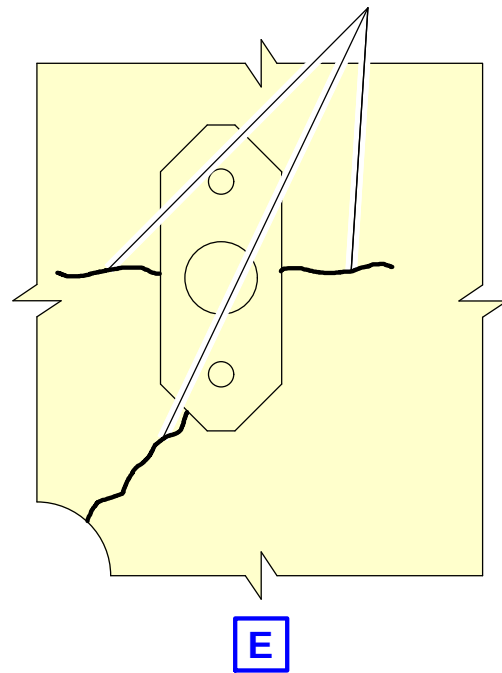
VIEW LOOKING UP

NOTE:



INSPECTION AREA

ANY CRACKS ARE NOT ACCEPTABLE



VIEW LOOKING UP

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01 SHEET 2/2

CONF 003
(3) Task A320-A-25-XX-1BKJ-02003-346A-A - Do an SDI of the Lower Central Support of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.00 h
Minimum Number of person	1
Elapsed Time	1.00 h
Skills	Non-Destructive Testing

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)
NTM	51-90-00

Procedure
(a) Standard Practices:

- 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
- 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.

(b) Do an SDI of the lower central support of the Electronics Rack FIN 80VU using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).

Refer to Ref. NTM 51-90-00

Refer to Ref. Fig. Flowchart

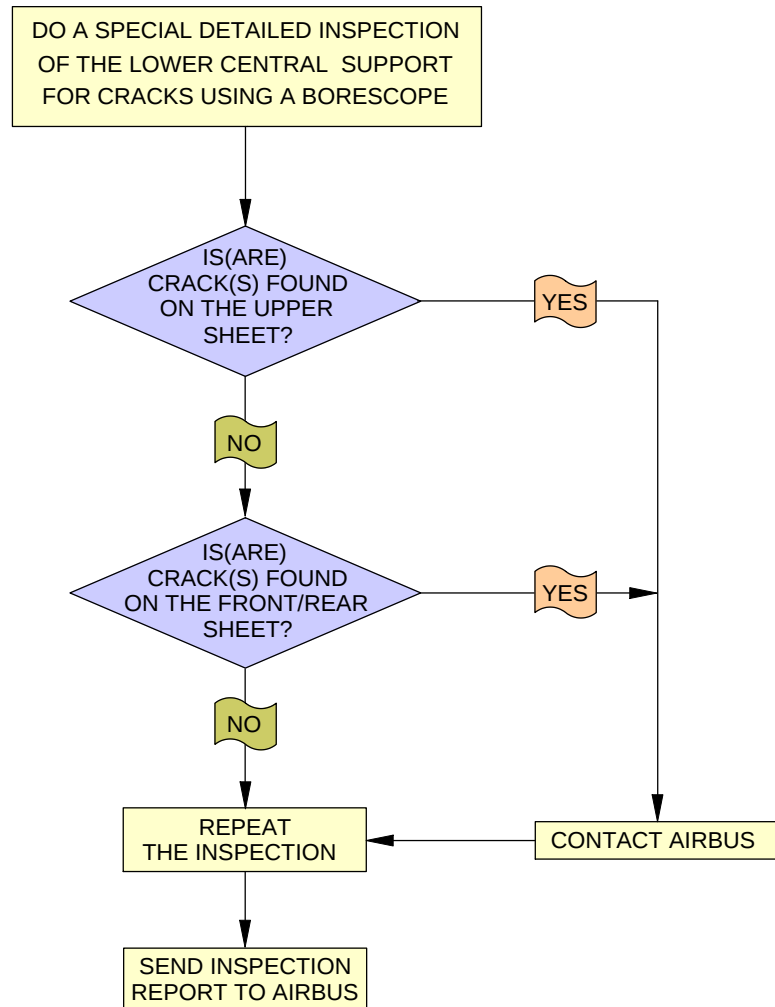
Refer to Ref. Fig. SDI of the Lower Central Support of the Electronics Rack FIN 80VU

- 1 If any crack(s) is(are) found on the upper sheet:
 - a Contact AIRBUS before next flight and follow their instructions.
- 2 If no crack(s) is(are) found on the upper sheet:
 - a Check the front/rear sheets for crack(s):
 - <1> If any crack(s) is(are) found on the front/rear sheets:
 - <a> Contact AIRBUS before next flight and follow their instructions.
 - <2> If no crack(s) is(are) found on the front/rear sheets:

<a> Repeat the inspection in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.

- (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF 003



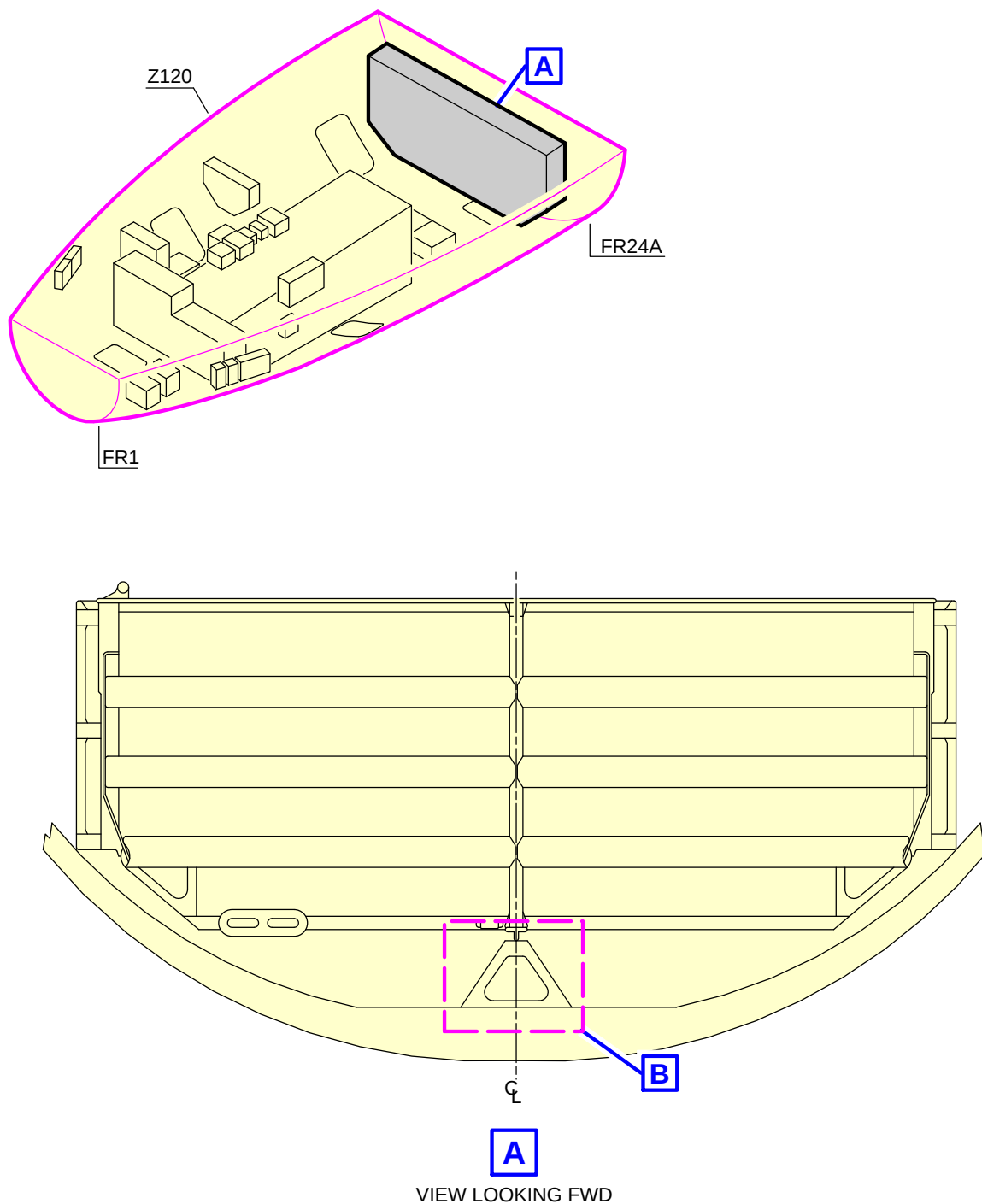
NOTE:
THE PURPOSE OF FLOWCHARTS
IS TO SUPPLEMENT THE
INFORMATION GIVEN IN THE
PROCEDURE AND COMPLIANCE
PARAGRAPHS AND NOT TO
SERVE AS THE PRIMARY SOURCE
FOR TASKS OR COMPLIANCE TIMES
GIVEN IN THIS SERVICE BULLETIN.

FOR THE COMPLIANCE TIMES,
REFER TO THE ACCOMPLISHMENT
TIMESCALE PARA 1.E.(2).

ICN-A320-A-25XX1BKJ-A-FAPE3-00EP6-A-008-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP6-A-008-01 SHEET 1/1

CONF 003



ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01

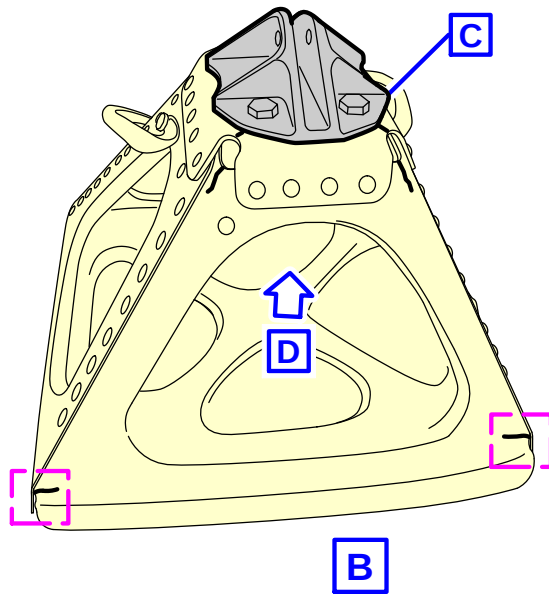
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOU-A-001-01 SHEET 1/2

REV 00 Aug 23/18
REV 02 Apr 09/20

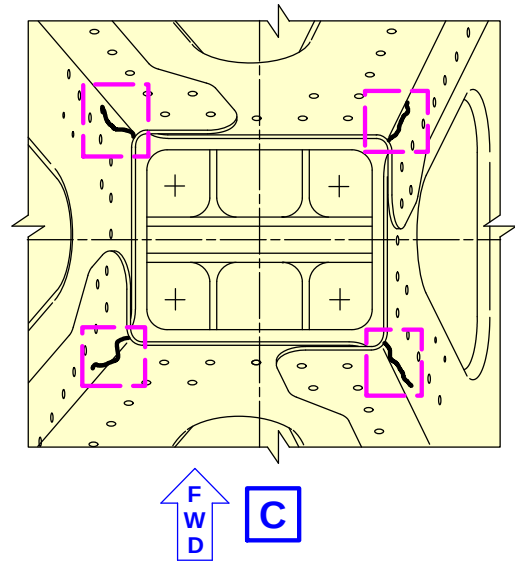
SDI of the Lower Central Support
of the Electronics Rack FIN 80VU

A320-25-1BKJ
Page 145

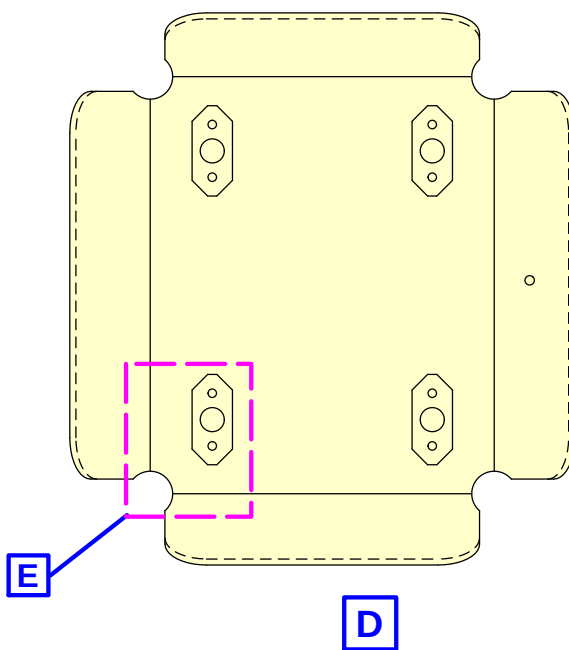
CONF 003



VIEW LOOKING FWD



VIEW LOOKING DOWN



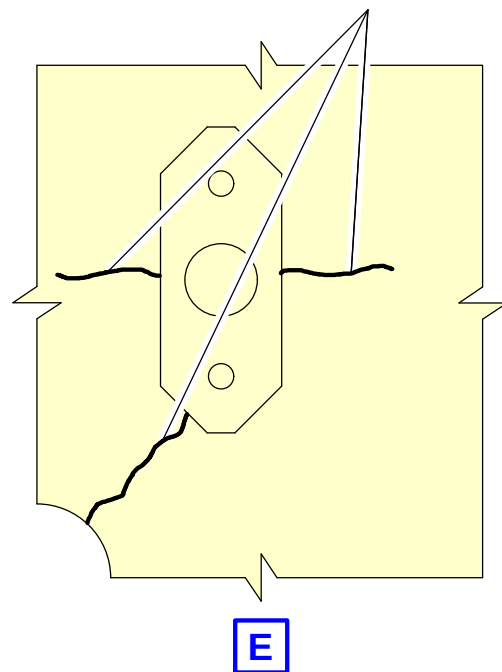
VIEW LOOKING UP

NOTE:



INSPECTION AREA

ANY CRACKS ARE NOT ACCEPTABLE



VIEW LOOKING UP

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOV-A-001-01 SHEET 2/2

CONF ALL
(4) Task A320-A-25-XX-1BKJ-04001-346A-A - Do an SDI of Central Post and Shelves Attachments of the Electronics Rack FIN 80VU
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.50 h
Minimum Number of person	1
Elapsed Time	1.50 h
Skills	Non-Destructive Testing

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 001 06-31-53, Page Block 001
CML	CONSUMABLE MATERIAL LIST (CML)

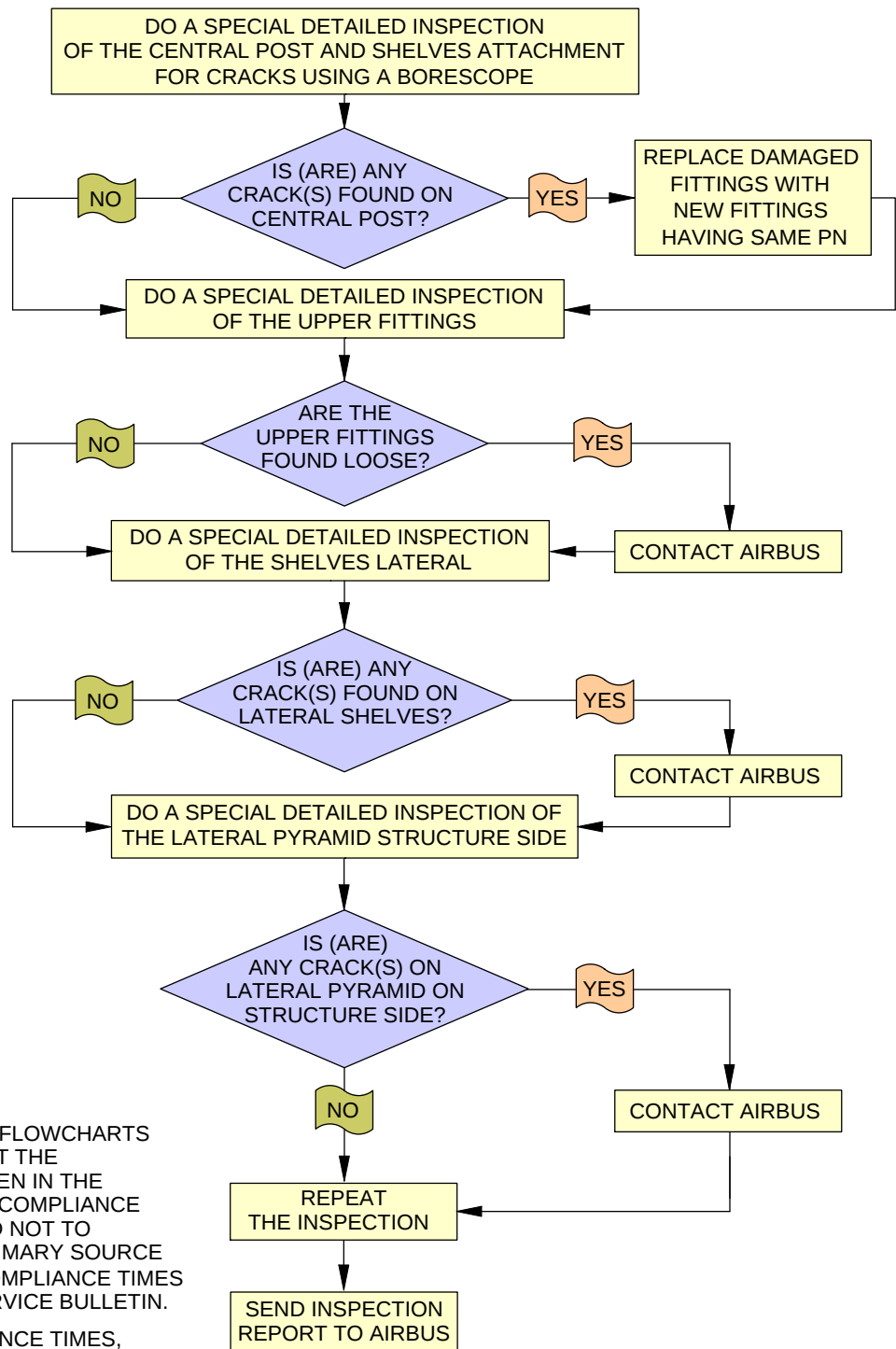
Procedure

- (a) Standard Practices:
 - 1 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 001.
 - 2 For the FR identification, refer to Ref. AMM 06-31-53, Page Block 001.
 - 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.
- (b) Do an SDI of the central post of the Electronics Rack FIN 80VU for cracks using a borescope within the threshold given in the PLANNING INFORMATION paragraph E.(2).
 Refer to Ref. Fig. Flowchart
 Refer to Ref. Fig. SDI of the Central Post and Shelves Attachments of the Electronics Rack FIN 80VU
 Refer to Ref. Appendix Inspection Condition of the Central Post
 Refer to Ref. Appendix Inspection Condition of the Lateral Shelves Attachment
 Refer to Ref. Appendix Inspection Condition of the Lateral Pyramid Structure
 - 1 If any crack(s) is(are) found on the central post:
 - a Replace the damaged fittings with new fittings which have the same PN.
 - 2 If no crack(s) is(are) found on the central post:
 - a Continue with the next workstep.
 - 3 Do an SDI of the upper fittings.
 - a If the upper fittings are found loose:
 - <1> Contact AIRBUS before next flight and follow their instructions.

- If the upper fittings are not found loose:
 - <1> Continue with the next workstep.
 - 4 Do an SDI of the lateral shelves.
 - a If any crack(s) is(are) found on the lateral shelves:
 - <1> Contact AIRBUS before next flight and follow their instructions.
 - b If no crack(s) is(are) found on the lateral shelves:
 - <1> Continue with the next workstep.
 - 5 Do an SDI of the lateral pyramid structure side.
 - a If crack(s) is(are) found on the lateral pyramid structure side:
 - <1> Contact AIRBUS before next flight and follow their instructions.
 - b If no crack(s) is(are) found on lateral pyramid structure side:
 - <1> Continue with the next workstep.
 - 6 Repeat the inspection within the threshold given in the PLANNING INFORMATION paragraph E.(2) in accordance with Ref. Task A320-A-25-XX-1BKJ-01ZZZ-346Z-A.
 - (c) After completion of all inspections, complete and send the Inspection Report sheet in accordance with Ref. Task A320-A-25-XX-1BKJ-03ZZZ-90AZ-A.

CONF ALL

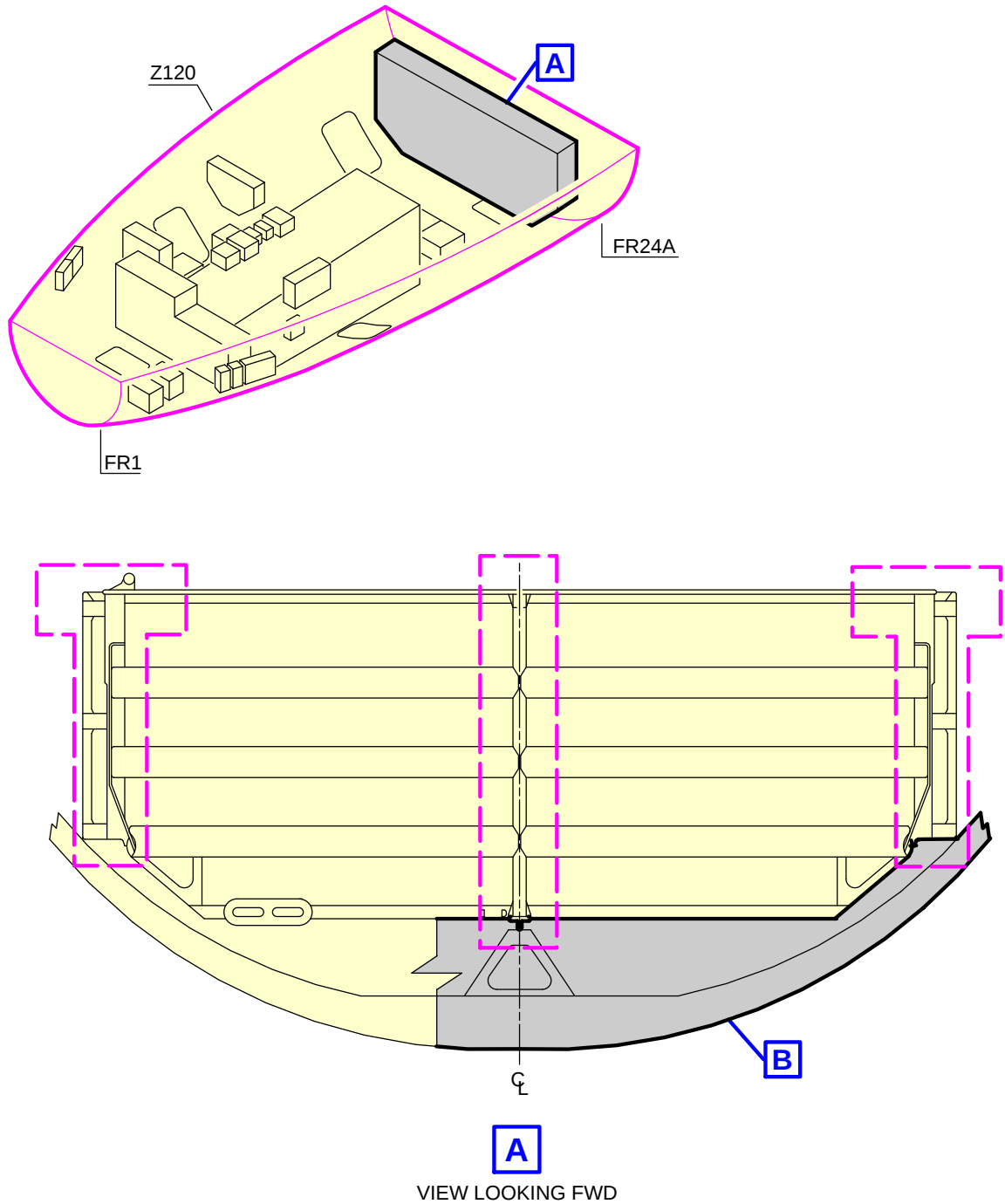
NOTE:
THE PURPOSE OF FLOWCHARTS IS TO SUPPLEMENT THE INFORMATION GIVEN IN THE PROCEDURE AND COMPLIANCE PARAGRAPHS AND NOT TO SERVE AS THE PRIMARY SOURCE FOR TASKS OR COMPLIANCE TIMES GIVEN IN THIS SERVICE BULLETIN. FOR THE COMPLIANCE TIMES, REFER TO THE ACCOMPLISHMENT TIMESCALE PARA 1.E.(2).



ICN-A320-A-25XX1BKJ-A-FAPE3-00EP8-A-009-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EP8-A-009-01 SHEET 1/1

CONF ALL



NOTE:

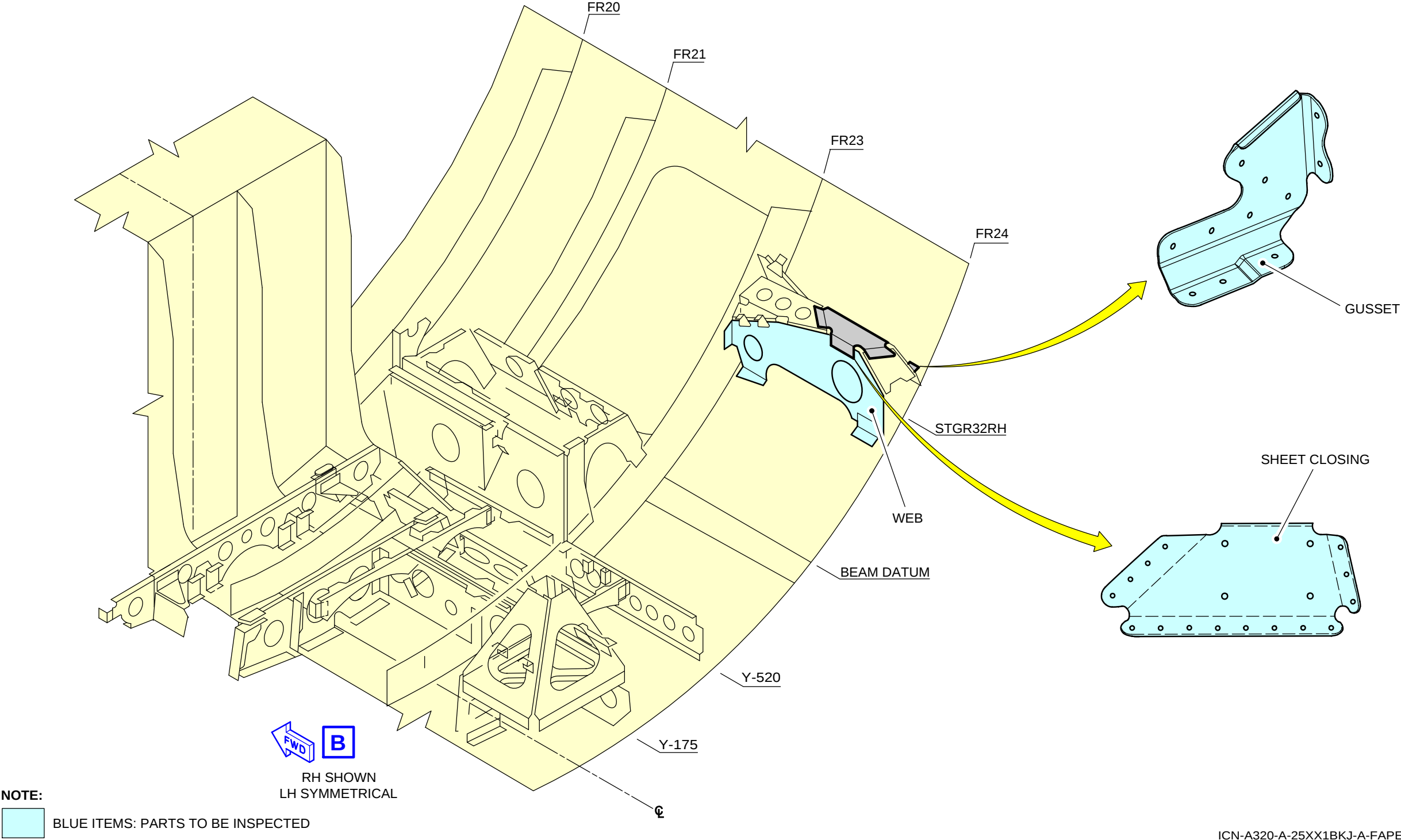


INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOW-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOW-A-001-01 SHEET 1/2

CONF ALL



ICN-A320-A-25XX1BKJ-A-FAPE3-00EOX-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOX-A-001-01 SHEET 2/2
SDI of the Central Post and Shelves Attachments of the Electronics Rack FIN 80VU

(5) Task A320-A-25-XX-1BKJ-03001-90AA-A - Complete the Inspection Report sheet
Job Set-Up Information
Manpower/Resources/Skills

Man-hours	0.25 h
Minimum Number of person	1
Elapsed Time	0.25 h

Referenced Information

Manuals/Drawings	References/Sheets
ISI	00.00.00179

Procedure

- (a) Complete the Inspection Report sheet, refer to Ref. Fig. Inspection Report sheet.
- (b) Send the completed Inspection Report sheet in accordance with Ref. ISI 00.00.00179.

NOTE: Sending the Inspection Report sheet is required for Compliance.
 The compliance time is 90 days since Service Bulletin accomplishment.

CONF 001-003

INSPECTION REPORT					
SERVICE BULLETIN No : A320-25-1BKJ			UPLOAD THE COMPLETED INSPECTION REPORT SHEET IN ACCORDANCE WITH ISI 00.00.00179		
A/C MSN :					
FLIGHT CYCLES :					
FLIGHT HOURS :					
DATE OF INSPECTION :					
INSPECTION METHOD : ☐ SPECIAL DETAILED INSPECTION					

	LOWER LATERAL 80VU LH SIDE		LOWER LATERAL 80VU RH SIDE		REPAIR WITH NEW FITTINGS APPLYING SB A320-25-1557 OR RI R53113174000 LH AND RI R53113174001 RH
	YES	NO	YES	NO	
FITTING BROKEN OR BOLT MISSING?					
RACK 80VU IN CONTACT WITH STRUCTURE FITTING?					
BUSH MIGRATED OR MATERIAL BURRED?					

ICN-A320-A-25XX1BKJ-A-FAPE3-01W92-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W92-A-001-01 SHEET 1/2

CONF 001-003

INSPECTION REPORT																	
SERVICE BULLETIN No : A320-25-1BKJ												UPLOAD THE COMPLETED INSPECTION REPORT SHEET IN ACCORDANCE WITH ISI 00.00.00179					
A/C MSN :																	
FLIGHT CYCLES :																	
FLIGHT HOURS :																	
DATE OF INSPECTION :																	
INSPECTION METHOD : ☐ SPECIAL DETAILED INSPECTION																	
	ON THE UPPER SHEET OF THE CENTRAL SUPPORT		ON THE FRONT SHEET OF THE CENTRAL SUPPORT				ON THE REAR SHEET OF THE CENTRAL SUPPORT				ON THE SHELVES ATTACHMENT		ON THE UPPER FITTING		ON THE LATERAL PYRAMID STRUCTURE		
			UPPER		BOTTOM		UPPER		BOTTOM								
	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	
CRACK(S) FOUND?																	
CRACK(S) LENGTH		N/A		N/A		N/A		N/A		N/A							
CRACK(S) LENGTH MORE THAN 40 mm (1.574 in)?			N/A														
CRACK(S) LENGTH LESS THAN OR EQUAL 40 mm (1.574 in)?			N/A														
REPAIR THE LOWER CENTRAL SUPPORT WITH: SB A320-53-1215☐																	
REMARKS :																	

ICN-A320-A-25XX1BKJ-A-FAPE3-01W93-A-002-01

CONF 004

INSPECTION REPORT					
SERVICE BULLETIN No : A320-25-1BKJ			UPLOAD THE COMPLETED INSPECTION REPORT SHEET IN ACCORDANCE WITH ISI 00.00.00179		
A/C MSN :					
FLIGHT CYCLES :					
FLIGHT HOURS :					
DATE OF INSPECTION :					
INSPECTION METHOD : ☐ SPECIAL DETAILED INSPECTION					

	LOWER LATERAL 80VU LH SIDE		LOWER LATERAL 80VU RH SIDE		REPAIR WITH NEW FITTINGS APPLYING SB A320-25-1557 OR RI R53113174002 LH AND RI R53113174003 RH
	YES	NO	YES	NO	
FITTING BROKEN OR BOLT MISSING?					
RACK 80VU IN CONTACT WITH STRUCTURE FITTING?					
BUSH MIGRATED OR MATERIAL BURRED?					

ICN-A320-A-25XX1BKJ-A-FAPE3-01W90-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W90-A-001-01 SHEET 1/2

CONF 004

INSPECTION REPORT																		
SERVICE BULLETIN No : A320-25-1BKJ										UPLOAD THE COMPLETED INSPECTION REPORT SHEET IN ACCORDANCE WITH ISI 00.00.00179								
A/C MSN :																		
FLIGHT CYCLES :																		
FLIGHT HOURS :																		
DATE OF INSPECTION :																		
INSPECTION METHOD : ☐ SPECIAL DETAILED INSPECTION																		
	ON THE UPPER SHEET OF THE CENTRAL SUPPORT		ON THE RODS AND FITTINGS OF THE CENTRAL SUPPORT		ON THE FRONT SHEET OF THE CENTRAL SUPPORT		ON THE REAR SHEET OF THE CENTRAL SUPPORT		ON THE SHELVES ATTACHMENT		ON THE UPPER FITTING		ON THE LATERAL PYRAMID STRUCTURE					
					UPPER		BOTTOM		UPPER		BOTTOM							
	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO	YES	NO
CRACK(S) FOUND?																		
CRACK(S) LENGTH		N/A				N/A		N/A		N/A		N/A						
CRACK(S) LENGTH MORE THAN 40 mm (1.574 in)?			N/A															
CRACK(S) LENGTH LESS THAN 40 mm (1.574 in)?			N/A															
REPAIR THE LOWER CENTRAL SUPPORT WITH: SB A320-53-1215☐																		
REMARKS :																		

ICN-A320-A-25XX1BKJ-A-FAPE3-01W91-A-001-01

CONF ALL
D. PREPARATION FOR TEST

None

E. TEST

None

F. CLOSE-UP
CONF 001
(1) Task A320-A-25-XX-1BKJ-01001-740A-A - Close-up
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.50 h
Minimum Number of person	2
Elapsed Time	1.00 h

Referenced Information

Manuals/Drawings	References/Sheets
AMM	Task 25-54-11-400-001 Task 52-30-00-860-002 Task 52-41-00-410-002

Procedure

- (a) Make sure that the work areas are clean and clear of tools and other items of equipment.
- (b) Install the FWD cargo compartment panels, Panel 131DW and/or Panel 131CW and/or Panel 132EW, refer to Ref. AMM Task 25-54-11-400-001.
- (c) Close the FWD cargo compartment door, Door 825, refer to Ref. AMM Task 52-30-00-860-002.
- (d) Close the avionics compartment door, Door 824, refer to Ref. AMM Task 52-41-00-410-002.
- (e) Put the aircraft back to its initial configuration.

CONF 002-004
(1) Task A320-A-25-XX-1BKJ-01002-740A-A - Close-up
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.50 h
Minimum Number of person	2
Elapsed Time	1.00 h

Referenced Information

Manuals/Drawings	References/Sheets
AMM	Task 25-54-11-400-001 Task 52-30-00-860-002 Task 52-41-00-410-002

Procedure

NOTE: If you intend to accomplish an associated Repair Task during the same layover period, the subsequent Close-up procedure is not necessary.

- (a) Make sure that the work areas are clean and clear of tools and other items of equipment.
- (b) Install the FWD cargo compartment panels, Panel 131DW and/or Panel 131CW and/or Panel 132EW, refer to Ref. AMM Task 25-54-11-400-001.
- (c) Close the FWD cargo compartment door, Door 825, refer to Ref. AMM Task 52-30-00-860-002.
- (d) Close the avionics compartment door, Door 824, refer to Ref. AMM Task 52-41-00-410-002.
- (e) Put the aircraft back to its initial configuration.

CONF ALL

G. ADMINISTRATIVE
(1) Task A320-A-25-XX-1BKJ-01001-90AA-A - Administrative
Procedure

Confirm in the applicable aircraft records that you have completed all the work defined in this Service Bulletin task set as required.

CONF 002-004

Task set A320-A-25-XX-1BKJ-01000-93CA-A

Repair of the Electronics Rack FIN 80VU

WARNING: MAKE SURE THAT YOU OBEY ALL THE WARNINGS AND ALL THE CAUTIONS INCLUDED IN THE REFERENCED PROCEDURES.

CAUTION: ALWAYS OBEY THE PRECAUTIONS THAT FOLLOW TO KEEP ELECTRICAL WIRING IN A SATISFACTORY CONDITION (ELECTRICALLY AND MECHANICALLY SERVICEABLE). WHEN YOU DO MAINTENANCE WORK, REPAIRS MODIFICATIONS, ALWAYS KEEP ELECTRICAL WIRING, COMPONENTS AND THE WORK AREA AS CLEAN AS POSSIBLE. TO DO THIS:

- PUT PROTECTION, SUCH AS PLASTIC SHEETING, CLOTHS, ETC., AS NECESSARY ON WIRING AND COMPONENTS.
- REGULARLY REMOVE ALL SHAVINGS, UNWANTED MATERIAL AND OTHER CONTAMINATION.

**THESE PRECAUTIONS WILL DECREASE THE RISK OF CONTAMINATION AND DAMAGE TO THE ELECTRICAL WIRING INSTALLATION.
IF THERE IS CONTAMINATION, REFER TO Ref. ESPM 20-55-00.**

NOTE: The accomplishment instructions of this Service Bulletin include procedures given in other documents or in other sections of the Service Bulletin. When the words "refer to" are used and the operator has a procedure accepted by the local authority he belongs to, the accepted alternative procedure can be used. When the words "in accordance with" are used then the given procedure must be followed.

NOTE: The access and close-up instructions, not comprising return to service tests, in this Service Bulletin do not constitute or affect the technical intent of the Service Bulletin. Operators can therefore, as deemed necessary, omit or add access and/or close-up steps to add flexibility to their maintenance operations as long as the technical intent of the Service Bulletin is met within the set parameters.

NOTE: Manual titles given in the accomplishment instructions are referred to by acronyms. Refer to paragraph 1.J., References, for the definition of acronyms.

NOTE: This Service Bulletin contains the instructions for the on-aircraft maintenance necessary to ensure the continued airworthiness of the aircraft.
For any deviations to the instructions, including RC paragraph contained herein, contact AIRBUS for further instructions and approval.

NOTE: This Service Bulletin is classified mandatory or expected to be classified mandatory by an Airworthiness Directive (AD). The paragraphs 3.C. and 3.E. in these accomplishment instructions are Required for Compliance (RC) and must be done to comply with the AD. To allow more flexibility, the rule "refer to" and "in accordance with" will apply to these paragraphs 3.C. and 3.E. Other paragraphs are recommended and may be deviated from, done as part of other actions or done with accepted methods different from those given in the Service Bulletin, as long as the RC paragraphs can be done and the aircraft can be put back into a serviceable condition.

NOTE: Discarded parts must be managed at the operator's discretion.

A. TASK SET ASSOCIATED INFORMATION

Task Manpower

CONF 002

TOTAL MAN-HOURS	122.00 h
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CONF 003

TOTAL MAN-HOURS	122.00 h
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CONF 004

TOTAL MAN-HOURS 122.00 h

CONF 002

ELAPSED TIME (HOURS) 73.50 h

CONF 003

ELAPSED TIME (HOURS) 73.50 h

CONF 004

ELAPSED TIME (HOURS) 73.50 h

CONF 002-004

B. PREPARATION

(1) Task A320-A-25-XX-1BKJ-02002-540A-A - Job Set-up

Job Set-Up Information

Working Area

Impacted Zone	Access / Work Location	
126	Work location	From Frame 9 to Frame 20
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	8.25 h
Minimum Number of person	2
Elapsed Time	4.25 h
Skills	Airframe
	Electrical

Referenced Information

Manuals/Drawings	References/Sheets
AMM	Task 12-34-24-869-002 21-26-34, Page Block 401 22-66-34, Page Block 401 22-83-34, Page Block 401 23-11-33, Page Block 401 23-12-33, Page Block 401 23-24-34, Page Block 401 23-28-34, Page Block 401 23-32-34, Page Block 401 23-32-35, Page Block 401 23-33-31, Page Block 401 23-33-37, Page Block 401 23-33-39, Page Block 401 23-33-52, Page Block 401 23-34-34, Page Block 401 23-51-34, Page Block 401 23-72-31, Page Block 401 23-73-34, Page Block 401 Task 24-38-51-000-001 Task 25-54-12-000-003 26-12-34, Page Block 401 26-13-34, Page Block 401

Manuals/Drawings	References/Sheets
	27-51-34, Page Block 401 27-93-34, Page Block 401 27-94-34, Page Block 401 27-95-34, Page Block 401 31-32-34, Page Block 401 31-33-34, Page Block 401 31-33-52, Page Block 401 31-36-34, Page Block 401 31-36-52, Page Block 401 31-38-34, Page Block 401 31-53-34, Page Block 401 31-55-34, Page Block 401 31-63-34, Page Block 401 34-35-34, Page Block 401 34-43-34, Page Block 401 34-48-34, Page Block 401 34-51-33, Page Block 401 34-52-33, Page Block 401 34-52-35, Page Block 401 34-53-31, Page Block 401 34-55-31, Page Block 401 46-21-34, Page Block 401 Task 53-12-13-000-001 Task 53-22-14-000-001 73-25-34, Page Block 401 77-32-34, Page Block 401
ESPM	20-54-50

Procedure

NOTE: For the basic Aircraft configuration, refer to AMM Chapter "Introduction - Aircraft status for maintenance" in Manual Front Matter section.

- (a) Make sure that the aircraft is electrically grounded, refer to Ref. AMM Task 12-34-24-869-002.
- (b) Make sure that the external connector(s) is (are) not connected to the aircraft receptacle(s) EXT PW.
- (c) Make sure that there is (are) warning notice(s) on external power receptacle to tell persons not to connect an external power source.
- (d) Make sure that the EMER EXIT LT pushbutton switch on the Electrical Panel 25VU is in OFF position.
- (e) Put a warning notice on the refuel/defuel panel, Electrical Panel 800VU, to tell persons not to operate the BATTERY switch.
- (f) Put a warning notice in the cockpit to tell persons not to start the APU.
- (g) Put a warning notice in the cockpit to tell persons not to start the ENG 1 and/or ENG 2.
- (h) For the protection of the connectors, contacts and terminals, refer to Ref. ESPM 20-54-50.
- (i) Tag all the parts you must remove and retain to make the subsequent installation easier.

- (j) Put all the retained hardware in identified plastic bags and attach the bags to the related unit.
- (k) Battery disconnection:
 - 1 Do the Job Set-up procedure as specified in the removal/installation of the batteries, refer to Ref. AMM Task 24-38-51-000-001.
 - 2 Turn the knurled nut a quarter turn.
 - 3 Disconnect the connectors.
 - 4 Put blanking caps on the disconnected electrical connectors.
- (l) Remove the sidewall panels, Panel 131GW, refer to Ref. AMM Task 25-54-12-000-003.
- (m) Remove the FWD cargo compartment floor panels, Panel 131CF or Panel 131RG, Panel 131BF or Panel 131QG and Panel 131AF or Panel 131LF and Panel 132MF or Panel 131PG, refer to Ref. AMM Task 53-22-14-000-001.
- (n) Remove all the equipment from the Electronics Rack FIN 80VU, depending on the aircraft configuration:
 - 1 In the Shelf FIN 81VU, remove:
 - a Transceiver HF (3RE1), refer to Ref. AMM 23-11-33, Page Block 401.
 - b ATSU (1TX1), refer to Ref. AMM 46-21-34, Page Block 401.
 - c ATC/Mode S Transponder (1SH1), refer to Ref. AMM 34-52-33, Page Block 401.
 - d DME Interrogator (2SD1), refer to Ref. AMM 34-51-33, Page Block 401.
 - e VOR/MKR Receiver (3RS1), refer to Ref. AMM 34-55-31, Page Block 401.
 - f Transceiver VHF (1RC3), refer to Ref. AMM 23-12-33, Page Block 401.
 - g Transceiver VHF (1RC1), refer to Ref. AMM 23-12-33, Page Block 401.
 - h AMU (1RN), refer to Ref. AMM 23-51-34, Page Block 401.
 - i ADF Receiver (2RP1), refer to Ref. AMM 34-53-31, Page Block 401.
 - 2 In the Shelf FIN 82VU, remove:
 - a ADF Receiver (2RP2), refer to Ref. AMM 34-53-31, Page Block 401.
 - b ACARS Management Unit (MU) (1RB), refer to Ref. AMM 23-24-34, Page Block 401.
 - c VHF Transceiver (1RC2), refer to Ref. AMM 23-12-33, Page Block 401.
 - d DME Interrogator (2SD2), refer to Ref. AMM 34-51-33, Page Block 401.
 - e ATC/Mode S Transponder (1SH2), refer to Ref. AMM 34-52-33, Page Block 401.
 - f T2CAS Computer (100SG), refer to Ref. AMM 34-43-34, Page Block 401.
 - g TCAS Computer (1SG), refer to Ref. AMM 34-43-34, Page Block 401.
 - h Transceiver HF (3RE2), refer to Ref. AMM 23-11-33, Page Block 401.
 - i VOR/MKR Receiver (3RS2), refer to Ref. AMM 34-55-31, Page Block 401.
 - 3 In the Shelf FIN 83VU, remove:

- a Elevator Aileron (ELAC) (2CE1), refer to Ref. AMM 27-93-34, Page Block 401.
 - b Spoiler Elevator (SEC) (1CE1), refer to Ref. AMM 27-94-34, Page Block 401.
 - c Flight Control Data (FCDC) (3CE1), refer to Ref. AMM 27-95-34, Page Block 401.
 - d Flight Management and Guidance (FMGC) (1CA1), refer to Ref. AMM 22-83-34, Page Block 401.
 - e Flight Augmentation (FAC) (1CC1), refer to Ref. AMM 22-66-34, Page Block 401.
- 4 In the Shelf FIN 84VU, remove:
 - a FAC (1CC2), refer to Ref. AMM 22-66-34, Page Block 401.
 - b FMGC (1CA2), refer to Ref. AMM 22-83-34, Page Block 401.
 - c FCDC (3CE2), refer to Ref. AMM 27-95-34, Page Block 401.
 - d SEC (1CE2), refer to Ref. AMM 27-94-34, Page Block 401.
 - e ELAC (2CE2), refer to Ref. AMM 27-93-34, Page Block 401.
- 5 In the Shelf FIN 85VU, remove:
 - a Slat and Flap Control (SFCC) (21CV), refer to Ref. AMM 27-51-34, Page Block 401.
 - b Engine interface Unit (EIU) (1KS1), refer to Ref. AMM 73-25-34, Page Block 401.
 - c Display Management Computer (DMC) (1WT1), refer to Ref. AMM 31-63-34, Page Block 401.
 - d DMC FIN (1WT3), refer to Ref. AMM 31-63-34, Page Block 401.
 - e Flight Warning Computer (FWC) (1WW1), refer to Ref. AMM 31-53-34, Page Block 401.
 - f System Data Acquisition (SDAC) (1WV1), refer to Ref. AMM 31-55-34, Page Block 401.
 - g Cockpit Door Surveillance (CDSS) (9RA), refer to Ref. AMM 23-72-31, Page Block 401.
- 6 In the Shelf FIN 86VU, remove:
 - a SDAC (1WV2), refer to Ref. AMM 31-55-34, Page Block 401.
 - b FWC (1WW2), refer to Ref. AMM 31-53-34, Page Block 401.
 - c DMC (1WT2), refer to Ref. AMM 31-63-34, Page Block 401.
 - d Avionics Equipment Ventilation Computer (AEVC) (10HQ), refer to Ref. AMM 21-26-34, Page Block 401.
 - e Head Up Display Computer (HUDC) (1FH), refer to Ref. AMM 34-35-34, Page Block 401.
 - f Engine Vibration Monitoring Unit (EVMU) (2EV), refer to Ref. AMM 77-32-34, Page Block 401.
 - g EIU (1KS2), refer to Ref. AMM 73-25-34, Page Block 401.
 - h SFCC (22CV), refer to Ref. AMM 27-51-34, Page Block 401.

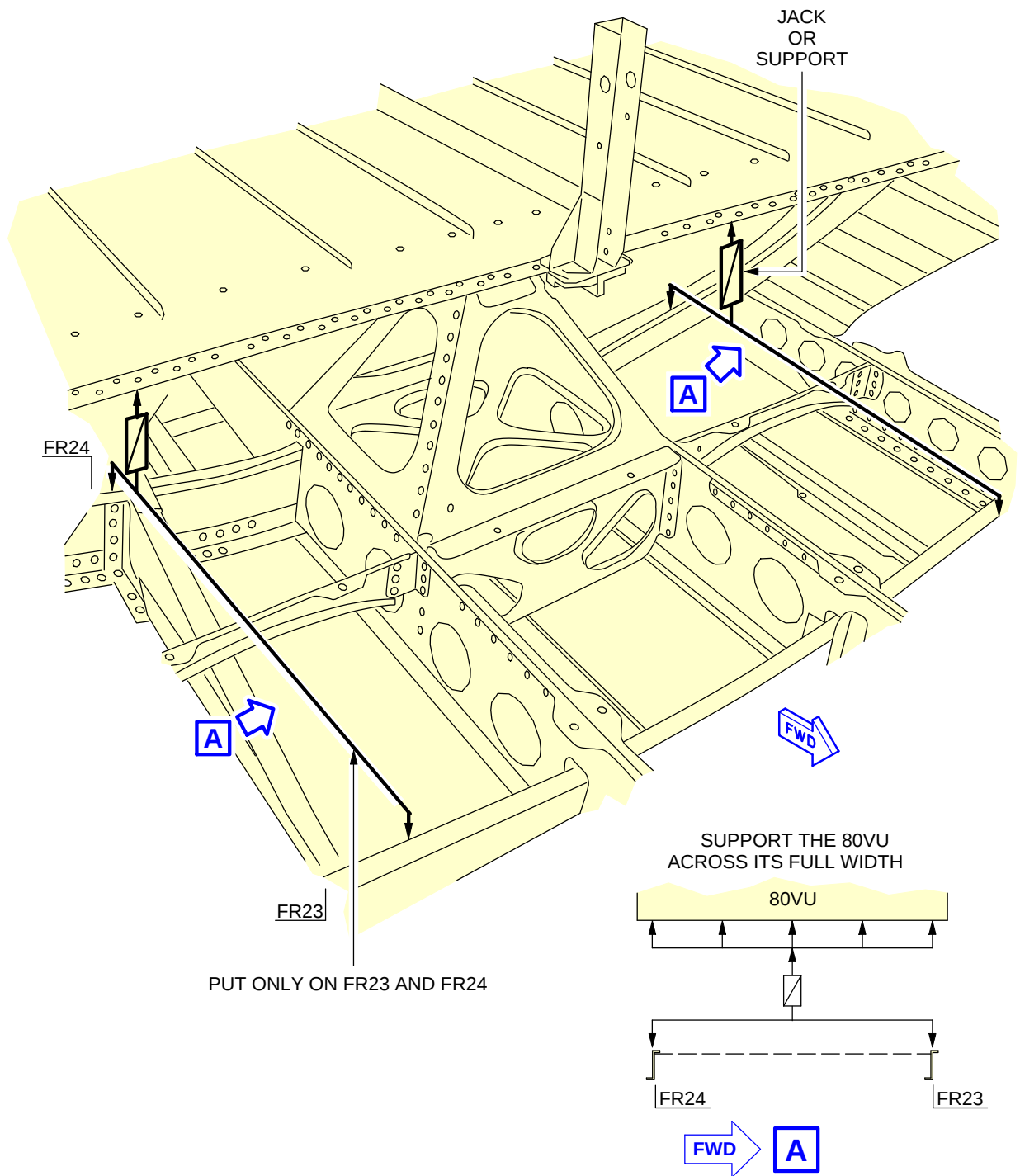
- 7 In the Shelf FIN 87VU, remove:
- a Digital AIDS Recorder (DAR) (2TV), refer to Ref. AMM 31-36-52, Page Block 401.
 - b AEVC (10HQ), refer to Ref. AMM 21-26-34, Page Block 401.
 - c Flight Data interface Unit (FDIU)/DMU (FDIMU) (10TV), refer to Ref. AMM 31-36-34, Page Block 401.
 - d FDIU (2TU), refer to Ref. AMM 31-33-34, Page Block 401.
 - e Quick Access Recorder (QAR) (3TU), refer to Ref. AMM 31-33-52, Page Block 401.
 - f DMU (1TV), refer to Ref. AMM 31-36-34, Page Block 401.
 - g Centralized Fault Display (CFDIU) (1TW), refer to Ref. AMM 31-32-34, Page Block 401.
 - h CIDS Director (101RH), refer to Ref. AMM 23-73-34, Page Block 401.
- 8 In the Shelf FIN 88VU, remove:
- a Prerecorded Announcement Voice Synthesizer Unit (PAVSU) (11RX), refer to Ref. AMM 23-32-35, Page Block 401.
 - b CIDS Director (102RH), refer to Ref. AMM 23-73-34, Page Block 401.
 - c Passenger Entertainment System Main Multiplexer (8MK), refer to Ref. AMM 23-33-31, Page Block 401.
 - d AEVC (10HQ), refer to Ref. AMM 21-26-34, Page Block 401.
 - e Remote ATC Control Box (8SH), refer to Ref. AMM 34-52-35, Page Block 401.
 - f Ground Proximity Warning System (GPWS) (1WZ), refer to Ref. AMM 34-48-34, Page Block 401.
 - g Satellite Data Unit (SDU) (105RV1), refer to Ref. AMM 23-28-34, Page Block 401.
 - h Digital Interface Unit (DIU) (190MH), refer to Ref. AMM 23-34-34, Page Block 401.
 - i Passenger Entertainment System (10MK), refer to Ref. AMM 23-33-52, Page Block 401.
 - j Announcement/Music Tape Reproducer (10RX), refer to Ref. AMM 23-32-34, Page Block 401.
 - k IFE Audio/Video Controller (314MK), refer to Ref. AMM 23-33-37, Page Block 401.
 - l IFE Audio/Video Controller (313MK), refer to Ref. AMM 23-33-37, Page Block 401.
 - m IFE Digital Server Unit (312MK), refer to Ref. AMM 23-33-39, Page Block 401.
 - n IFE Audio/Video Controller (311MK), refer to Ref. AMM 23-33-37, Page Block 401.
- (o) Remove all the equipment under the Electronics Rack FIN 80VU, depending on the configuration:
- 1 Fire Detection Unit (FDU) (2WD1), refer to Ref. AMM 26-12-34, Page Block 401.

- 2 FDU (2WD2), refer to Ref. AMM 26-12-34, Page Block 401.
- 3 FDU (13WG), refer to Ref. AMM 26-13-34, Page Block 401.
- 4 Data Loading Routing Box (102TD), refer to Ref. AMM 31-38-34, Page Block 401.
- (p) Remove the avionics compartment floor panels, (Panel 127BF, Panel 127CF, Panel 128CF and Panel 128EG or Panel 128EF), refer to Ref. AMM Task 53-12-13-000-001.
- (q) Jack the FIN 80VU in accordance with the principle given in Ref. Fig. Principle jacking of the 80VU.

NOTE: The jacking device is used to maintain and support the FIN 80VU in its original position. This position should remain the same during the modification.

- 1 The jack or support must be put:
 - a On Frame FR23 and Frame FR24,
 - b Or, on the panels, Panel 128CF, Panel 127CF and Panel 131AF or Panel 131LF, Panel 132MF or Panel 131PG (if installed).
- 2 Support the FIN 80VU across its full width.
- 3 If necessary, move the wire or pipe without disconnection.

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TM5-A-002-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM5-A-002-01 SHEET 1/2

CONF 002-004

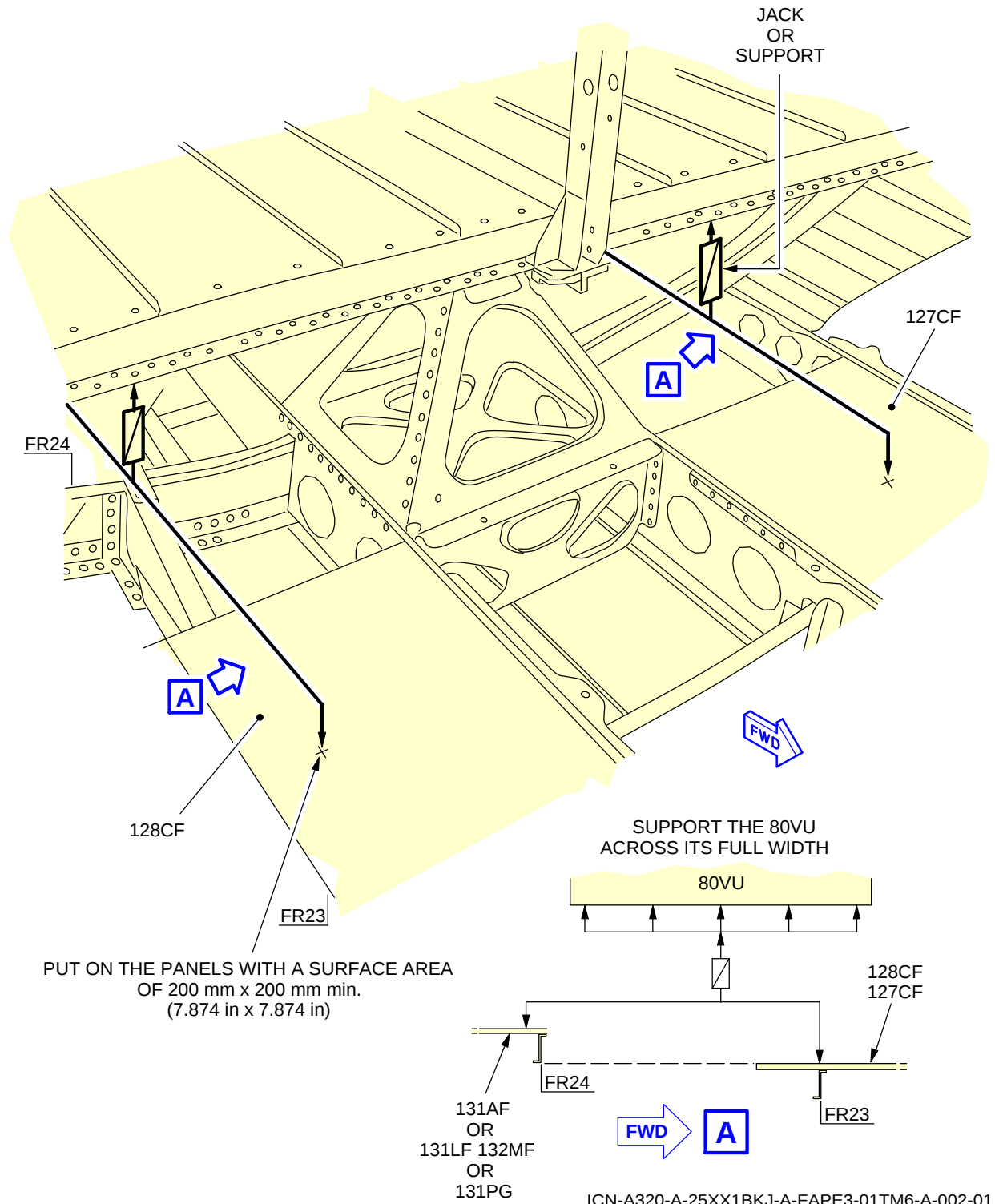


Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM6-A-002-01 SHEET 2/2

(2) Task A320-A-25-XX-1BKJ-03002-540A-A - Remove the ducts
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	2.00 h
Minimum Number of person	1
Elapsed Time	2.00 h
Skills	Airframe

Material necessary to do the job

Consumables				
ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	08BAA9	Non Aqueous Cleaner - General		
	14SBA1	Textile - Lint free Cotton		

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000 06-31-53, Page Block 000 Task 21-20-00-010-001
SRM	51-42-00

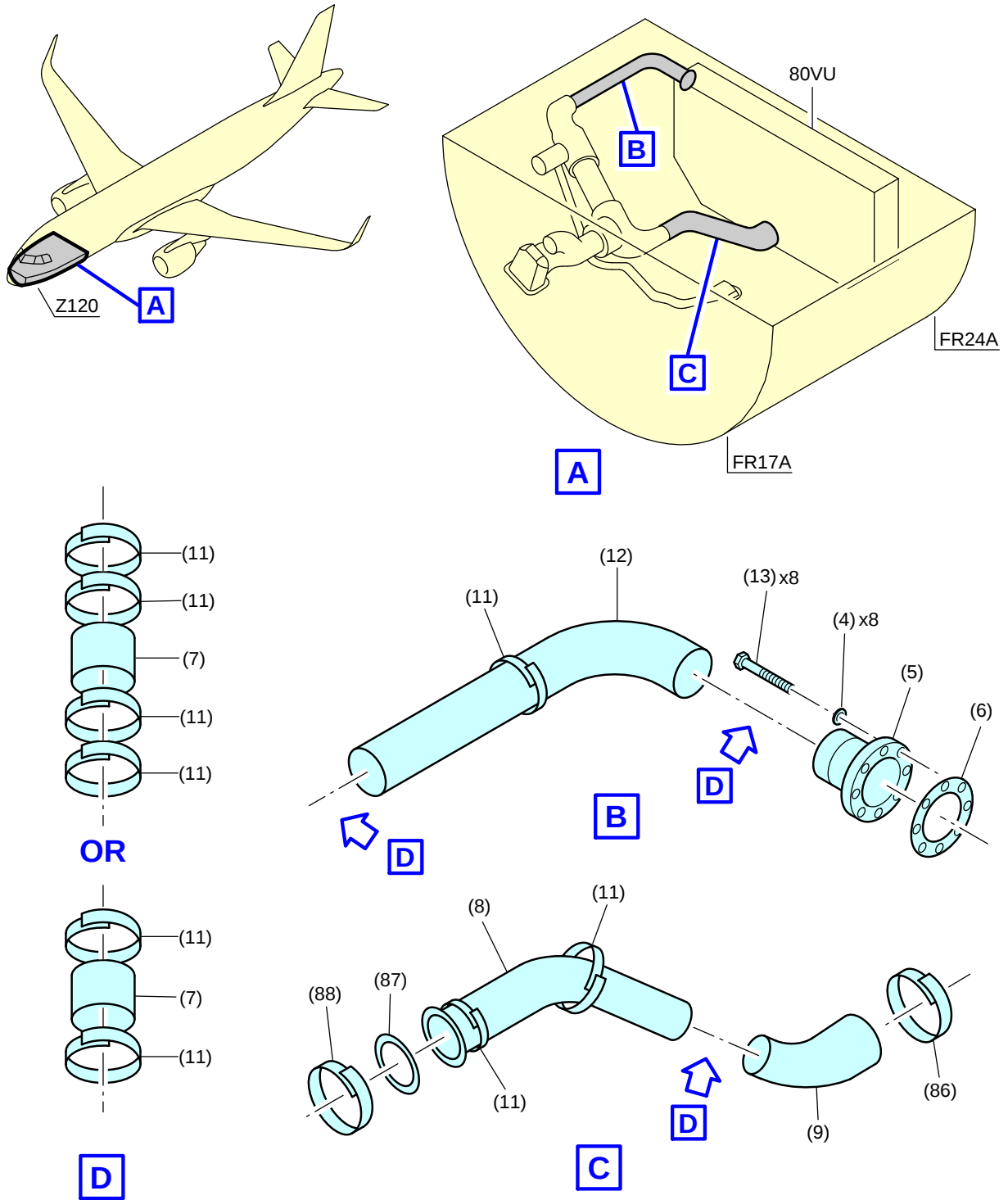
Procedure
(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.
- 3 Use a non-metallic spatula to remove the sealant and clean the areas with:
Non Aqueous Cleaner - 08BAA9
General
Textile - Lint free Cotton 14SBA1
- 4 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 5 Tag all the parts you must remove and retain to make the subsequent installation easier.
- 6 Put all the retained hardware in identified plastic bags and attach the bags to the related unit.
- 7 For the general removal of air-conditioning distribution ducts, refer to Ref. AMM Task 21-20-00-010-001.

(b) Remove, at the RH side of the FIN 80VU, refer to Ref. Fig. Removal of the RH ducts:			
1	DUCT ASSY	Item (5)	Retain
1	DUCT ASSY	Item (12)	Retain
1	SEAL attached with	Item (6)	Retain
8	SCREW	Item (13)	Retain
8	WASHER	Item (4)	Discard
2	SLEEVE	Item (7)	Retain
9	CLAMP	Item (11)	Retain
or			
5	CLAMP	Item (11)	Retain
(c) Remove from under the FIN 80VU, refer to Ref. Fig. Removal of the RH ducts:			
1	DUCT ASSY	Item (8)	Retain
1	DUCT ASSY	Item (9)	Retain
1	SEAL attached with	Item (87)	Discard
1	SLEEVE	Item (7)	Retain
1	CLAMP	Item (86)	Retain
1	CLAMP	Item (88)	Retain
6	CLAMP	Item (11)	Retain
or			
4	CLAMP	Item (11)	Retain
(d) Remove, at the LH side of the FIN 80VU, refer to Ref. Fig. Removal of the LH duct:			
1	DUCT attached with	Item (30)	Retain
6	CLAMP	Item (11)	Retain
1	SLEEVE	Item (7)	Retain
or			
4	CLAMP	Item (11)	Retain
1	SLEEVE	Item (7)	Retain
(e) Remove, at the LH side of the FIN 80VU, refer to Ref. Fig. Removal of the upper LH duct:			
1	DUCT	Item (17)	Retain
1	DUCT ASSY	Item (18)	Retain
1	DUCT ASSY	Item (22)	Retain

1	DIAPHRAGM	Item (10)	Retain
1	SEAL	Item (19)	Retain
1	SEAL attached with	Item (6)	Retain
8	SCREW	Item (14)	Retain
10	SCREW	Item (13)	Retain
18	WASHER	Item (4)	Discard
3	SLEEVE	Item (25)	Retain
1	CLAMP	Item (16)	Retain
12	CLAMP	Item (24)	Retain
	or		
6	CLAMP	Item (24)	Retain

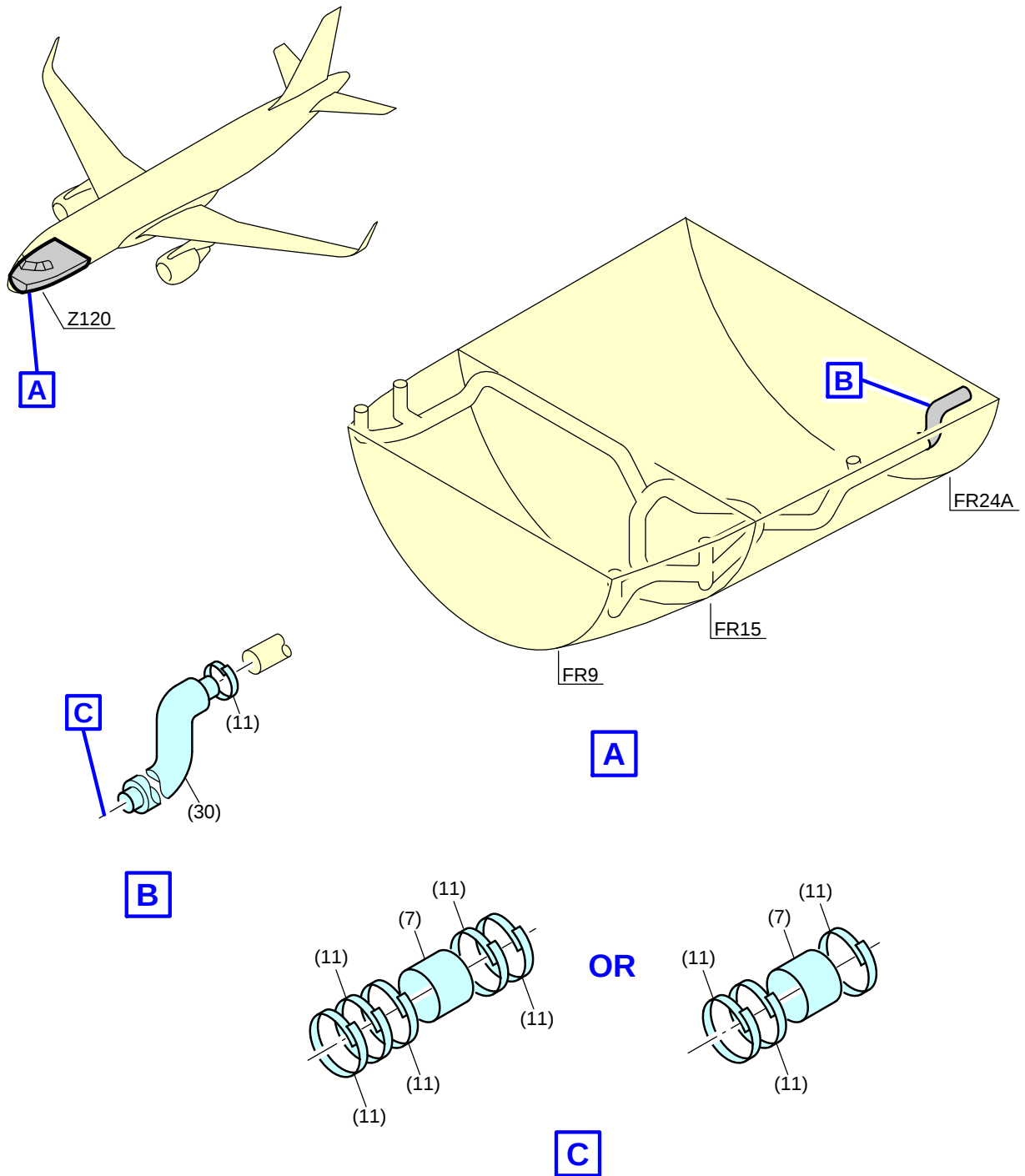
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TMC-A-003-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TMC-A-003-01 SHEET 1/1

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TMF-A-002-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TMF-A-002-01 SHEET 1/1

CONF 002-004

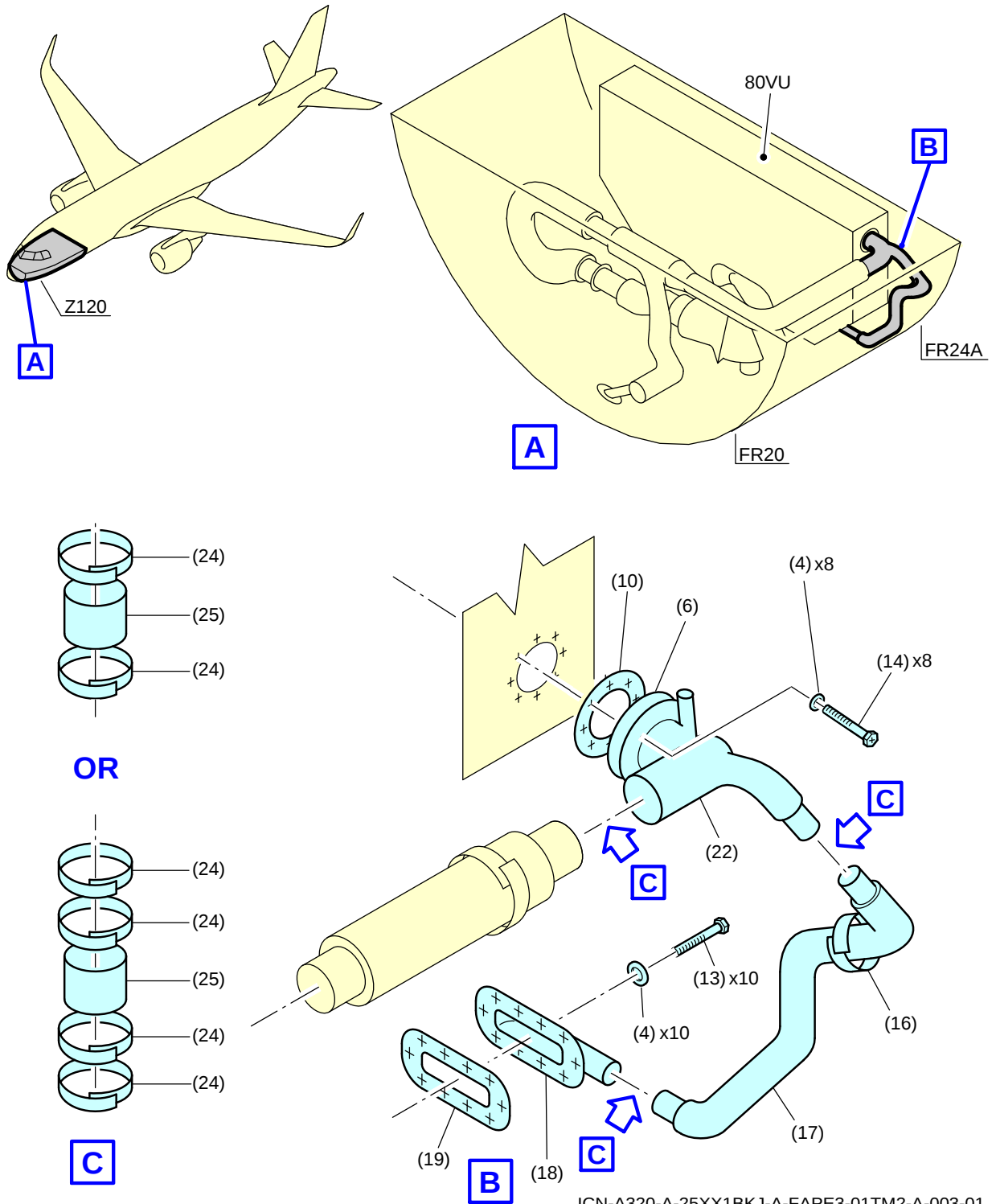


Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM2-A-003-01 SHEET 1/1

(3) Task A320-A-25-XX-1BKJ-04002-540A-A - Remove structure and struts
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	5.75 h
Minimum Number of person	1
Elapsed Time	5.75 h
Skills	Airframe

Material necessary to do the job

Consumables				
ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	08BAA9	Non Aqueous Cleaner - General		
	14SBA1	Textile - Lint free Cotton		

Referenced Information

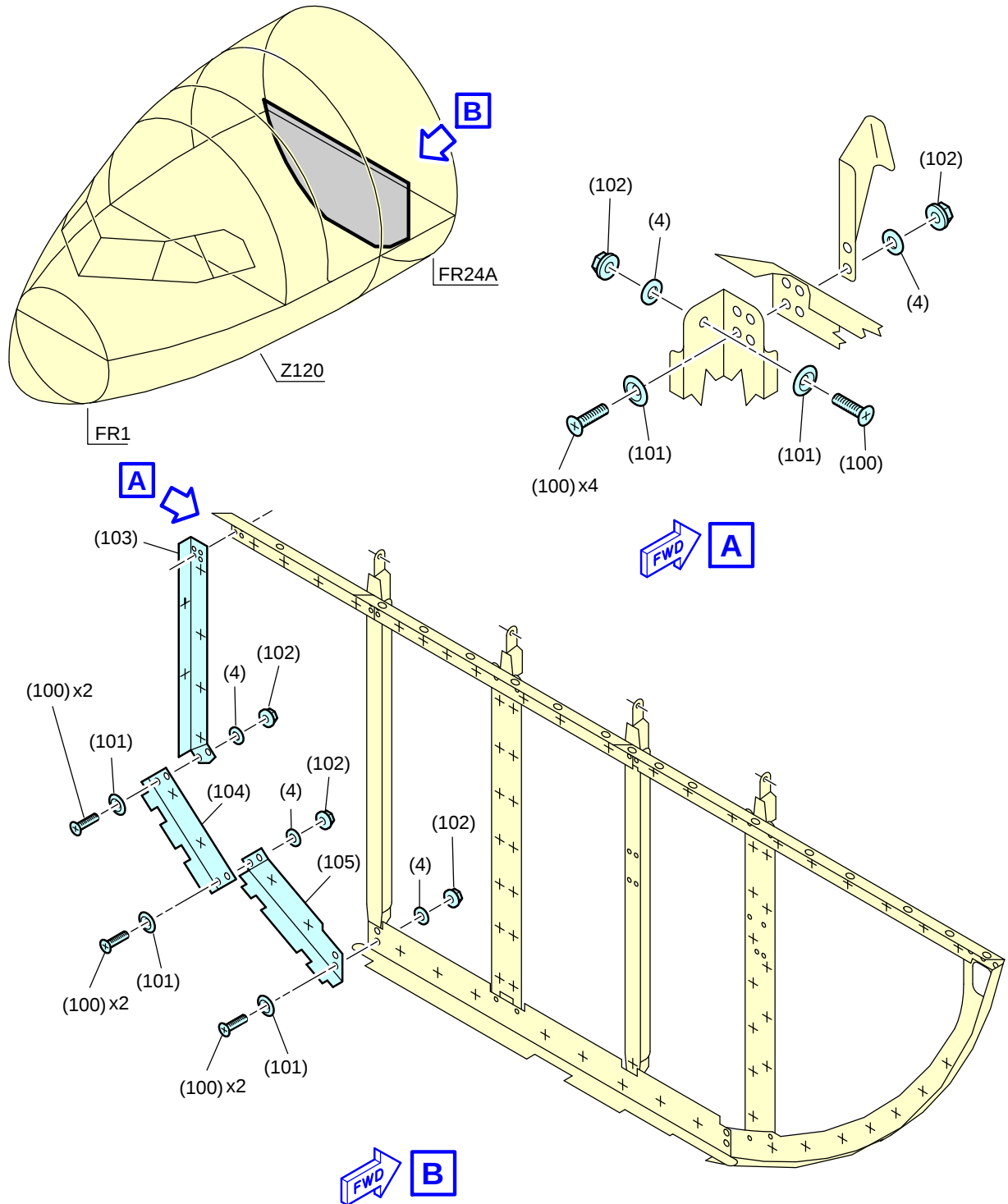
Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000 06-31-53, Page Block 000 Task 21-20-00-010-001
SRM	51-42-00

Procedure
(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.
- 3 Use a non-metallic spatula to remove the sealant and clean the areas with:
Non Aqueous Cleaner - 08BAA9
General
Textile - Lint free Cotton 14SBA1
- 4 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 5 Tag all the parts you must remove and retain to make the subsequent installation easier.
- 6 Put all the retained hardware in identified plastic bags and attach the bags to the related unit.
- 7 For the general removal of air-conditioning distribution ducts, refer to Ref. AMM Task 21-20-00-010-001.

(b) Remove, at Frame FR24A, refer to Ref. Fig. Removal of the structure at FR24A:			
1	SECTION ASSY	Item (103)	Retain
1	BEAM-CROSS	Item (104)	Retain
1	BEAM-CROSS attached with	Item (105)	Retain
11	SCREW	Item (100)	Retain
11	WASHER	Item (101)	Discard
11	WASHER	Item (41)	Discard
11	NUT	Item (102)	Discard
(c) Remove, at Frame FR24 at the LH side of the FIN 80VU, refer to Ref. Fig. Removal of the LH strut at FR24A:			
1	STRUT attached with	Item (56)	Retain
15	SCREW	Item (57)	Retain
15	NUT	Item (51)	Discard
1	SCREW	Item (110)	Retain
1	NUT	Item (111)	Discard
(d) Remove, at Frame FR24 at the RH side of the FIN 80VU, refer to Ref. Fig. Removal of the RH strut at FR24A:			
1	STRUT attached with	Item (115)	Retain
8	SCREW	Item (116)	Retain
11	SCREW	Item (57)	Retain
19	NUT	Item (51)	Discard
1	SCREW	Item (110)	Retain
1	NUT	Item (111)	Discard

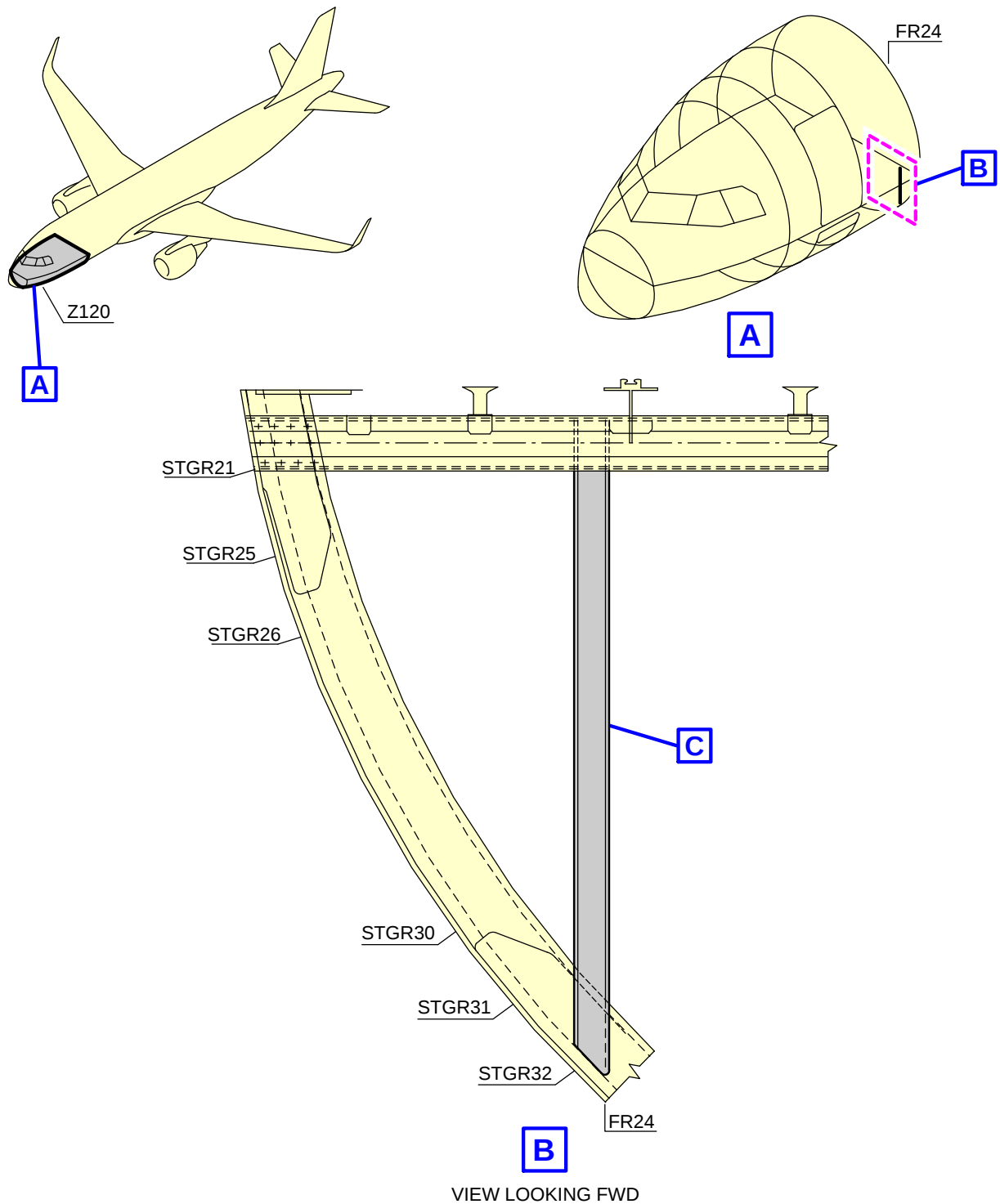
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TMA-A-002-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TMA-A-002-01 SHEET 1/1

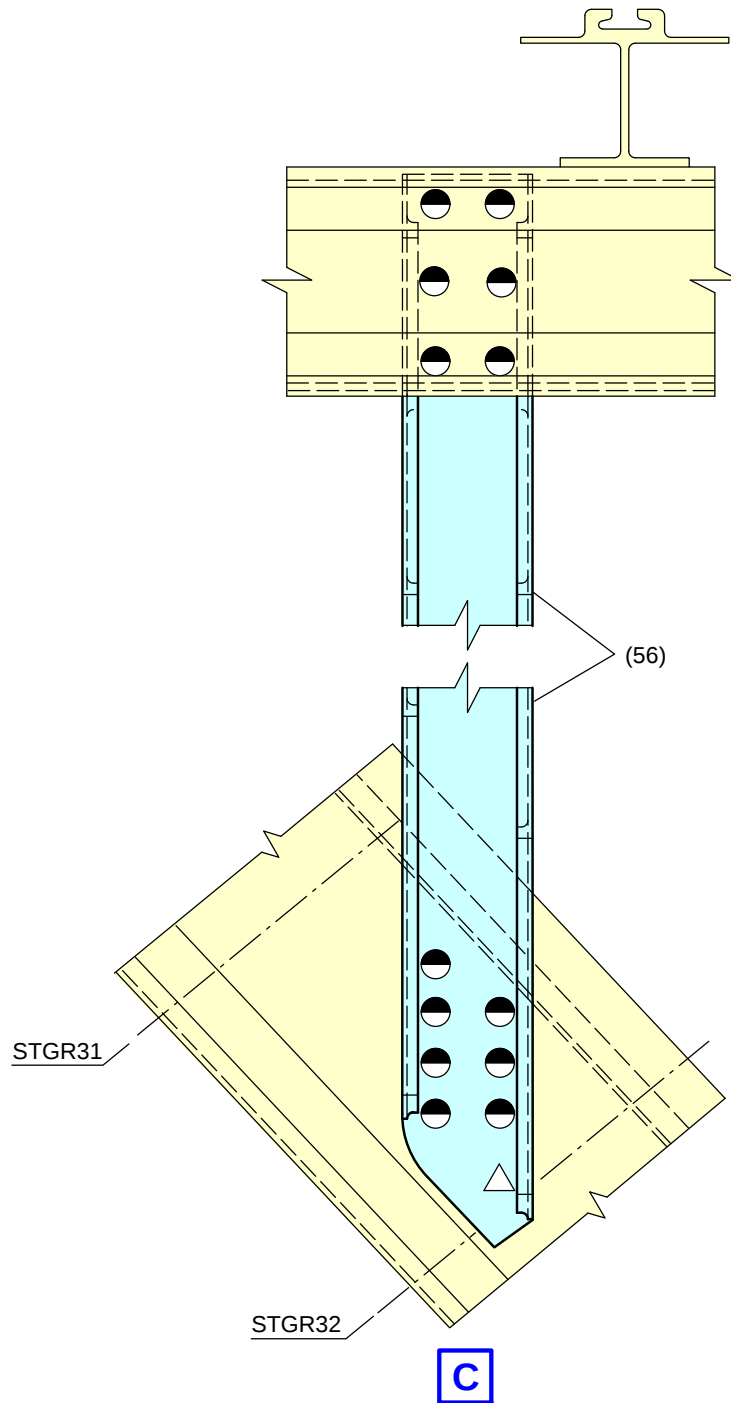
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TM7-A-002-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM7-A-002-01 SHEET 1/3

CONF 002-004



NOTE:
REFER TO SHEET 03
FOR APPLICABLE FASTENERS

VIEW LOOKING FWD

ICN-A320-A-25XX1BKJ-A-FAPE3-01TM8-A-003-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM8-A-003-01 SHEET 2/3

CONF 002-004

HOLE SYMBOL	OLD ITEM	P/N	DESCRIPTION	SIZE / FIT TYPE	NOTE
	(57)		SCREW		
	(51)	ASNA2528-3	NUT		
	(110)		SCREW		
	(111)	ASNA2529-3	NUT		

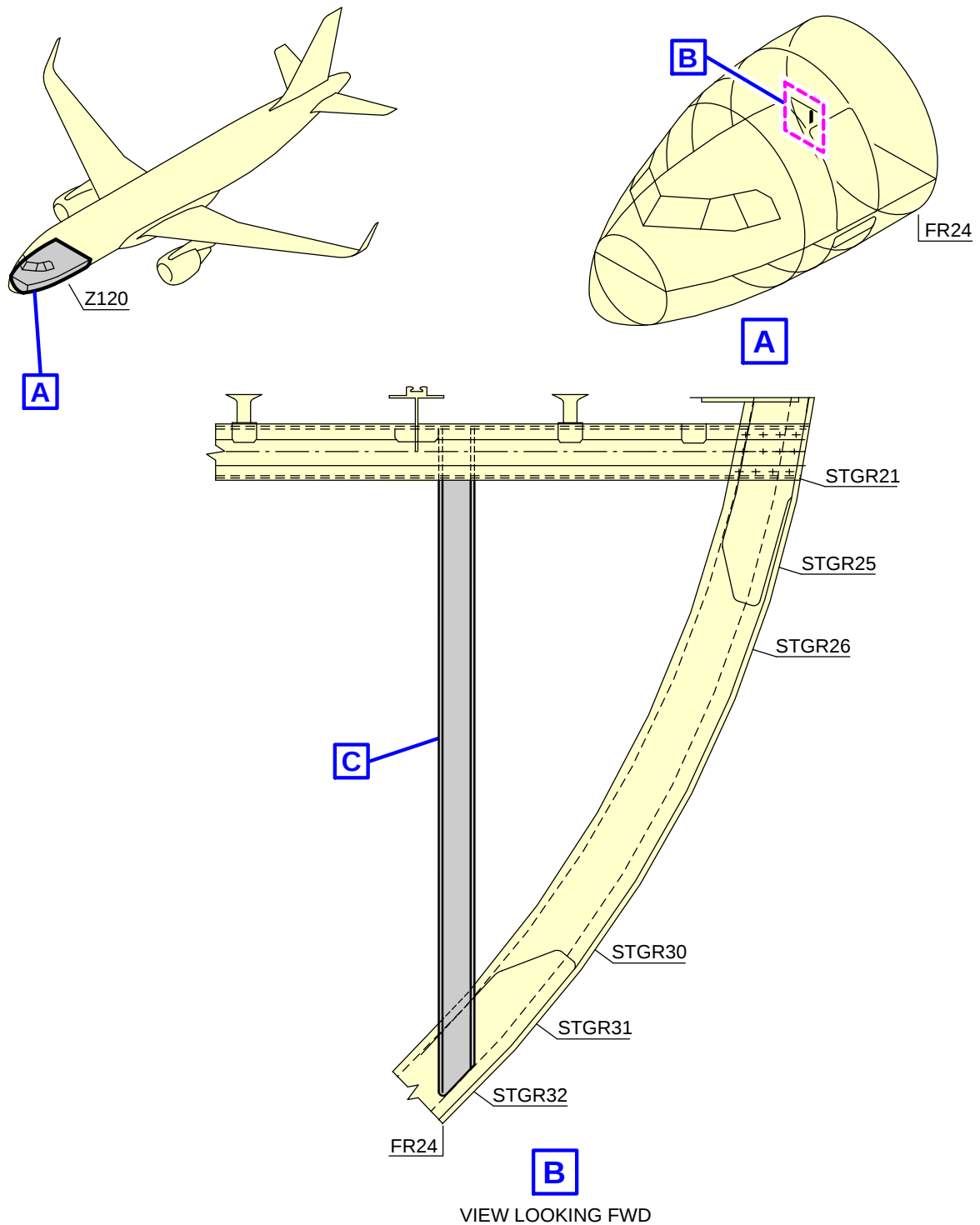
NOTE:

-  RETAIN THIS PARTNUMBER
-  DISCARD THIS PARTNUMBER

ICN-A320-A-25XX1BKJ-A-FAPE3-01TM9-A-003-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM9-A-003-01 SHEET 3/3

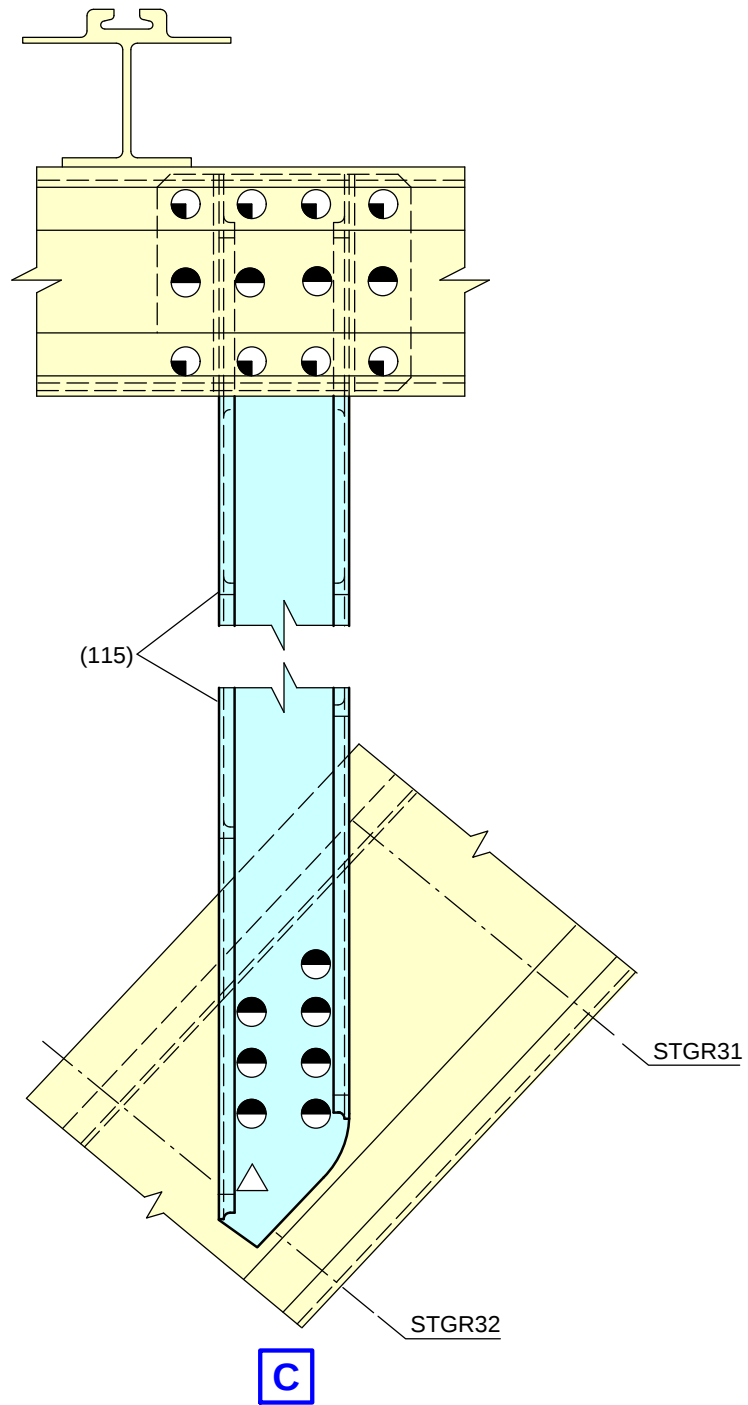
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TMG-A-002-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TMG-A-002-01 SHEET 1/3

CONF 002-004



NOTE:
REFER TO SHEET 03
FOR APPLICABLE FASTENERS

VIEW LOOKING FWD

ICN-A320-A-25XX1BKJ-A-FAPE3-01TMH-A-003-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TMH-A-003-01 SHEET 2/3

CONF 002-004

HOLE SYMBOL	OLD ITEM	P/N	DESCRIPTION	SIZE / FIT TYPE	NOTE
	(57)		SCREW		
	(51)	ASNA2528-3	NUT		
	(110)		SCREW		
	(111)	ASNA2529-3	NUT		
	(116)		SCREW		
	(51)	ASNA2528-3	NUT		

NOTE:

-  RETAIN THE PARTNUMBER
-  DISCARD THE PARTNUMBER

ICN-A320-A-25XX1BKJ-A-FAPE3-01TMI-A-003-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TMI-A-003-01 SHEET 3/3

(4) Task A320-A-25-XX-1BKJ-06002-540A-A - Remove the support
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.00 h
Minimum Number of person	1
Elapsed Time	1.00 h
Skills	Airframe

Material necessary to do the job

Consumables				
ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	08BAA9	Non Aqueous Cleaner - General		
	14SBA1	Textile - Lint free Cotton		

Referenced Information

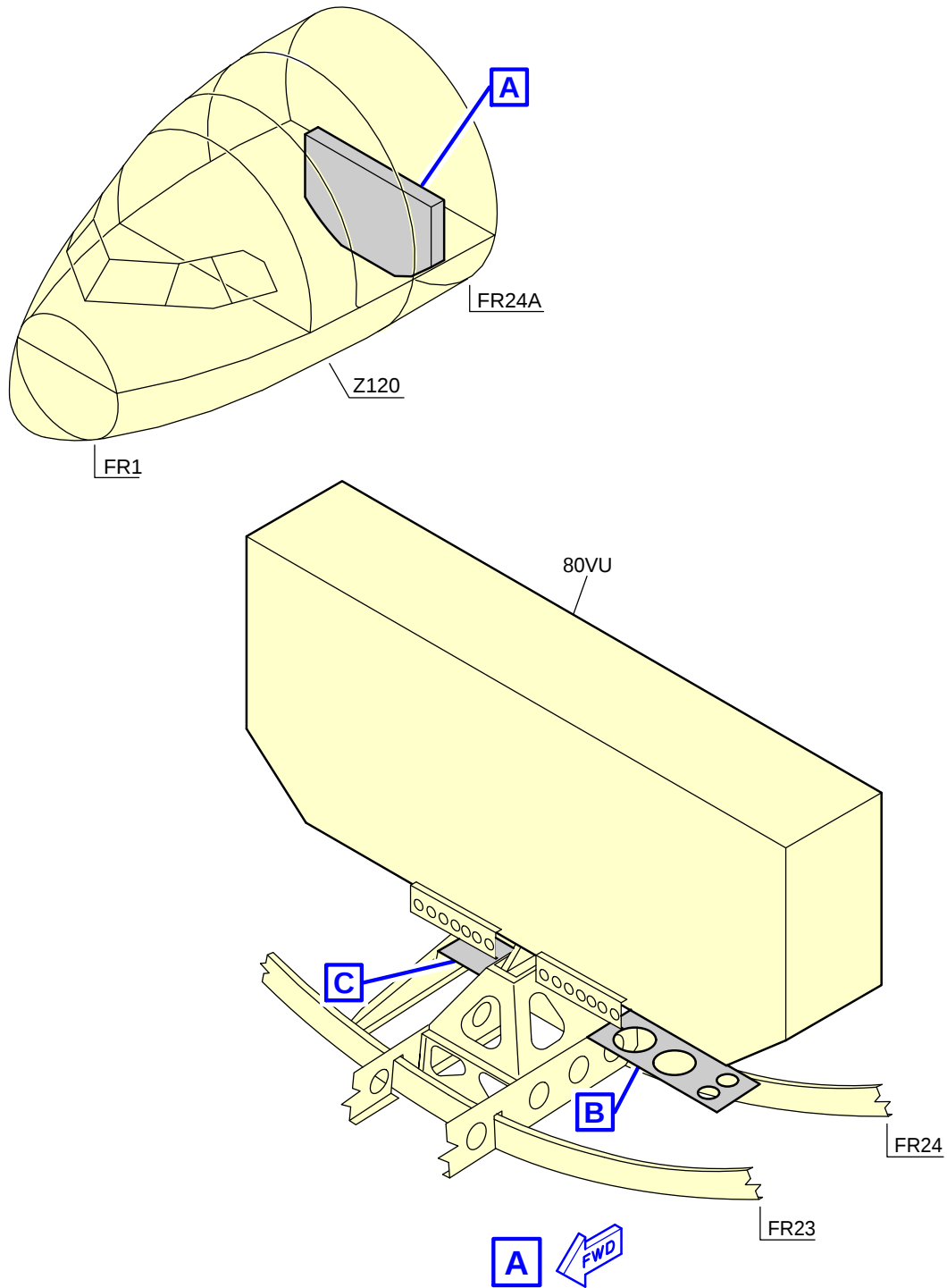
Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000 06-31-53, Page Block 000 Task 21-20-00-010-001
SRM	51-42-00

Procedure
(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.
- 3 Use a non-metallic spatula to remove the sealant and clean the areas with:
Non Aqueous Cleaner - 08BAA9
General
Textile - Lint free Cotton 14SBA1
- 4 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 5 Tag all the parts you must remove and retain to make the subsequent installation easier.
- 6 Put all the retained hardware in identified plastic bags and attach the bags to the related unit.
- 7 For the general removal of air-conditioning distribution ducts, refer to Ref. AMM Task 21-20-00-010-001.

(b) Remove the support of the equipment FIN 13WG and FIN 2WD1, refer to Ref. Fig. Removal of the support located under the 80VU:			
1	SUPPORT attached with	Item (48)	Retain
2	SCREW	Item (39)	Retain
2	SCREW	Item (37)	Retain
2	WASHER	Item (4)	Discard
(c) Remove the support of the equipment FIN 102TD and FIN 2WD2, refer to Ref. Fig. Removal of the support located under the 80VU:			
1	SUPPORT attached with	Item (49)	Retain
4	SCREW	Item (37)	Retain
4	WASHER	Item (4)	Discard
(d) If installed, remove the supports fixed under the FIN 80VU, refer to Ref. Fig. Removal of the support located under the 80VU:			
2	SUPPORT attached with	Item (50)	Retain
10	SCREW	Item (40)	Retain
10	WASHER	Item (4)	Discard

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TMD-A-002-01

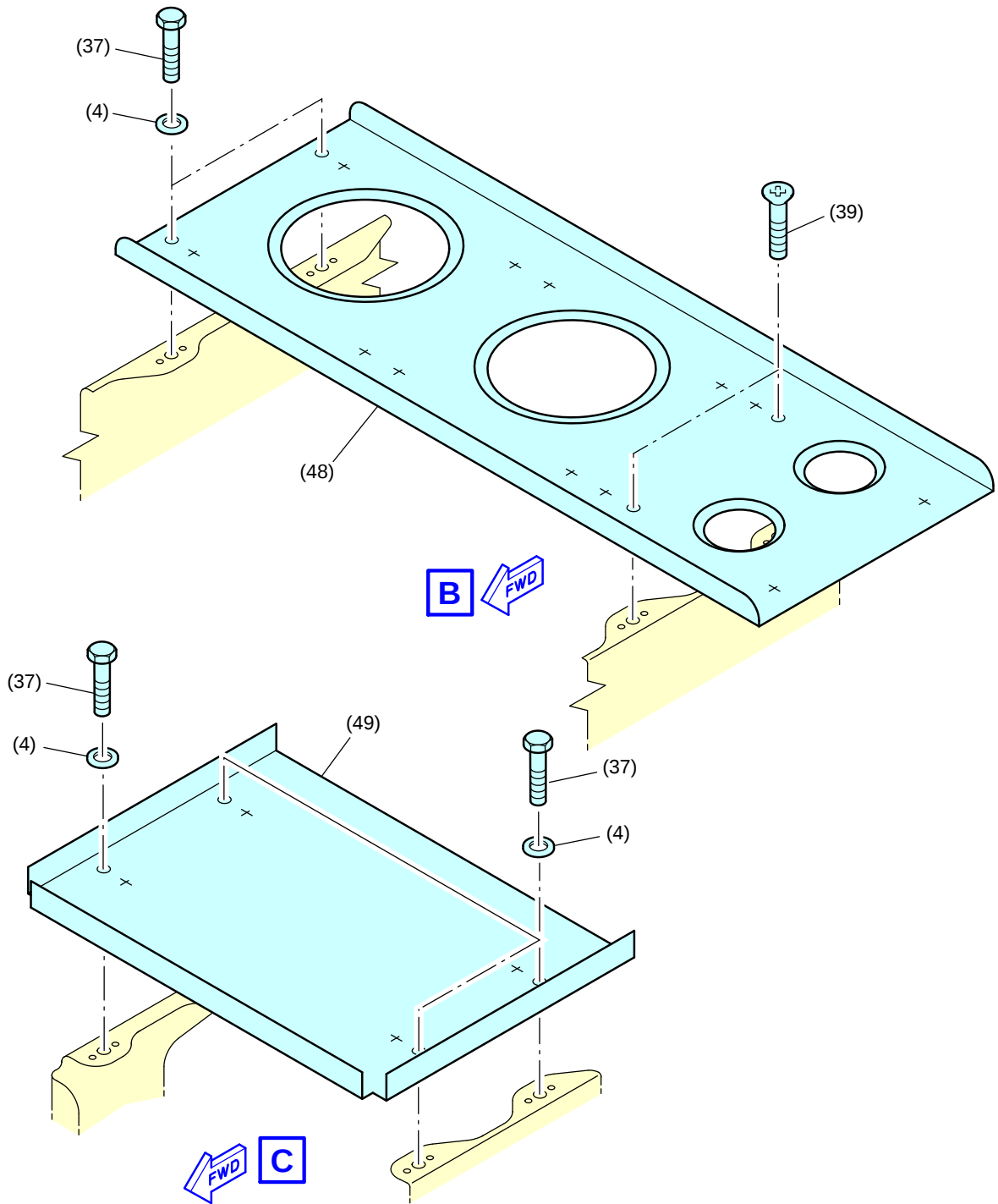
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TMD-A-002-01 SHEET 1/2

REV 00 Aug 23/18
REV 02 Apr 09/20

Removal of the support located under the 80VU

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Page 185

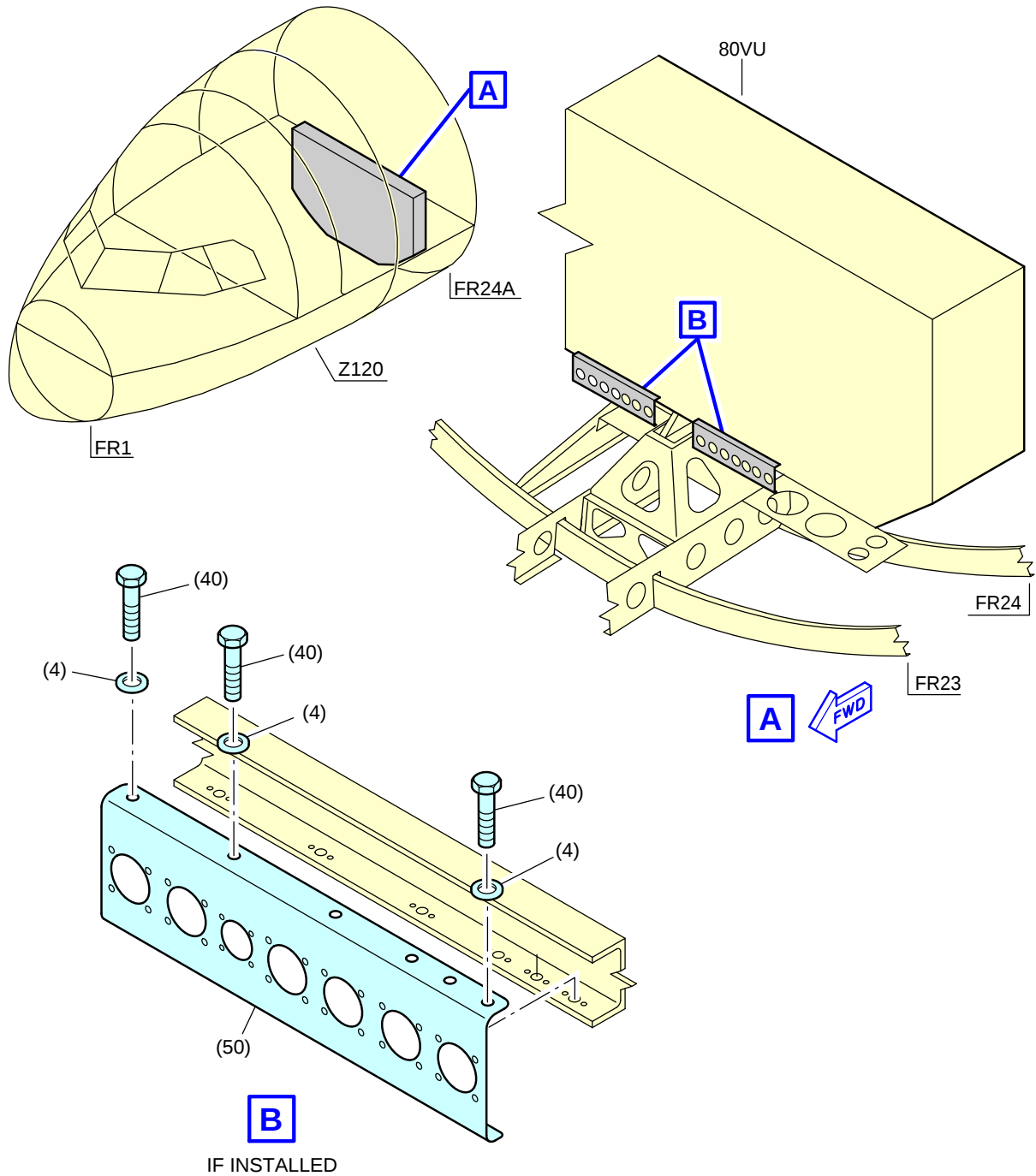
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TME-A-003-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TME-A-003-01 SHEET 2/2

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01TM3-A-003-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM3-A-003-01 SHEET 1/1

C. PROCEDURE
CONF 002 003
(1) Task A320-A-25-XX-1BKJ-01002-600A-A - Accomplish the Repair Instruction
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	67.00 h
Minimum Number of person	2
Elapsed Time	40.00 h
Skills	Airframe

Material necessary to do the job

Part kit 251BKJD01R00					
ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
0	D5391065720000	1	FORKEND		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
634	NAS6603-4	4	BOLT		
644	NAS6604-4	4	BOLT		
813	NAS1726-3E	12	NUT		
814	NAS1726-4E	25	NUT		

Part kit 251BKJD02R00					
ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
1	D5391065720100	1	FORKEND		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
634	NAS6603-4	4	BOLT		
644	NAS6604-4	4	BOLT		
813	NAS1726-3E	12	NUT		
814	NAS1726-4E	25	NUT		

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000 06-31-53, Page Block 000 Task 20-21-11-911-001 20-28-00, Page Block 200
CML	CONSUMABLE MATERIAL LIST (CML)
ESPM	20-51-20
SRM	51-24-00 51-42-00 51-43-00 51-44-00 51-46-00 51-47-00 51-75-12
R53113174000	all sheets
R53113174001	all sheets

Procedure

(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.
- 4 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 5 Get alternative and substitute fastener data in accordance with Ref. SRM 51-43-00.
- 6 Get fastener hole diameters and drill data in accordance with Ref. SRM 51-44-00.
- 7 For fasteners spacing and margin, refer to Ref. SRM 51-47-00.
- 8 For the countersinking, refer to Ref. SRM 51-46-00.
- 9 To torque tighten the standard threaded fasteners, refer to Ref. AMM Task 20-21-11-911-001.
- 10 For the electrical bonding, refer to Ref. ESPM 20-51-20 and Ref. AMM 20-28-00, Page Block 200.
- 11 Apply sealant in accordance with Ref. SRM 51-24-00.
- 12 Renew the protective finish in accordance with Ref. SRM 51-75-12.

(b) To accomplish the Repair Instruction Ref. R53113174000 all sheets, use the following parts:

1	FORK END	D5391065720000	Item 0
1	FITTING	D9251650700095	Item 2
1	BOLT	D9251642620000	Item 3
10	RIVET	EN6069AD4-13	Item 32

17	RIVET	ASNA0078A402	Item 85
14	RIVET	EN6081AD4-14	Item 132
4	BOLT	NAS6603-4	Item 634
4	BOLT	NAS6604-4	Item 644
12	NUT	NAS1726-3E	Item 813
25	NUT	NAS1726-4E	Item 814

- (c) To accomplish the Repair Instruction Ref. R53113174001 all sheets, use the following parts:

1	FORK END	D5391065720100	Item 1
1	FITTING	D9251650700095	Item 2
1	BOLT	D9251642620000	Item 3
10	RIVET	EN6069AD4-13	Item 32
17	RIVET	ASNA0078A402	Item 85
14	RIVET	EN6081AD4-14	Item 132
4	BOLT	NAS6603-4	Item 634
4	BOLT	NAS6604-4	Item 644
12	NUT	NAS1726-3E	Item 813
25	NUT	NAS1726-4E	Item 814

- (d) Installation principle to adjust the lower lateral fitting with the LH or RH side post.

Refer to Ref. Fig. Installation principle of the lower lateral fitting

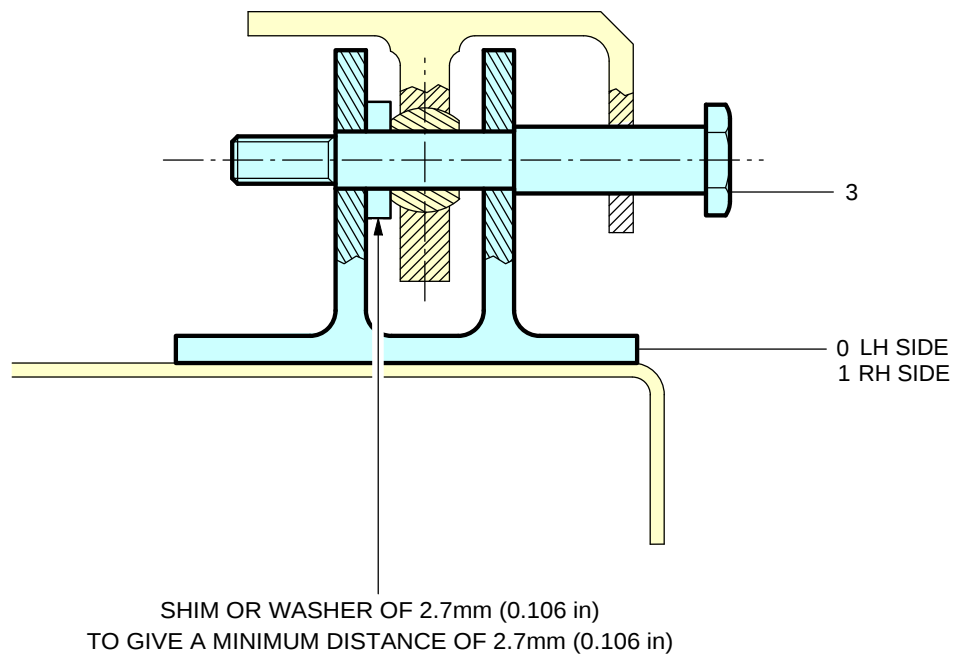
- 1 Make a shim with a thickness of 2.7mm (0.106 in.) to get the required minimum distance.
- 2 Temporarily put in position the bolt, item 3, with the shim of 2.7mm (0.106 in.) to the fitting, item 59 or 79.
- 3 Adjust the fitting, item 0 or 1.
- 4 Tighten the nuts and the bolts of the fitting, item 0 or 1.
- 5 Remove the shim.
- 6 Tighten the bolt, item 3, and the nut.

CONF 002 003

INSTALLATION PRINCIPLE

VIEW LOOKING FWD

LH SIDE SHOWN
RH SIDE SYMMETRICAL



ICN-A320-A-25XX1BKJ-A-FAPE3-01TM4-A-002-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM4-A-002-01 SHEET 1/1

REV 00 Aug 23/18
REV 02 Apr 09/20

Installation principle of the lower lateral fitting

A320-25-1BKJ
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CONF 004
(1) Task A320-A-25-XX-1BKJ-01003-600A-A - Accomplish the Repair Instruction
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	67.00 h
Minimum Number of person	2
Elapsed Time	40.00 h
Skills	Airframe

Material necessary to do the job

Part kit 251BKJD03R00					
ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
0	D5392408320090	1	CLEVIS		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
644	NAS6604-4	4	BOLT		
657	ABS0114T4-7	4	BOLT		
813	NAS1726-3E	8	NUT		
814	NAS1726-4E	25	NUT		

Part kit 251BKJD04R00					
ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
1	D5392408320190	1	CLEVIS		
2	D9251650700095	1	FITTING		
3	D9251642620000	1	BOLT		
32	EN6069AD4-13	10	RIVET		
85	ASNA0078A402	17	RIVET		
132	EN6081AD4-14	14	RIVET		
644	NAS6604-4	4	BOLT		
657	ABS0114T4-7	4	BOLT		
813	NAS1726-3E	8	NUT		
814	NAS1726-4E	25	NUT		

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000

Manuals/Drawings	References/Sheets
	06-31-53, Page Block 000 Task 20-21-11-911-001 20-28-00, Page Block 200
CML	CONSUMABLE MATERIAL LIST (CML)
ESPM	20-51-20
SRM	51-24-00 51-42-00 51-43-00 51-44-00 51-46-00 51-47-00 51-75-12
R53113174002	all sheets
R53113174003	all sheets

Procedure

(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.
- 4 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 5 Get alternative and substitute fastener data in accordance with Ref. SRM 51-43-00.
- 6 Get fastener hole diameters and drill data in accordance with Ref. SRM 51-44-00.
- 7 For fasteners spacing and margin, refer to Ref. SRM 51-47-00.
- 8 For the countersinking, refer to Ref. SRM 51-46-00.
- 9 To torque tighten the standard threaded fasteners, refer to Ref. AMM Task 20-21-11-911-001.
- 10 For the electrical bonding, refer to Ref. ESPM 20-51-20 and Ref. AMM 20-28-00, Page Block 200.
- 11 Apply sealant in accordance with Ref. SRM 51-24-00.
- 12 Renew the protective finish in accordance with Ref. SRM 51-75-12.

(b) To accomplish the Repair Instruction Ref. R53113174002 all sheets, use the following parts:

1	CLEVIS	D5392408320090	Item 0
1	FITTING	D9251650700095	Item 2
1	BOLT	D9251642620000	Item 3
10	RIVET	EN6069AD4-13	Item 32
17	RIVET	ASNA0078A402	Item 85
14	RIVET	EN6081AD4-14	Item 132

4	BOLT	NAS6604-4	Item 644
4	BOLT-HEXAGON HEAD	ABS0114T4-7	Item 657
8	NUT	NAS1726-3E	Item 813
25	NUT	NAS1726-4E	Item 814

- (c) To accomplish the Repair Instruction Ref. R53113174003 all sheets, use the following parts:

1	CLEVIS	D5392408320190	Item 1
1	FITTING	D9251650700095	Item 2
1	BOLT	D9251642620000	Item 3
10	RIVET	EN6069AD4-13	Item 32
17	RIVET	ASNA0078A402	Item 85
14	RIVET	EN6081AD4-14	Item 132
4	BOLT	NAS6604-4	Item 644
4	BOLT-HEXAGON HEAD	ABS0114T4-7	Item 657
8	NUT	NAS1726-3E	Item 813
25	NUT	NAS1726-4E	Item 814

- (d) Installation principle to adjust the lower lateral fitting with the LH or RH side post.

Refer to Ref. Fig. Installation principle of the lower lateral fitting

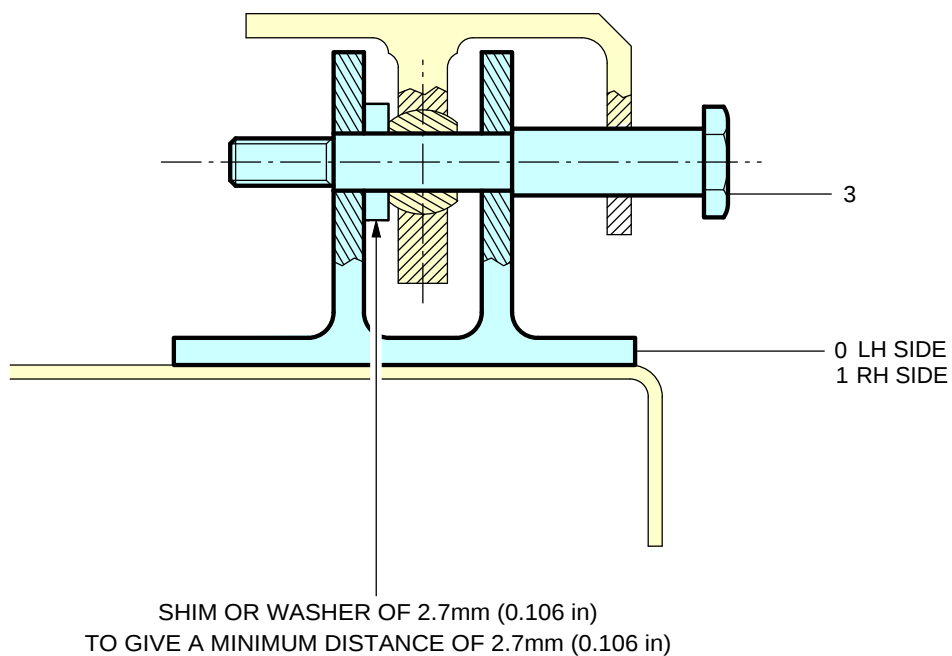
- 1 Make a shim with a thickness of 2.7mm (0.106 in.) to get the required minimum distance.
- 2 Temporarily put in position the bolt, item 3, with the shim of 2.7mm (0.106 in.) to the fitting, item 59 or 79.
- 3 Adjust the fitting, item 0 or 1.
- 4 Tighten the nuts and the bolts of the fitting, item 0 or 1.
- 5 Remove the shim.
- 6 Tighten the bolt, item 3, and the nut.

CONF 004

INSTALLATION PRINCIPLE

VIEW LOOKING FWD

LH SIDE SHOWN
RH SIDE SYMMETRICAL



ICN-A320-A-25XX1BKJ-A-FAPE3-01TM4-A-002-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01TM4-A-002-01 SHEET 1/1

CONF 002-004
D. PREPARATION FOR TEST
(1) Task A320-A-25-XX-1BKJ-01002-900A-A - Preparation for Test
Job Set-Up Information
Manpower/Resources/Skills

Man-hours	4.00 h
Minimum Number of person	2
Elapsed Time	2.00 h
Skills	Cabin
	Electrical

Referenced Information

Manuals/Drawings	References/Sheets
AMM	21-26-34, Page Block 401
	22-66-34, Page Block 401
	22-83-34, Page Block 401
	23-11-33, Page Block 401
	23-12-33, Page Block 401
	23-24-34, Page Block 401
	23-28-34, Page Block 401
	23-32-34, Page Block 401
	23-32-35, Page Block 401
	23-33-31, Page Block 401
	23-33-37, Page Block 401
	23-33-39, Page Block 401
	23-33-52, Page Block 401
	23-34-34, Page Block 401
	23-51-34, Page Block 401
	23-72-31, Page Block 401
	23-73-34, Page Block 401
	Task 24-38-51-000-001
	26-12-34, Page Block 401
	26-13-34, Page Block 401
	27-51-34, Page Block 401
	27-93-34, Page Block 401
	27-94-34, Page Block 401
	27-95-34, Page Block 401
	31-32-34, Page Block 401
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	31-36-34, Page Block 401
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	31-38-34, Page Block 401
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	31-55-34, Page Block 401
	31-63-34, Page Block 401
	34-35-34, Page Block 401
	34-43-34, Page Block 401
	34-48-34, Page Block 401
	34-51-33, Page Block 401
	34-52-33, Page Block 401

Manuals/Drawings	References/Sheets
	34-52-35, Page Block 401 34-53-31, Page Block 401 34-55-31, Page Block 401 46-21-34, Page Block 401 73-25-34, Page Block 401 77-32-34, Page Block 401
ESPM	20-52-10

Procedure

- (a) Remove the jacking of the FIN 80VU.
 - 1 If moved, put the wire or pipe in place.
- (b) Do a check of the electrical connectors before you connect them, refer to Ref. ESPM 20-52-10.
- (c) Install all the computers of the Electronics Rack FIN 80VU, depending on the aircraft configuration:
 - 1 In the Shelf FIN 88VU, install:
 - a PAVSU (11RX), refer to Ref. AMM 23-32-35, Page Block 401.
 - b CIDS Director (102RH), refer to Ref. AMM 23-73-34, Page Block 401.
 - c Passenger Entertainment System Main Multiplexer (8MK), refer to Ref. AMM 23-33-31, Page Block 401.
 - d AEVC (10HQ), refer to Ref. AMM 21-26-34, Page Block 401.
 - e Remote ATC Control Box (8SH), refer to Ref. AMM 34-52-35, Page Block 401.
 - f GPWS (1WZ), refer to Ref. AMM 34-48-34, Page Block 401.
 - g SDU (105RV1), refer to Ref. AMM 23-28-34, Page Block 401.
 - h DIU (190MH), refer to Ref. AMM 23-34-34, Page Block 401.
 - i Passenger Entertainment System (10MK), refer to Ref. AMM 23-33-52, Page Block 401.
 - j Announcement/Music Tape Reproducer (10RX), refer to Ref. AMM 23-32-34, Page Block 401.
 - k IFE Audio/Video Controller (314MK), refer to Ref. AMM 23-33-37, Page Block 401.
 - l IFE Audio/Video Controller (313MK), refer to Ref. AMM 23-33-37, Page Block 401.
 - m IFE Digital Server Unit (312MK), refer to Ref. AMM 23-33-39, Page Block 401.
 - n IFE Audio/Video Controller (311MK), refer to Ref. AMM 23-33-37, Page Block 401.
 - 2 In the Shelf FIN 87VU, install:
 - a DAR (2TV), refer to Ref. AMM 31-36-52, Page Block 401.
 - b AEVC (10HQ), refer to Ref. AMM 21-26-34, Page Block 401.
 - c FDIU/DMU (FDIMU) (10TV), refer to Ref. AMM 31-36-34, Page Block 401.

- d FDIU (2TU), refer to Ref. AMM 31-33-34, Page Block 401.
 - e QAR (3TU), refer to Ref. AMM 31-33-52, Page Block 401.
 - f DMU (1TV), refer to Ref. AMM 31-36-34, Page Block 401.
 - g CFDIU (1TW), refer to Ref. AMM 31-32-34, Page Block 401.
 - h CIDS Director (101RH), refer to Ref. AMM 23-73-34, Page Block 401.
- 3 In the Shelf FIN 86VU, install:
 - a SDAC (1WV2), refer to Ref. AMM 31-55-34, Page Block 401.
 - b FWC (1WW2), refer to Ref. AMM 31-53-34, Page Block 401.
 - c DMC (1WT2), refer to Ref. AMM 31-63-34, Page Block 401.
 - d AEVC (10HQ), refer to Ref. AMM 21-26-34, Page Block 401.
 - e HUDC (1FH), refer to Ref. AMM 34-35-34, Page Block 401.
 - f EVMU (2EV), refer to Ref. AMM 77-32-34, Page Block 401.
 - g EIU (1KS2), refer to Ref. AMM 73-25-34, Page Block 401.
 - h SFCC (22CV), refer to Ref. AMM 27-51-34, Page Block 401.
- 4 In the Shelf FIN 85VU, install:
 - a SFCC (21CV), refer to Ref. AMM 27-51-34, Page Block 401.
 - b EIU (1KS1), refer to Ref. AMM 73-25-34, Page Block 401.
 - c DMC (1WT1), refer to Ref. AMM 31-63-34, Page Block 401.
 - d DMC FIN (1WT3), refer to Ref. AMM 31-63-34, Page Block 401.
 - e FWC (1WW1), refer to Ref. AMM 31-53-34, Page Block 401.
 - f SDAC (1WV1), refer to Ref. AMM 31-55-34, Page Block 401.
 - g CDSS (9RA), refer to Ref. AMM 23-72-31, Page Block 401.
- 5 In the Shelf FIN 84VU, install:
 - a FAC (1CC2), refer to Ref. AMM 22-66-34, Page Block 401.
 - b FMGC (1CA2), refer to Ref. AMM 22-83-34, Page Block 401.
 - c FCDC (3CE2), refer to Ref. AMM 27-95-34, Page Block 401.
 - d SEC (1CE2), refer to Ref. AMM 27-94-34, Page Block 401.
 - e ELAC (2CE2), refer to Ref. AMM 27-93-34, Page Block 401.
- 6 In the Shelf FIN 83VU, install:
 - a ELAC (2CE1), refer to Ref. AMM 27-93-34, Page Block 401.
 - b SEC (1CE1), refer to Ref. AMM 27-94-34, Page Block 401.
 - c FCDC (3CE1), refer to Ref. AMM 27-95-34, Page Block 401.
 - d FMGC (1CA1), refer to Ref. AMM 22-83-34, Page Block 401.
 - e FAC (1CC1), refer to Ref. AMM 22-66-34, Page Block 401.
- 7 In the Shelf FIN 82VU, install:
 - a ADF Receiver (2RP2), refer to Ref. AMM 34-53-31, Page Block 401.
 - b ACARS MU (1RB), refer to Ref. AMM 23-24-34, Page Block 401.
 - c VHF Transceiver (1RC2), refer to Ref. AMM 23-12-33, Page Block 401.

- d DME Interrogator (2SD2), refer to Ref. AMM 34-51-33, Page Block 401.
- e ATC/Mode S Transponder (1SH2), refer to Ref. AMM 34-52-33, Page Block 401.
- f T2CAS Computer (100SG), refer to Ref. AMM 34-43-34, Page Block 401.
- g TCAS Computer (1SG), refer to Ref. AMM 34-43-34, Page Block 401.
- h Transceiver HF (3RE2), refer to Ref. AMM 23-11-33, Page Block 401.
- i VOR/MKR Receiver (3RS2), refer to Ref. AMM 34-55-31, Page Block 401.
- 8 In the Shelf FIN 81VU, install:
 - a Transceiver HF (3RE1), refer to Ref. AMM 23-11-33, Page Block 401.
 - b ATSU (1TX1), refer to Ref. AMM 46-21-34, Page Block 401.
 - c ATC/Mode S Transponder (1SH1), refer to Ref. AMM 34-52-33, Page Block 401.
 - d DME Interrogator (2SD1), refer to Ref. AMM 34-51-33, Page Block 401.
 - e VOR/MKR Receiver (3RS1), refer to Ref. AMM 34-55-31, Page Block 401.
 - f Transceiver VHF (1RC3), refer to Ref. AMM 23-12-33, Page Block 401.
 - g Transceiver VHF (1RC1), refer to Ref. AMM 23-12-33, Page Block 401.
 - h AMU (1RN), refer to Ref. AMM 23-51-34, Page Block 401.
 - i ADF Receiver (2RP1), refer to Ref. AMM 34-53-31, Page Block 401.
- (d) Install all the computers under the Electronics Rack FIN 80VU, depending on the configuration:
 - 1 FDU (2WD1), refer to Ref. AMM 26-12-34, Page Block 401.
 - 2 FDU (2WD2), refer to Ref. AMM 26-12-34, Page Block 401.
 - 3 FDU (13WG), refer to Ref. AMM 26-13-34, Page Block 401.
 - 4 Data Loading Routing Box (102TD), refer to Ref. AMM 31-38-34, Page Block 401.
- (e) Battery re-connection:
 - 1 Do the Job Set-up procedure as specified in the removal/installation of the batteries, refer to Ref. AMM Task 24-38-51-000-001.
 - 2 Remove the blanking caps from the electrical connectors.
 - 3 Make sure that the electrical connectors are clean and in correct condition.
 - 4 Connect the connectors to the battery.
 - 5 Turn the knurled nut a quarter turn.
- (f) Remove all warning notices.

(g) Restore the systems and the aircraft to normal operating condition.

E. TEST
(1) Task A320-A-25-XX-1BKJ-01002-345A-A - Test
Job Set-Up Information
Manpower/Resources/Skills

Man-hours	20.00 h
Minimum Number of person	2
Elapsed Time	10.00 h
Skills	Airframe
	Cabin
	Electrical

Referenced Information

Manuals/Drawings	References/Sheets
AMM	21-26-00, Page Block 500
	21-26-34, Page Block 400
	22-66-34, Page Block 400
	22-83-34, Page Block 400
	22-96-00, Page Block 500
	22-97-00, Page Block 500
	23-11-33, Page Block 400
	23-12-33, Page Block 400
	23-24-34, Page Block 400
	23-28-34, Page Block 400
	23-32-34, Page Block 400
	23-32-35, Page Block 400
	23-33-31, Page Block 400
	23-33-37, Page Block 400
	23-33-39, Page Block 400
	23-33-52, Page Block 400
	23-34-34, Page Block 400
	23-51-34, Page Block 400
	23-72-31, Page Block 400
	23-73-34, Page Block 400
	26-12-34, Page Block 400
	26-13-34, Page Block 400
	27-51-34, Page Block 400
	27-93-34, Page Block 400
	27-94-34, Page Block 400
	27-95-34, Page Block 400
	31-32-34, Page Block 400
	31-33-34, Page Block 400
	31-33-52, Page Block 400
	31-36-34, Page Block 400
	31-36-52, Page Block 400
	31-38-34, Page Block 400
	31-50-00, Page Block 500
	31-53-34, Page Block 400
	31-55-34, Page Block 400
	31-60-00, Page Block 500
	31-63-34, Page Block 400
	33-13-00, Page Block 500

Manuals/Drawings	References/Sheets
	33-14-00, Page Block 500 34-35-34, Page Block 400 34-43-34, Page Block 400 34-48-34, Page Block 400 34-51-33, Page Block 400 34-52-33, Page Block 400 34-52-35, Page Block 400 34-53-31, Page Block 400 34-55-31, Page Block 400 46-21-34, Page Block 400 73-25-34, Page Block 400 77-32-34, Page Block 400
ESPM	20-52-10

Procedure

- (a) Check the following initial conditions:

- 1 Do a visual check on each disconnected connector, refer to Ref. ESPM 20-52-10, to make sure that :
 - a The label of the plug is the same as the label on the receptacle.
 - b The plug is correctly locked.

- (b) Do the following preliminary tests:

NOTE: If the aircraft is operated in CAT3 conditions, you must also do the Land CAT3 Capability Test, refer to Ref. AMM 22-97-00, Page Block 500.

- 1 Do an operational test of the instrument and panel integral-lighting, refer to Ref. AMM 33-13-00, Page Block 500.
- 2 Do an operational test of the annunciator light test system in the cockpit, refer to Ref. AMM 33-14-00, Page Block 500.
- 3 Do an operational test of the avionics equipment ventilation system, refer to Ref. AMM 21-26-00, Page Block 500.
- 4 Do an operational test of the AFS, refer to Ref. AMM 22-96-00, Page Block 500.
- 5 Do an operational test of the EIS, refer to Ref. AMM 31-60-00, Page Block 500.
- 6 Do an operational test of the ECAM control panel, refer to Ref. AMM 31-50-00, Page Block 500.
- 7 Do a ground scanning of the central warning system, refer to Ref. AMM 31-50-00, Page Block 500.

- (c) Do the following tests after removal/installation of the equipment, depending on the configuration:

- 1 Do the tests after removal/installation of the HF Transceiver (FINs 3RE1, 3RE2), refer to Ref. AMM 23-11-33, Page Block 400.
- 2 Do the tests after removal/installation of the ATSU (FIN 1TX1), refer to Ref. AMM 46-21-34, Page Block 400.
- 3 Do the tests after removal/installation of the ATC/Mode S Transponder (FINs 1SH1, 1SH2), refer to Ref. AMM 34-52-33, Page Block 400.

- 4 Do the tests after removal/installation of the DME Interrogator (FINs 2SD1, 2SD2), refer to Ref. AMM 34-51-33, Page Block 400.
- 5 Do the tests after removal/installation of the VOR/MKR Receiver (FINs 3RS1, 3RS2), refer to Ref. AMM 34-55-31, Page Block 400.
- 6 Do the tests after removal/installation of the VHF Transceiver (FINs 1RC1, 1RC2, 1RC3), refer to Ref. AMM 23-12-33, Page Block 400.
- 7 Do the tests after removal/installation of the AMU (FIN 1RN), refer to Ref. AMM 23-51-34, Page Block 400.
- 8 Do the tests after removal/installation of the ADF Receiver (FINs 2RP1, 2RP2), refer to Ref. AMM 34-53-31, Page Block 400.
- 9 Do the tests after removal/installation of the ACARS MU (FIN 1RB), refer to Ref. AMM 23-24-34, Page Block 400.
- 10 Do the tests after removal/installation of the TCAS Computer (FIN 1SG), refer to Ref. AMM 34-43-34, Page Block 400.
- 11 Do the tests after removal/installation of the T2CAS Computer (FIN 100SG), refer to Ref. AMM 34-43-34, Page Block 400.
- 12 Do the tests after removal/installation of the ELAC (FINs 2CE1, 2CE2), refer to Ref. AMM 27-93-34, Page Block 400.
- 13 Do the tests after removal/installation of the SEC (FINs 1CE1, 1CE2, 1CE3), refer to Ref. AMM 27-94-34, Page Block 400.
- 14 Do the tests after removal/installation of the FCDC (FINs 3CE1, 3CE2), refer to Ref. AMM 27-95-34, Page Block 400.
- 15 Do the tests after removal/installation of the FMGC (FINs 1CA1, 1CA2), refer to Ref. AMM 22-83-34, Page Block 400.
- 16 Do the tests after removal/installation of the FAC (FINs 1CC1, 1CC2), refer to Ref. AMM 22-66-34, Page Block 400.
- 17 Do the tests after removal/installation of the SFCC (FINs 21CV, 22CV), refer to Ref. AMM 27-51-34, Page Block 400.
- 18 Do the tests after removal/installation of the EIU (FINs 1KS1, 1KS2), refer to Ref. AMM 73-25-34, Page Block 400.
- 19 Do the tests after removal/installation of the DMC (FINs 1WT1, 1WT2, 1WT3), refer to Ref. AMM 31-63-34, Page Block 400.
- 20 Do the tests after removal/installation of the FWC (FINs 1WW1, 1WW2), refer to Ref. AMM 31-53-34, Page Block 400.
- 21 Do the tests after removal/installation of the SDAC (FINs 1WV1, 1WV2), refer to Ref. AMM 31-55-34, Page Block 400.
- 22 Do the tests after removal/installation of the CDSS (FIN 9RA), refer to Ref. AMM 23-72-31, Page Block 400.
- 23 Do the tests after removal/installation of the AEVC (FIN 10HQ), refer to Ref. AMM 21-26-34, Page Block 400.
- 24 Do the tests after removal/installation of the HUDC (FIN 1FH), refer to Ref. AMM 34-35-34, Page Block 400.
- 25 Do the tests after removal/installation of the EVMU (FIN 2EV), refer to Ref. AMM 77-32-34, Page Block 400.

- 26 Do the tests after removal/installation of the DAR (FIN 2TV), refer to Ref. AMM 31-36-52, Page Block 400.
- 27 Do the tests after removal/installation of the DMU (FIN 1TV), refer to Ref. AMM 31-36-34, Page Block 400.
- 28 Do the tests after removal/installation of the FDIMU (FIN 10TV), refer to Ref. AMM 31-36-34, Page Block 400.
- 29 Do the tests after removal/installation of the Flight Data Interface Unit (FIN 2TU), refer to Ref. AMM 31-33-34, Page Block 400.
- 30 Do the tests after removal/installation of the QAR (FIN 3TU), refer to Ref. AMM 31-33-52, Page Block 400.
- 31 Do the tests after removal/installation of the CFDIU (FIN 1TW), refer to Ref. AMM 31-32-34, Page Block 400.
- 32 Do the tests after removal/installation of the CIDS Director (FINs 101RH, 102RH), refer to Ref. AMM 23-73-34, Page Block 400.
- 33 Do the tests after removal/installation of the Passenger Entertainment System Main Multiplexer (FIN 8MK), refer to Ref. AMM 23-33-31, Page Block 400.
- 34 Do the tests after removal/installation of the Remote ATC Control Box (FIN 8SH), refer to Ref. AMM 34-52-35, Page Block 400.
- 35 Do the tests after removal/installation of the GPWC (FIN 1WZ), refer to Ref. AMM 34-48-34, Page Block 400.
- 36 Do the tests after removal/installation of the SDU (FIN 105RV1), refer to Ref. AMM 23-28-34, Page Block 400.
- 37 Do the tests after removal/installation of the DIU (FIN 190MH), refer to Ref. AMM 23-34-34, Page Block 400.
- 38 Do the tests after removal/installation of the PES Reproducer (FIN 10MK), refer to Ref. AMM 23-33-52, Page Block 400.
- 39 Do the tests after removal/installation of the Announcement/Music Tape Reproducer (FIN 10RX), refer to Ref. AMM 23-32-34, Page Block 400.
- 40 Do the tests after removal/installation of the IFE Audio/Video Controller (FIN 311MK), refer to Ref. AMM 23-33-37, Page Block 400.
- 41 Do the tests after removal/installation of the IFE Digital Server Unit (FIN 312MK), refer to Ref. AMM 23-33-39, Page Block 400.
- 42 Do the tests after removal/installation of the Data Loading Routing Box (FIN 102TD), refer to Ref. AMM 31-38-34, Page Block 400.
- 43 Do the tests after removal/installation of the Engine FDU (FINs 2WD1, 2WD2), refer to Ref. AMM 26-12-34, Page Block 400.
- 44 Do the tests after removal/installation of the APU FDU (FIN 13WG), refer to Ref. AMM 26-13-34, Page Block 400.

- 45 Do the test after removal/installation of the PAVSU (FIN 11RX), refer to Ref. AMM 23-32-35, Page Block 400.

F. CLOSE-UP
(1) Task A320-A-25-XX-1BKJ-02002-740A-A - Install the support
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	1.00 h
Minimum Number of person	1
Elapsed Time	1.00 h
Skills	Airframe

Material necessary to do the job

Consumables				
ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	06AEB2	Polysulfide Sealant - Low Adhesion Fillet Structure		

Part kit 251BKJD05R00					
ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
4	NAS1149F0332P	16	WASHER		

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000 06-31-53, Page Block 000 Task 20-21-11-911-001 20-28-00, Page Block 200 Task 21-20-00-410-001
CML	CONSUMABLE MATERIAL LIST (CML)
ESPM	20-51-20
SRM	51-24-00 51-42-00 51-43-00 51-75-12

Procedure
(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.

- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.
- 4 For the general installation of air-conditioning distribution ducts, refer to Ref. AMM Task 21-20-00-410-001.
- 5 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 6 Get alternative and substitute fastener data in accordance with Ref. SRM 51-43-00.
- 7 To torque tighten the standard threaded fasteners, refer to Ref. AMM Task 20-21-11-911-001.
- 8 For the electrical bonding, refer to Ref. ESPM 20-51-20 and Ref. AMM 20-28-00, Page Block 200.
- 9 Apply sealant in accordance with Ref. SRM 51-24-00.
- 10 Renew the protective finish in accordance with Ref. SRM 51-75-12.

- (b) If removed, install the supports fixed under the FIN 80VU, refer to Ref. Fig. Installation of the support located under the 80VU:

2	SUPPORT attached with		Item (50)	Retained
10	WASHER	NAS1149F0332P	Item 4	
10	SCREW wet installed with		Item (40)	Retained
	Polysulfide Sealant - Low Adhesion Fillet Structure	06AEB2		

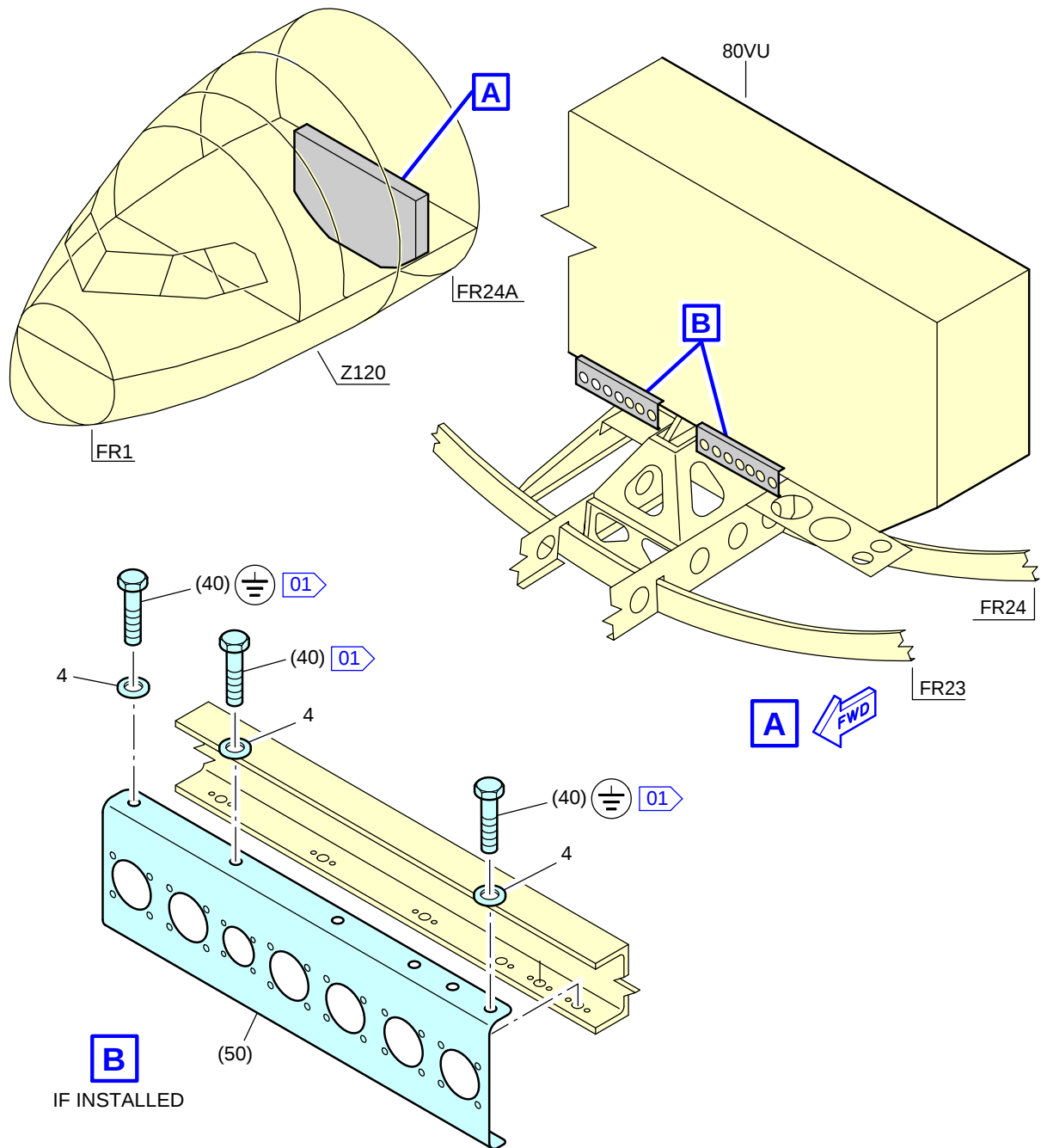
- (c) Install the support of the equipment FIN 102TD and FIN 2WD2, refer to Ref. Fig. Installation of the support located under the 80VU:

1	SUPPORT attached with		Item (49)	Retained
4	WASHER	NAS1149F0332P	Item 4	
4	SCREW		Item (37)	Retained

- (d) Install the support of the equipment FIN 13WG and FIN 2WD1, refer to Ref. Fig. Installation of the support located under the 80VU:

1	SUPPORT attached with		Item (48)	Retained
2	WASHER	NAS1149F0332P	Item 4	
2	SCREW		Item (39)	Retained
2	SCREW		Item (37)	Retained

CONF 002-004



NOTE:

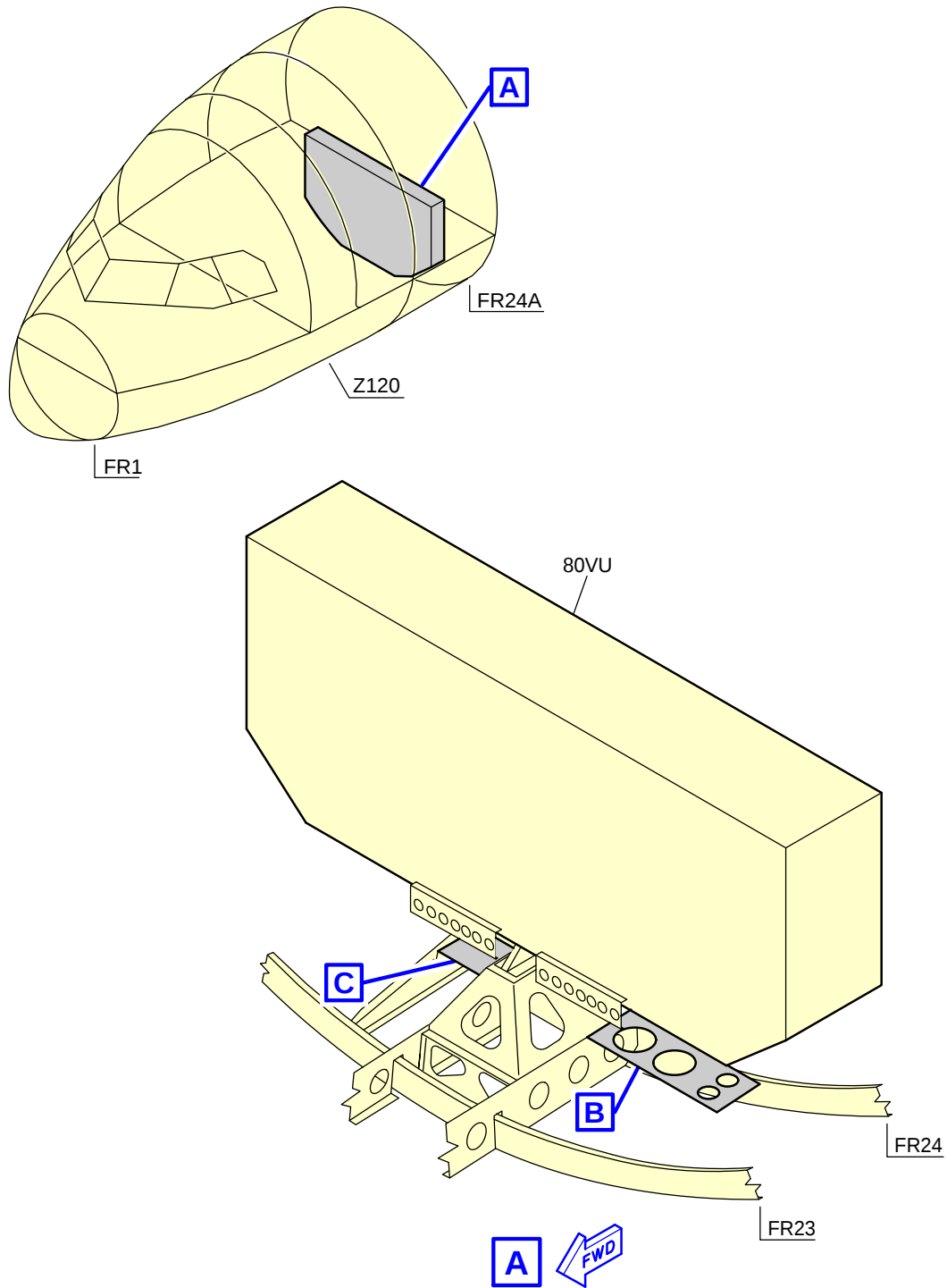
[01] WET INSTALLED WITH
SEALANT MATERIAL No 06AEB2

[E] ELECTRICAL BONDING

ICN-A320-A-25XX1BKJ-A-FAPE3-01W0I-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0I-A-001-01 SHEET 1/1

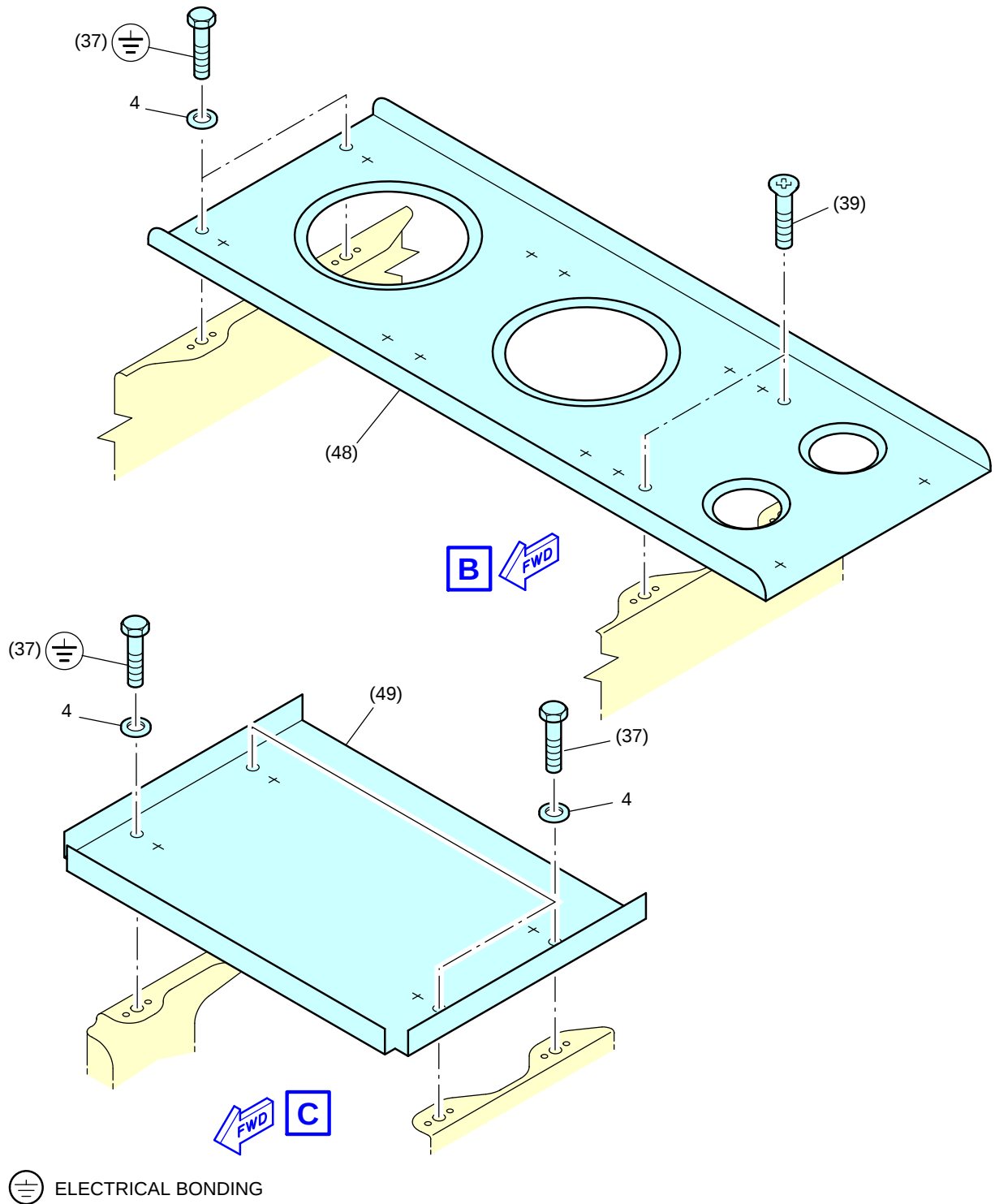
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0J-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0J-A-001-01 SHEET 1/2

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0K-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0K-A-001-01 SHEET 2/2

(2) Task A320-A-25-XX-1BKJ-04002-740A-A - Install structure and struts
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	5.75 h
Minimum Number of person	1
Elapsed Time	5.75 h
Skills	Airframe

Material necessary to do the job

Consumables				
ITEM	REFERENCE CML	DESCRIPTION	DRAWING	SEE NOTES
	06AAA1	Polysulfide Sealant - General Purpose Brushable		
	06AAC1	Polysulfide Sealant - General Purpose Interfay		
	06BHG1	Silicone Sealant - Firewall High Temp.		

Part kit 251BKJD05R00					
ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
4	NAS1149F0332P	11	WASHER		
93	ASNA2528-3	34	NUT		
101	NAS1169DD10	11	WASHER		
102	ASNA2531-3	11	NUT		
111	ASNA2529-3	2	NUT		

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000 06-31-53, Page Block 000 Task 20-21-11-911-001 20-28-00, Page Block 200 Task 21-20-00-410-001
CML	CONSUMABLE MATERIAL LIST (CML)
ESPM	20-51-20
SRM	51-24-00 51-42-00 51-43-00 51-75-12

Procedure

(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.
- 4 For the general installation of air-conditioning distribution ducts, refer to Ref. AMM Task 21-20-00-410-001.
- 5 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 6 Get alternative and substitute fastener data in accordance with Ref. SRM 51-43-00.
- 7 To torque tighten the standard threaded fasteners, refer to Ref. AMM Task 20-21-11-911-001.
- 8 For the electrical bonding, refer to Ref. ESPM 20-51-20 and Ref. AMM 20-28-00, Page Block 200.
- 9 Apply sealant in accordance with Ref. SRM 51-24-00.
- 10 Renew the protective finish in accordance with Ref. SRM 51-75-12.

(b) Install at Frame FR24 at the RH side of the FIN 80VU, refer to Ref. Fig. Installation of the RH strut at FR24A:

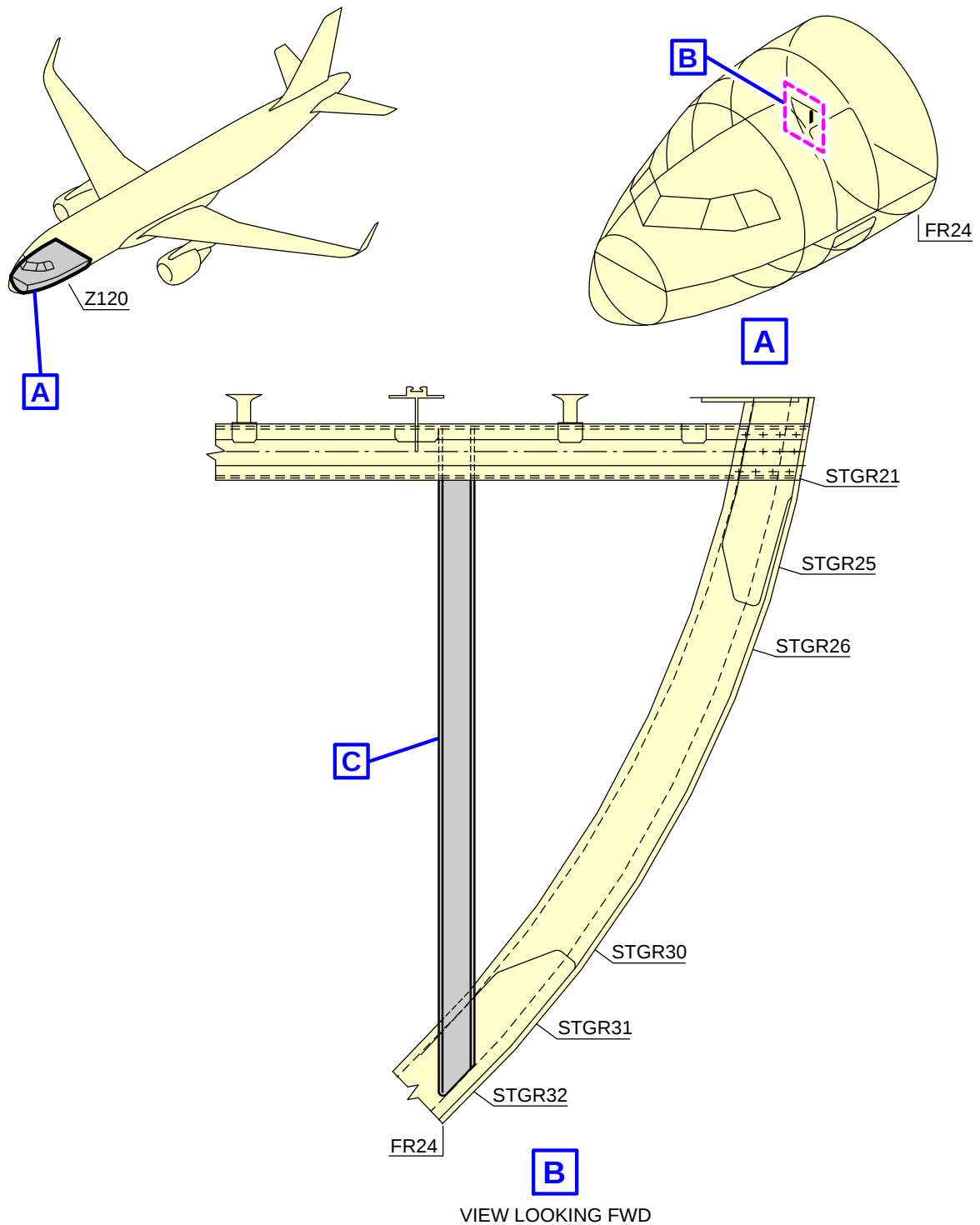
1	STRUT		Item (115)	Retained
	apply to the interface of the strut and the structure			
	Polysulfide Sealant	06AAC1		
	- General Purpose			
	Interfay			
	and attache the strut with			
8	SCREW		Item (116)	Retained
11	SCREW		Item (57)	Retained
19	NUT	ASNA2528-3	Item 93	
1	SCREW		Item (110)	Retained
1	NUT	ASNA2529-3	Item 111	
	wet installed with			
	Polysulfide Sealant	06AAA1		
	- General Purpose			
	Brushable			

(c) Install at Frame FR24 at the LH side of the FIN 80VU, refer to Ref. Fig. Installation of the LH strut at FR24A:

1	STRUT		Item (56)	Retained
	apply to the interface of the strut and the structure			
	Polysulfide Sealant	06AAC1		
	- General Purpose			
	Interfay			
	and attache the strut with			

15	SCREW		Item (57)	Retained
15	NUT	ASNA2528-3	Item 93	
1	SCREW		Item (110)	Retained
1	NUT	ASNA2529-3	Item 111	
	wet installed with			
	Polysulfide Sealant	06AAA1		
	- General Purpose			
	Brushable			
	(d) Install at the Frame FR24A, refer to Ref. Fig. Installation of the structure at FR24A:			
1	SECTION ASSY		Item (103)	Retained
1	BEAM-CROSS		Item (104)	Retained
1	BEAM-CROSS		Item (105)	Retained
	apply to the junction of the sheets			
	Silicone Sealant -	06BHG1		
	Firewall High Temp.			
	and attache with			
11	SCREW		Item (100)	Retained
11	WASHER	NAS1169DD10	Item 101	
11	WASHER	NAS1149F0332P	Item 4	
11	NUT	ASNA2531-3	Item 102	

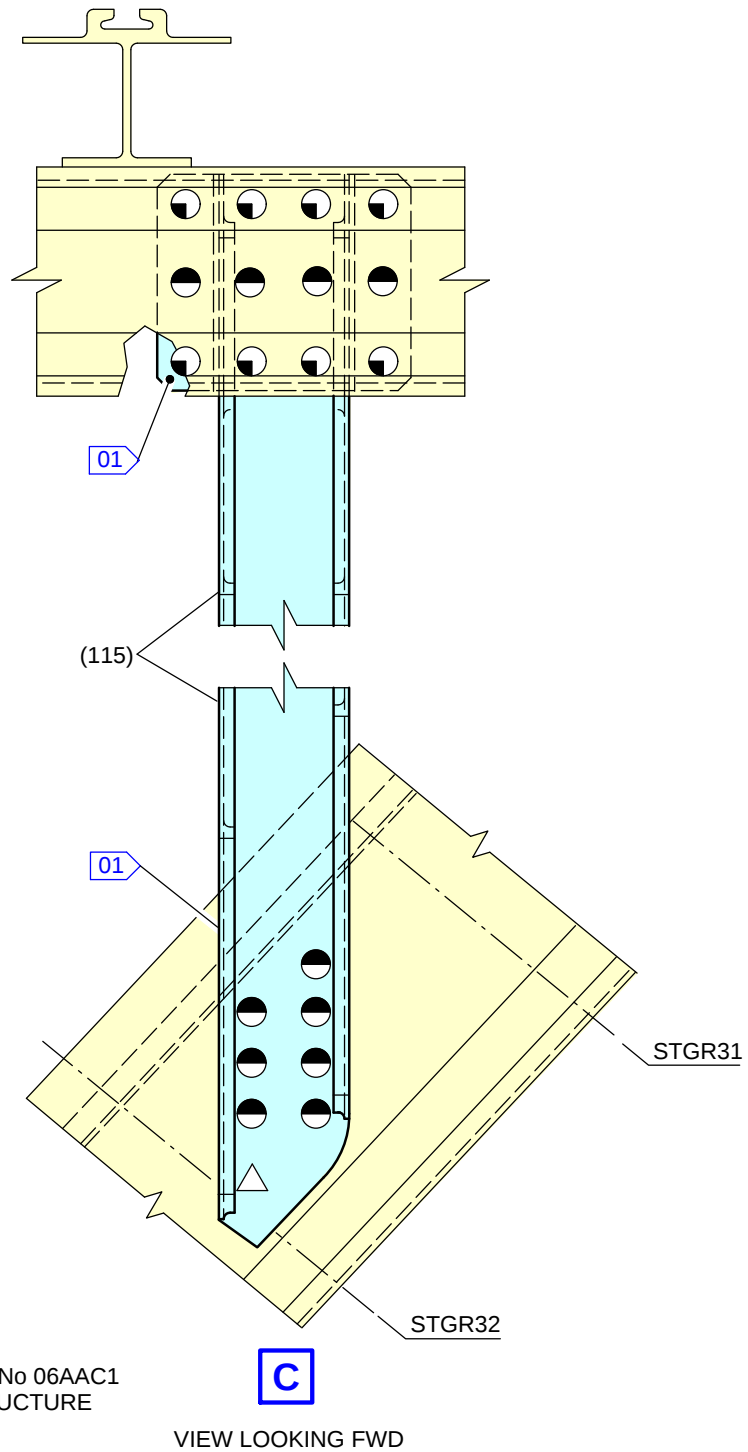
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0A-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0A-A-001-01 SHEET 1/3

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0B-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0B-A-001-01 SHEET 2/3

CONF 002-004

HOLE SYMBOL	NEW ITEM	P/N	DESCRIPTION	SIZE / FIT TYPE	NOTE
	(57)		SCREW	NOMINAL SIZE TRANSITION FIT	
	93	ASNA2528-3	NUT		
	(110)		SCREW	NOMINAL SIZE TRANSITION FIT	
	111	ASNA2529-3	NUT		
	116		SCREW	NOMINAL SIZE TRANSITION FIT	
	93	ASNA2528-3	NUT		

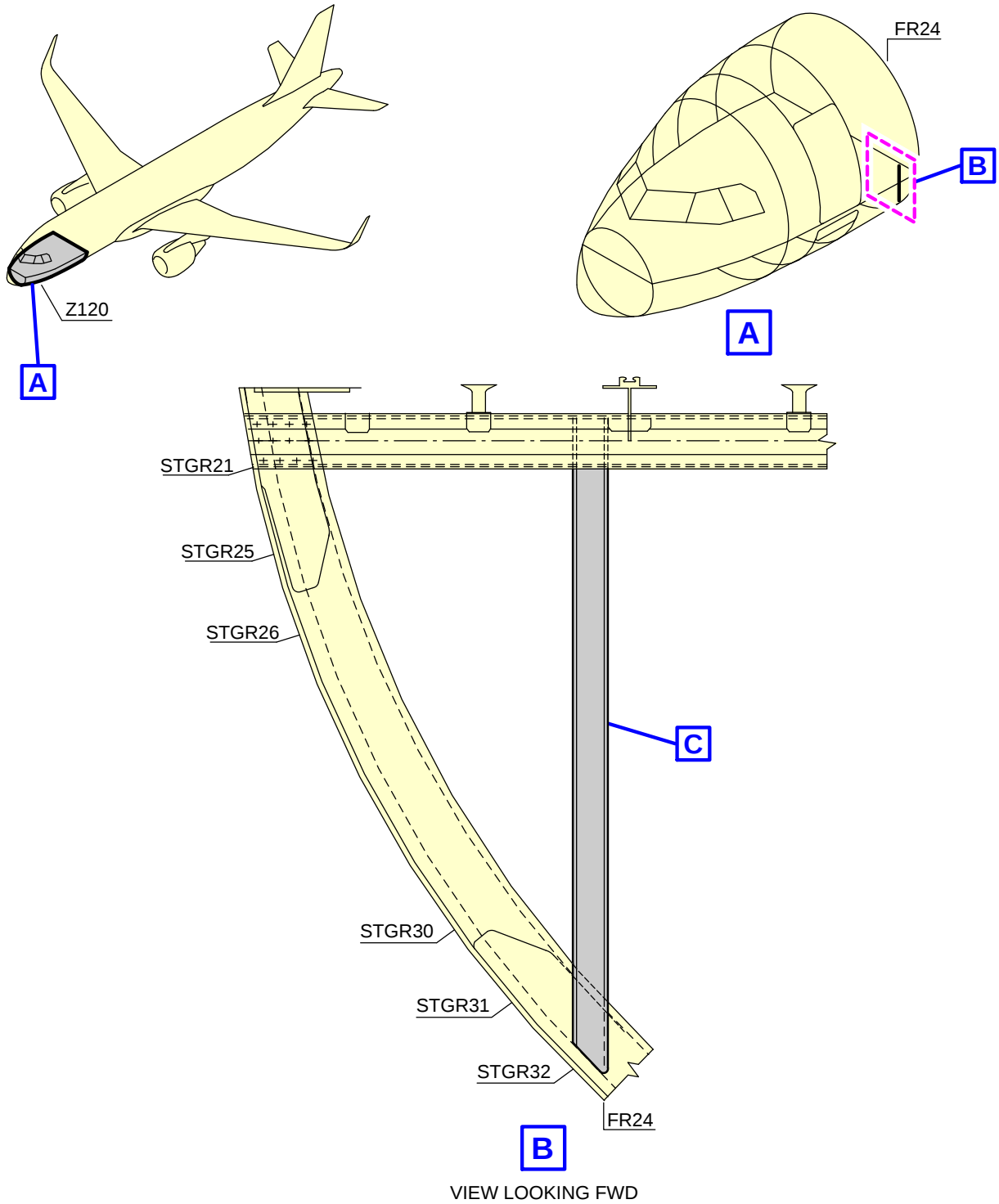
NOTE:

 WET INSTALLED WITH
MATERIAL No 06AAA1

ICN-A320-A-25XX1BKJ-A-FAPE3-01W0C-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0C-A-001-01 SHEET 3/3

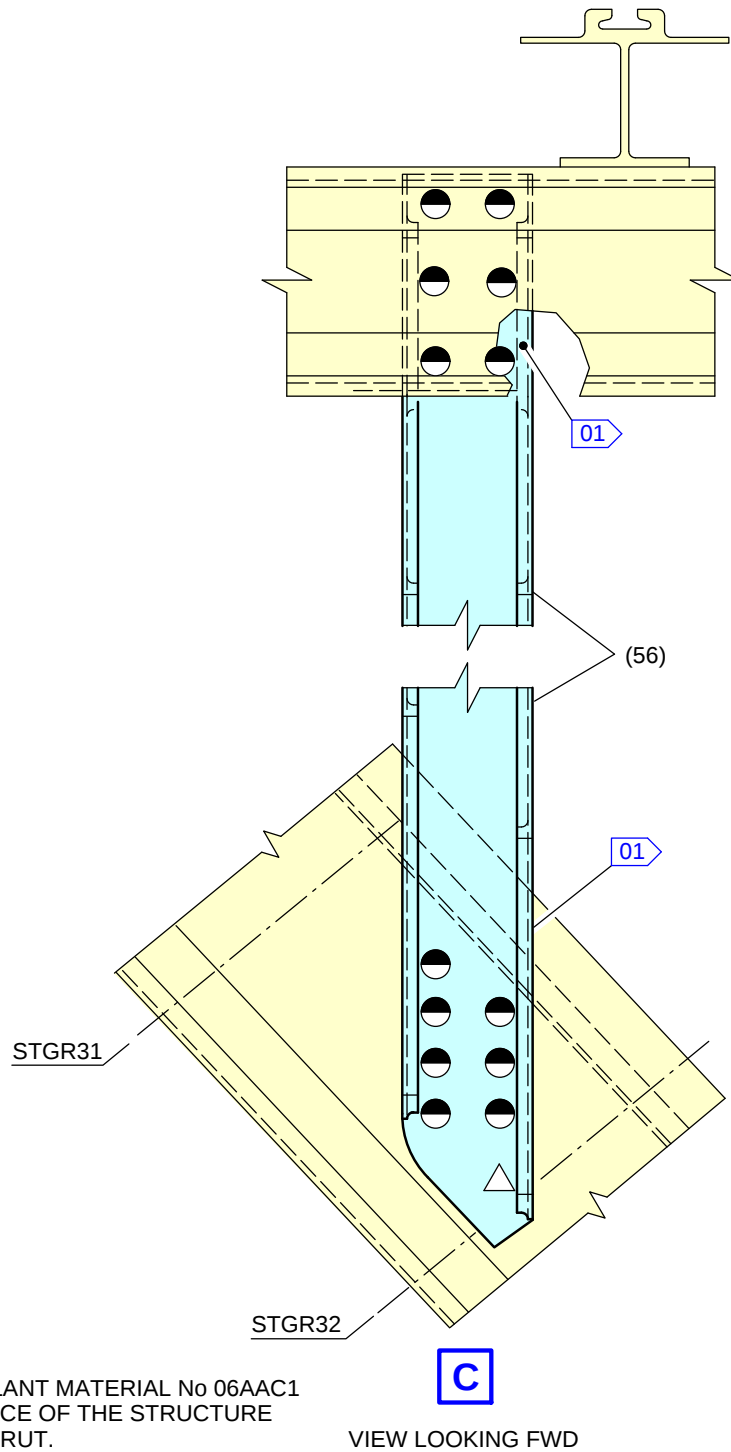
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0E-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0E-A-001-01 SHEET 1/3

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0F-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0F-A-001-01 SHEET 2/3

CONF 002-004

HOLE SYMBOL	NEW ITEM	P/N	DESCRIPTION	SIZE / FIT TYPE	NOTE
	(57)		SCREW	NOMINAL SIZE TRANSITION FIT	
	93	ASNA2528-3	NUT		
	(110)		SCREW	NOMINAL SIZE TRANSITION FIT	
	111	ASNA2529-3	NUT		

NOTE:

 WET INSTALLED WITH
MATERIAL No 06AAA1

ICN-A320-A-25XX1BKJ-A-FAPE3-01W0G-A-001-01

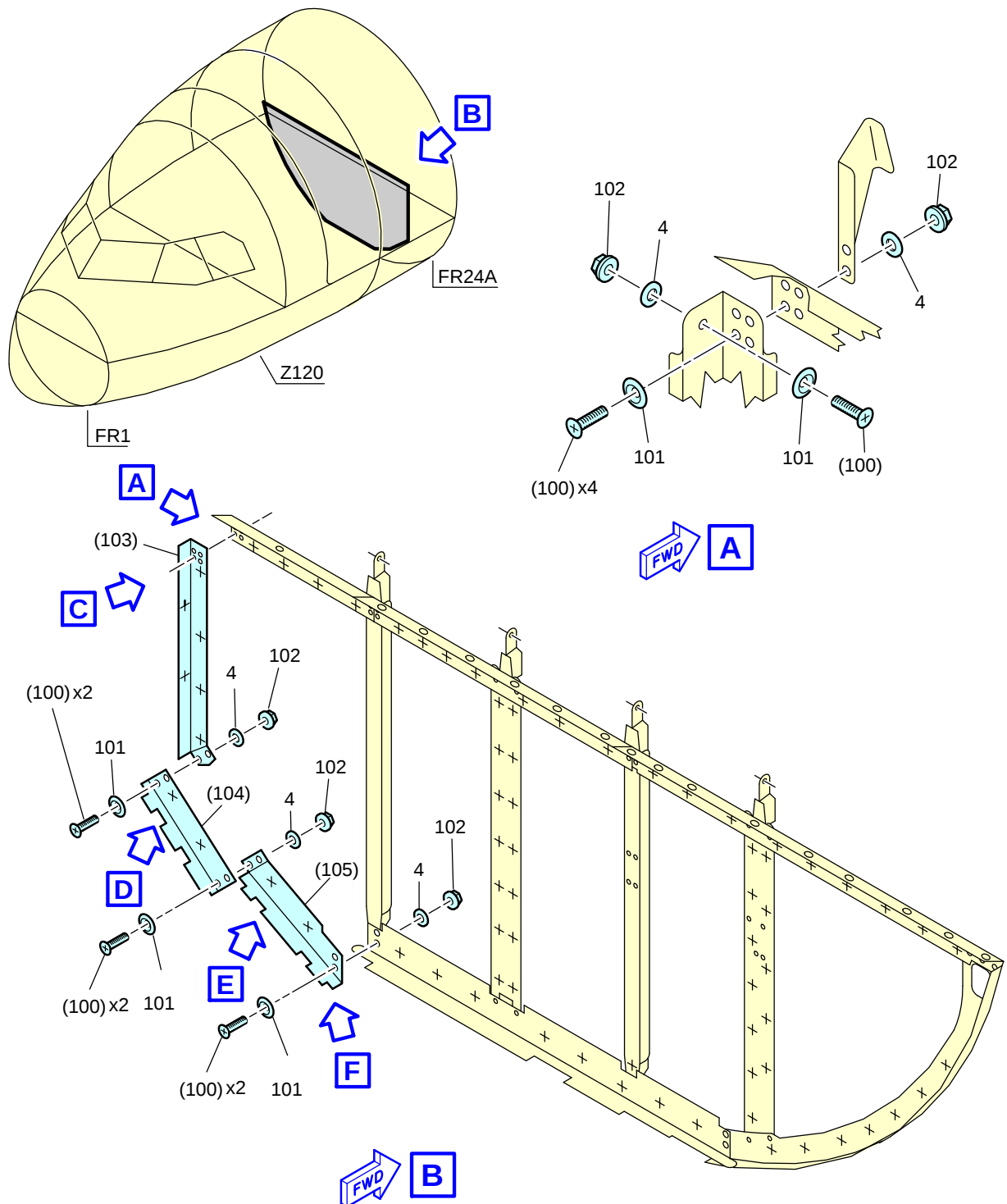
Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0G-A-001-01 SHEET 3/3

REV 00 Aug 23/18
REV 02 Apr 09/20

Installation of the LH strut at FR24A

A320-25-1BKJ
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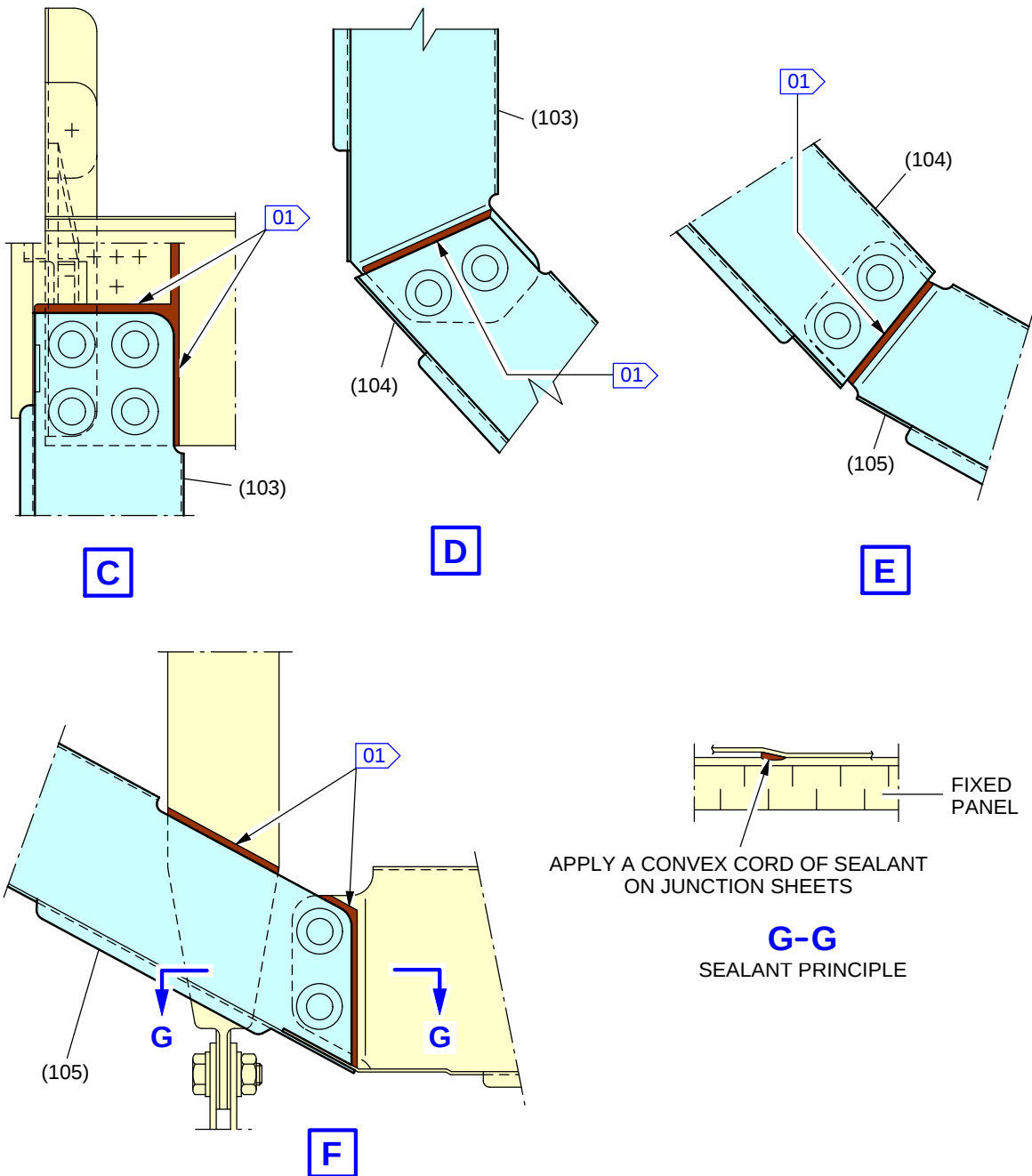
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0L-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0L-A-001-01 SHEET 1/2

CONF 002-004



NOTE:

01 APPLY SEALANT MATERIAL No 06BHG1 TO THE JUNCTION OF THE SHEETS.

ICN-A320-A-25XX1BKJ-A-FAPE3-01W0M-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0M-A-001-01 SHEET 2/2

(3) Task A320-A-25-XX-1BKJ-05002-740A-A - Install the ducts
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	2.00 h
Minimum Number of person	1
Elapsed Time	2.00 h
Skills	Airframe

Material necessary to do the job

Part kit 251BKJD05R00					
ITEM	PART No.	QTY (UM)	KEYWORD	DRAWING	SEE NOTES
4	NAS1149F0332P	26	WASHER		
87	NSA8203-258	1	ORING		

Referenced Information

Manuals/Drawings	References/Sheets
AMM	06-20-00, Page Block 000 06-31-53, Page Block 000 Task 20-21-11-911-001 20-28-00, Page Block 200 Task 21-20-00-410-001
CML	CONSUMABLE MATERIAL LIST (CML)
ESPM	20-51-20
SRM	51-24-00 51-42-00 51-43-00 51-75-12

Procedure
(a) Standard Practices

- 1 For the Frame (FR) identification, refer to Ref. AMM 06-31-53, Page Block 000.
- 2 For the identification of zones, refer to Ref. AMM 06-20-00, Page Block 000.
- 3 For the specification of Material application codes given in this Service Bulletin, refer to the Ref. CML.
- 4 For the general installation of air-conditioning distribution ducts, refer to Ref. AMM Task 21-20-00-410-001.
- 5 Remove and install fasteners in accordance with Ref. SRM 51-42-00.
- 6 Get alternative and substitute fastener data in accordance with Ref. SRM 51-43-00.

- 7 To torque tighten the standard threaded fasteners, refer to Ref. AMM Task 20-21-11-911-001.
- 8 For the electrical bonding, refer to Ref. ESPM 20-51-20 and Ref. AMM 20-28-00, Page Block 200.
- 9 Apply sealant in accordance with Ref. SRM 51-24-00.
- 10 Renew the protective finish in accordance with Ref. SRM 51-75-12.

(b) Install at the LH side of the FIN 80VU, refer to Ref. Fig. Installation of the upper LH duct:

1	DUCT		Item (17)	Retained
1	DUCT ASSY		Item (18)	Retained
1	DUCT ASSY		Item (22)	Retained
1	DIAPHRAGM		Item (10)	Retained
1	SEAL		Item (19)	Retained
1	SEAL attached with		Item (6)	Retained
8	SCREW		Item (14)	Retained
10	SCREW		Item (13)	Retained
18	WASHER	NAS1149F0332P	Item 4	
3	SLEEVE		Item (25)	Retained
1	CLAMP		Item (16)	Retained
12	CLAMP		Item (24)	Retained
	or			
6	CLAMP		Item (24)	Retained

(c) Install at the LH side of the FIN 80VU, refer to Ref. Fig. Installation of the LH duct:

1	DUCT attached with		Item (30)	Retained
1	SLEEVE		Item (7)	Retained
6	CLAMP		Item (11)	Retained
	or			
1	SLEEVE		Item (7)	Retained
4	CLAMP		Item (11)	Retained

(d) Install under the FIN 80VU, refer to Ref. Fig. Installation of the RH ducts:

1	DUCT ASSY		Item (8)	Retained
1	DUCT ASSY		Item (9)	Retained
1	O-RING attached with	NSA8203-258	Item 87	
1	SLEEVE		Item (7)	Retained

1	CLAMP		Item (86)	Retained
1	CLAMP		Item (88)	Retained
6	CLAMP		Item (11)	Retained
	or			
4	CLAMP		Item (11)	Retained
(e)	Install at the RH side of the FIN 80VU, refer to Ref. Fig. Installation of the RH ducts:			
1	DUCT ASSY		Item (5)	Retained
1	DUCT ASSY		Item (12)	Retained
1	SEAL		Item (6)	Retained
	attached with			
8	WASHER	NAS1149F0332P	Item 4	
8	SCREW		Item (13)	Retained
2	SLEEVE		Item (7)	Retained
9	CLAMP		Item (11)	Retained
	or			
5	CLAMP		Item (11)	Retained

CONF 002-004

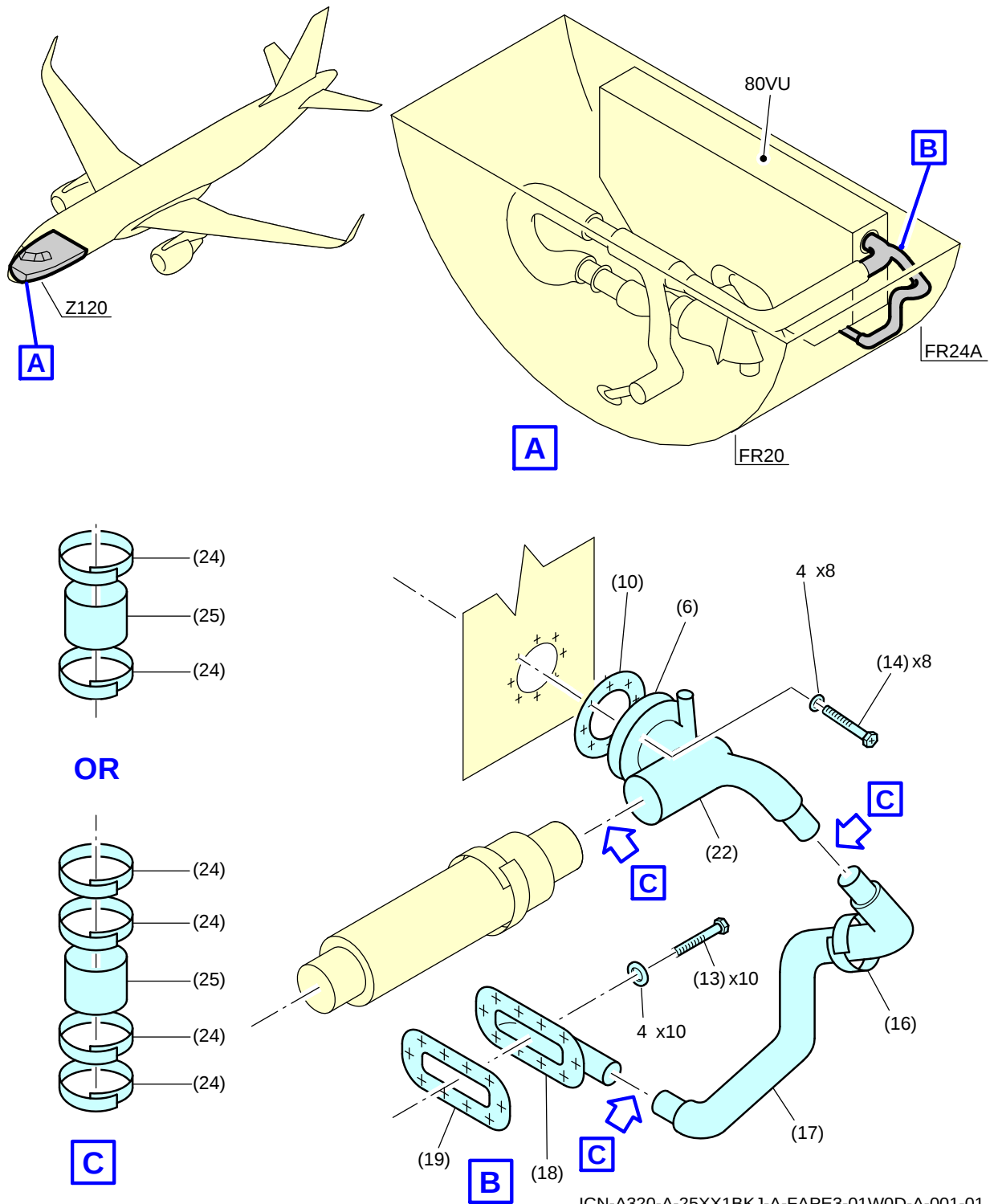
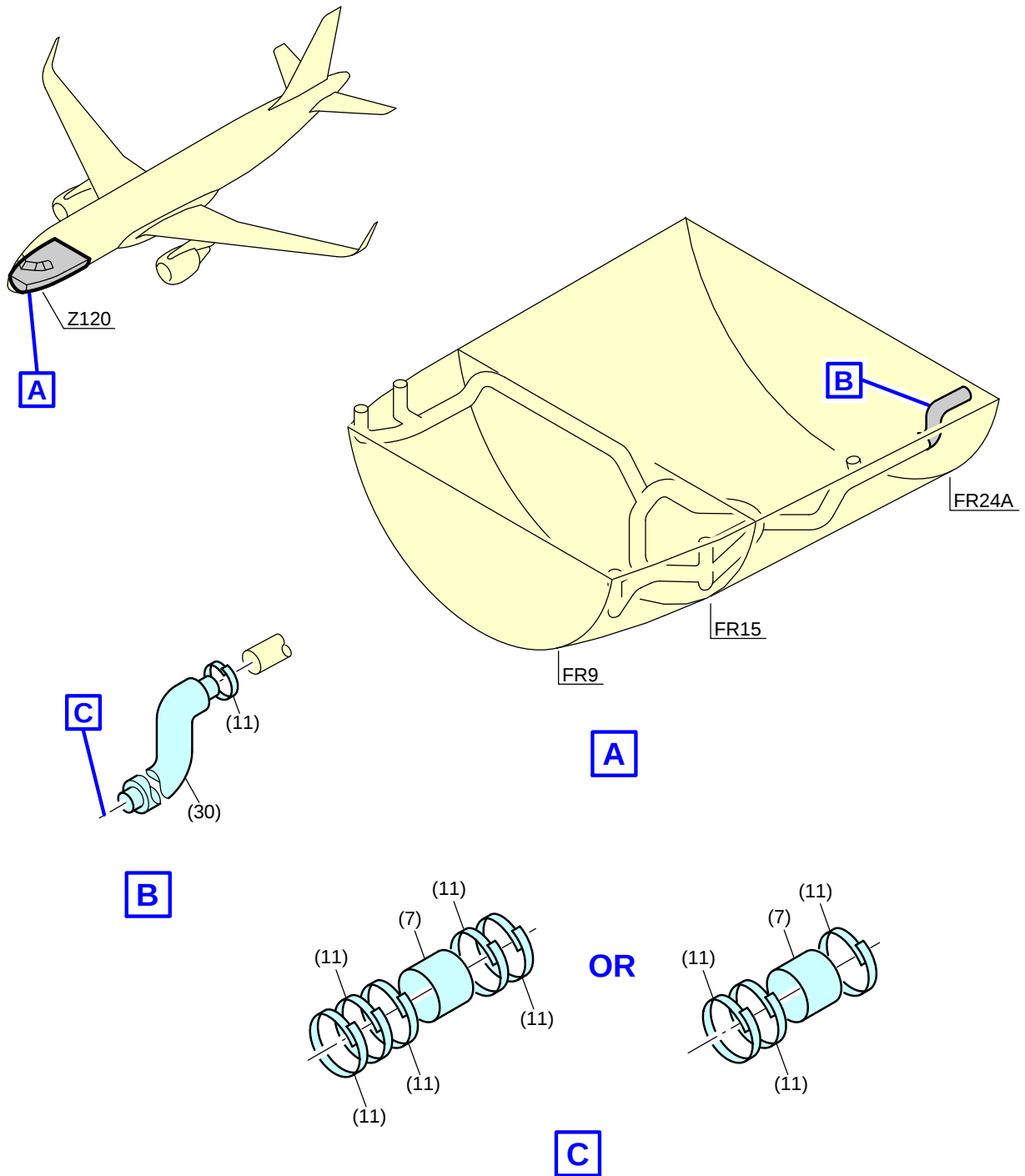


Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0D-A-001-01 SHEET 1/1

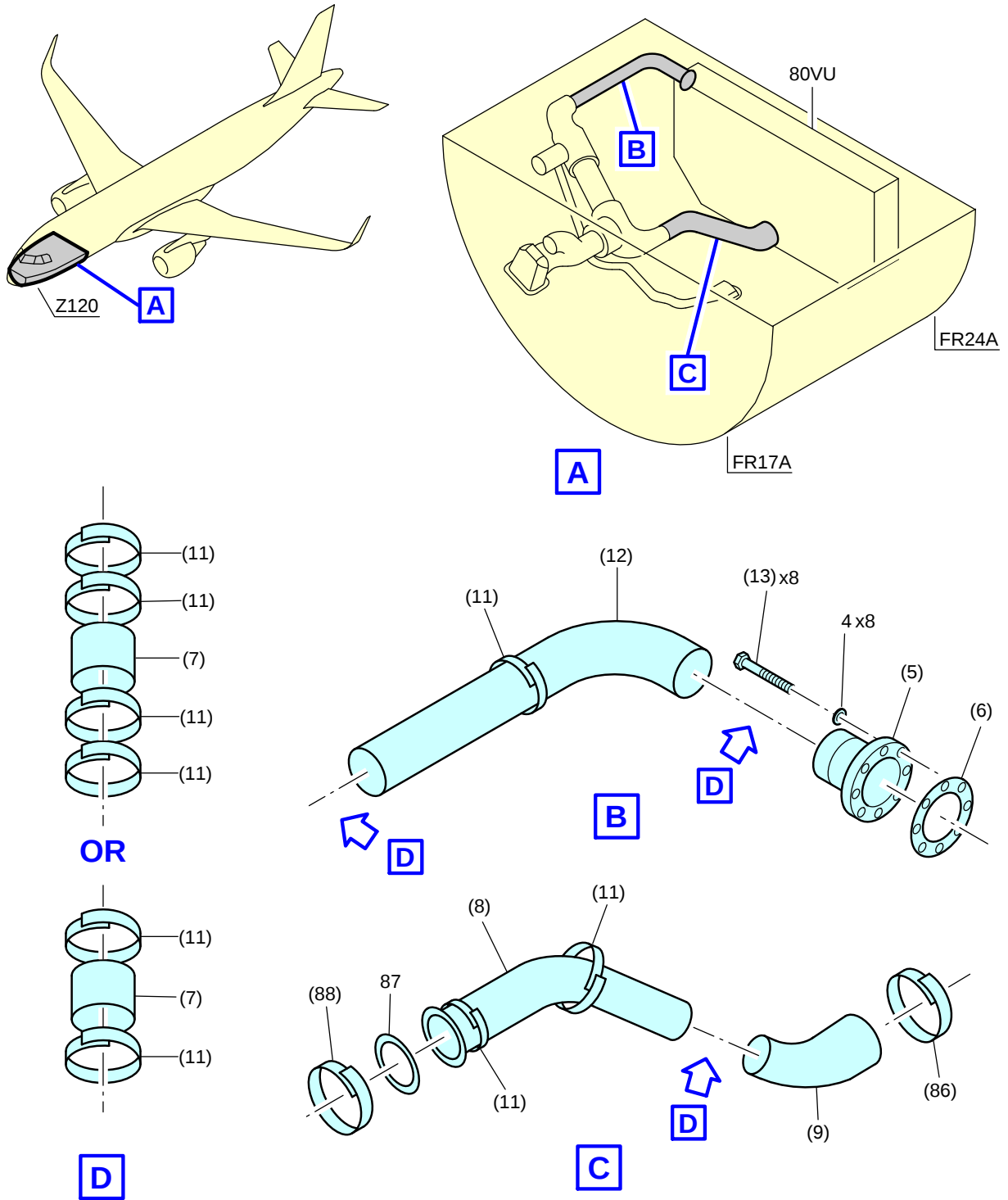
CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W8Z-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W8Z-A-001-01 SHEET 1/1

CONF 002-004



ICN-A320-A-25XX1BKJ-A-FAPE3-01W0H-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-01W0H-A-001-01 SHEET 1/1

(4) Task A320-A-25-XX-1BKJ-03002-740A-A - Close-up
Job Set-Up Information
Working Area

Impacted Zone	Access / Work Location	
127	Work location	From Frame 20 to Frame 24
128	Work location	From Frame 20 to Frame 24
130	Work location	From Frame 24 to Frame 36

Manpower/Resources/Skills

Man-hours	5.25 h
Minimum Number of person	2
Elapsed Time	2.75 h
Skills	Airframe

Referenced Information

Manuals/Drawings	References/Sheets
AMM	Task 25-54-11-400-001 Task 25-54-12-400-003 Task 52-30-00-860-002 Task 52-41-00-410-002 Task 53-12-13-400-001 Task 53-22-14-400-001

Procedure

- (a) Make sure that the work areas are clean and clear of tools and other items of equipment.
- (b) Install the FWD cargo compartment floor panels, (Panel 131CF or Panel 131RG, Panel 131BF or Panel 131QG and Panel 131AF or (Panel 131LF and Panel 132MF) or Panel 131PG), refer to Ref. AMM Task 53-22-14-400-001.
- (c) Install the sidewall panel, Panel 131GW, refer to Ref. AMM Task 25-54-12-400-003.
- (d) Install the FWD cargo compartment panels, Panel 131DW and/or Panel 131CW and/or Panel 132EW, refer to Ref. AMM Task 25-54-11-400-001.
- (e) Install the avionics compartment floor panels, (Panel 127BF, Panel 127CF, Panel 128CF and Panel 128EG or Panel 128EF), refer to Ref. AMM Task 53-12-13-400-001.
- (f) Close the FWD cargo compartment door, Door 825, refer to Ref. AMM Task 52-30-00-860-002.
- (g) Close the avionics compartment door, Door 824, refer to Ref. AMM Task 52-41-00-410-002.
- (h) Restore the systems and the aircraft to normal operating condition.

G. ADMINISTRATIVE
(1) Task A320-A-25-XX-1BKJ-02002-90AA-A - Administrative
Procedure

Confirm in the applicable aircraft records that you have completed all the work defined in this Service Bulletin task set as required.

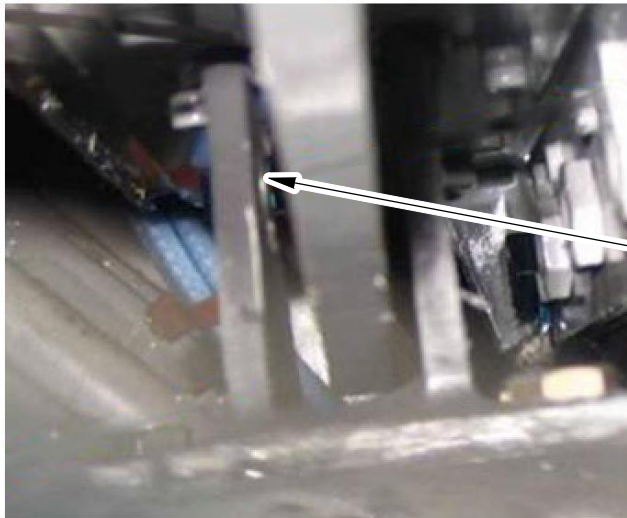
CONF ALL
APPENDIX**A. Inspection Condition of the Lateral Support**

Inspection Condition of the Lateral Support

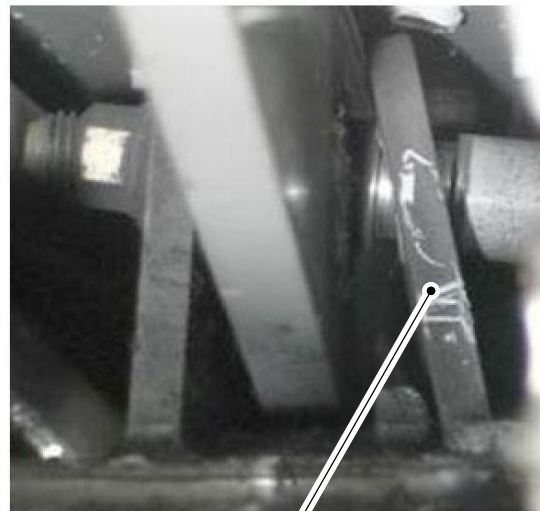
For an explanation of the lateral support inspection, refer to the next page(s).

CONF ALL

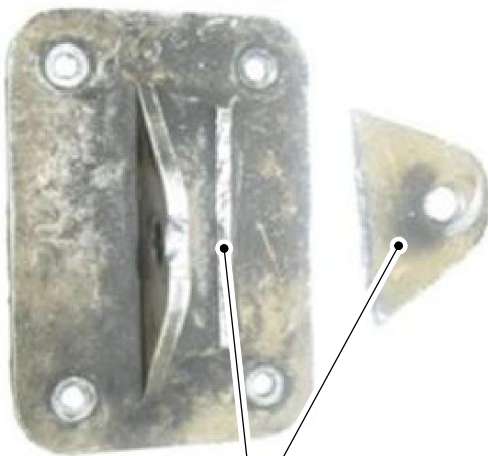
FITTINGS BROKEN OR MISSING BOLT



MISSING BOLT



FITTING BROKEN



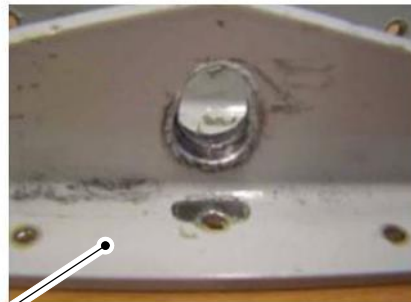
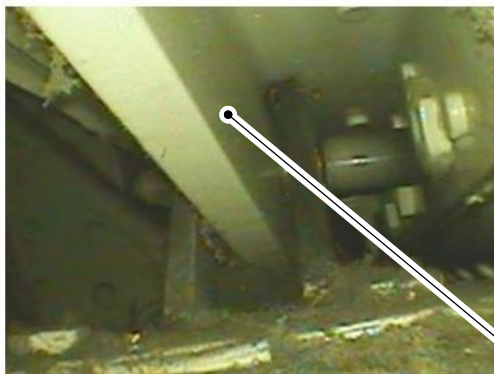
FITTING BROKEN

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOJ-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOJ-A-001-01 SHEET 1/2

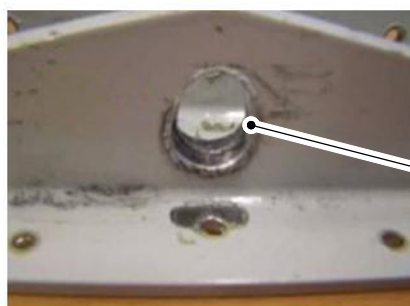
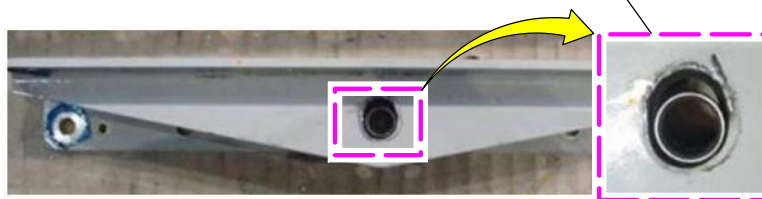
CONF ALL

BUSH MIGRATED OR MATERIAL BURRED



RACK IN CONTACT WITH STRUCTURE

MATERIAL BURRED (OVALIZATION)



MATERIAL BURRED
(OVALIZATION)

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOK-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOK-A-001-01 SHEET 2/2

B. Inspection Condition of the Central Post

Inspection Condition of the Central Post

For an explanation of the central support inspection, refer to the next page(s).

CONF ALL

CRACKS ON CENTRAL POST



CRACK LOCATION

NOTE:



INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOL-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOL-A-001-01 SHEET 1/1

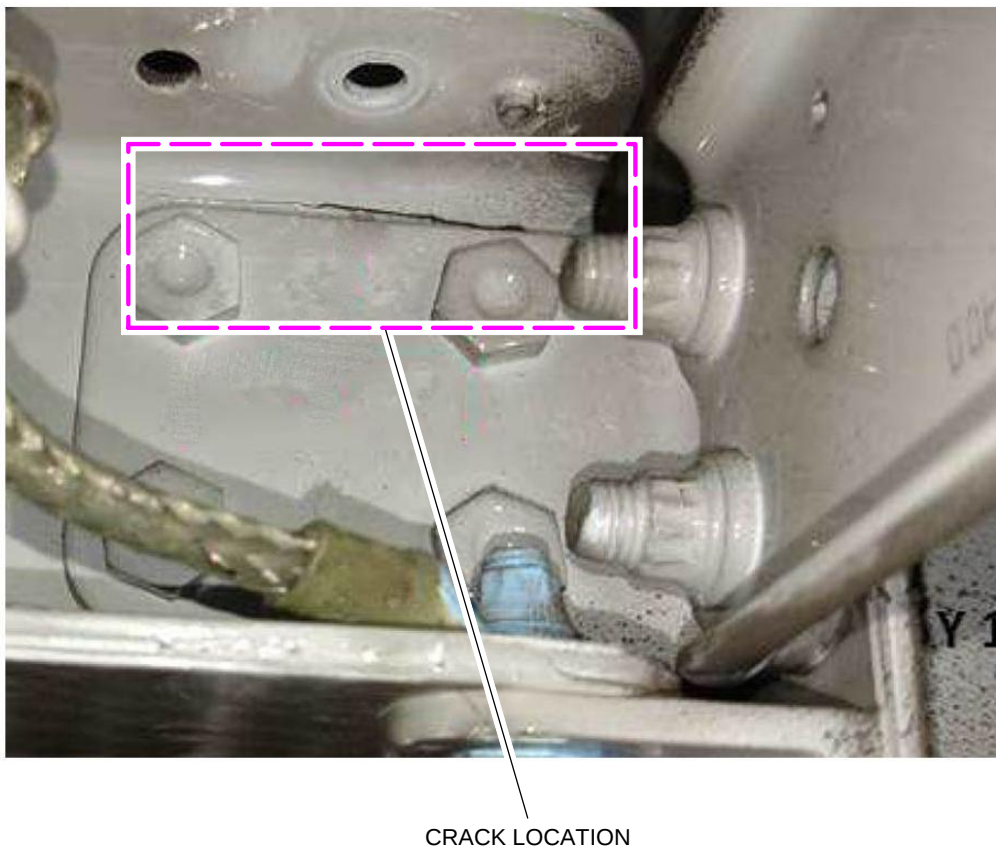
C. Inspection Condition of the Lateral Shelves Attachment

Inspection of Lateral Shelves Attachment

For an explanation of the lateral shelves attachment inspection, refer to the next page(s).

CONF ALL

SHELVES LATERAL ATTACHMENTS



NOTE:



INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EOM-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EOM-A-001-01 SHEET 1/1

D. Inspection Condition of the Lateral Pyramid Structure

Inspection of Lateral Pyramid Structure

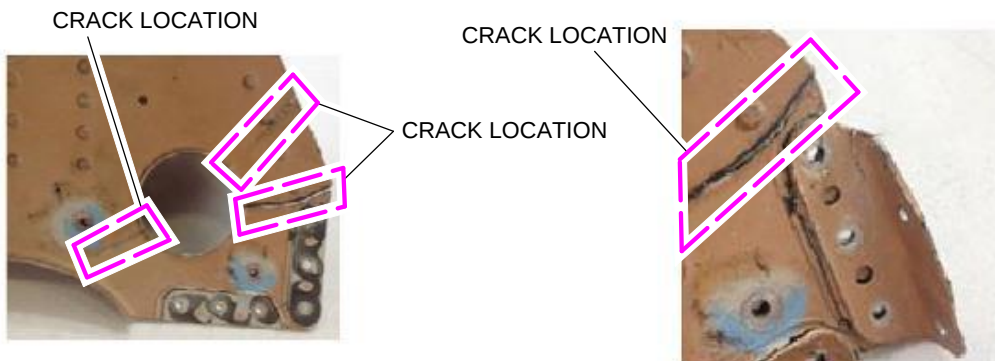
For an explanation of the lateral pyramid structure inspection, refer to the next page(s).

CONF ALL

LATERAL PYRAMID STRUCTURE



CRACK LOCATION



CRACK LOCATION

CRACK LOCATION

CRACK LOCATION



CRACK LOCATION

NOTE:



INSPECTION AREA

ICN-A320-A-25XX1BKJ-A-FAPE3-00EON-A-001-01

Figure ICN-A320-A-25XX1BKJ-A-FAPE3-00EON-A-001-01 SHEET 1/1

E. REPORTING SHEET

TITLE:	EQUIPMENT/FURNISHINGS - FORWARD FUSELAGE EQUIPMENT - SPECIAL DETAILED INSPECTION OF THE RACK 80VU
MODIFICATION No.:	None

(1) **SB Reporting status**

This Service Bulletin will only be incorporated into affected customized Maintenance and Flight Operations Products if it is duly reported to Airbus.

(2) **GENERAL information**

The SB Reporting policy is published in ISI 00.00.00135. This document provides information regarding: the SB Reporting means and benefits, the update of customized Maintenance Products and customized Flight Operations Products.

If this Service Bulletin has an operational impact, refer to paragraph "Flight Operations Products" for further information on the update of affected Flight Operations Products.

(3) **SB Reporting procedure**

Standard practice to report SB is to use the online reporting application on AirbusWorld.

Refer to ISI 00.00.00179:

- For complete information on reporting means to Airbus, Question 2 "What are the standard practices for SB Reporting?"
- If this SB requires previous or simultaneous accomplishment of other SBs, Question 3 "What about the reporting of prerequisite Service Bulletins?"

NOTE: ISI 00.00.00179 covers all the frequently asked questions about SB Reporting. It is revised on a regular basis.

F. QUALITY PERCEPTION FORM

Use this form to tell us what is your perception of the quality of this Service Bulletin.

The reported data that you provide us will be used to analyse areas of difficulties and to take corrective action to further improve the quality of our Service Bulletin.

We thank you for the time you have taken in completing this form.

Quality rating of this SB	4	3	2	1
Quality rating of the Accomplishment Instructions	4	3	2	1
Quality rating of the Illustrations	4	3	2	1
Is this SB easy to understand ?	Y / N			

NOTE: Please rate on a scale of 1 to 4, with 4 being the highest score

If you have difficulties in the accomplishment of this SB please quote below the area(s) and give a short description of the issue.

	Planning		Material		Instructions
X	Applicability	X	Tooling	X	Preparation
X	Reason	X	List of Materials Operator Supplied	X	Mod/Inspection
X	Manpower	X	Kit content	X	Test
X	References	X	Re-identification	X	Close-up
X	Publication			X	Illustrations
X	Industry Support				

Comments :

Operator :

Date :

Name/Title

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Or via your Resident Customer Support Office.